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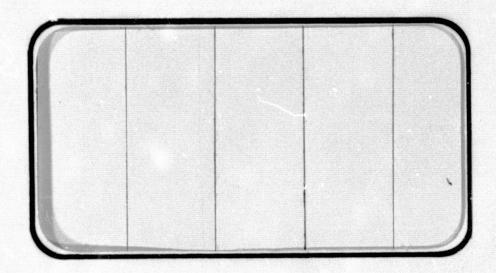
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## NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



(NASA-CR-141845) AERODYNAMIC RESULTS OF A SEPARATION TEST (CA 20) CONDUCTED AT THE BOEING TRANSONIC WIND TUNNEL USING 0.030-SCALE MODELS OF THE CONFIGURATION 140A/B (MODIFIED) SSV ORBITER (MODEL NO.

N76-16034 HC \$28.25

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT



JOHNSON SPACE CENTER HOUSTON, TEXAS

SPACE DIVISION CHRYSLER CORPORATION

DMS-DR-2217 NASA CR-141,845 VOLUME 2 of 3

AERODYNAMIC RESULTS OF A SEPARATION TEST (CA20)

CONDUCTED AT THE BOEING TRANSONIC WIND TUNNEL

USING 0.030-SCALE MODELS OF THE CONFIGURATION

140A/B (MODIFIED) SSV ORBITER (MODEL NO. 45-0) AND

THE BOEING 747 CARRIER (MODEL NO. AX 1319 I-1)

bу

T. Dziubala, V. Esparza, R. L. Gillins and M. Petrozzi Shuttle Aero Sciences Rockwell International Space Division

Propared under NASA Contract Number NAS9-13247

by

Data Management Services Chrysler Corporation Space Division New Orleans, La. 70189

for

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Johnson Space Center National Aeronautics and Space Administration Houston, Texas

### WIND TUNNEL TEST SPECIFICS:

Test Number:

BTWT 1431/AX 1319 I-1

NASA Series Number:

**CA20** 

Model Number:

45-0 Mod/747 Carrier AX 1319 I-1

Test Dates:

9 through 16 October 1974

Occupancy Hours:

115

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CONDUCTED AT THE BOEING TRANSONIC WIND TUNNEL

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#### **ABSTRACT**

An experimental aerodynamic investigation (CA20) was conducted in the Boeing Transonic Wind Tunnel from October 9 through October 16, 1974. A Rockwell built 0.030-scale 45-0 modified SSV Orbiter Configuration 140A/B model and a Boeing built 0.030-scale 747 carrier model were tested to provide six component force and moment data for each vehicle in proximity to the other at a matrix of relative positions, attitudes and test conditions. Orbiter model support system tare effects were determined for corrections to obtain support-free aerodynamics.

In addition to the balance force data, pressures were measured. Pressure orifices were located at the base of the Orbiter, on either side of the vertical blade strut, and at the mid-root chord on either side of the vertical tail. Strain gages were installed on the Boeing 747 vertical tail to indicate buffet onset.

The 747 carrier was varied through angles of attack (measured with respect to its FRL) of 0°, 2°,4°,6°,8°, and 10° and varied through sideslip

angles of 0°, +5°, and -5°. Elevator settings were also varied.

The SSV Orbiter model was varied through angles of attack of  $6^{\circ}$ ,  $8^{\circ}$ ,  $10^{\circ}$ ,  $12^{\circ}$ ,  $14^{\circ}$ ,  $16^{\circ}$ , and  $18^{\circ}$  and varied through sideslip angles of  $2.5^{\circ}$ ,  $0^{\circ}$ ,  $-2.5^{\circ}$ ,  $-5^{\circ}$ ,  $-7.5^{\circ}$ ,  $-10^{\circ}$ , and  $-15^{\circ}$ .

Vertical displacements of 0", 1", 2", 3", 5", 7", 9", 11", 13", 15", 18", and 21.6" (model scale) were tested. Longitudinal movements of 0", 3.6", and 7.2" (model scale) and lateral displacements of 0" and 3.6" (model scale) were tested to simulate various separation positions. Orbiter elevon deflections were also varied.

Orbiter support system tare and interference effects were determined utilizing various support and image support strut configurations. Carrier support system tare and interference effects were determined during test CA5.

The Orbiter tail cone and carrier models were provided by The Boeing Company. The Orbiter model was provided by Rockwell. These were the same models used earlier in test CA5.

This report for CA20 consists of three volumes: Volume 1 - data figures 1 through 25; Volume 2 - data figures 26 through 39; Volume 3 - tabulated source data.

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# NOMENCLATURE

Symbol	Plot Symbol	Description
b	BREF	reference span, in
BSTA	XC	longitudinal carrier station, in
BWL	ZC	vertical carrier station, in
<del>c</del>	LREF	mean aerodynamic chord, in
c <sub>A</sub>	CA	axial force coefficient
c <sub>D</sub>	CD	drag coefficient
$c_{k_B}$	CBL	body axis rolling moment coefficient
$c_{ls}$	CSL	stability axis rolling moment coefficient
$c_L$	CL	lift coefficient
Cm	CLM	pitching moment coefficient
$c_{n_B}$	CYN	body axis yawing moment coefficient
c <sub>ns</sub>	CLN	stability axis yawing moment coefficient
CN	CN	normal force coefficient
c <sub>PB1</sub>	PB1	Orbiter base pressure coefficient for orifice no. 1, see Figure 2c
C <sub>PB2</sub>	PB2	Orbiter base pressure coefficient for orifice no. 2, see Figure 2c
c <sub>PB4</sub>	PB4	Orbiter base pressure coefficient for orifice no. 4, see Figure 2c
CPCAV	PCAV	Orbiter cavity pressure coefficient

Symbol	Plot Symbol	<u>Definition</u>
c <sub>b</sub> e <sup>B1</sup>	LHLS	coefficient of pressure measured on fuselage at left side of vertical tail
c <sub>p</sub> EB2	RHLS	coefficient of pressure measured on fuselage at right side of vertical tail
c <sub>P</sub> sc	PSC	carrier cavity pressure coefficient
c <sub>p</sub> s <sub>1</sub>	LHVERT	coefficient of pressure measured on left side of Orbiter strut
c <sub>p</sub> s <sub>2</sub>	RHVERT	coefficient of pressure measured on right side of Orbiter strut
c <sub>Y</sub>	СҮ	side force coefficient
C.G.		center of gravity
C.R.		center of rotation
FRL		fuselage reference line
ℓ <sub>o</sub>	IORB	Orbiter incidence relative to carrier FRL, deg.
LB	LREF	reference body length, in
MACH	MACH	Mach number
M.R.C.	XMRP,YMRP ZMRP	moment reference center, in
MS		model station, in
P <sub>Bi</sub>		base pressure measured at station i, i=1,2,4, psia
PEB1		pressure measured on Orbiter fuselage surface on left side vertical tail/fuselage juncture, psia

Symbol	Plot Symbol	<u>Definition</u>
P <sub>EB2</sub>		pressure measured on Orbiter fuselage surface on right side vertical tail/fuselage juncture, psia
PSI		pressure measured on left side of Orbiter strut S <sub>1</sub> , psia
P <sub>S2</sub>		pressure measured on right side of Orbiter strut S <sub>1</sub> , psia
q	Q(PSF)	freestream dynamic pressure, psf
RN/ET	RN/L	freestream unit Reynolds no., 10 <sup>6</sup> per foot
V		mean freestream velocity, ft/sec
S	SREF	wing area or reference area, ft <sup>2</sup>
WL	Z	water line, in
<b>X</b>		longitudinal Orbiter separation distance, measured from nominal mated position, ft
Х <sub>с</sub>	хс	carrier longitudinal station, in
X <sub>MRP</sub>	XMRP	longitudinal location of MRC, in
Χo	хо	Orbiter longitudinal station, in
<b>Y</b>		Orbiter lateral separation distance, measured from nominal mated position, ft
YC	YC	carrier lateral station, in
YMRP	YMRP	lateral location of MRC, in
Yo	YO	Orbiter lateral station, in
<b>Z</b>		Orbiter vertical separation distance, measured from nominal mated position, ft
z <sub>c</sub>	ZC	carrier vertical station, in

Symbol -	Plot Symbol	<u>Definition</u>
Z <sub>MRP</sub>	ZMRP	vertical location of MRC, in
Z <sub>o</sub>	ZO	Orbiter vertical station, in
1/( <sub>\(\DZ</sub> +10)	1/Z+10	separation parameter, inverse of vertical separation distance plus 10 ft, per foot
α	ALPHA	angle of attack, deg.
<sup>α</sup> C	ALPHAC	carrier fuselage angle of attack, $\alpha_W^{-2^\circ}$ , deg.
<sup>α</sup> 0	ALPHA0	Orbiter angle of attack, deg.
$\alpha_{W}$	ALPHAW	carrier wing angle of attack, $\alpha_c^+$ 2°, deg.
αWall	ALPWAL	wind tunnel wall correction to carrier angle of attack, deg.
β	ВЕТА	angle of sideslip, deg.
βС	BETAC	carrier sideslip angle, deg.
<sup>β</sup> 0	BETA0	Orbiter sideslip angle, deg.
$^{\delta}$ a	AILRON	aileron deflection angle, deg.
δ <b>e</b>	ELEVON	Orbiter elevon deflection angle, deg.
<b>δe</b> I	ELV-IB	inboard carrier elevator panel deflection angle, deg.
$^{\delta}\mathbf{e}_{\phi}$	ELV-OB	outboard carrier elevator panel deflection angle, deg.
δeγ	ELEVTR	carrier elevator deflection angle, deg.
$^{\delta}\mathbf{r}$	RUDDER	carrier rudder deflection angle, deg.
δ <b>r</b> L	RUD-L	carrier lower rudder panel deflection angle, deg.

Symbol .	Plot Symbol	D <u>efinitio</u> n
δ <sub>ru</sub>	RUD-U	carrier upper rudder panel deflection angle, deg.
δ <sub>S</sub>		spoiler deflection angle, deg.
ΔCA	DCA	incremental axial force coefficient
ΔCD	DCD	incremental drag coefficient
∆C <sub>&amp;B</sub>	DCBL	incremental body axis rolling moment coefficient
ΔC <sub>l</sub> S	DCSL	incremental stability axis rolling moment coefficient
ΔCL	DCL	incremental lift coefficient
ΔC <sub>m</sub>	DCLM	incremental pitching moment coefficient
$\Delta c_{n_{B}}$	DCYN	incremental body axis yawing moment coefficient
ΔC <sub>n</sub> S	DCLN	incremental stability axis yawing moment coefficient
ΔCN	DCN	incremental normal force coefficient
ΔСγ	DCY	incremental side force coefficient
ΔΧ	DX	Orbiter longitudinal separation distance from nominal mated position, ft
Δ <b>Υ</b>	DY	Orbiter lateral separation distance from nominal mated position, ft
ΔΖ	DZ	Orbiter vertical separation distance from nominal mated position, ft
Δα	DALFA	incremental angle of attack between Orbiter and carrier FRL, $\alpha_{\rm o}$ - $\alpha_{\rm c}$ , deg.

# NOMENCLATURE (Concluded)

Symbol .	Plot Symbol	<u>Definition</u>
Δβ	DBETA	incremental angle of sideslip between Orbiter and carrier, $\beta_{0}$ - $\beta_{c},$ deg.
Δφ	DPHI	incremental roll angle between Orbiter and carrier, deg.
ф	PHI	Orbiter roll angle, deg.
c <sub>LC</sub>	CL-C	carrier lift coefficient with test mounting system corrections
CDC	CD-C	carrier drag coefficient with test mounting system corrections
CmC	CLM-C	carrier pitching moment coefficient with test mounting system corrections
CYC	CY-C	carrier side force coefficient with test mounting system corrections
C <sub>nBC</sub>	CYN-C	carrier body yaw moment coefficient with test mounting system corrections
c <sub>nSC</sub>	CLN-C	carrier stability yaw moment coefficient with test mounting system corrections
C <sub>LSC</sub>	CSL-C	carrier stability roll moment coefficient with test mounting system corrections
C <sub>&amp;BC</sub>	CBL-C	carrier body roll moment coefficient with test mounting system corrections
c <sub>A</sub> c	CA-C	carrier axial force coefficient with test mounting system corrections
c <sub>NC</sub>	CN-C	carrier normal force coefficient with test mounting system corrections

#### REMARKS

The Orbiter axial force, measured during this test, exhibits the following trend:

- at low angles of attack, axial force decreases with increasing angle of attack, as would normally be expected,
- at high angles of attack, axial force increases with increasing angle of attack, contrary to normal expectations.

Extensive investigations and analysis, conducted during the test, indicated that trend number (2) was not caused by model fouling or other test problems and was, indeed, representative of aerodynamic characteristics.

Vertical tail pressure instrumentation ( $P_{EB_1}$  and  $P_{EB_2}$ ) was disconnected during runs 588 through 599.

Configuration D (as described in figure 2f) was not at  $\phi$  = 90°, as planned, because of support system deflections (caused by the Orbiter model touching strut S3).

### CONFIGURATIONS INVESTIGATED

The Orbiter model was an 0.030-scale representation of the Space Shuttle Orbiter VL70-000140A/B lines with modified OMS pods and elevons as shown in figure 2a. The basic Orbiter is a blended wing-body design with a double delta wing (75° and 45° leading edge sweeps). The Orbiter model was tested both with and without a tail cone fairing. The tail cone fairing covered the MPS nozzles, OMS nozzles, and base, as shown in Figure 3b. The Orbiter model was mounted in the tunnel using several blade strut configurations as follows:

S<sub>1</sub> = Orbiter support blade strut, upper entry position,

S<sub>2</sub> = Orbiter support blade strut, lower entry position,

 $S_3 = Orbiter dummy support blade strut.$ 

Figure 2f shows the strut arrangements. Orbiter elevon and aileron deflection angles were varied. The Orbiter was tested both isolated and in the presence of the carrier at various separation locations. The following Orbiter configurations were tested:

$$O_1 = B_{26} C_9 E_{43} F_8 M_{16}$$
  $W_{116} T_{5.1}$   
 $O_2 = B_{26} C_9 E_{43} F_8 M_{16} N_{28} N_{24}$   $W_{116}$  (with strut  $S_1$ )  
 $O_3 = B_{26} C_9 E_{43} F_8 M_{16}$   $R_5 V_8 W_{116} T_{5.1}$   
 $O_4 = B_{26} C_9 E_{43} F_8 M_{16}$   $W_{116} T_{5.1}$   
 $O_5 = B_{26} C_9 E_{43} F_8 M_{16} N_{28} N_{24} R_5 V_8 W_{116}$   
 $O_6 = B_{26} C_9 E_{43} F_8 M_{16} N_{28} N_{24} R_5 V_8 W_{116}$  MPS cover plate off

# CONFIGURATIONS INVESTIGATED (Continued)

 $0_7$  =  $B_{26}$   $C_9$   $E_{43}$   $F_8$   $M_{16}$   $N_{28}$   $R_5$   $V_8$   $W_{116}$  strut,  $S_2$  cover plate #1 off MPS  $0_8$  =  $B_{26}$   $C_9$   $E_{43}$   $F_8$   $M_{16}$   $N_{28}$   $R_5$   $V_8$   $W_{116}$  strut,  $S_2$  cover plate #2 off MPS  $0_9$  =  $0_9$   $0_9$  =  $0_9$ 

### where:

Component	<u>Description</u>
<sup>B</sup> 26	Orbiter fuselage per Rockwell lines VL70-000140A/B, model drawing SS-A01360
c <sub>9</sub>	Orbiter canopy per Rockwell lines VL70-000140A/B, model drawing SS-A01360
E <sub>43</sub>	Orbiter full-span, unswept hingeline, 6" gapped elevons per Rockwell lines VL70-000200, model drawing SS-A01360
F <sub>8</sub>	Orbiter body flap per Rockwell lines VL70-000200, model drawing SS-A01360
M <sub>16</sub>	Orbiter OMS/RCS pods per Rockwell lines VL70-000203A, VL70-008401, model drawing SS-A01360
N <sub>24</sub>	Orbiter main propulsion system (MPS) nozzles - VL70-000140A, VL70-005030A, model drawing SS-A01360
N <sub>28</sub>	Orbiter OMS nozzles - VL70-000140A model drawing SS- A01360
R <sub>5</sub>	Orbiter rudder per Rockwell lines VL70-000146A, model drawing SS-A01360
TC <sub>5.1</sub>	Orbiter tail cone fairing which covers the MPS nozzles and the OMS nozzles and base, built by the Boeing Company, also used in CA5
<b>V</b> <sub>8</sub>	Orbiter centerline vertical tail per Rockwell lines VL70-000146A, model drawing SS-A01360
W116	Orbiter double delta wing per Rockwell lines VL70-000200, model drawing SS-A01360

### CONFIGURATIONS INVESTIGATED (Continued)

Effects of simulated attach hardware were investigated using the following model components attached to the carrier.

AT 38 Forward attach structure between the Orbiter and carrier model used for  $i_0$  of 3 to 10 degrees for  $\Delta Z = 0$  feet

AT 39 Aft attach structure between the Orbiter and carrier model for  $\Delta Z = 0$  feet

The carrier model was an 0.030-scale representation of the Boeing 737-100 aircraft with surface contours built to represent the 747 under loads it would experience with a 600,000 pound gross weight flying at Mach 0.86 at an altitude of 35,000 feet. The model also had a built in 0.64° leading edge up wing tip twist to compensate for model aeroelastic effects, which are estimated to produce a 0.64° leading edge down twist. The carrier had 200 square foot tip fins on its horizontal tail. Spoilers were deflected to 45° and flaps were retracted during most of the test. Several runs were made with spoilers retracted. Elevator and rudder deflections were varied during the test. The carrier was tested both isolated and in the presense of the Orbiter at various separation conditions. Carrier configurations investigated were:

$$747/0 = B_{27.8} F_0 H_{15.6} M_{26}^{25} N_{58}^{57} T_{19} V_{9.1} W_{44.1}$$
  
 $747/1 = B_{27.8} F_0 H_{15.6} M_{26}^{25} N_{58}^{57} S_{1-12} T_{19} V_{9.1} W_{44.1}, \delta_S = 45^{\circ}$ 

where:

## CONFIGURATIONS INVESTIGATED (Concluded)

H <sub>15.6</sub>	horizontal tail (H <sub>15</sub> ) with 200 ft <sup>2</sup> tip fins
M <sup>25</sup> <sub>26</sub>	inboard $(M_{25})$ and outboard $(M_{26})$ nacelle struts
N <sub>58</sub>	inboard ( $N_{57}$ ) and outboard ( $N_{58}$ ) nacelles
s <sub>1-12</sub>	12 spoiler panels located on wing upper surface, all deflected 45°
T <sub>19</sub>	flap track fairing
v <sub>9.1</sub>	vertical tail
W <sub>44.1</sub>	wing

Orbiter base pressures were measured, for configurations without tail cone, at locations as shown by figure 2c. Pressures were measured on both sides of Orbiter support strut when  $S_1$  was used and pressures were measured on the fuselage near the vertical tail when the vertical tail was installed as shown by figure 2d. Pressures were measured in the Orbiter and carrier balance cavity.

#### TEST FACILITY DESCRIPTION

The Boeing Transonic Wind Tunnel (BTWT) is a continuous flow, closed circuit, single return, atmospheric facility with the following characteristics:

Test Section F	low Parameters	Test Section Dimer	sions
Freestream Condition	Range	Description	Value
Mach number	0 thru 1.15	Cross-section (minus	
Dynamic pressure, psia	0 thru 6.3	corner fillets), ft.	8 x 10
Static pressure, psia	15 to 5.4	Length, ft.	14.5
Stagnation pressure	atmospheric	Area, ft. <sup>2</sup>	88
Maximum unit Reynolds number, per foot	4 x 10 <sup>6</sup>		
Maximum total temperature, °F	160		

The test section can be operated with either solid or slotted walls. The slotted wall configuration consists of 16 slots which can vary wall porosity from 3.5% to 11%.

Test data acquistion, recording, computations, and display are done by an XDS-9300 computer and Astro data sub-system.

#### DATA REDUCTION

Force and moment data were reduced in both body and stability axes using standard Boeing data reduction procedures. The following data reduction constants were used:

		<u>Carri</u>	er	<u>Orbiter</u>	•
Symbol	<u>Description</u>	Model Scale	Full Scale	Model Scale	Full Scale
S	reference area, ft. <sup>2</sup>	4.950	5500	2.421	2690.0
b	reference span, in	70.441	2348.04	28.100	936.68
ē	reference mac, in	9.833	327.78	14.244	474.81
MRC	moment reference center, in				
	XC or XO	40.197	1339.90	33.270	1109.0
	YC or YO	0.0	0.0	0.0	0.0
	ZC or ZO	5.723	190.80	11.250	375.0

No base or cavity corrections were applied to the data.

Wind tunnel data were interpolated versus the applicable separation parameters ( $\alpha_0$ ,  $\Delta Z$ ,  $\Delta X$ ,  $\alpha_W$ ,  $\Delta Y$ ,  $\beta_0$ ,  $\beta_C$ , and  $\phi$ ) as summarized by Table VII. These interpolated data were used to compute interference increments by subtracting isolated data from interference data as summarized by Table VIII. A special interpolation routine was used for datasets with simulated attach hardware as summarized by Tables IX and X. Interpolated carrier data were corrected for support strut tare and interference using corrections obtained during test CA5 as summarized by Table XI. Basic data, interpolated data, incremental data, and carrier data with tare and interference corrections, are presented in this report. Tables IV through VI describe data presentation formats.

#### REFERENCES

### Reports and Internal Letters

Speed Letter, SAS/WTO/74-365, "Fabrication of a new 0.03-scale Orbiter Model," dated July 3, 1974

- IL, SAS/WTO/74-173, Addendum #1, "Updated Model Design Requirements for Model 45-0", dated July 24, 1974
- IL, SAS/WTO/74-173, Addendum #2, "Additional Requirements for Model 45-0," dated July 24, 1974
- IL, SAS/AERO/74-493, "Piggyback Separation Tests Orbiter Support Configurations and Corrections," dated August 9, 1974
- IL, SAS/AERO/74-552, "Orbiter Model Support and Instrumentation Requirements"
- IL, SAS/AERO/74-617, "Test Requirements for Separation Test CA20," dated August 20, 1974
- NA-74-541, "Structural Analysis of the 0.03-scale SSV model 45-0", dated July 23, 1974

DMS-DR-2211, "Results of a 0.03-scale Aerodynamic Characteristics Investigation of a Boeing 747 Carrier (Model AX 1319 I-D) Mated with a Space Shuttle Orbiter (Model 45-0) conducted in the Boeing Transonic Wind Tunnel (CA5)", by 747 Aerodynamics, 747 Flight Controls, and Wind Tunnel Test Group, Boeing Aerospace Company

#### Drawings

#### Rockwell International - SSV Orbiter

SS-A01360 - Model Assy., 45-0, 0.03 Sc. SSV Orbiter (140A/B) Revision B, dated August 1, 1974

SS-A013 Model Instl. 45-0, 0.03 Sc. SSV Ferry Separation, Release 1, dated August 12, 1974

SS-A01362 - Blade Strut Assy., 0.03 Sc. 45-0 SSV Model, dated July 29, 1974

## The Boeing Company - 747 Carrier

65-69716 - Model Assy., TE 1007 I-1, dated August 23, 1973

65-89585 - Wing W44.1 AX 1319 I-1, dated August 1, 1974

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- 747-MD-572 Structural Arrangement Forward "A" Frame Support Orbiter 747 MOD, dated June 25, 1974
- 747-MD-461 General Arrangement 747 Space Shuttle Orbiter Carrier Aircraft (Piggyback Configuration), dated July 15, 1974
- 747-MD-576 Structural Arrangement Orbiter Aft Support, 747 MOD, dated August 1, 1974
- 1319-6, "Inbd Main Flap," dated 7-26-74
- 1319-15, "Wing Coves," dated 7-29-74
- 1319-24, "Outbd Fore-Flap," dated 8-5-74
- 1319-25, "Outbd Fore-Flap," dated 8-5-74
- 1319-33, "Inner Body Orbiter (Bal #660)" dated 8-13-74
- 1319-34, "Spoiler, dated 8-14-74
- 1319-35, "Balance Holder Orbiter (Bal #660)" dated 8-14-74
- 1319-36, "Rear Mtg. Parts Orbiter," dated 8-28-74
- 1319-37, "Aft Support and Balance Adapter Assy. Orbiter," dated 8-28-74
- 1319-38, "Inbd Flap Assy 20° F8.7," dated 8-17-74
- 1319-39, "Inbd Flap Brkts 20° F8.1," dated 8-19-74
- 1319-40, "Setting Temp L.E. Flaps," dated 8-17-74
- 1319-41, "Outbd Flap Brkts 20° F8.2," dated 8-20-74
- 1319-42, "Outbd Flap Assy 20° F<sub>8.2</sub>," dated 8-20-74
- 1319-43, "Fwd Orbiter Support Parts & Assy," dated 8-21-74
- 1319-44, "L. E. Kruger & Flap Instl.," dated 8-21-74
- 1319-45, "BTWT Orbiter Alone Mtg Parts & Assy," dated 8-22-74
- 1319-47, "Template-Stabilizer Tip Fin," dated 8-22-74

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- 1319-55, "Stabilizer Fins," dated 8-23-74
- 1319-57, "Stabilizer Fin Brkts," dated 8-24-74
- 1319-60, "Stabilizer Fin Instl," dated 8-26-74
- 1319-63, "Orbiter Modif, & Inner Body Instl," dated 8-29-74
- 1319-64, "Model Support Mat'l," dated 9-3-74
- D6-25552, "Model Geometry Estimated Loads and Stress Analysis, Model AX13181-1," dated 9-11-74

TEST : CA20			DATE: 11-20-74
	TEST CON	DITIONS	
MACH NUMBER	REYNOLDS NUMBER (per unit length)	DYNAMIC PRESSURE (pounds/sq.ft.)	STAGNATION TEMPERATURE (degrees Rankine)
0.3	1.93 x 10 <sup>6</sup> /FT	126	548
0.48	2.81 x 10 <sup>6</sup> /FT	293	559
0.50	2.94 x 10 <sup>6</sup> /FT	315	555
0.60	3.30 x 10 <sup>6</sup> /FT	422	563
		,	
		The state of the s	
BALANCE UTILIZED:		#660F 2.074 inch External Balance	dia.
		CITY	COEFFICIENT
	Orbiter	Carrier	TOLERANCE:
NF		10,000 lb.	
SF	1335 lb.	5,000 lb.	
AF	301.5 lb.	1,000 lb.	
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RM	and the second s	25,000 in1b.	
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	8	_			14	-5	9/3	0	5	0	0.6	)	7.5	10	10			752	755	
	8	MAKESON SERVICE			8	-5	9/3	0	5	0	0.6	2	7.5	0	10			798		
	9	3,			8	-5	0/3	0	5	١	5.6	)	7.5	10	10			753	754	
١	1 8	4		V	4	0	0/3	0	5	U	0.6	כ	7.5	0	0	Y		705	704	
		7		13 19			25		31		37	43	49		55		61		67	7
•		-		للبيب	•••		111				ICENTS		للب				100	AR (1)	16:48	(2)

3

TABLE II. (Continued)

DATA SET	AZO		_	CA	RRI		I		WOLK C	OLLATIO					1	Ctn :		
IDENTIFIER	CONFIG	URATION	- America	le.	<b>Marcheologicus</b>	-	Se	Ea	MACH	30	I do	ΔX	DY	DZ	6/10	Minute Street, St. Co., vol. 65, 65, 650.	1	٦
RGN085	747/1	0,5,	14	0	9/3	0	5	0	0.6	0	7.5	10	0	A	68	CONTRACTOR CONTRACTOR	agros nazran	٦
86			8	0	9/3	0	5	0	0.6	0	7.5	0	0		70	2 703		
87			8	0	9/3	0	5	0	0.6	U	7.5	10	0		68	3 684		
88			14	0	0/3	0	5	0	0.6	0	7.5	0	10		79	0 793		
89			4	0	0/3	0	5	0	0.6	0	7.5	10	10		74	8 751		
90			3	0	0/3	0	5	Э	0.5	0	7,5	0	10		79	9 796		
91			3	0	0/3	0	5	٦,	2,13	O	7.5	10	10		Control of the Contro	19 750	March 100 (1970)	
92			4	5	9/3	S	5	0	0.6	0	7,5	0	10		79	9 794		
93			4	5	0/3	0	5	0	000	0	7.5	10	10		75	6 759		
94			8	5	0/3	0	5	0	0.6	0	7,5	0	10		80	0 795		
95			8	5	0/3	0	5	0	0.6	0	7.5	10	10		75	7 758		
96			4	-5	0/3	0	5	0	0.6	5	7.5	0	10		80	4 805		
97			8	-5	9/3	0	5	٥	0.6	-5	7,5	0	10		811	810	- 1 To	
98			4	0	0/3	0	5	0	5.0	-5	7.5	0	10		80.	3 806		
99			8	0	0/3	0	5	0	1.6	-5	7.5	0	10		812	2 809		
100			4	5	93	0	5	0	5.6	-5	7,5	0	10		803	2 807		
101			8	5	9/3	0	5	0	0.6	-5	7.5	0	10		813	3 808		
V 102	1	1	4	-5	9/3	0	5	0	0.6	-5	7.5	0	10	V		915		
	7 13	1 19			25		31	100	37	43	49		55		61	67		-
a OR				لد					CENTS	<u></u>	<u>ш</u>	•••	ш.	•••		I IC.A	1 (2)	<u> </u>

DATA SET			Т	CA	RR	IER	Г			ORBIT	TER					. 0	60	
IDENTIFIER	CONFIG	URATION	વ્ય.	. Cc	Sev	80	હેહ	8a	MACH	80	\$0	ΔX	DY	02	6	10	14	
REN103	747/1	0,5,	4	0	0/3	0	5	0	0.6	-5	7.5	0	10	A		amin .	814	
104			4	-5	%3	0	5	0	0.6	5	0	10	0			830	935	
105		and the second of the second	8	-5	9/3	0	5	0	0.6	- 5	0	10	0			841	836	
106	The second second		4	-5	0/3	Ö	5	0	0.6	-5	0	0	10			844		
107			4	-5	9/3	0	5	0	0.6	-5	0	10	10			819	820	
108	AND THE PERSON NAMED IN	parental de la companya de la companya de la companya de la companya de la companya de la companya de la compa	8	-5	9/3	0	5	0	0.6	-5	0	10	10	age of		828	823	
109			4	0	9/3	0	5	0	0.6	-5	0	10	0			THE RESERVE OF THE PERSON NAMED IN	834	
110	Tanan da kanan da 1975.	1 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	8	0	9/3	0	5	)	0.6	5	0	10	٥			840	837	
111	The state of the s		4	0	0/3	0	5	0	0.6	- 5	)	0	10			843		
112		and the second	4	0	9/3	)	5	٥	0.6	- 5	0	10	10			819	158	
113			8	0	2/3	)	13	O	0.6	5	2	0	10				846	
114	And the second of the	apple and the state of the state of	9	0	13	0	5	0	0.6	5	0	10	10			827	824	
5			4	5	0/3	0	5	0	0.6	-5	0	10	0	The second		832	833	
116		The second secon	8	5	9/3	0	5	0	0.6	-5	O	10	0			839	838	* a - 40 (0.31)
117	read the Computer of 1997		4	5.	4/3	0	5	0	0.6	-5	0	0	10			845		
118			4	5	9/3	0	15	0	0.6	-5	0	10	10			817	822	
119			8	5	9/3	0	5	0	0.6	-5	0	10	10		8	326	825	
Y 120	nina da mara N	/	4	-5	0	0	5	0	0.6	-5	0	0	10	Ý		765	768	
	12	3 19			25		31		37	43	49		55		61		67	,

TABLE II. (Continued)

-	CAZO		~~			1/KU	N NU		DLLATION		IARY			_		15/75	
DATA SET	CONFIGURATION			Sev		Se	Sa		RBIT E.	do	AX	DY	42	6	10	14	
GN121	747/1 0,51	8	-5	0	0	5	0	0.6	-5	0	0	10	4	Ť	ACCURATION OF THE	767	
7 122		4	0	0	0	5	0	0.6	-5	0	0	10		İ	761	764	
123		8	0	0	0	5	0	0.6	-5	0	0	10	1		762	763	
124		4	5	0	o	5	0	0,6	-5	0	0	10			769	772	
125	4	3	5	0	0	5	0	0.6	-5	0	0	13			770	771	
126	747/1 025	4	-5	9/3	0	5	0	0.6	0	0	0	0			656		
127		4	-5	9/3	0	5	0	0.6	0	0	10	0			657		
128		4	-5	0/3	0	5	0	0.6	0	0	20	0			669		
129		4	0	9/3	0	5	0	0.6	0	Э	٥	0			652	653	
130		4	0	0/3	O	5	0	0.6	0	0	10	0			661	659	
131		4	0	0/3	0	5	0	0.6	0	0	20	0			665	666	
132		8	0	0/3	ن	5	0	2.6	0	0	0	0			655	654	
133		8	0	0/3	0	5	0	0.6	0	)	10	0			658	660	
134		8	0	9/3	0	5	0	0.6	0	0	40	2			668	667	
135		4	-5	Per Consession	0	5	0	0.6	0	0	O	10			728		
136		4	-5	9/3	0	5	ن	0.6	0	0	10	10	$\Box$		732		
137		4	0	9/3	0	5	0	0,6	0	2	0	10			727		
Y 138	<u> </u>	4	0	9/3	0	5	0	0.6	0	-	10	10	V		731		
a OR		9		25	<u></u>			37 LILLI CENTS	43	49 باب		55			AR (1)	67 I	121

		-			A	RIE	R	Г			ORBI	TER	-			I	. 0	40	
	TA SET	CONFIC	SURATION	a.	and the last	Sev	10000	Se	Sa	MACH	€0	90	DX	AY	ΔZ	6	ACCOUNTS NAMED IN	14-	
C	1139	747/1	025,	4	5		O	5	0	0.6	0	0	0	10	A		729		
T	140	1		4	5	9/3	0	5	0	0.6	0	0	10	10			733	Construction of the last of th	
T	141	747/1	0,51	4	0	10/13	0	5	0	0.6	0	0	0	0			and the factor of the factor o	708	
T	142			4	0	-1%-7	O	5	0	0.6	0	0	0	0	Ш		709	710	
T	143	4		4	0	43	15	5	၁	0,6	0	0	2	2			7/1	712	
†	144	747/1	025,	4	0	9/3	15	5	0	0,6	0	0	0	0			725		
T	145	747/1	0,51	4	0	0/3	0	0	0	0.6	9	0	0	2			719	CONTRACTOR OF THE PARTY OF THE	
1	146			1	0	0/3	၁	10	0	0.6	2)	9	J	U			714	715	
1	147			1	0	9/3	С	10	9	0.3	)	0	)	0			7/7		
T	148			4	0	03	5	10	0	5.7	)	9	3	Ġ,			716		
¥	149	Ý		4	0	0/3	Э	5	-10	0.6	9	3	٥	0	V		722	723	
				$\pm$															
				$\pm$															
				$\dagger$	L														
-		7	13 19	L	L_	25		31		37	43	<u>l</u>		55		61	<u> </u>	67	
			ىلىيىد			Mark and distributed on the Sale		LL.	OEFF	ICENTS						100	AR (1)	IC', AR	(2)

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# TABLE II. - DATA SET/RUN NUMBER COLLATION SUMMARY (Continued)

#### Symbol Definition

#### Orbiter

		<del></del>	and the second second second second			
01	=	Vertical tail off Tail cone on	(V <sub>8</sub> ) (TC <sub>5.1</sub> )	05	=	Vertical tail on (V <sub>8</sub> ) Tail cone off
02	=	Vertical tail off Tail cone off with strut S <sub>1</sub>	(v <sub>8</sub> )	<sup>0</sup> 6	=	MPS base plate off Vertical tail on (V <sub>8</sub> )
03	=	Vertical tail on Tail cone on	(V <sub>8</sub> ) (TC <sub>5.1</sub> )	<sup>0</sup> 7	=	MPS base plate off, S <sub>2</sub> cover plate #1 off, Vertical tail on
04	=	Tail cone on Vertical Tail simulating dummy strut	(TC <sub>5.1</sub> )			MPS base plate off, S <sub>2</sub> cover plate #2 off, Vertical tail on
						MPS base plate on, Vertical tail off with strut S <sub>2</sub>

#### Orbiter Support Strut

 $S_1$  = Orbiter support blade strut, upper entry position

 $S_2$  = Orbiter support blade strut, lower entry position

 $S_3$  = Orbiter dummy support blade strut

#### Carrier

747/0 = Carrier with spoilers and flaps retracted

747/1 = Carrier with spoilers deflected 45° and flaps retracted

### $\alpha$ , $\beta$ , and $\Delta Z$ Schedules

$$\triangle$$
  $\alpha_{c} = 0^{\circ}, 2^{\circ}, 4^{\circ}, 6^{\circ}, 8^{\circ}, 10^{\circ}$ 

$$\triangle$$
  $\beta_c = -10^{\circ}, -7^{\circ}, -5^{\circ}, -3^{\circ}, -2^{\circ}, -1^{\circ}, 0^{\circ}, +1^{\circ}, +2^{\circ}, +3^{\circ}, +5^{\circ}, +10^{\circ}$ 

$$\triangle \alpha_0 = 6^{\circ}, 8^{\circ}, 10^{\circ}, 12^{\circ}, 14^{\circ}, 16^{\circ}, 18^{\circ}$$

#### TABLE II. (Concluded)

$$\triangle$$
  $\Delta Z = 0^*, 3^*, 7.5^*, 15^*, 30^*, 45^*, 60^*$ 

$$\triangle$$
  $\beta_0 = 2.5^{\circ}, 0^{\circ}, -2.5^{\circ}, -5^{\circ}, -7.5^{\circ}, -10^{\circ}, -15^{\circ}$ 

\* minimum attainable

### Table III. - MODEL DIMENSIONAL DATA A. Carrier Model

MODEL COMPONENT: BO	DY - <sup>B</sup> 27.8		
GENERAL DESCRIPTION:_	Body 74-7 Project	with A.P.V.	
Model Scale: 0.03			
Drawing Number:	65-69716		
Dimensions:			
		Full Scale	Model Scale
Length, in		2702	81.06
Max. Width, in			7.66
Area			
Wetted, ft <sup>2</sup>			12.71

# Table IIIA - Continued.

MODEL COMPONENT: Fo	
GENERAL DESCRIPTION Clean Wing	
Flaps Up	

## Table IIIA - Continued

MODEL COMPONENT: Horizontal	Tail H <sub>15.0</sub>	5	· · · · · · · · · · · · · · · · · · ·	<del></del>
GENERAL DESCRIPTION: Horizon	tal Tail wi	th Vertical Fins	s on each	
Tip at Body B. L. 12.82				
Model Scale 0.03				
Drawing Number 1319-55 1/2	- 60			
Dimension:		Full Scale	Mod	del Scale
EXPOSED DATA (one side)				
Area-ft <sup>2</sup>		200		

#### Table IIIA - Continued.

MODEL COMPONENT: "25		<del> </del>	
GENERAL DESCRIPTION: Inboard 747, JT9D nacelle	strut		
Model Scale: 0.03		· · · · · · · · · · · · · · · · · · ·	
Dimensions	Full Scale		Model Scale
Wing B.L. of nacelle C <sub>L</sub> , in. Cont angle deg. inboard	470.0		14.100

MODEL COMPONENT: M26	<del>, , , , , , , , , , , , , , , , , , , </del>		
GENERAL DESCRIPTION: Outboard	747, JT9D		
Strut			
Model Scale: 0.03			
Drawing Number: 937-590			
<u>Dimensions</u>		Full Scale	Model Scale
W L of C <sub>1</sub> , in			25.020
Cant angle, deg inboard		2	

# Table IIIA - Continued.

MODEL COMPONENT:	<sup>N</sup> 57								<del></del>
GENERAL DESCRIPTION_	Inboard	Fan	Cow1	and	Primary	747	Nacel	le,	
Flow Through Typ	e								
Model Scale: 0.03		<del>-</del> '.							
Drawing Number: S.O.	1007-96	-97							

## Table IIIA - Continued

MODEL COMPONENT:	N <sub>58</sub>	-						
GENERAL DESCRIPTION	N: Outboard	Fan	Cow1	and	Primary	747	Nacelle,	
Flow Through Typ	oe							
Model Scale: 0	.03							
Drawing Number 5.0	. 1.007-96,-	-97						

## Table III A - Continued.

Chord

MODEL COMPONENT: Spoilers S1-12		
GENERAL DESCRIPTION: Multi-panel flight	spoilers. Four outb	oard and
two inboard spoiler per side. Subscrip	t denotes spøiler pa	nel <sup>S</sup> lis
the most outboard L.H. panel and S12 is	most outboard R.H.	panel,
747 Model Scale: 0.03 Mo	odel: 1065	
Drawing No.: 65-71450, S.O. 1065-51,	<u>-59, -81, -173</u>	
Dimensions: (One panel)	Full Scale Ft.	Model Scale
Outboard $S_{1-4}$ and $S_{9-12}$ (Ft <sup>2</sup> )	21.48	0.019 ft <sup>2</sup>
Span (equivalent)	6.25	2.25
Chord	3.44	1.238
Inboard $S_{5-6}$ and $S_{7-8}$ (Ft <sup>2</sup> )	35.31	7
Span (equivalent)	7.50	2.70
Chord	4.71	1.696

## Table III A - Continued

MODEL COMPONENT: T19			
GENERAL DESCRIPTION:	Flap Track F	airings,	
4 on each side			
Model Scale: 0.03			
Drawing Number: S.O. 1	007-403		

DIMENSIONS	Full Scale	Model Scale
WBL of Track no. 1, in.	235.3	7.06
2, in.	353.0	10.59
3, in.	652.0	19.56
4, in	743.6	22.31
Distance from wing	50.0	1.5
Trailing edge to:		

Track Trailing

edge, in.

#### Table IIIA - Continued.

MODEL COMPONENT: Vertical V9.1		
GENERAL DESCRIPTION: Swept Vertical Tail		
Model Scale: 0.03		
Drawing Number: 65-6.9716; 1007-26,-610;	937-319	
Dimensions:	<u>Full Scale</u>	Model Scale
TOTAL DATA		
Area (Theo) Ft <sup>2</sup>	630.0	.567
Span (theo) - In.	_386.5	11.595
Sweep-Back Angles, Degrees Leading Edge	50.12	50.12
Chords:		
Root (Theo) WP-in.	461.67	13.85
Tip (Theo) WP-in.	157.0	4.71
Cup Sta of 25 MAC	2529.6	75.888

## Table IIIA - Continued.

MODEL COMPONENT: WING-W44.1		
GENERAL DESCRIPTION: Swept 747 Wing		
Model Scale: 0.03		
7.000		
Test No.	DWG. No. 65-89585	
<u>Dimensions:</u>	Full Scale	Model Scale
<u>Total Data</u>		
Area (Theo.) Ft <sup>2</sup> Planform	5500	4.95
Span (Theo In.	2348.04	70.441
Aspect Ratio	6.96	6.96
Incidence Angle, degrees		7
Chords:		
MAC	327.78	9.833
Fus. Sta. of .25 MAC	1339.90	40.197
W.P. of .25 MAC	190.80	5.723

MODEL COMPONENT: ATTACH STRUCTURE - AT38

GENERAL DESCRIPTION: Orbiter to carrier forward attach

struts.

MODEL SCALE: 0.030

DRAWING NO.: BOEING 1319-43

	SCA	LE
DIMENSIONS:	FULL	MODEL
AT <sub>38</sub>	15.6	0.465
AT'38.1	91.67	2.75
AT <sub>38.2</sub>	75.00	2.25
AT38.2A	75.0	2.25
AT38.3	ROD REMOVED	ROD REMOVED

#### TABLE IIIA - Concluded.

MODEL COMPONENT: ATTACH STRUCTURE - AT39

GENERAL DESCRIPTION: Orbiter to carrier aft attachment, pitch

adjustable from 0 to 10 deg.

MODEL SCALE: 0.030

DRAWING NO.: Boeing 50 1319-37.

DIMENSIONS:	FULL SCALE	MODEL SCALE
Pivot location:		
In., X <sub>C</sub>	400.0	12.0
In., Z <sub>C</sub>	160.7	4.821
Equivalent Span (At 0 deg iorb):		
Centerline orbiter	521.0	15.63

# TABLE III MODEL DIMENSIONAL DATA B. Orbiter

MODEL COMPONENT : BODY - B26		
GENERAL DESCRIPTION: Configuration 1	40A/B orbiter	fuselage.
NOTE: B26 is identical to B24 except unde	erside of fusela	ige has been
refaired to accept W <sub>116</sub> .		
MODEL SCALE: 0.030 MODEL	DWG: SS-A00	0147, Release 12
DRAWING NUMBER: VL70-000143B, -00 VL70-000140A, -00	0200, -000205, 00140B	-006089, -000145
DIMENSIONS:  Length (OML: Fwd Sta $X_0 = 235$ ), In Length (IML: Fwd Sta $X_0 = 238$ ), In		MODEL SCALE 38.799 38.709
Max Width (At $X_0 = 1528.3$ ), In.	264.0	7.920
Max Depth (At $X_0 = 1464$ ), In.	250.0	7.500
Fineness Ratio Area - Ft <sup>2</sup>	0.264	0.264
Max. Cross—Sectional	340.88	0.307
Planform		
Wetted Base		

MODEL COMPONENT : CANOPY - C9		
GENERAL DESCRIPTION : Configuration	3A. Canopy u	sed with fuselage B
MODEL SCALE: 0.030 MOI	DEL DWG: SS-	A00147, Release 12
DRAWING NUMBER: VL70-000143A	Particology and the second control of the se	
DIMENSIONS:	FULL SCALE	MODEL SCALE
Length ( $X_0 = 434.643$ to 578), In.	143,357	4.301
Max Width (At $X_0 = 513.127$ ), In.	152.412	4.572
Max Depth (At $X_0 = 485.0$ ), In.	25.00	0.750
Fineness Ratio		
Area		*****
Max. Cross—Sectional		
Planform		
Wetted		·
Base		

MODEL COMPONENT: SLOTTED ELEVON (	6 INCH GAP) - E43	
GENERAL DESCRIPTION: Configuration 1404	•	
NOTE: E43 is a slotted version of E24	. Data are for one	side
MODEL SCALE: 0.030		
DRAWING NUMBER: VL70-00020	0, -006089, -00609	2
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Area - Ft <sup>2</sup>	210,00	0, 189
Span (equivalent) , In.	349.2	10. 476
Inb'd equivalent chord , In.	118.004	3.540
Outb'd equivalent chord, In.	55. 192	1.656
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.2096	0.2096
At Outb'd equiv. chord	0.4004	0,4004
Sweep Back Angles, degrees		
Leading Edge	0.00	0.00
Tailing Edge	-10.056	-10.056
Hingeline (Product of Area & c)	0.00	0.00
Area Moment (Normal×toxhingextine),	Ft <sup>3</sup> 1587.25	0.043
Mean Aerodynamic Chord, In.	90.7	2. 721

MODEL COMPONENT : BODY FLAP -	F <sub>8</sub>	
GENERAL DESCRIPTION : Configuratio	n 140A/B orbite	r body flap
NOTE: Hingeline located at X = 1	528.3, Z = 284	3
MODEL SCALE: 0.030 M	ODEL DWG: SS	-A00147, Release
DRAWING NUMBER: VL70-000140A,		
DIMENSIONS :	FULL SCALE	MODEL SCALE
Length ( $X_0 = 1520 - 1613$ ) In.	93.00	2,79
Max Width , IN.	262.00	7.86
Max Depth ( $X_0 = 1520$ ), In.	23.00	0.69
Fineness Ratio		
Area - Ft <sup>2</sup>		
Max. Cross—Sectional		
Planform	150.525	0.1355
Wetted		
Base	41.847	0.0377

MODEL COMPONENT : OMS POD - M16		
GENERAL DESCRIPTION : Configuratio	n 140C	
Orbiter OMS pod - Short pod		
MODEL SCALE: 0.030		
DRAWING NUMBER: VL70-008401, -00	8410	
DIMENSIONS:	FULL SCALE	MODEL SCALE
Length (OMS Fwd Sta. Xo=1310.	5) 258.50	7.755
Max Width(At $X_0 = 1511$ ), In.	136.8	4.104
Max Depth (At $X_0 = 1511$ ), In.	74.70	2.241
Fineness Ratio	2.484	2.484
Area - Ft <sup>2</sup>		
Max. Cross-Sectional	58.864	0.053
Planform		
Wetted		
Base		

MODEL COMPONENT: MPS NOZZLES - N24	
GENERAL DESCRIPTION: Configuration	n 140A/B orbiter MPS nozzles
MODEL SCALE: 0.030	MODEL DWG: SS-A00147, Release 1
DRAWING NUMBER: VL70-005030A, -00	00140A
DIMENSIONS:	FULL SCALE MODEL SCALE
MACH NO.	
Length - In.  Gimbal Point to Exit Plane  Throat to Exit Plane	$\begin{array}{c c}     \hline                                $
Diameter - In. Exit Throat Inlet	91.00 2.73
Area - ft <sup>2</sup> Exit Throat	45.166 0.0407
Gimbal Point (Station) - In.	
Upper Nozzle X Y Z	$\begin{array}{c cccc} 1445.00 & 43.35 \\ \hline 0.0 & 0.0 \\ \hline 443.00 & 13.29 \end{array}$
Lower Nozzles X Y Z	$ \begin{array}{rrrr} 1468.170 & 44.045 \\                                    $
Null Position - Deg.	
Upper Nozzle Pitch	<u>16</u> <u>16</u>
Yaw	0
Lower Nozzle Pitch Yaw	$\begin{array}{cccc}  & 10 & & 10 \\ \hline  & 3.5 & & 3.5 \end{array}$

	IS NOZZLES - N	<u> </u>		
ENERAL DESCRIPTION:	Configuration 1	40A/B orbiter	OMS No.	zzies
ODEL SCALE: 0.030	· · · · · · · · · · · · · · · · · · ·			
RAWING NUMBER: VI	70-000140A (Loc	cation), SS-A0	0106, Re	lease 5 (Conto
imensions:		<u>FUL</u>	L SCALE	MODEL SCALE
MACH NO.				
Length - In. Gimbal Point t Throat to Exit				
Diameter - In. Exit Throat Inlet				
Area - ft <sup>2</sup> Exit Throat				
Gimbal Point (Stat Left XXXXX Nozzle	cion) · In.			
X o Y o Z o			518.00 -88.0 492.0	45. 54 -2. 64 14. 76
Right kamer Nozzles	ORIGINAL PAG	ie is	<b>710</b> 0	Ar ru
Х о <b>Y</b> о Zo	ORIGINAL PAG OF POOR QUA	TILES	518.0 88.0 492.0	45.54 2.64 14.76
Null Position - D Left XFFEX Nozzle	<b>eg.</b>		-8	±8
	eg.	13 <sup>0</sup> 17 Outb <u>'d</u> ,	=8 , 2 <sup>0</sup> 30' In	
Left XXXX Nozzle Pitch			<u>, 2 <sup>0</sup>30' In</u> ±8	b'd Same ±8

MODEL COMPONENT: RUDDER - R5		
GENERAL DESCRIPTION: Configuration 140C o	erbiter rudder (	identical to
configuration 140A/B rudder)		
MODEL SCALE: 0.030		
DRAWING NUMBER: VL70-000146B,	-000095	
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Area - Ft <sup>2</sup>	100.15	_0.090
Span (equivalent), In.	201.00	6. 03
Inb'd equivalent chord, In.	91.585	2. 748
Outb'd equivalent chord, In.	50.833	1.525
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.400	0.400
At Outb'd equiv. chord	0,400	0.400
Sweep Back Angles, degrees		
Leading Edge		
Tailing Edge	26. 25	26. 25
Hingeline (Product of area and c)	34.83	34.83
Area Moment (Mermanxtexappexkine), Ft3	610.92	0.0165
Mean Aerodynamic Chord, Inches	73.2	2. 196

MODEL COMPONENT : ORBITER TA	ALLCONE - TC <sub>5</sub>	1
GENERAL DESCRIPTION : Fairing moun	ited on orbiter fi	uselage base for
ferry missions.		
MODEL SCALE: 0.030		
DRAWING NUMBER: Boeing Dwg No.: 1	319-71	
DIMENSIONS:	FULL SCALE	MODEL SCALE
Length	445.83	13.375
Max Width	303.33	9.10
Max Deptox Height	265.00	7.95
Fineness Ratio		
Area - Ft <sup>2</sup> Projected frontal area Max. Cross-Sectional	324.105	0.2917
Planform Wetted		
Wetted		

MODEL COMPONENT: VERTICAL - V8		*
GENERAL DESCRIPTION: Configuration 140A	/B orbiter vertical	tail.
dia dia dia dia dia dia dia dia dia dia		
		and the same of th
MODEL SCALE: 0.030 MODE	EL DWG: SS-A0014	18, Release 6
DRAWING NUMBER: VL70-000146A		
DIMENSIONS:	FULL SCALE	MODEL SCALE
TOTAL DATA		
Area (Theo) - Ft <sup>2</sup> Planform Span (Theo) - In. Aspect Ratio Rate of Taper Taper Ratio Sweep-Back Angles, Degrees. Leading Edge Trailing Edge 0.25 Element Line  Chords: Root (Theo) WP Tip (Theo) WP MAC Fus. Sta. of .25 MAC W.P. of .25 MAC	413, 253 315, 720 1, 675 0, 507 0, 404  45, 000 26, 25 41, 13  268, 50 108, 47 199, 81 1463, 35 635, 52	0.372 9.472 1.675 0.507 0.404 45.000 26.25 41.13 8.055 3.254 5.994 43.901 19.066 0.00
B.L. of .25 MAC  Airfoil Section Leading Wedge Angle - Deg. Trailing Wedge Angle - Deg. Leading Edge Radius	10,00 14,92 2,00	10.00 14.92 0.060
Void Area	13.17	0,119
Blanketed Area	0.00	0.00



MODEL SCALE: 0.030   DMG. NO. VL70-000140A, -	ODEL COMPONENT: WING-W <sub>116</sub> ENERAL DESCRIPTION: Configuration 4		
MODEL SCALE: 0.030   DMG. NO. VL70-000140A,		D'1 1 1	1
MODEL SCALE: 0.030  EST NO. DWG. NO. VL70-000140A, -  IMENSIONS: FULL-SCALE MODEL SCALE  TOTAL DATA Area (Theo.) Ft2 Planform Span (Theo In. 936.68 28.10 Aspect Ratio 2.265 2.265 Rate of Taper 1.1.77 1.177 Taper Ratio 0.200 0.200 0.200 Dihedral Angle, degrees 3.5.00 3.500 Incidence Angle, degrees 0.500 0.500 Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge 45.00 45.00 0.500 Chords: Root (Theo) B.P.O.O. 10.056 10.056 0.25 Element Line 35.209 35.209 Chords: Root (Theo) B.P.O.O. 689.24 20.677 Tip. (Theo) B.P. 137.85 4.136 MAC 0.000 FIGURAL PAGE B. 1136.83 34.105 W.P. of .25 MAC POOR QUALITY  EXPOSED DATA Area (Theo) Ft2 Span, (Theo) In. BP108 Aspect Ratio 1.576 Taper Ratio Chords Root BP108 Root BP108 Tip 1.00 b 1.576 Fus. Sta. of .25 MAC 2.059 2.059 Taper Ratio Chords Root BP108 Tip 1.00 b 137.85 4.136 Fus. Sta. of .25 MAC 2.059 2.059 W.P. of .25 MAC 3.05.25 MAC 2.059 1.576 Aspect Ratio 1.576 Fus. Sta. of .25 MAC 2.059 2.059 Taper Ratio Chords Root BP108 Tip 1.00 b 137.85 4.136 AC 2.059 2.059 Taper Ratio 2.059 1.576 ASPECT RATIO 1.58 MAC 2.059 2.059 Taper Ratio 2.059 1.37.85 4.136 ASPECT RATIO 1.55 MAC 2.059 2.059 Taper Ratio 2.059 1.37.85 4.136 ASPECT RATIO 1.55 MAC 2.059 2.059 Taper Ratio 2.059 1.37.85 4.136 ASPECT RATIO 1.55 MAC 2.059 2.059 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.37.85 4.136 ASPECT RATIO 1.00 b 1.00			le is along
IMENSIONS:   FULL-SCALE   MODEL SCALE	trailing edge of wing. Geometric twist = 0	),	
TOTAL DATA	MODEL SCALE: 0.030		
TOTAL DATA   Area (nec.)   Ft2   Planform   2690.00   2.421     Span (Theo In.   936.68   28.10     Aspect Ratio   2.265   2.265     Rate of Taper   1.177   1.177     Taper Ratio   0.200   0.200   0.200     Dihedral Angle, degrees   3.500   3.500     Incidence Angle, degrees   0.500   0.500     Aerodynamic Twist, degrees   45.00   45.00     Area (Theo Barren   10.056   10.056   10.056     0.25 Element Line   35.209   35.209     Chords:   Root (Theo) B.P.O.O.   689.24   20.677     Tip, (Theo) B.P.   137.85   4.136     MAC   Fus. Sta. of .25 MAC   PAGE   B   136.83   34.105     W.P. of .25 MAC   POOR   PAGE   B   182.13   5.464     EXPOSED DATA   Area (Theo)   Ft2   1751.50   1.576     Span, (Theo) In. BP108   720.68   21.620     Aspect Ratio   2.059   2.059     Taper Ratio   2.059   2.059     Taper Ratio   2.059   2.059     Taper Ratio   392.83   11.785     Fus. Sta. of .25 MAC   137.85   4.136     Root BP108   562.09   16.863     Tip 1.00 b   137.85   4.136     MAC   392.83   11.785     Fus. Sta. of .25 MAC   185.98   35.579     W.P. of .25 MAC   294.30   8.829     B.L. of .25 MAC   294.30   8.829     Airfoil Section (Rockwell Mod NASA)   XXXX-64     Root b   2	EST NO.	DWG. NO. VL	70-000140A, -0
Area (.neo.) Ft2 Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Aerodynamic Twist, degrees Leading Edge Trailing Edge O.25 Element Line Chords: Root (Theo) B.P.O.O. Tip, (Theo) B.P. Area (Theo) Ft2 Span, (Theo) Ft2 Span, (Theo) In. BP108 Aspect Ratio Chords Root BP108 Tip 1.00 b Area (Theo) Ft2 Fus. Sta. of .25 MAC Aspect Ratio Chords Root BP108 Tip 1.00 b Root BL. of .25 MAC Airfoil Section (Rockwell Mod NASA)  XXXX-64 Root b Tip b Tip b Tip b Tip b Tip b Tip C.25 Ides  Aspect Ratio Chords Root BP108 Tip b Tip b Tip b Tip b Tip c Tip b Tip b Tip C.25 Ides  Aspect Ratio Chords Root BP108 Tip b Tip b Tip b Tip b Tip b Tip C.25 Ides  Aspect Ratio Chords Root BP108 Tip b Tip b Tip b Tip b Tip C.25 Ides	IMENSIONS:	FULL-SCALE	MODEL SCALE
Span (Theo In.   936.68   28.10   28.20   28.10   22.265   22.26	TOTAL DATA		
Span (Theo In.	Area (Theo.) Ft2		
Aspect Ratio Rate of Taper Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) B.P.O.O. Tip, (Theo) B.P. MAC Fus. Sta. of .25 MAC B.L. of .25 MAC Aspect Ratio Aspect Ratio Chords  EXPOSED DATA Area (Theo) In. BP108 Aspect Ratio Chords Root BP108 Root BP108 Root Data Root Data Area Sta. of .25 MAC Root BP108 Root Data Area Chords Root BP108 Root Data Area Chords Root BP108 Root Data Area Chords Root BP108 Root Data Area Chords Root BP108 Root Data Area Chords Root BP108 Root Data Area Chords Root BP108 Root			2.421
Rate of Taper Taper Ratio Dihedral Angle, degrees O. 200 Dihedral Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge Trailing Edge O.25 Element Line Chords: Root (Theo) B.P.O.O. Tip, (Theo) B.P.  W.P. of .25 MAC Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Root BP108 Root BP108 Tip 1.00 b L. of .25 MAC Root BP108 Root DATA Area (Theo) Ft Root OR CALL PAGE Aspect Ratio Chords Root BP108 Root BP108 Tip 1.00 b L. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA) XXXXX-64 Root b  Root D I. D I. Sides  D. 120 Data for (1) of (2) Sides		936.68 2.265	28, 10
Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) B.P.O.O. Tip, (Theo) B.P. MAC Fus. Sta. of .25 MAC B.L. of .25 MAC Area (Theo) Ft Span, (Theo) In. BP108 Appect Ratio Chords Root BP108 Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 b Tip 1.00 c Tip 2 Tip 5 Tip 5 Tip 2 Tip 5 Tip 5 Tip 2 Tip 5 Tip 5 Tip 1.00 c Tip			
Dinedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) B.P.O.O. B.P. OOR POOR PAGE IS BL. of .25 MAC Aspect Ratio Chords Root BP108 Root BP108 Tip 1.00 b L. of .25 MAC Airfoil Section (Rockwell Mod NASA) XXXXX-64 Root b =  Data of .25 MAC  Root BP108			
Aerodynamic Twist, degrees  Sweep Back Angles, degrees  Leading Edge Trailing Edge 0.25 Element Line Chords:  Root (Theo) B.P.O.O. Tip, (Theo) B.P.  MAC Fus. Sta. of .25 MAC Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Aspect Ratio Chords  Root BP108 Root BP108 Root BP108 Root BP108 Root BP108 Root BP108 Root BP108 Area (Theo) BP108 ARC Area (Theo) BP108 ARC Area (Theo) BP108 ARC ARC BROOT BP108 ARC ARCA BROOT BP108 ARCA BROOT		3,500	
Sweep Back Angles, degrees Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) B.P.O.O. Tip, (Theo) B.P. MAC Fus. Sta. of .25 MAC Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Root BP108 Root BP108 Root BP108 Fus. Sta. of .25 MAC Root BP108 Root BP108 Root BP108 Fus. Sta. of .25 MAC Root BP108 Root			
Leading Edge Trailing Edge 0.25 Element Line Chords:  Root (Theo) B.P.O.O. Tip, (Theo) B.P. MAC Fus. Sta. of .25 Mac B.L. of .25 MAC Area (Theo) Ft Span, (Theo) In. BP108 Aspect Ratio Root BP108 Root BP108 Tip 1.00 b Taper Ratio Chords Root BP108 Root BP108 Fus. Sta. of .25 MAC Root BP108 Fus. Sta. of .25 MAC Root BP108 Fus. Sta. of .25 MAC Root BP108 Fus. Sta. of .25 MAC B.L. of .25 MAC Root BP108 Fus. Sta. of .25 MAC B.L. of .25 MAC Root BP108 Fus. Sta. of .25 MAC B.L. of .25 MAC Root BP108 Fus. Sta. of .25 MAC B.L. of .25 MAC B.L. of .25 MAC Root BP108 Fus. Sta. of .25 MAC B.L. of			<del></del>
Trailing Edge 0,25 Element Line Chords:  Root (Theo) B.P.0.0.  Root (Theo) B.P.  ORIGINAL PAGE Span, (Theo) In. BP108 Aspect Ratio Chords  Root BP108 Root BP108 Root b =  Tip b =  Tip b =  Tip b =  To compare the side of t		AE 00	45 00
0.25 Element Line Chords: Root (Theo) B.P.0.0. Root (Theo) B.P.0.0. Tip, (Theo) B.P.  MAC Fus. Sta. of .25 MAG B.L. of .25 MAC Area (Theo) Span, (Theo) Span, (Theo) Root BP108 Root BP108 Root BP108 Tip 1.00 b Root BP108 Root BP108 Aspect Ratio Chords Root BP108 Root BP108 Root BP108 Fus. Sta. of .25 MAC Root BP108 Root BP108 Aspect Root Root Root Root Root Root Root Ro			
Chords:  Root (Theo) B.P.O.O.  Tip, (Theo) B.P.  ORIGINAL PAGE IS  B.L. of .25 MAC  Aspect Ratio  Taper Ratio  Chords  Root BP108  Tip 1.00 b  MAC  Fus. Sta. of .25 MAC  Root .25 MAC  Root .25 MAC  Aspect Ratio  Tip 1.00 b  MAC  Fus. Sta. of .25 MAC  Root BP108  Acc  Fus. Sta. of .25 MAC  Root BP108  Tip 1.00 b  Aspect Ratio  Tip 1.00 b  N.P. of .25 MAC  B.L. of .25 MAC  Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b  Tip b  Tip b  On 120  On 120  On 120			
Root (Theo) B.P.O.O. Tip, (Theo) B.P. Tip, (Theo) B.P.  MAC  PAC  PAC  PAC  PAC  PAC  PAC  PAC			
Tip, (Theo) B.P.  MAC  MAC  Pus. Sta. of .25 MAC  W.P. of .25 MAC  B.L. of .25 MAC  Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108  Aspect Ratio  Taper Ratio  Chords  Root BP108  Tip 1.00 b  MAC  MAC  MAC  Area (Theo) Exposed MAC  Root BP108  Tip 1.00 b  MAC  ARC  Fus. Sta. of .25 MAC  Arca (Theo) BP108  Aspect Ratio  Tip 1.00 b  MAC  Fus. Sta. of .25 MAC  BL. of .25 MAC  Arca (Theo) BP108  Arca (Theo) In. BP108  Aspect Ratio  Tip 1.00 b  MAC  Fus. Sta. of .25 MAC  BL. of .25 MAC  Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b  Tip b  Online  Online  At 136  At 14.244  At 14.244  At 136  At 136  At 136  At 136  At 136  At 136  At 136  At 137  At 1751  At 1751  At 1751  At 1751  At 1761  At 136  At 1	Root (Theo) B.P.O.O.	689.24	20.677
Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Aspect Ratio Aspect Ratio Taper Ratio Chords Root BP108 Tip 1.00 b  MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b =  Tip b =  Ontage  1751.50 1.576 21.576 22.059 22.059 2.059	Tip, (Theo) B.P.	137.85	4.136
Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Aspect Ratio Aspect Ratio Taper Ratio Chords Root BP108 Tip 1.00 b  MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b =  Tip b =  Ontage  1751.50 1.576 21.576 22.059 22.059 2.059	MAC URIGIAT		والمرسوب والمرابع والمرابع والمرابع والمرابع والمرابع والمرابع والمرابع والمرابع والمرابع والمرابع والمرابع وا
Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Aspect Ratio Aspect Ratio Taper Ratio Chords Root BP108 Tip 1.00 b  MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b =  Tip b =  Ontage  1751.50 1.576 21.576 22.059 22.059 2.059	W D of 25 MAC PACE		
Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Aspect Ratio Aspect Ratio Taper Ratio Chords Root BP108 Tip 1.00 b  MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b =  Tip b =  Ontage  1751.50 1.576 21.576 22.059 22.059 2.059	B.1. of 25 MAC QUATER		
Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108 Aspect Ratio Aspect Ratio Taper Ratio Chords Root BP108 Tip 1.00 b  MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b =  Tip b =  Ontage  1751.50 1.576 21.576 22.059 22.059 2.059	EYPOSED DATA	100, 10	J. 707
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Area (Theo) Ft2	1751.50	1.576
Aspect Ratio $2.059$ $2.059$ Taper Ratio $0.245$ $0.245$ Chords  Root BP108 $562.09$ $16.863$ Tip 1.00 b $137.85$ $4.136$ MAC $392.83$ $11.785$ Fus. Sta. of .25 MAC $185.98$ $35.579$ W.P. of .25 MAC $294.30$ $8.829$ B.L. of .25 MAC $251.77$ $7.553$ Airfoil Section (Rockwell Mod NASA)  XXXX-64  Root b = $0.113$ $0.113$ Tip b = $0.120$ $0.120$	Span. (Theo) In. BP108	والمستقصين والمستقد و	CONTRACTOR OF THE PERSON NAMED IN COLUMN 1
Taper Ratio	Aspect Ratio		
Chords  Root BP108 Tip 1.00 b  MAC  Fus. Sta. of .25 MAC  W.P. of .25 MAC  B.L. of .25 MAC  Airfoil Section (Rockwell Mod NASA)  XXXXX-64  Root b =   Tip b =   Ontage    Ontage    16.863  137.85  4.136  392.83  11.785  1185.98  35.579  294.30  8.829  7.553  Ontage    Ontage			
Tip 1.00 $\frac{b}{2}$ MAC  MAC  Fus. Sta. of .25 MAC  W.P. of .25 MAC  B.L. of .25 MAC  Airfoil Section (Rockwell Mod NASA)  XXXX-64  Root $\frac{b}{2}$ Tip $\frac{b}{2}$ Data for (1) of (2) Sides $ \frac{137.85}{392.83}  \frac{11.785}{185.98}  35.579  294.30  8.829  7.553  0.113  0.113$	Chords		
MAC $\frac{2}{\text{Fus. Sta. of .25 MAC}}$ Fus. Sta. of .25 MAC $\frac{1185.98}{294.30}$ B.L. of .25 MAC $\frac{294.30}{251.77}$ Airfoil Section (Rockwell Mod NASA)  XXXX-64  Root $\frac{b}{2}$ Tip $\frac{b}{2}$ Data for (1) of (2) Sides			
Fus. Sta. of .25 MAC $\frac{1185.98}{294.30}$ $\frac{35.579}{8.829}$ B.L. of .25 MAC $\frac{294.30}{251.77}$ $\frac{8.829}{7.553}$ Airfoil Section (Rockwell Mod NASA)  XXXX-64  Root $\frac{b}{2}$ $\frac{0.113}{200}$ $\frac{0.120}{200}$ Data for (1) of (2) Sides	1γ <b>ρ. Ι. το</b> Ε. Το Ε. Ε. Ε. Ε. Ε. Ε. Ε. Ε. Ε. Ε. Ε. Ε. Ε.	_137.85_	4.136_
Fus. Sta. of .25 MAC $ \begin{array}{ccccccccccccccccccccccccccccccccccc$		392, 83	11.785
B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)		1185.98	35.579
Airfoil Section (Rockwell Mod NASA)  XXXX-64  Root $\frac{b}{2}$ = $\frac{0.113}{0.113}$ 0.113  Tip $\frac{b}{2}$ = $\frac{0.120}{0.120}$ 0.120  Data for (1) of (2) Sides			
		251.77	7.553
Tip $\frac{b}{2}$ = 0.120 0.120  Data for (1) of (2) Sides			
Tip $\frac{b}{2}$ = 0.120 0.120  Data for (1) of (2) Sides	Root <u>b</u> =	0.113	0.113
		0.120	0.120
	Data for (1) of (2) Sides		
	Leading Edge Cuff ,		
Planform Area Ft <sup>2</sup> 0,102	Planform Area Ft		
Leading Edge Intersects Fus M. L. 0 Sta 500.00 15.00 Leading Edge Intersects Wing 0 Sta 1024.00 30.72			

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#### Table IV. CA20 DATASET DESCRIPTION (Raw Data)

DATASET TYPE	DESCRIPTION
RGNXXX	Longitudinal coefficient schedule for 747 carrier balance data which contain "standard" wind tunnel corrections.
AGNXXX	Lateral coefficient schedule for 747 carrier balance data which contain "standard" wind tunnel corrections.
BGNXXX	Longitudinal coefficient schedule for orbiter balance data which contain "standard" wind tunnel corrections.
CGNXXX	Lateral coefficient schedule for orbiter balance data which contain "standard" wind tunnel corrections.
DGNXXX	Pressure coefficient data as follows:  Q(PSF) - dynamic pressure, psf PB1, PB2, PB3 - orbiter base pressure coefficients PCAV - orbiter cavity pressure coefficient PSC - carrier cavity pressure coefficient LHLS, RHLS - left and right hand pressure coefficients in proximity to orb. vert. tail for blade/sting support system. LHVERT, RHVERT - identical to LHLS and RHLS but for base sting support system.

Table V.
CA20 COEFFICIENT SCHEDULE
(Raw Data)

<u> </u>				<u> </u>				Coeffi	cients				
Dataset Type	Dataset <b>•</b> Sequence	1st ID	2nd ID	1	2	3	4	5	6	7	8	9	10
	034-036	МАСН	ALPHAW	ВЕТА	Q(PSF)	CL	CD	CLM , .	СҮ	CLN	CSL		
	037	масн	ALPHAO	ALPHAW	ВЕТА	DY	DZ	CL	CD	CLM	СҮ	CLN	CSL
	038 & 039	масн	вета	ALPHAW	ALPHAO	DY	DZ	CL	CD	CLM	CY	CLN	CSL
	040-149	ALPHA0	DZ	МАСН	ΧΩ	DY	ветао	РНІ	ALPHAW	ВЕТА	CL	CD	CLM
AGNXXX	040-149	ALPHAO	DZ	MACH	DX	DY	ВЕТАО	PHI	ALPHAW	ВЕТА	СУ	CLN	CSL
BGNXXX	001-011 & 015-033 & 037	MACH	ALPHA0	ВЕТАО	PHI	Q(PSF)	CL	CD	CLM	ĆΥ	CLN	CSL	
	012 & 013	МАСН	DZ	ALPHAO	ветао	PHI	Q(PSF)	CL	CD	CLM	СҮ	CLN	CSL
	014	MACH	BETA0	ALPHA0	PHI	Q(PSF)	CL	CD	CLM	CY	CLN	CSL	
	038 & 039	MACH	вета	ALPHAW	ALPHAO	DY	DZ	CL	CD	CLM	СҮ	CLN	CSL
	040-149	ALPHA0	DZ	MACH	DX	DY	BETAO	PHI	ALPHAW	вета	CL	CD	CLM
CGNXXX	040-149	ALPHA0	DZ	MACH	DX	DY	BETAO	PHI	ALPHAW	вета	СҮ	CLN	CSL
DGNXXX	001-011 & 015-019 & 037	MACH	ALPHA0	Q(PSF)	PB1	PB2	PB4	LHLS	RHLS	PCAV	1		
	012 & 013	MACH	DZ	Q(PSF)	PB1	PB2	PB4	LHLS	RHLS	PCAV			
	014	MACH	вета0	Q(PSF)	PB1	PB2	PB4	LHLS	PHLS	PCAV			
	020-033	МАСН	ALPHA0	Q(PSF)	PB1	PB2	PB4	LHVERT	RHVERT	PCAV			
	034-036	MACH	ALPHAW	PSC									
	038 & 039	масн	ВЕТА	Q(PSF)	PB1	PB2	PB4	LHLS	RHLS	PCAV			
	040-149	ALPHAO	DZ	Q(PSF)	PB1	PB2	PB4	LHLS	PHLS	PCAV	<u> </u>		

Note: ID--Independent variable

Table VI
CA20 DATASET DESCRIPTION
(INTERPOLATED/INCREMENTED DATASETS)

DATASET TYPE	DESCRIPTION
MGNXXX	Interpolated data for 747 carrier balance data in carrier reference dimensions.
NGNXXX	Interpolated data for orbiter balance data in orbiter reference dimensions.
UGNXXX	Incremental data - 747 carrier data in presence of orbiter (mated) minus 747 carrier alone data in 747 carrier reference dimensions.
VGNXXX	Incremental data - Orbiter data in presence of 747 carrier (mated) minus orbiter alone data in orbiter reference dimensions.

NOTE: Datasets M, N, U, and V contain the full  $\Delta Z$  array of 0 3 7.5 15 30 40 and 60 ft. Therefore, the datasets reflect extrapolations for some individual test arrays of  $\Delta Z$ . For subsequent data plotting, the full  $\Delta Z$  arrays were truncated to the actual tested arrays.

#### Table VII. CA20 INTERFULATED DATASET SUMMARY

(M AND N DATASET	151	,
------------------	-----	---

	(M AND N DATASETS)
DATASET(S)	INTERPOLATED VARIABLES (1) (2)
NGN001 - 011	MACH, ALPHAO
NGN012 → 013	MACH, DZ
NGNO14	MACH, BETAO
NGN015 → 033	MACH, ALPHAO
MGN034 → 036	MACH, ALPHAW
MGN037 NGN037	MACH, ALPHAO
MGN038 NGN038 → 039	MACH, BETAC
MGN040 NGN040 → 048	ALPHAO, DZ (SEE NOTE 3)
MGN049 NGN049 → 119 MGN126 NGN126 → 140	ALPHAO, DZ, DX, ALPHAW, DY, BETAO, BETAC, PHI
MGN126 MGN141 NGN141 → 149	ALPHAO, DZ
MGN120 NGN120 → 125	ALPHAO, DZ, BETAC, ALPHAW

#### NOTES:

#### (1) Interpolation procedure:

Number of Values Available for Interpolation	Interpolation Procedure
1 2 3 4 → ∞	Substitute actual test value with a nominal test value (Note 3 below) Straight line Parabolic spline fit Cubic spline fit

BETA = BETAC

(3)
Interpolation was versus DZ and ALPHAO; however, since each of these datasets (40 → 48) has only one ALPHAO there was therefore no ALPHAO interpolation. The recorded test ALPHAO was replaced with a nominal test ALPHAO (i.e., 8, 12, or 16) so that the only interpolation was versus DZ.

#### Table VII. Concluded.

(4)
Interpolation on DX was not performed on all datasets due to large data fluxuations from the nominal condition.

## Table VIII. CA20 INCREMENTAL DATASET SUMMARY

(INTERFERENCE) - (ISOLATED)

(IL AND V DATASETS)

BASE DATASET	VEHICLE	BETAC, deg.	ALPHAW, deg.	BETAO, deg.	ELEVON, deg.	AILRON, deg.
MGN034	CARRIER (1)	<b>-</b> 5	0 4 8	NA 	NA 	NA 
MGN035			0 4 8			
MGN036		5 <b>↓</b>	0 4 8			
NGN007 NGN010 NGN008 NGN011 NGN009 NGN018	ORBITER-0 <sub>1</sub> S <sub>1</sub> (2)  ORBITER-0 <sub>2</sub> S <sub>1</sub> (2)	NA ,,	NA   	-5 0 	.5 5 0 10 5 5	0

## NOTES:

(1)

ALPHAW Sweep (0, 4, 8°) ALPHAO Sweep (6, 8, 10, 12, 14, 16, 18)

Interpolate base datasets to various nominal  $\alpha$  and  $\beta$ Procedure - (a) combinations.

> Subtract appropriate interpolated base dataset from interpolated separation (mated) data, except for datasets 45 thru 48 which were utilized to provide the increment due to attach hardware as follows:

Resulting Dataset Number	First Dataset Number	Second Dataset Number
UGNO45	MGNO45	MGN049 @ ALPHAO = 8
VGNO45	NGNO45	NGN049 @ ALPHAO = 8
UGNO46	MGN046	MGN052 @ ALPHA0 = 12
VGNO46	NGN046	NGN052 @ ALPHA0 = 12
UGNO47	MGNO47	MGN055 @ ALPHAO = 16
VGNO47	NGNO47	NGN055 @ ALPHAO = 16
UGNO48	MGN048	MGN046
VGNO48	NGN048	NGN046

INCREMENT = (First Dataset) - (Second Dataset)

Datasets 45 thru 48 interpolated per note (3) on "Interpolated Dataset Summary".

Datasets 49, 52, and 55 interpolated versus ALPHAO, DZ, DX, ALPHAW, DY, BETAO, BETAC, PHI.



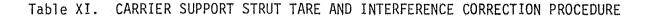
RESULTANT SGNO		CONFIGURATION	βο =	IPUT DA O°, βc	= 0°	
CARRIER	ORBITER		αW	8° 2°	12° 6°	16° 10°
A	В	747/0 0 <sub>1</sub> S <sub>1</sub> AT <sub>38</sub> AT <sub>39</sub>		40	41	42
E	F	747/1 0 <sub>1</sub> S <sub>1</sub> AT <sub>38</sub> AT <sub>39</sub>		45	46	47
I	J	747/1 0 <sub>1</sub> S <sub>1</sub>		49		
К	Ĺ	747/1 0 <sub>1</sub> S <sub>1</sub>			52	
M	N	747/1 0 <sub>1</sub> S <sub>1</sub>				55
			α(	) = 12°	$\alpha_W =$	5.83°
			β <sub>0</sub> β <sub>C</sub>	-5° -5°	0°	
С	D	747/0 0 <sub>1</sub> S <sub>1</sub> AT <sub>38</sub> AT <sub>39</sub>		43	41	
G	Н	747/1 0 <sub>1</sub> S <sub>1</sub> AT <sub>38</sub> AT <sub>39</sub>		48	46	
					<u> </u>	

## NOTES:

- (1) Orbiter data were interpolated versus  $\alpha_0$  and  $\Delta z$
- (2) Carrier data were interpolated versus  $\alpha_W$  and  $\Delta Z$
- (3) The interpolation assumes a constant incidence angle between Orbiter and Carrier even though they were mounted on separate support systems (see Configuration A, in Figure 2F).
- (4) Resultant datasets SGNO\_\_ 1 have both lateral and longitudinal coefficient data.

			INPUT DATASETS			
RESUL	$\beta_0 = 0^{\circ}  \beta_C = 0^{\circ}$					
DATA	αo	8°	12°	16°		
		$\alpha_{W}$	2 °	6°	10°	
CARRIER	ORBITER					
WGNR45	XGNB45		45- 49			
WGNR46	XGNB46			46-52		
WGNR47	XGNB47				47-55	
			$\alpha_0 = 12^\circ$ ,	$\alpha_W = 5$ .	.83°	
WGNR48	XGNB48		48-	46		
WGNR43	XGNB43		43-	41		
WGNRDB	XGNBDB	(	48, 46) -	(43, 47	1)	
			$\beta_0 = 0^\circ$ ,	$\beta_{\rm C} = 0^{\rm c}$	)	
WGNRCA	XGNBCA	(45,	46, 47) -	(40, 4	1, 42)	

NOTE: Resultant datasets have incremental lateral and longitudinal coefficient data.



α, deg.	β, deg.	CA5 Run with Image Strut	CA5 Run without Image Strut
1A	0	15	99
2		20	104
6		23	108
8		27	112

$$\alpha_{W} = 10^{-100} = 3^{\circ} \rightarrow 16^{\circ}, 1^{\circ} \text{ increments}$$

$$\beta = 10^{\circ} = -12^{\circ}, -10^{\circ}, -8^{\circ}, -6^{\circ}, -4^{\circ}, -3^{\circ}, -2^{\circ}, -1^{\circ}, 0^{\circ}, 1^{\circ}, 2^{\circ}, 3^{\circ}, 4^{\circ}, 6^{\circ}, 8^{\circ}, 10^{\circ}, 12^{\circ}$$

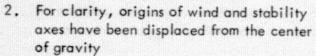
$$Correction = \left(\begin{array}{c} \text{Run with} \\ \text{Image Strut} \end{array}\right) - \left(\begin{array}{c} \text{Run without} \\ \text{Image Strut} \end{array}\right)$$

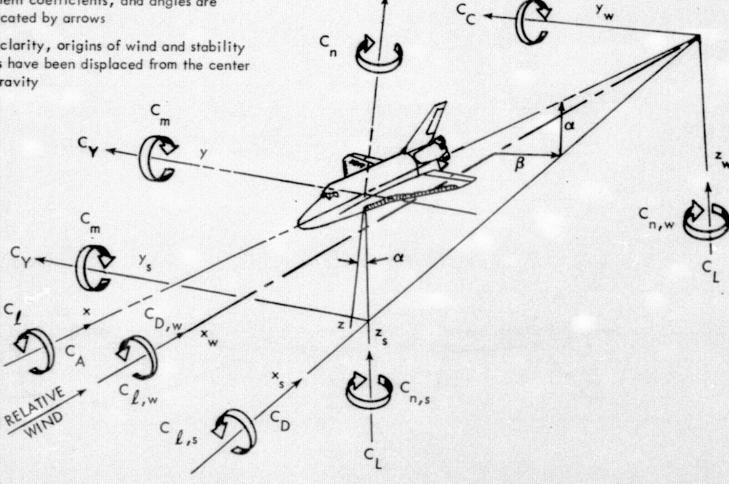
"Correction" datasets are 6GMDA4, 6GMDB4 and 6GMDC4, which were interpolated for  $\alpha_W$  = 2° to 12° and  $\beta$  = -5°, 0°, +5°, respectively.

"Corrected" datasets are 5GN034-149. For the DZ and  $\alpha_0$  sweeps (2nd independent variable), the "correction" is a constant value for all coefficients. For the  $\alpha_W$  and  $\beta$  sweeps (2nd independent variable), the "correction" is a function of  $\alpha_W$  and  $\beta$ , respectively.

#### Note:

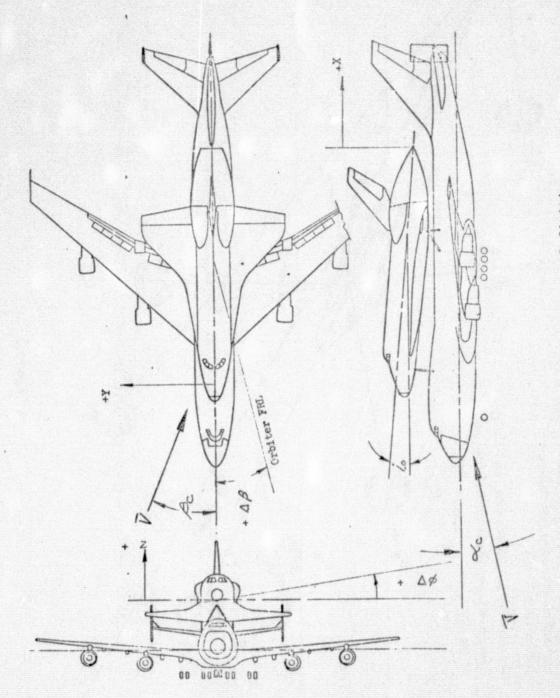
"Correction: and "corrected" data are shown in the Appendix.





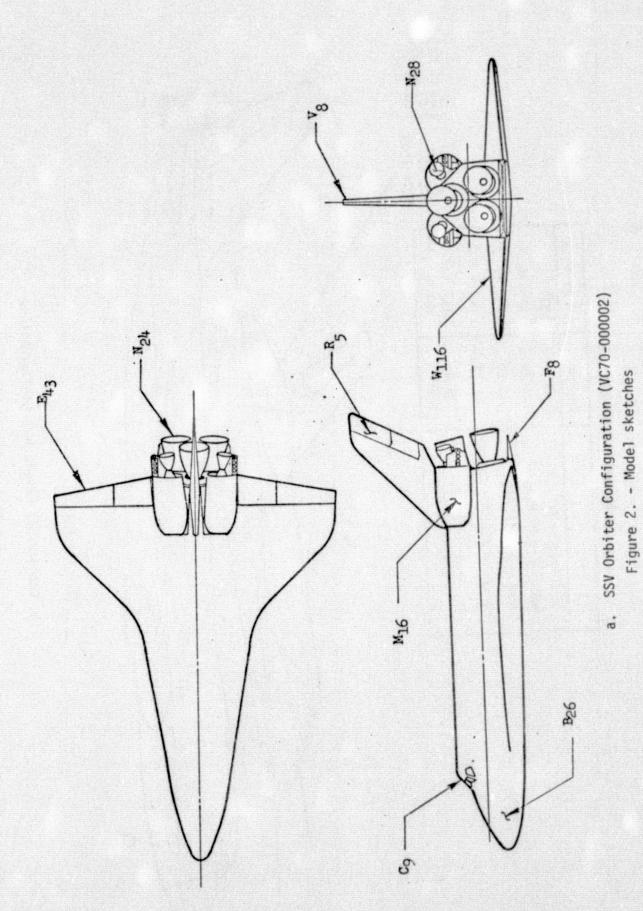
a. General

Figure 1. - Axis systems.



b. Orbiter/747 Axis System Definition Figure 1. - Concluded.

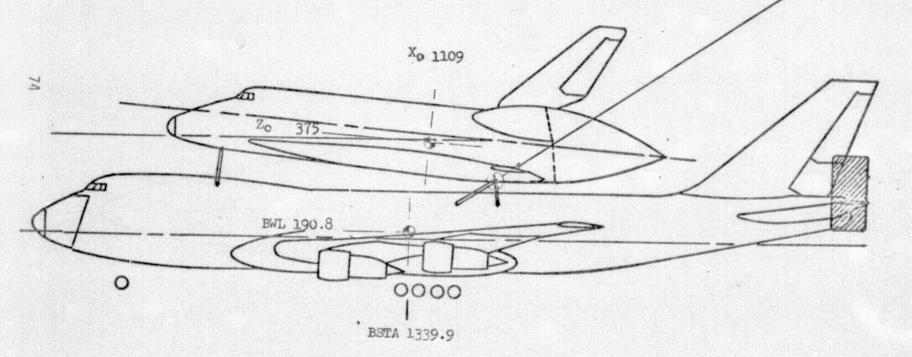
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# REFERENCE DIMENSIONS (FS)

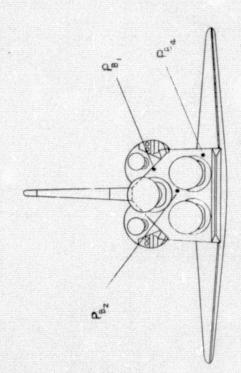
	ORBITER	747 CARPIER
WING AREA ∼Ft²	2690	5500
MAC (c) ~ INCHES	474.81.	327.78
SPAN (b) ~ INCHES	936.68	2348.04
MOMENT REFERENCE CENTER	67.5% LB	25.0 % €
F.S. ~ INCHES W.P. ~ INCHES	1109.0 375.0	1339.9

BWL 400 (% 96 51) BSTA 1607 (% 1317)



b. Orbiter/747 Flight Test Configurations

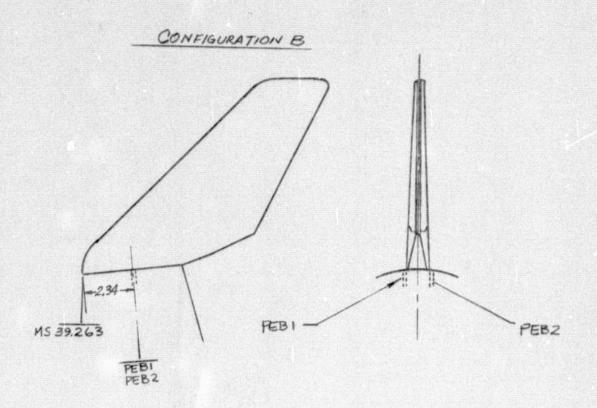
Figure 2. - Continued.



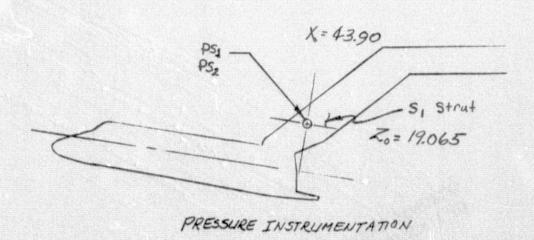
c. Base Pressure Locations

Figure 2. - Continued.

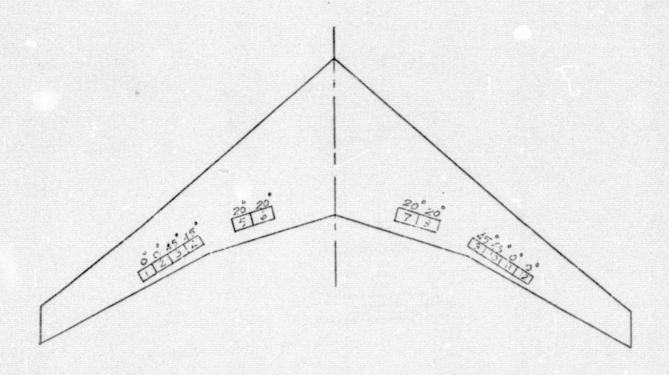
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OF POOR QUALITY



# CONFIGURATION A

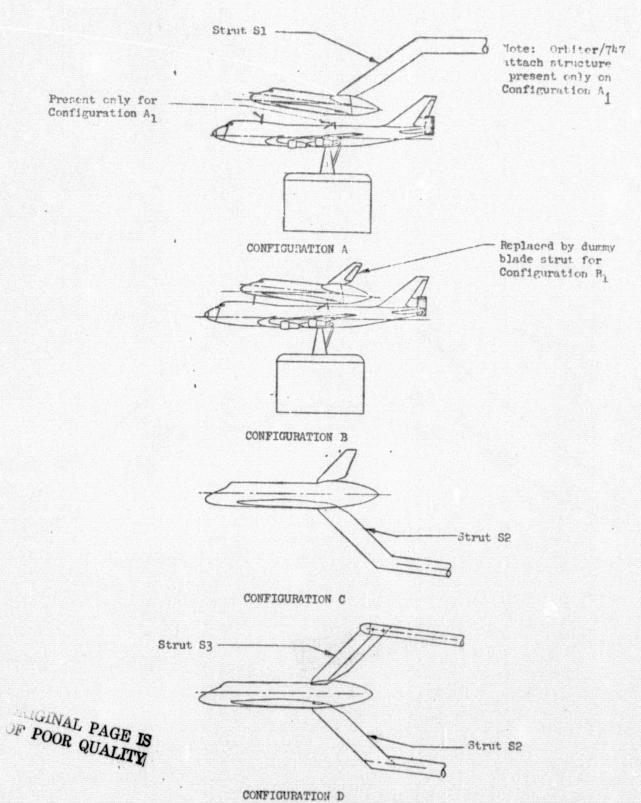


d. Blade Strut and Vertical Tail Pressure Locations
Figure 2. - Continued.

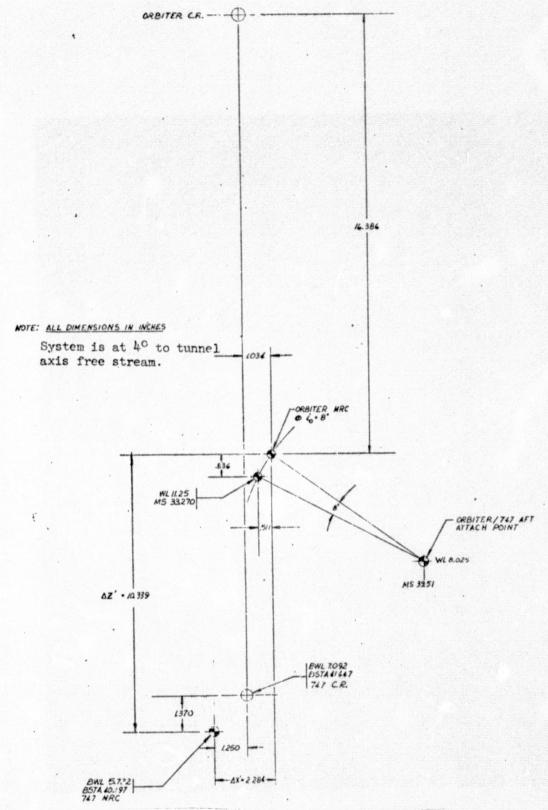


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e. Standard In-Flight Speed-BrakeFigure 2. - Continued.



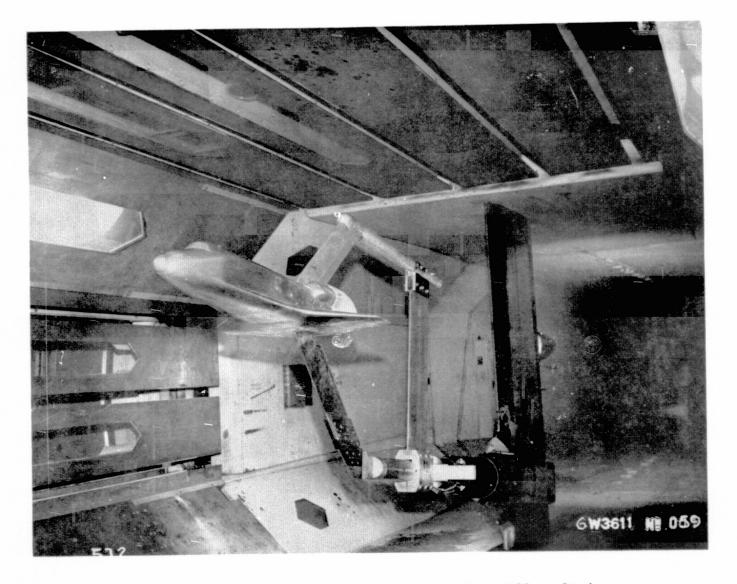
f. Test Support ConfigurationsFigure 2. - Continued.



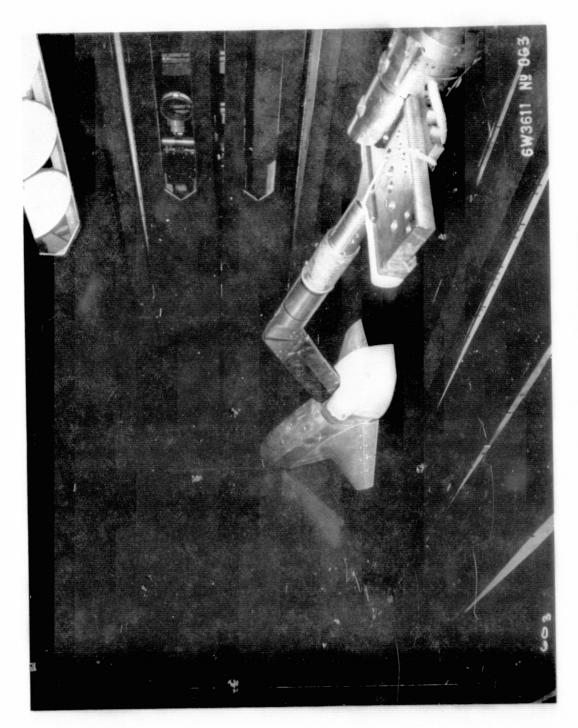
g. Orbiter/747 C.G. and C.R. Orientation

Figure 2. - Concluded.

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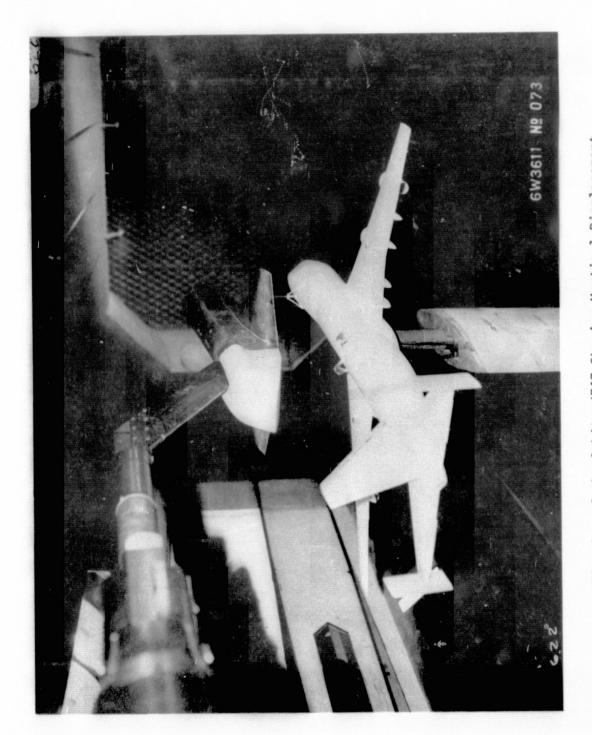


a. Orbiter Alone with Dummy Blade in Proximity for Sting Tare Effect Study  $\hbox{Figure 3. - Model photographs.}$ 

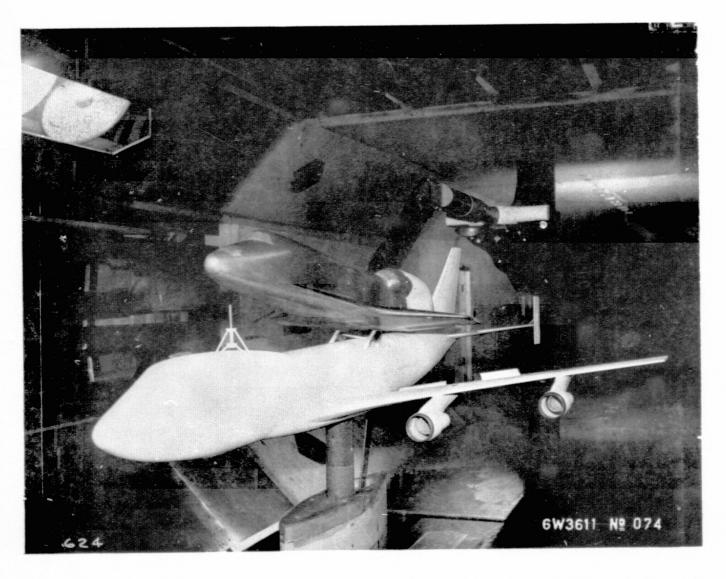


b. Orbiter Alone with Tail Cone  ${\rm TC}_{5.1}$ 

Figure 3. - Continued.



c. Aft View of the Orbiter/747 Showing Vertical Displacement Figure 3. - Continued.



d. Front View of the Orbiter at an Angle of Attack with Respect to the 747 Carrier Figure 3. - Concluded.

# DATA FIGURES

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VOLUME 2 Figures 26-39 Pages 832-1863

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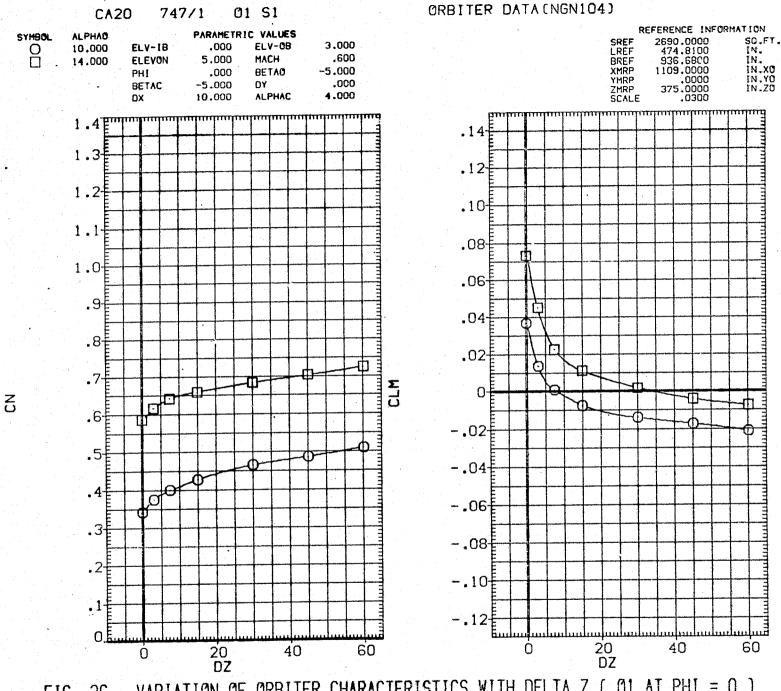


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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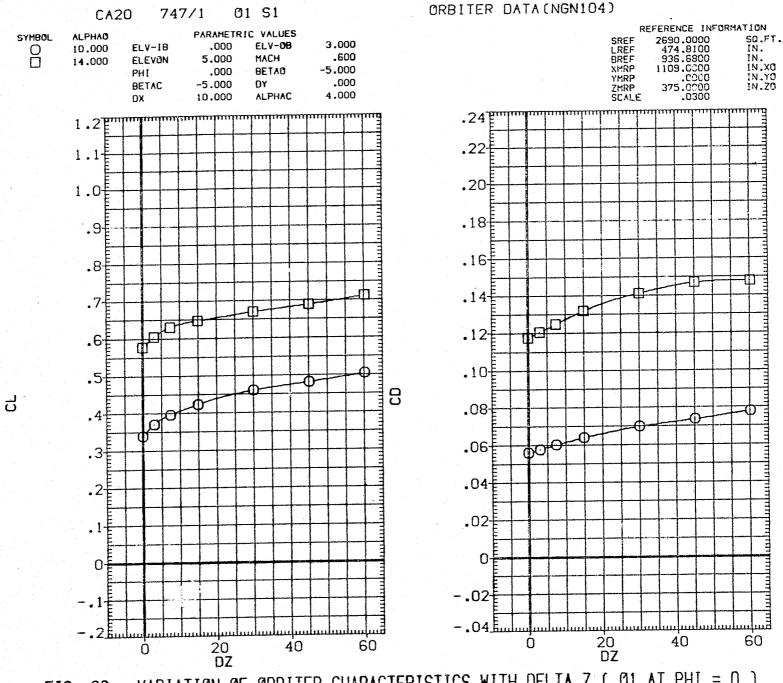


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 836

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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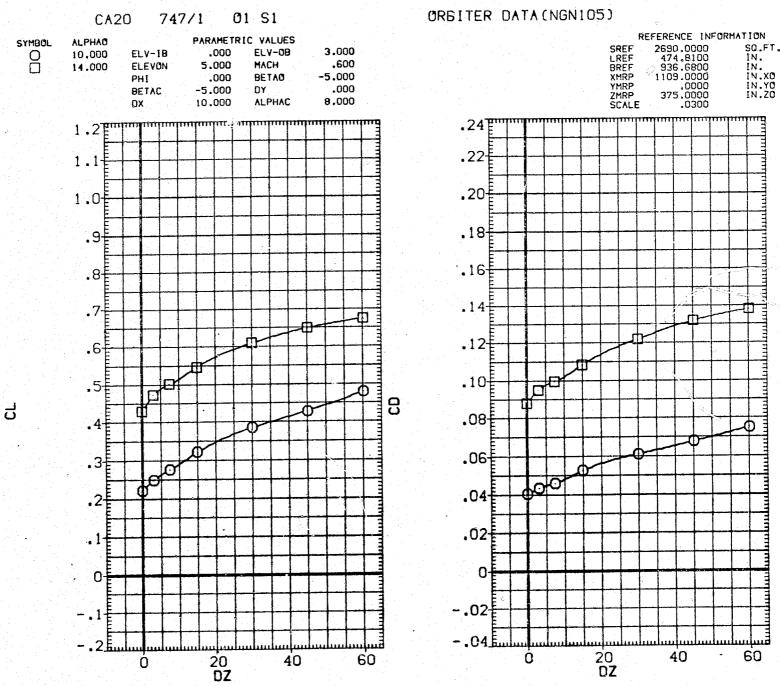
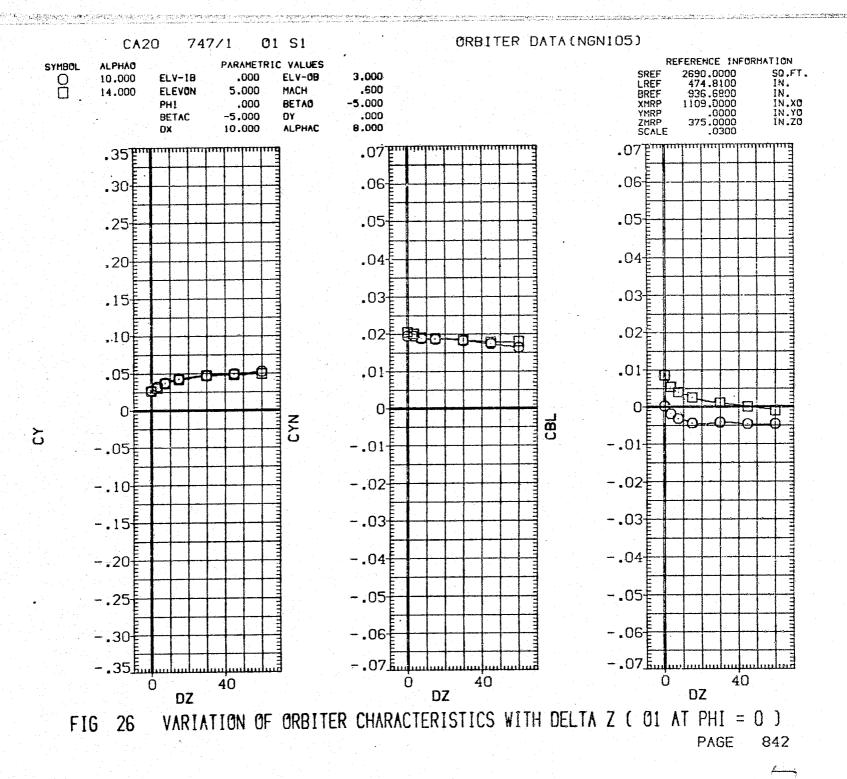


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )



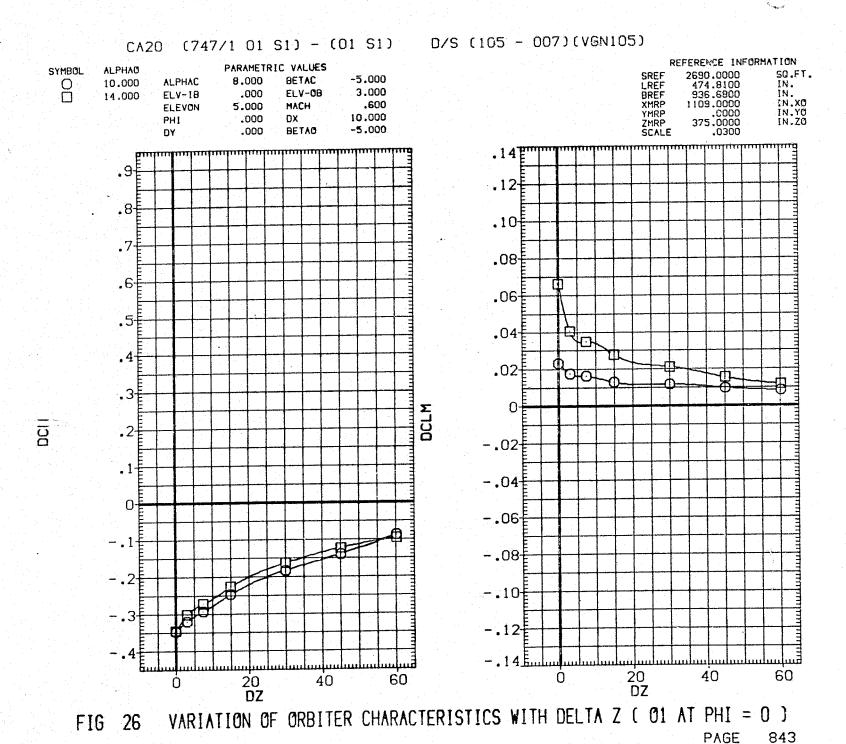


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )



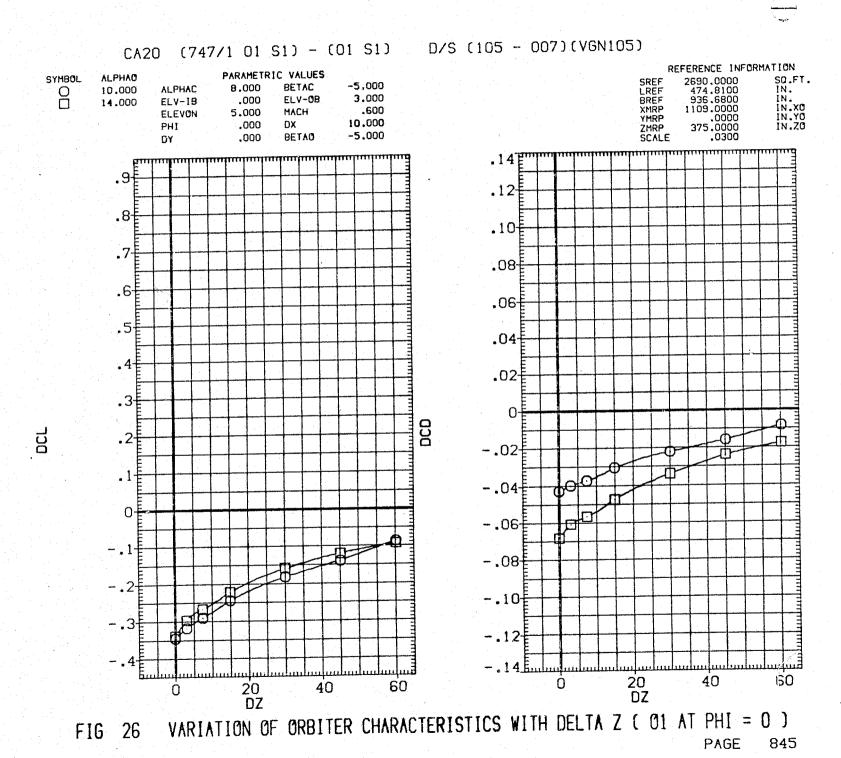


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

40

20 **DZ** 

0

-.04点

20 DZ 40

60

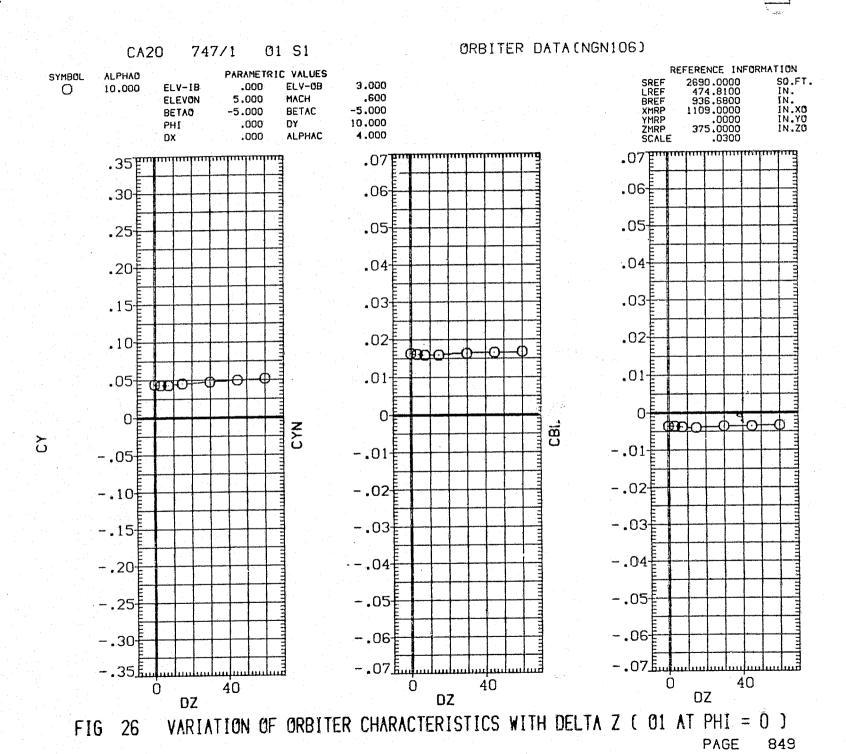


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 850

60

20 DZ

Ò

40

20 DZ

0

40

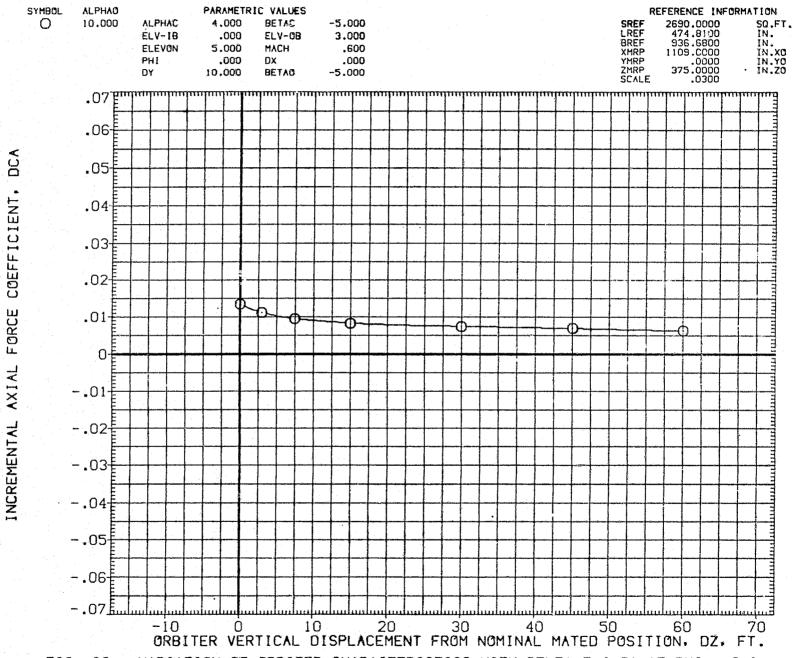


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 852

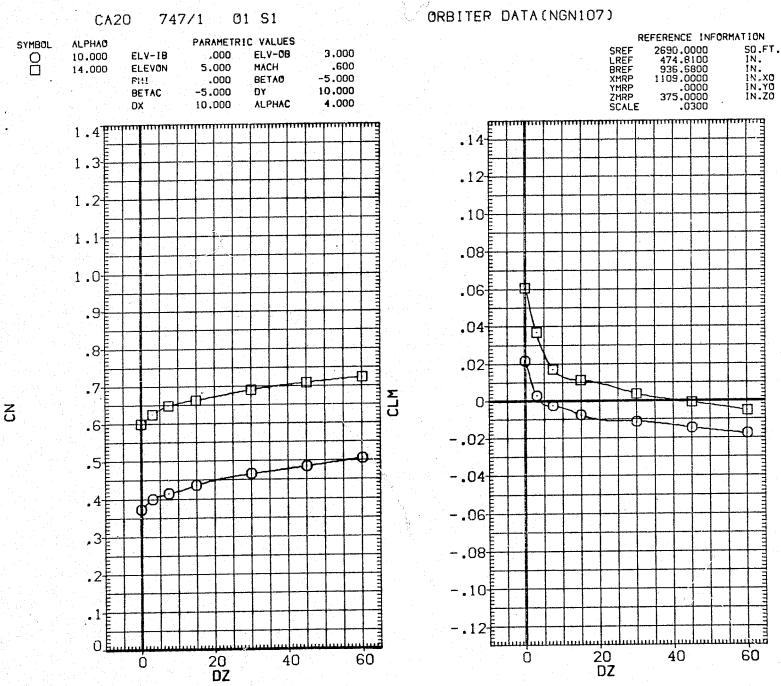
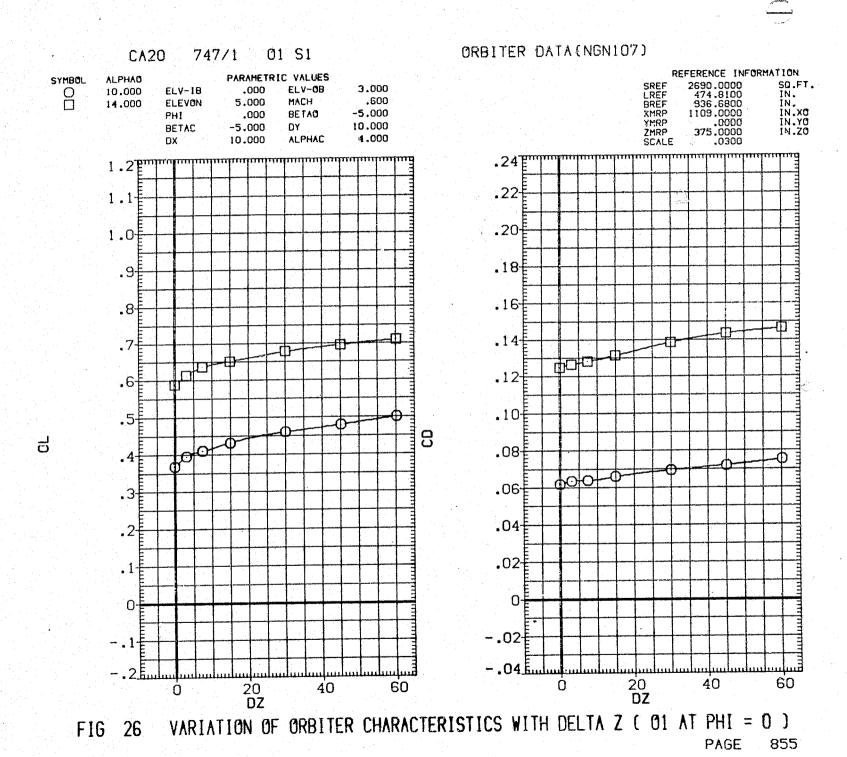
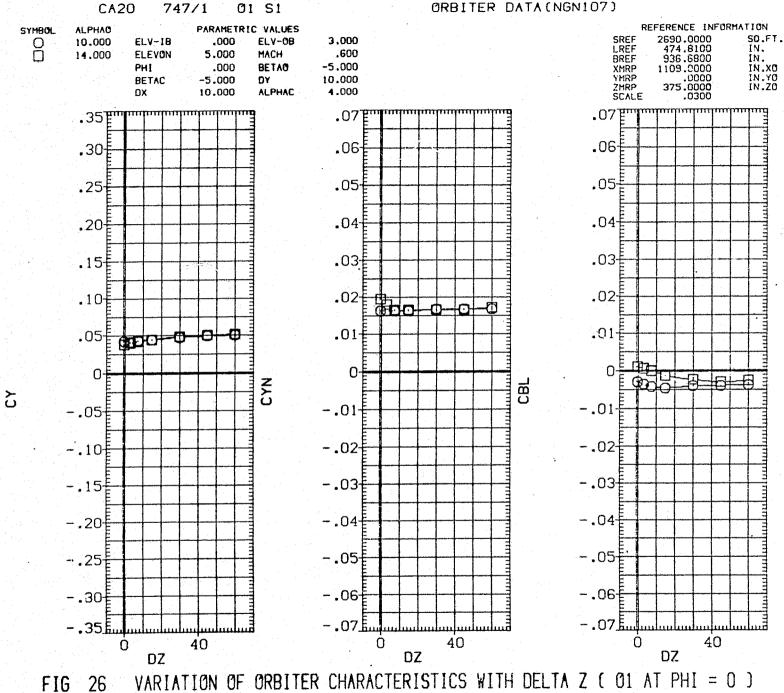


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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D/S (107 - 007)(VGN107) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES ALPHAO SYMBOL 2690.0000 474.8100 SQ.FT. SREF -5.000 4.000 BETAC 0 10.000 **ALPHAC** IN. LREF .000 ELV-0B 3.000 BREF 14.000 ELV-IB 936,6800 IN. IN.XO IN.YO IN.ZO 1109.0000 .600 5.000 **ELEVON** .0000 YMRP .000 DX 10.000 PHI ZMRP 375.0000 -5.000 10,000 BETAO .0300 .12 .8 .10E .08 .6<del>‡</del> .06+ .5<del>[</del> .04<del>E</del> .4 .02 0-DCLM DCN DCN -.02<del>[</del> -.04<del>[</del> 0--.06 -.08<del>-</del> -.10<del>[</del> -.3 -.12<del>-</del> 20 D**Z** 60 40 20 **DZ** 40 60 0 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG

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ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT.

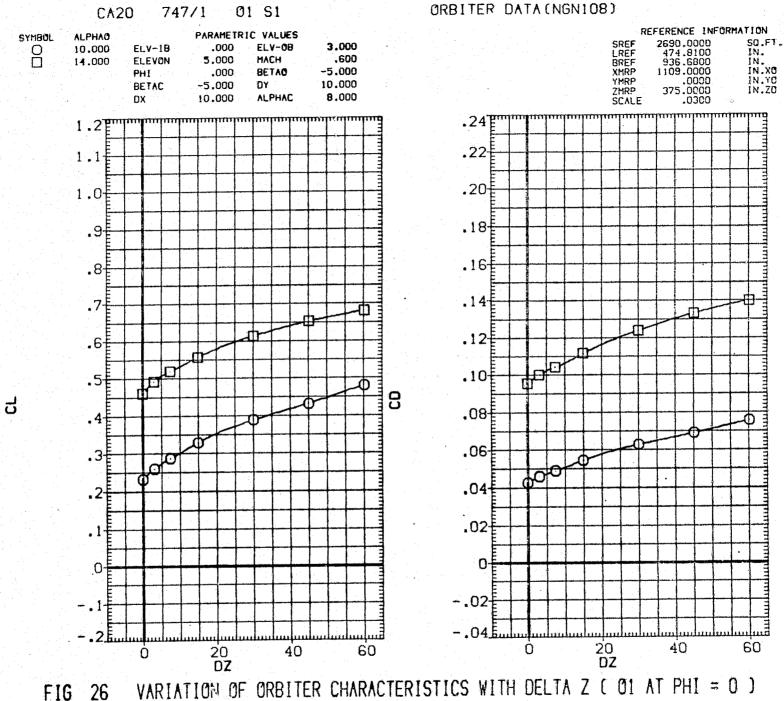
FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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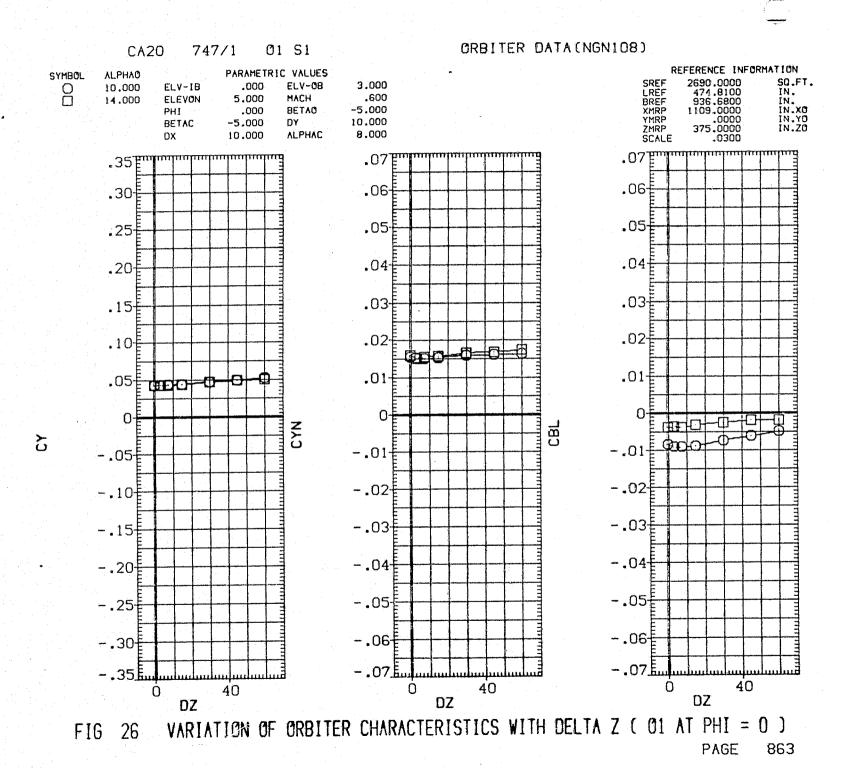
859

FIG

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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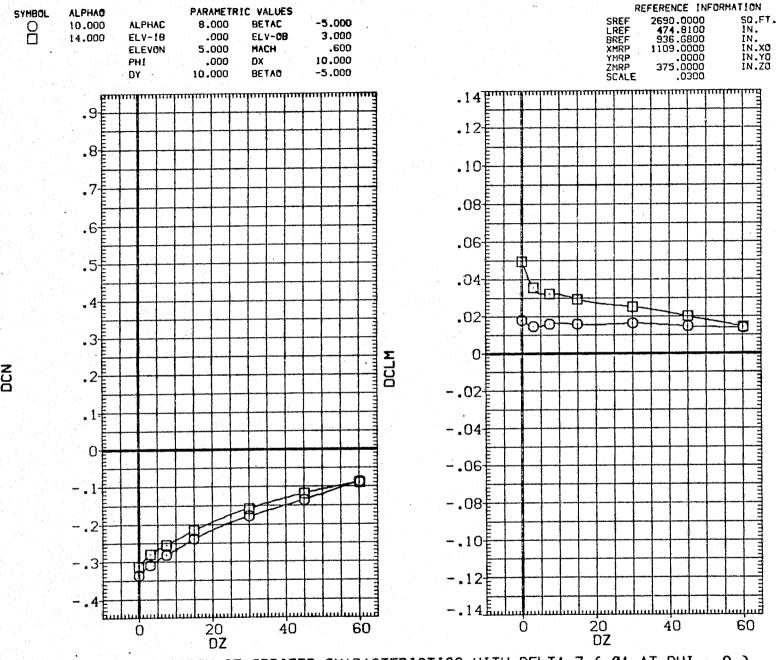
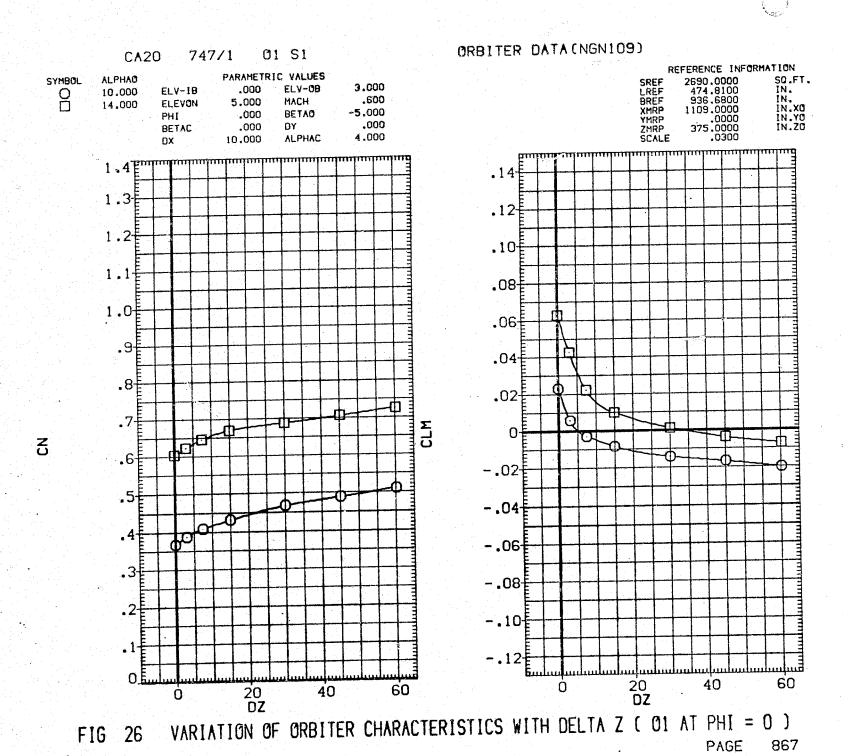


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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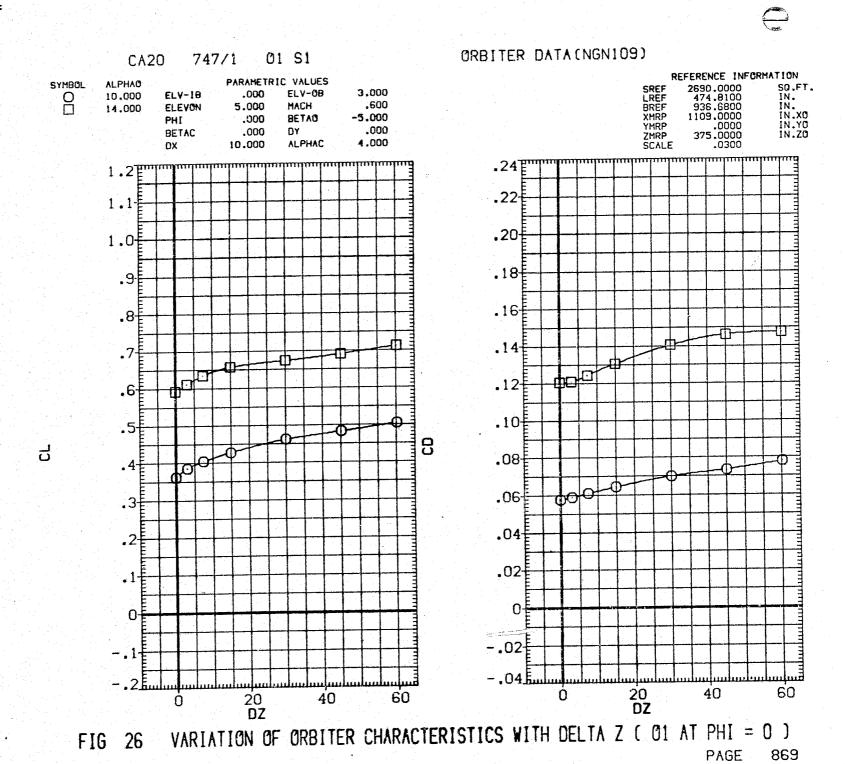
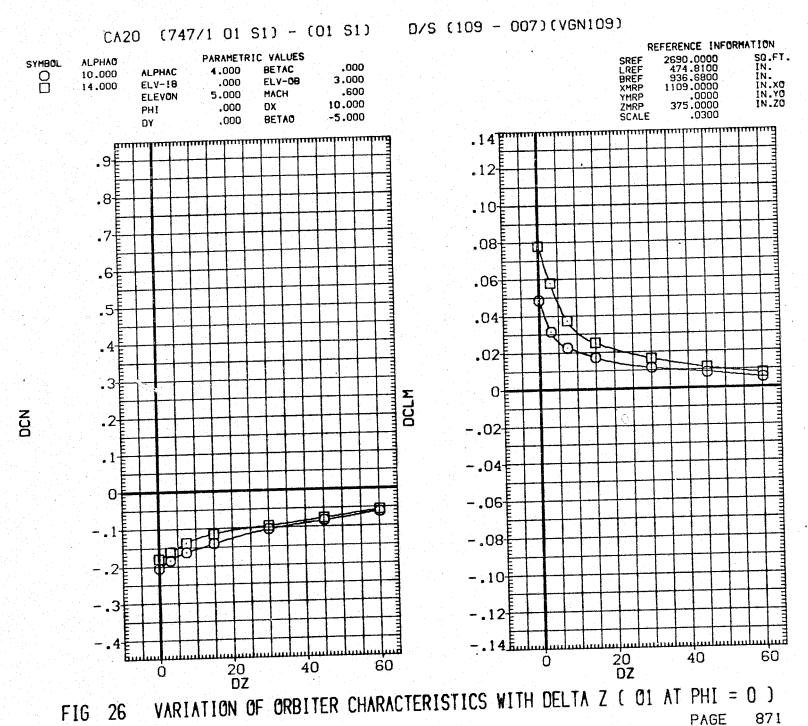


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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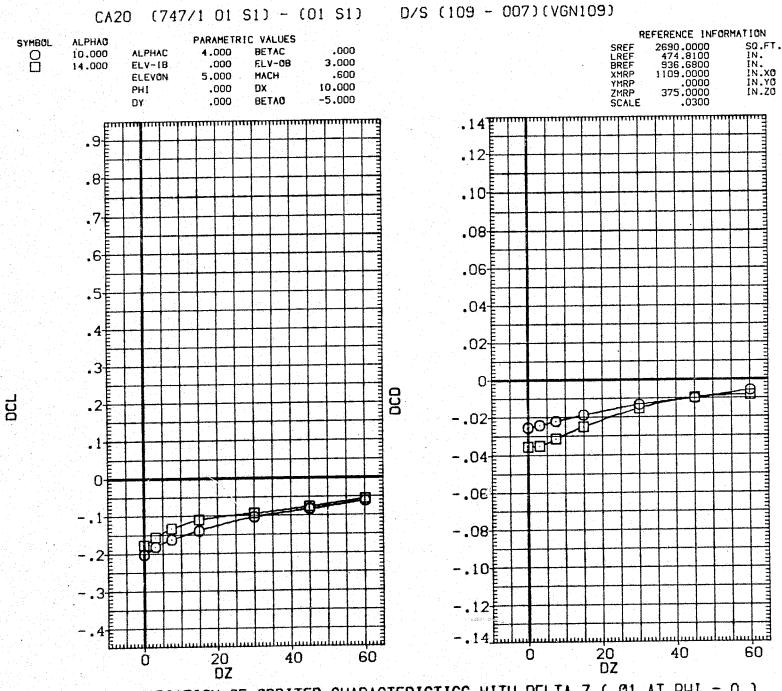


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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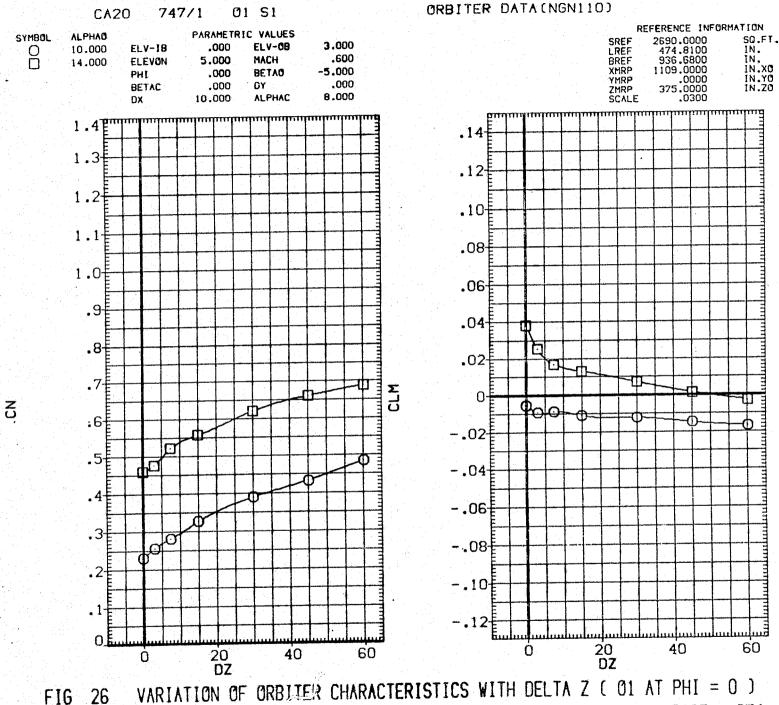


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CA20 747/1 01 S1

## ORBITER DATA (NGN110)

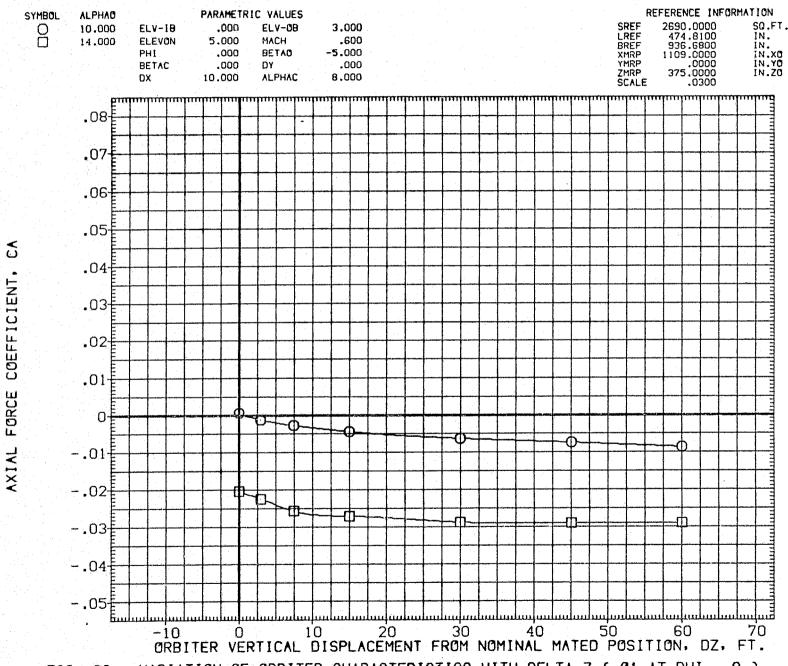
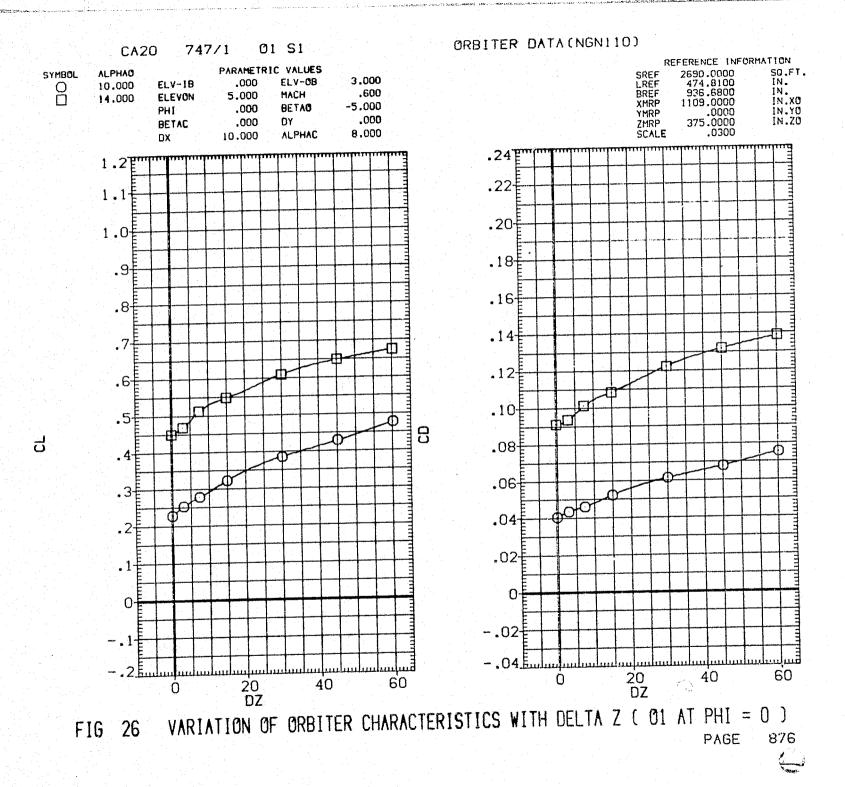
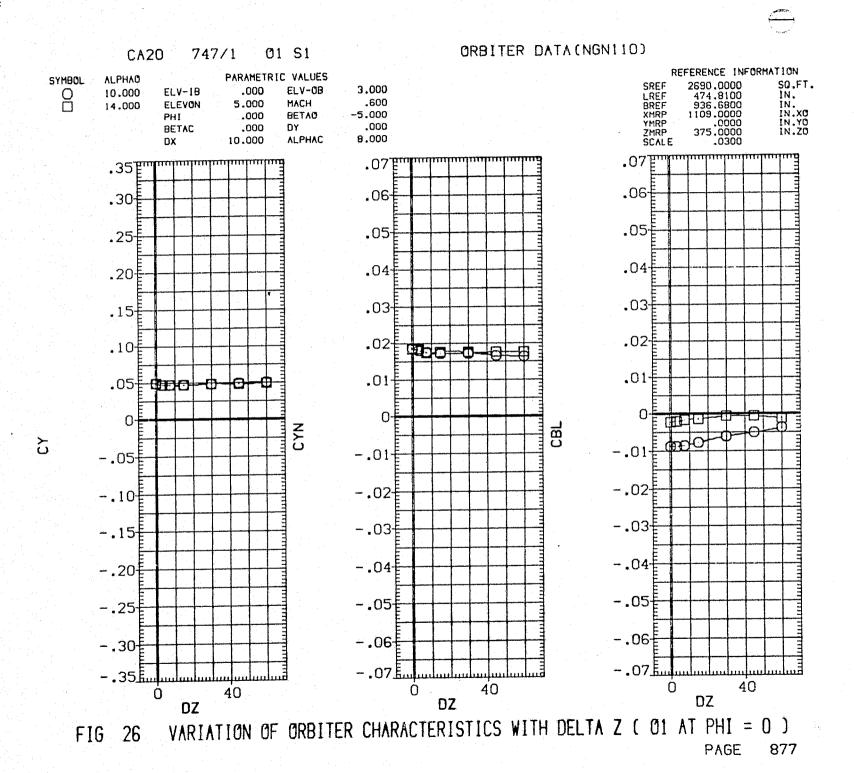


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )





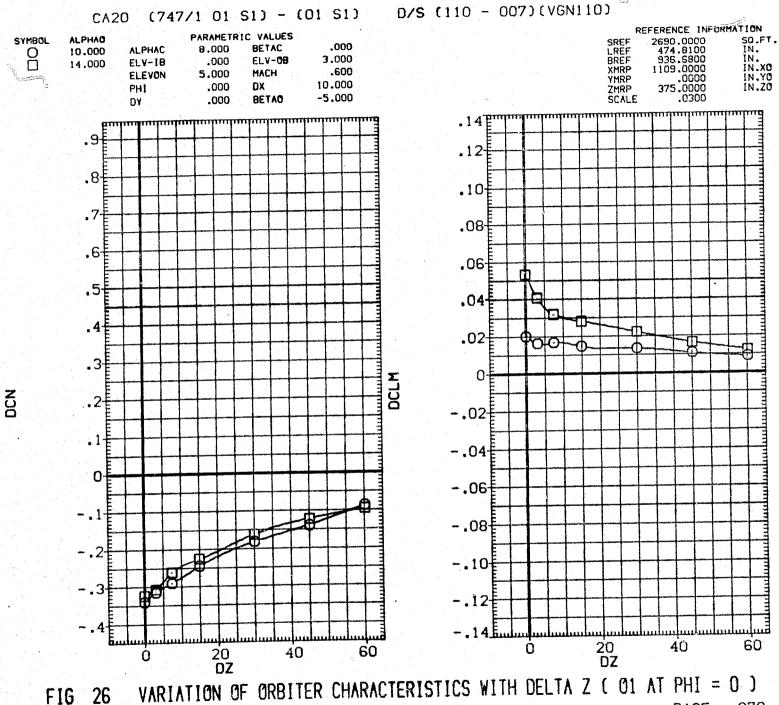


FIG 26 878 PAGE

CA20 (747/1 01 S1) - (01 S1) D/S (110 - 007)(VGN110)

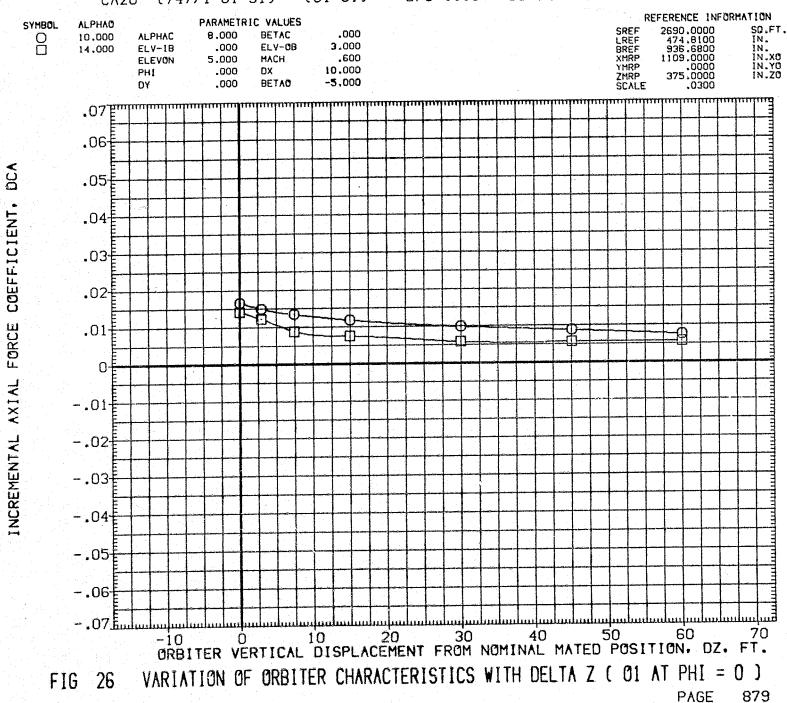


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 880

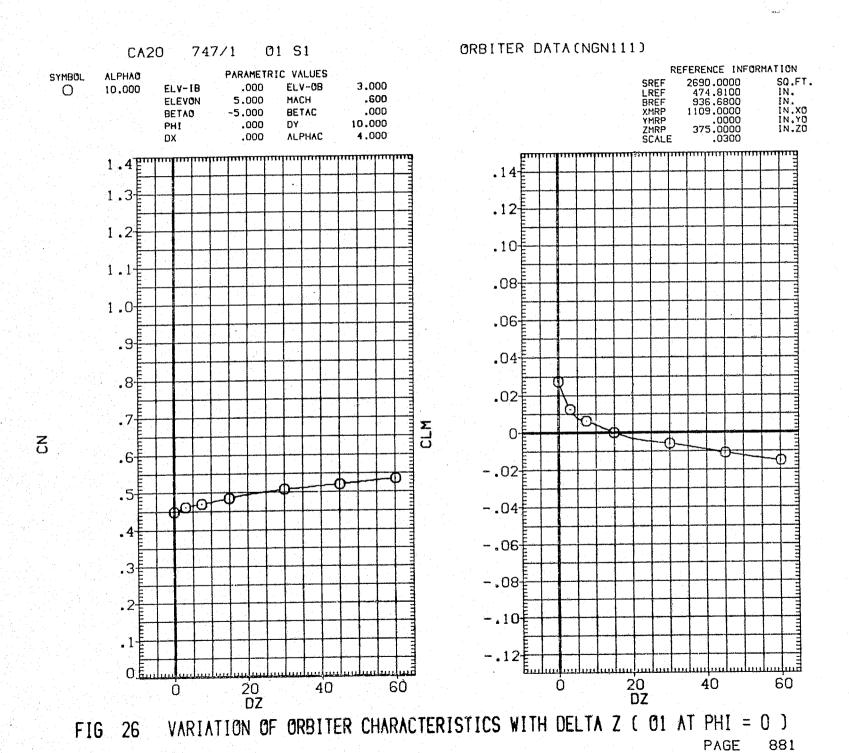
60

20 **DZ** 

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40

20 DZ 40



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26

ORBITER DATA (NGN111) 01 S1 **CA20** 747/1 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL ALPHAO 2690.0000 474.8100 936.6800 SO.FT. IN. IN. .000 ELV-08 3.000 10.000 0 ELV-1B .600 ELEVON 5.000 MÁCH .000 IN.XO BETAO -5.000 BETAC XMRP 1109.0000 YMRP ZMRP .0000 IN.YO 10.000 PHI .000 DY 4.000 .000 **ALPHAC** DX: SCALE .24 Em .22 1.1 .20 1.0 .18 .16 .14 .10 7 .08<del>[</del> .06 .04 .02 0--.02 سلسط 04. 20 DZ 20 DZ 40 60 40 60 0 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG

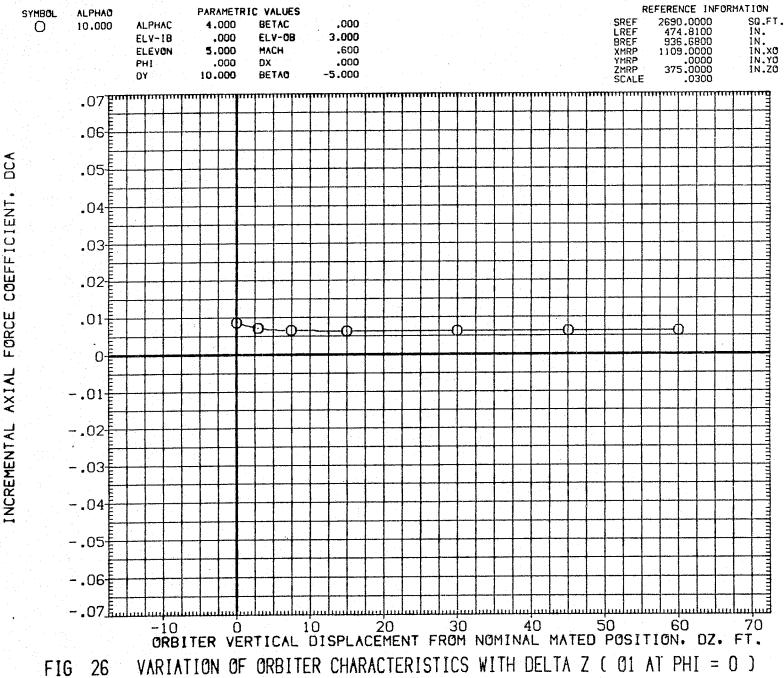
PAGE

FIG PAGE 884

D/S (111 - 007) (VGN111)

CA20 (747/1 01 S1) - (01 S1) SYMBOL ALPHAO PARAMETRIC VALUES REFERENCE INFORMATION 0 10.000 4.000 2690.0000 474.8100 **ALPHAC** BETAC .000 SREF LREF BREF SO.FT. IN. .000 3.000 ELV-IB ELV-OB 936.6800 ELEVON 5.000 MACH .600 XMRP YMRP ZMRP SCALE IN.XO IN.YO IN.ZO 1109.0000 PHI .000 DX .000 .0000 DY 10.000 BETAO -5.000 375.0000 .0300 .9-.12 .8-.10 .08 .6<del>-</del> .06 .04-.02 .3 DCLM 0-OCN OCN -.02 -.04<sup>‡</sup> -.06 <del>000</del> -.08 -.10<del>[</del> -.12<del>-</del> -.14 hm 20 DZ 20 DZ 0 40 60 40 60

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG PAGE 885





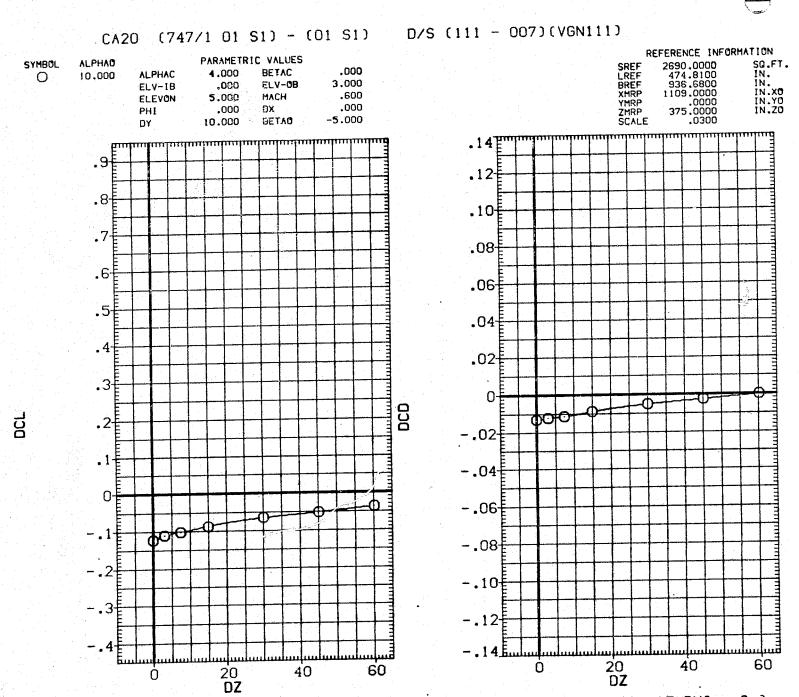


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 887

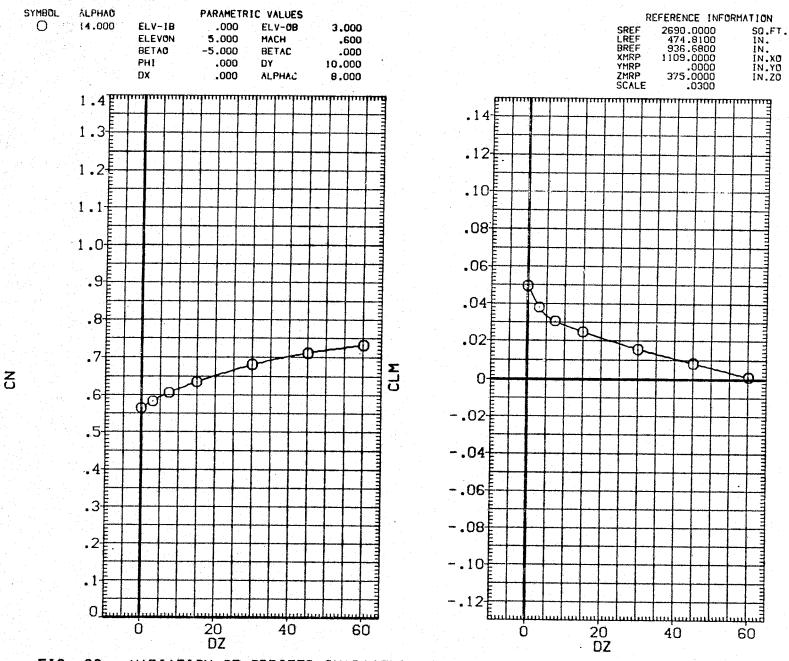


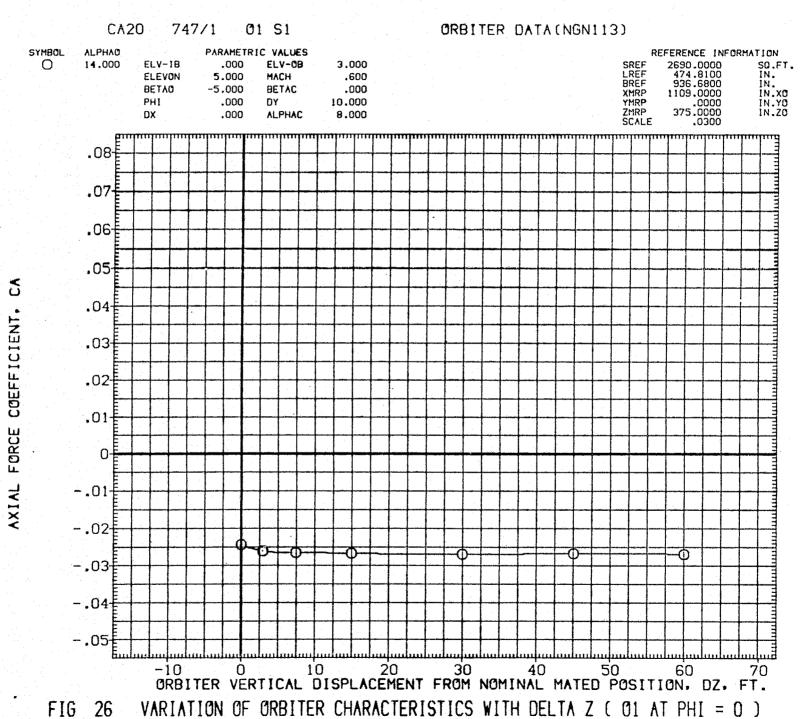
FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 888



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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 890

20 **DZ** 

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20 DZ





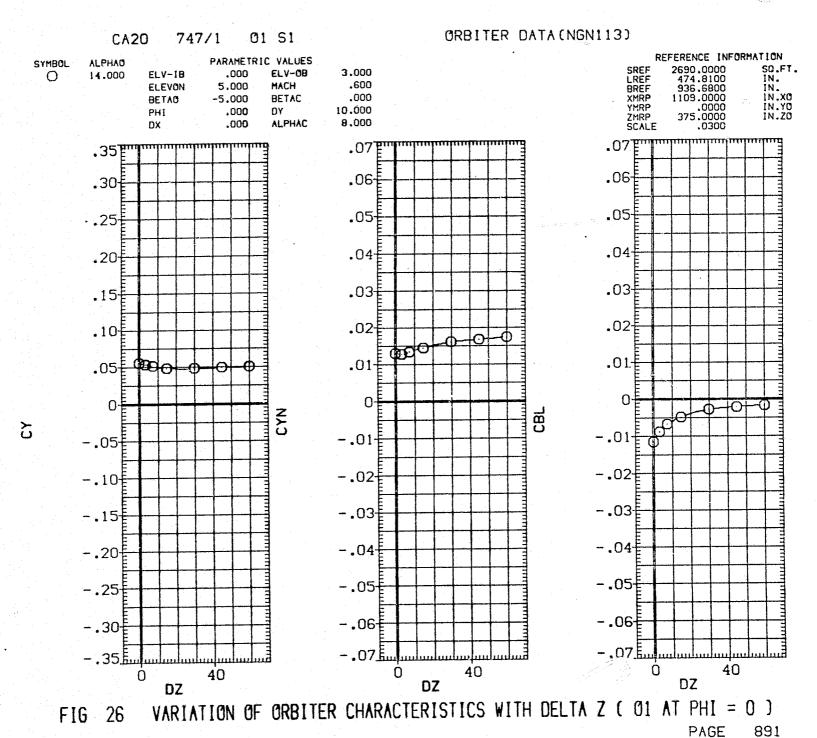
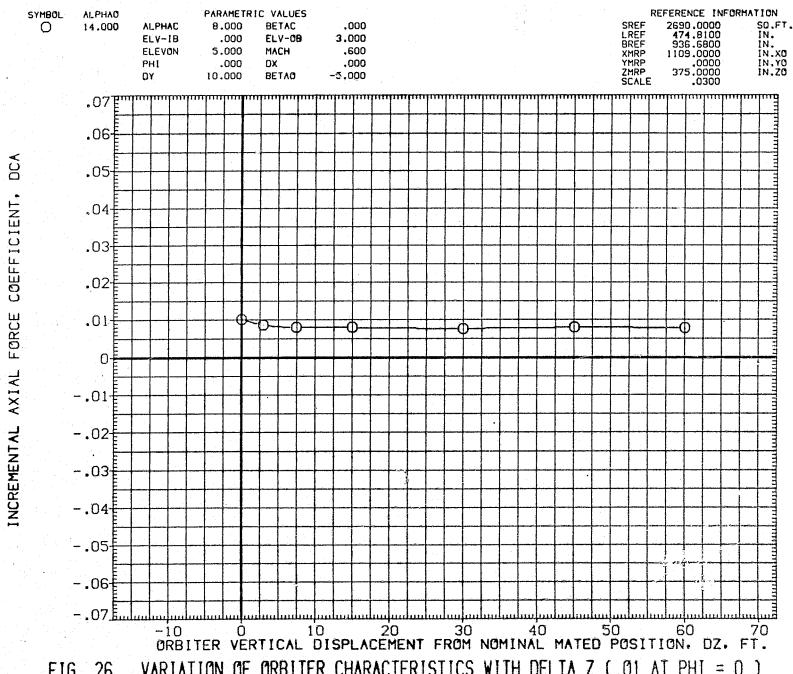


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 892



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26

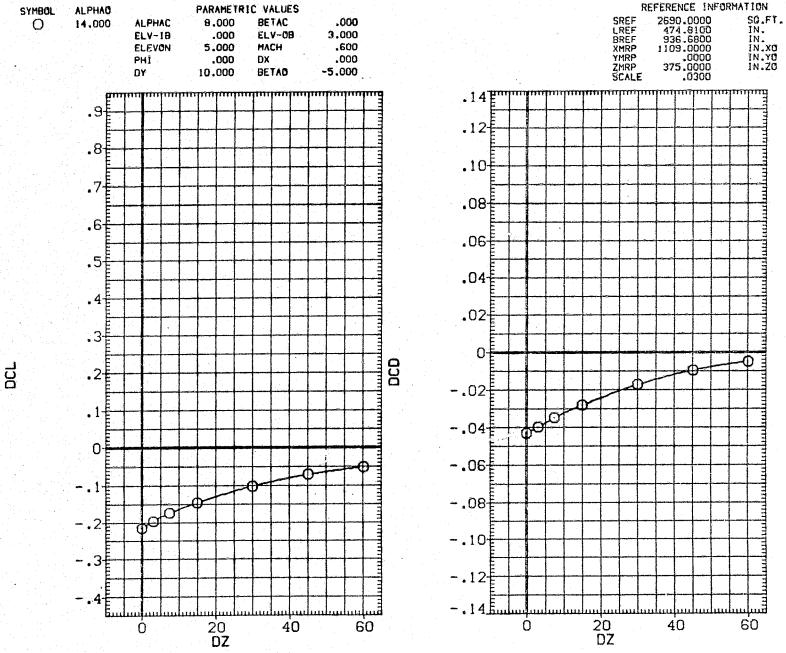
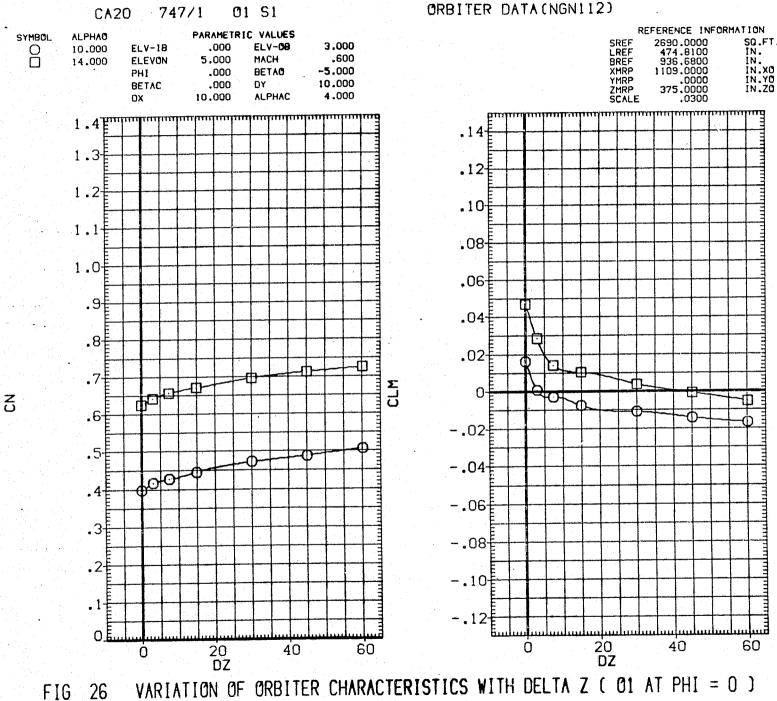


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 894



FIG



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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 896

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT.

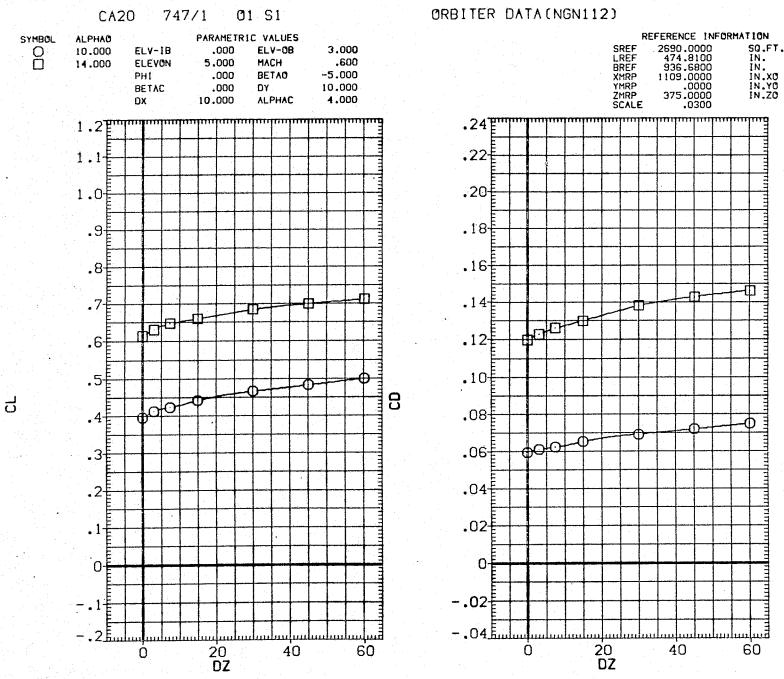
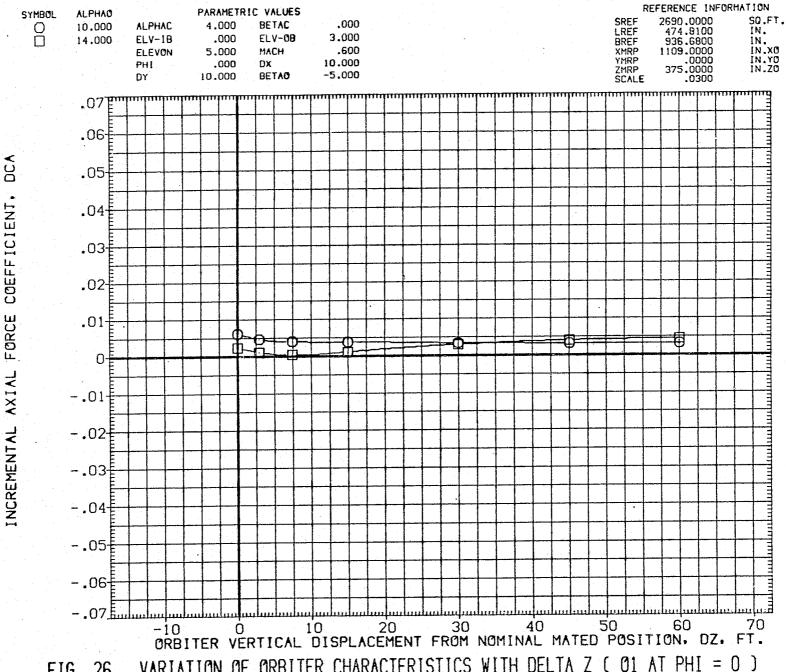


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 897

FIG 26 PAGE 898

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 899



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE

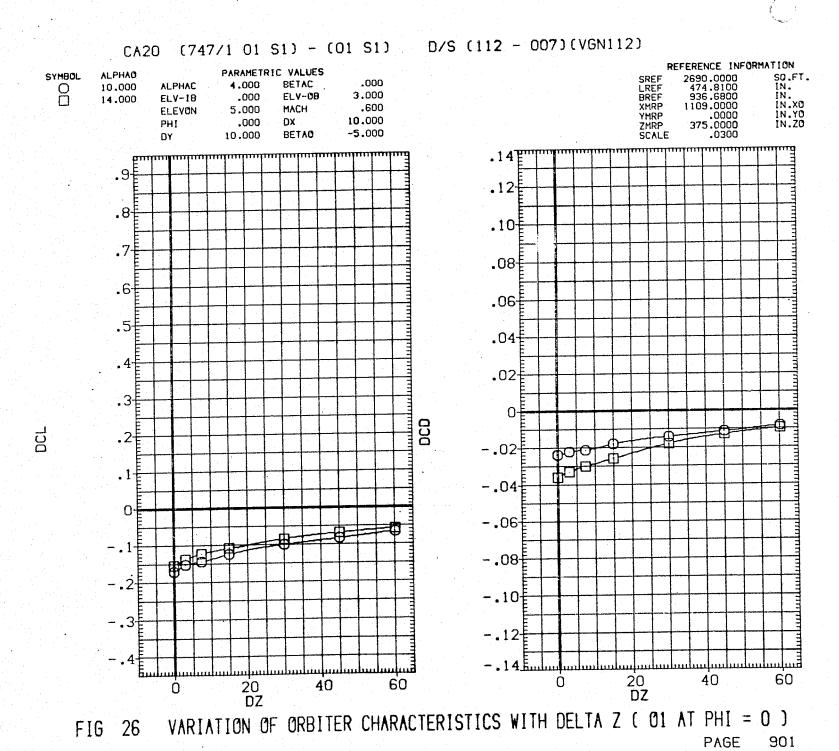


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 902



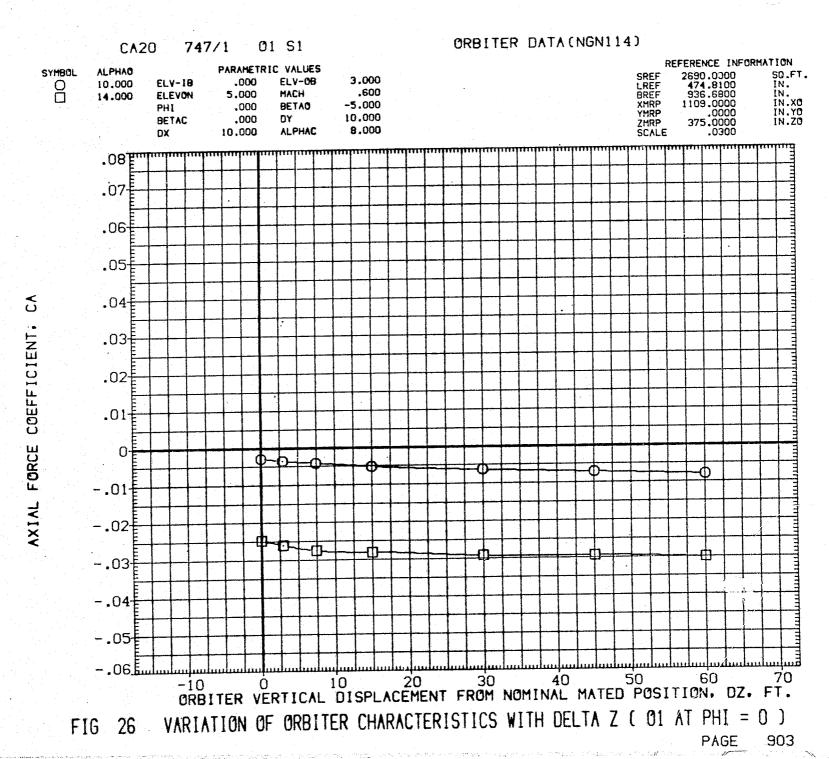


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 904

20 **DZ** 

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20 DZ

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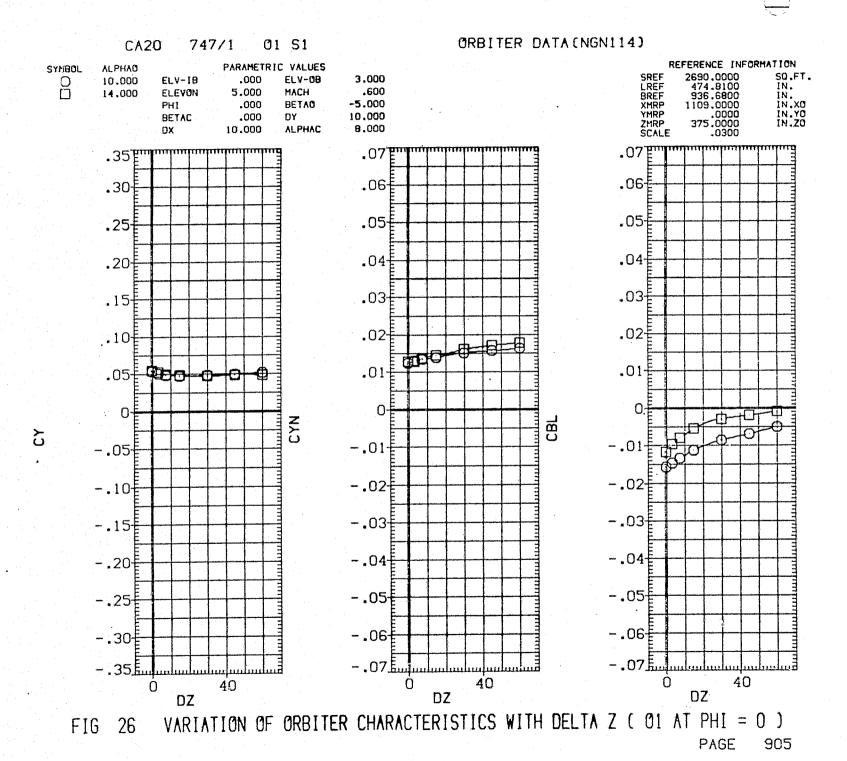
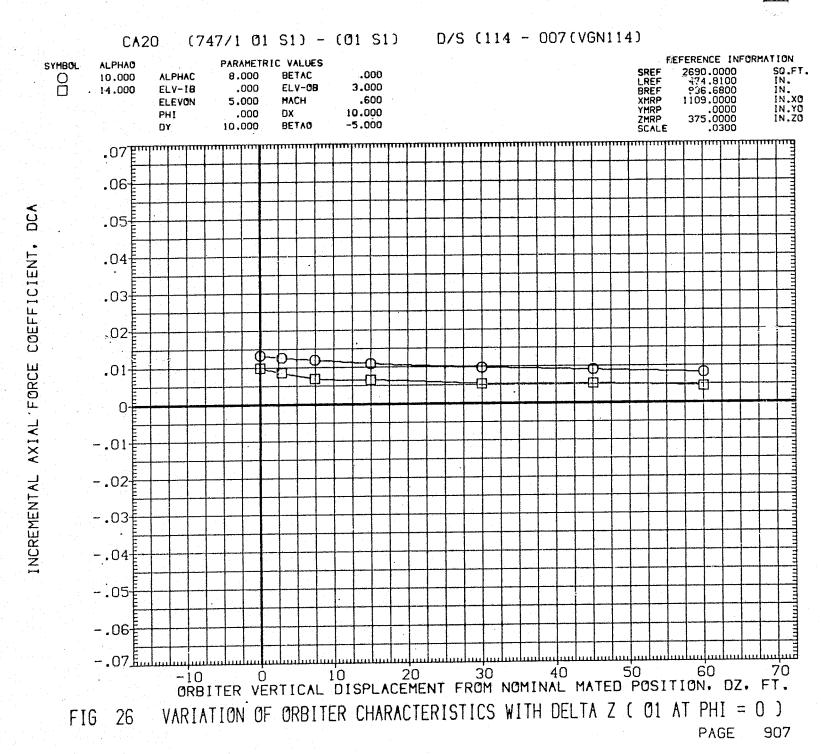
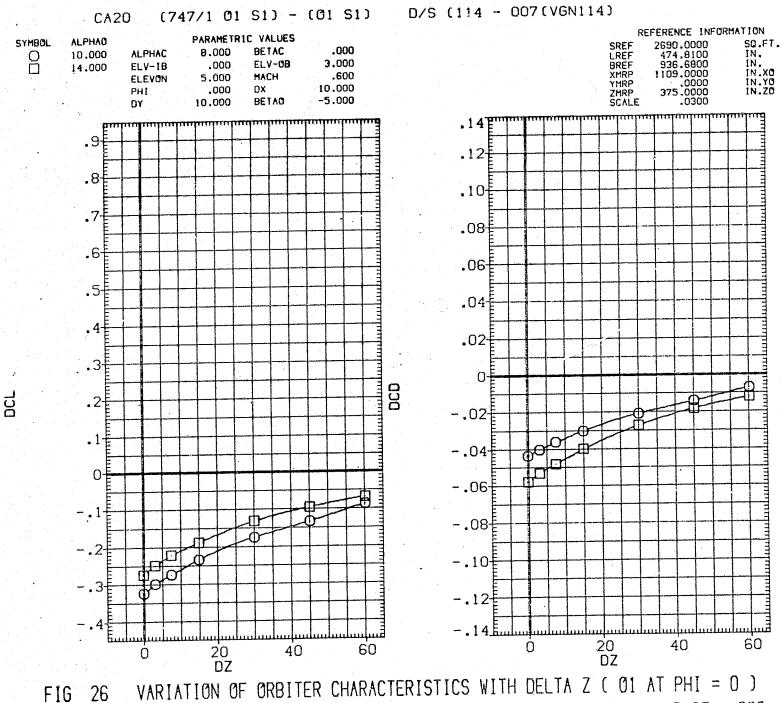


FIG 26 PAGE 906

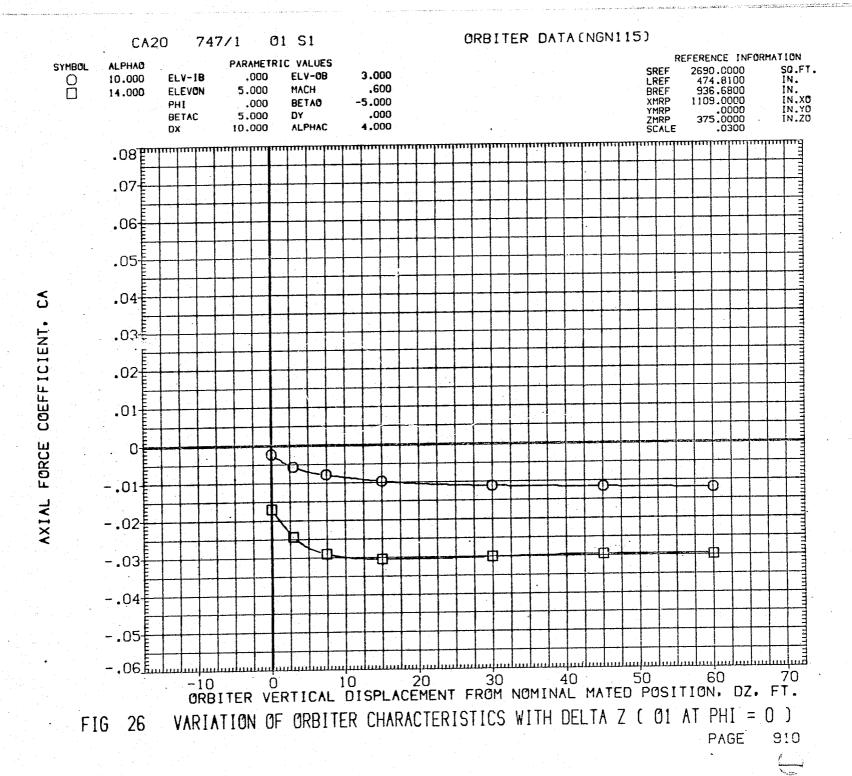
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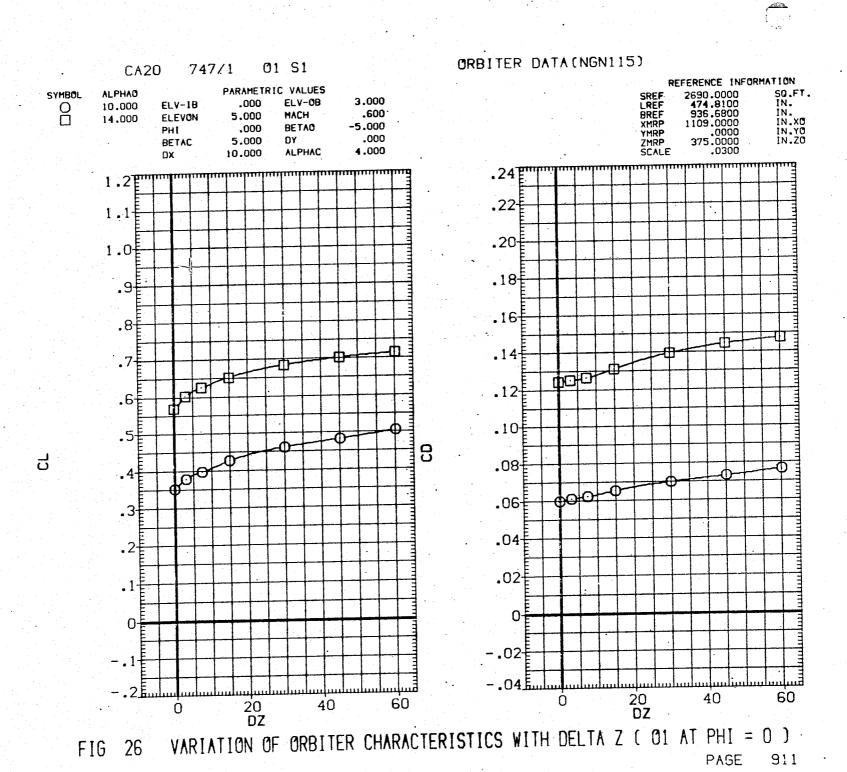






PAGE 908







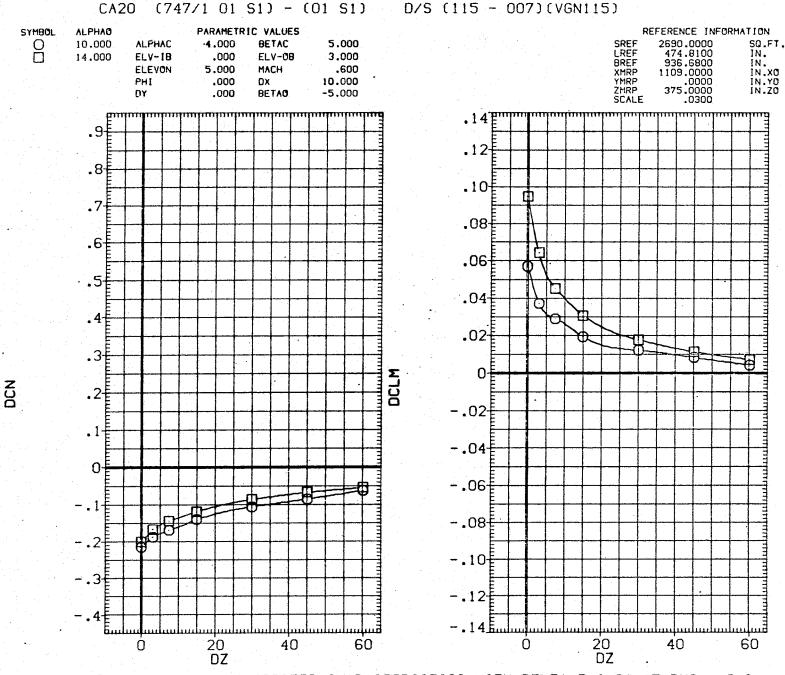


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 913

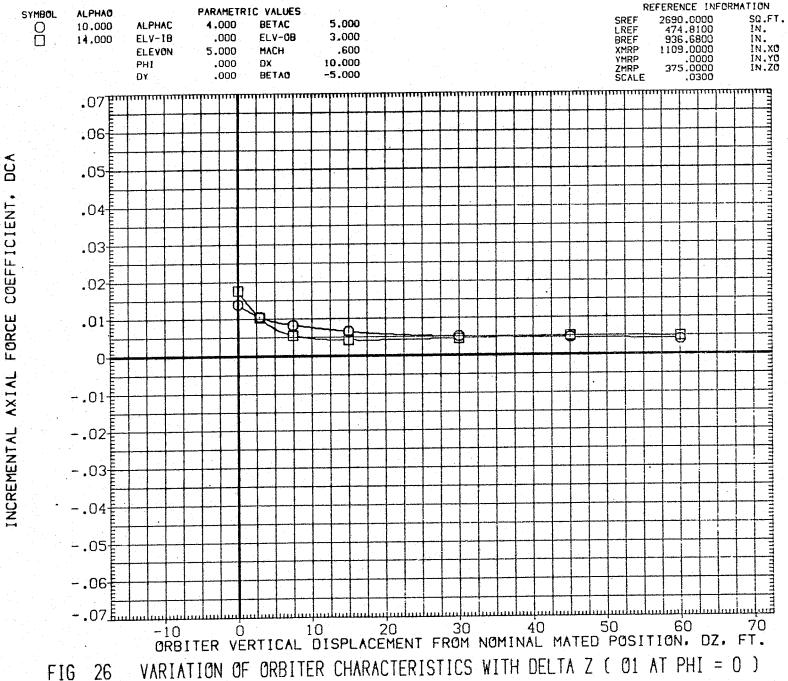


FIG 26

D/S (115 - 007)(VGN115) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL **ALPHAO** 2690.0000 474.8100 936,6800 SO.FT. IN. IN. SREF LREF BREF 4.000 BETAC 5.000 0 10.000 ALPHAC .000 3.000 ELV-08 14.000 ELV-18 IN.XO IN.YO IN.ZO 5.000 MACH .600 ELEVON **XMRP** 1109.0000 YMRP ZMRP .0000 375.0000 .000 DX 10.000 PHI -5.000 .000 BETAO DY SCALE .14 F''' .9₽ .12 .10 .08 .6<del>[</del> .06 .5<del>[</del> .04 .4<del>-</del> .02 0 000 DCL -.02‡ -.04-0 -.06 -.08 -.10 -.12<del>[</del> - . 14 <u>Emlandan</u> 60 20 **DZ** 40 20 40 60 0 0 DZ

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

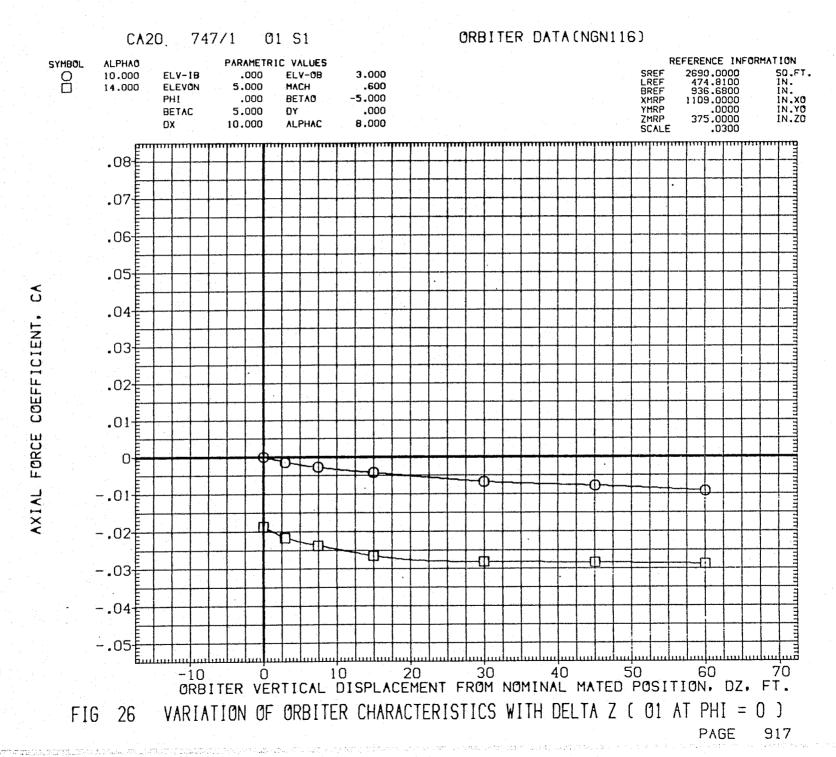
915

20 **DZ** VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG PAGE

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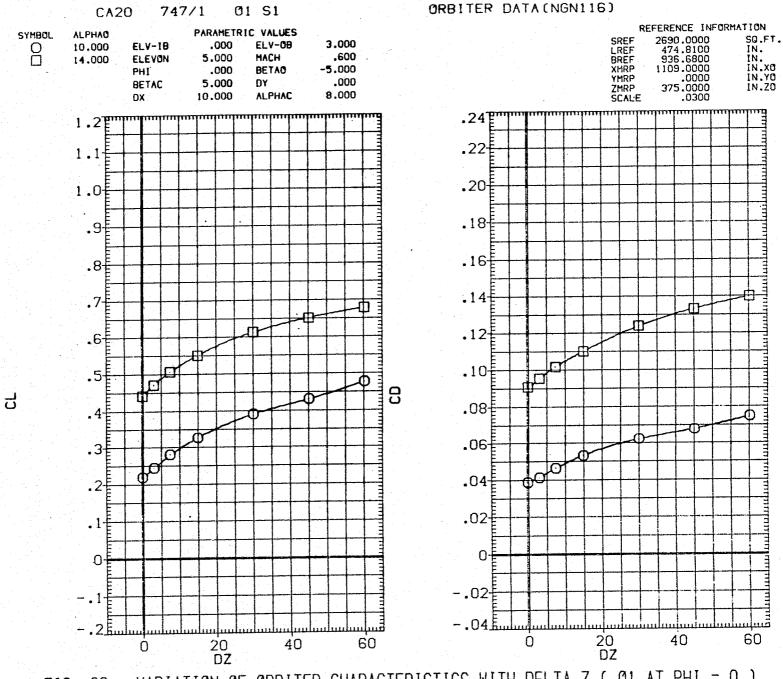


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

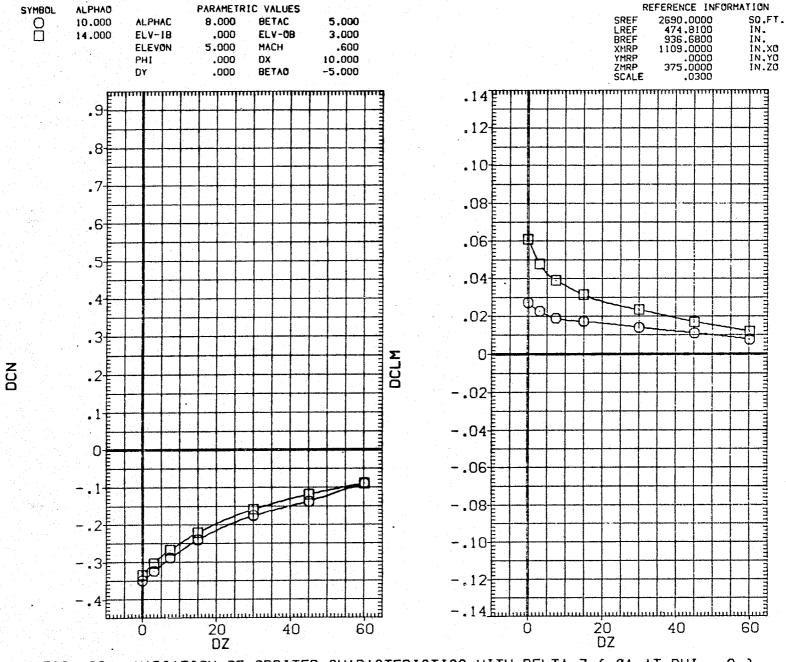


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

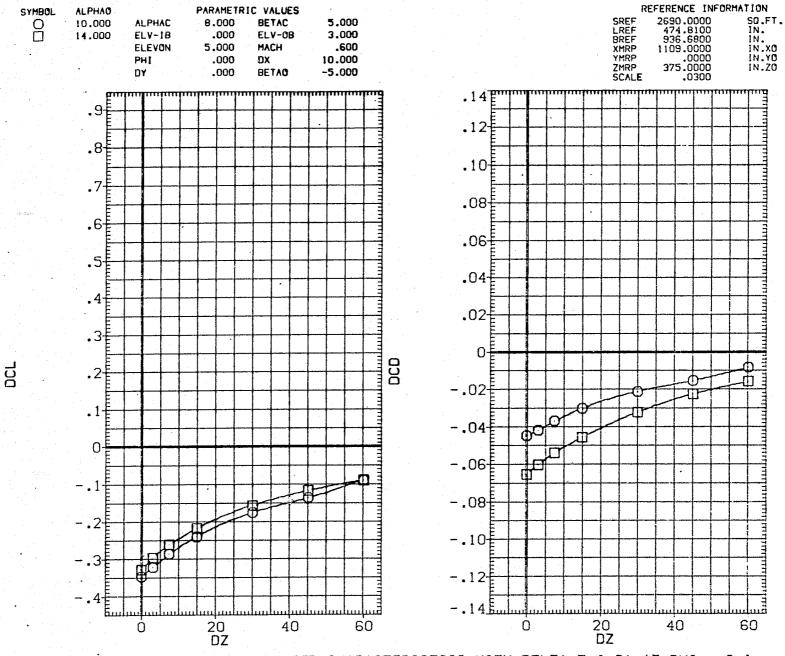


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT.

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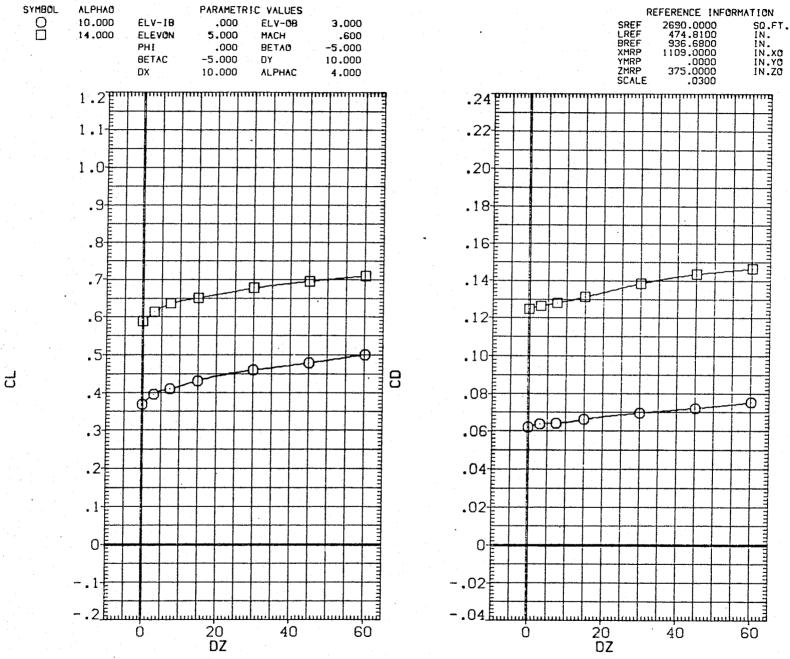
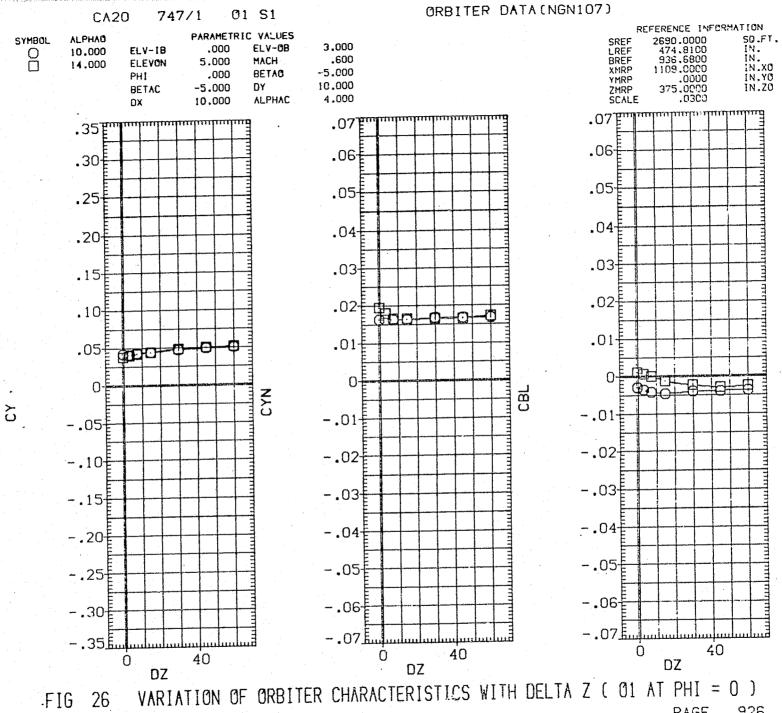


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 925



PAGE

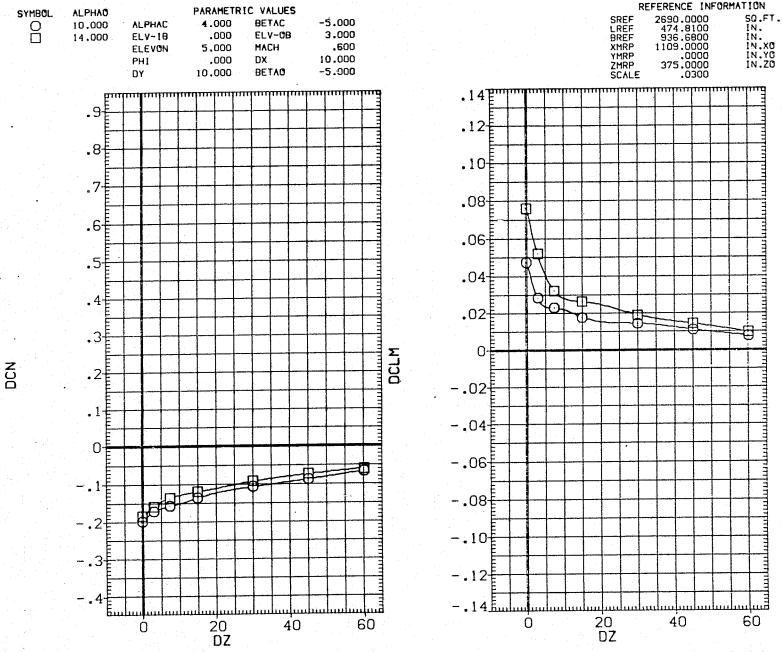
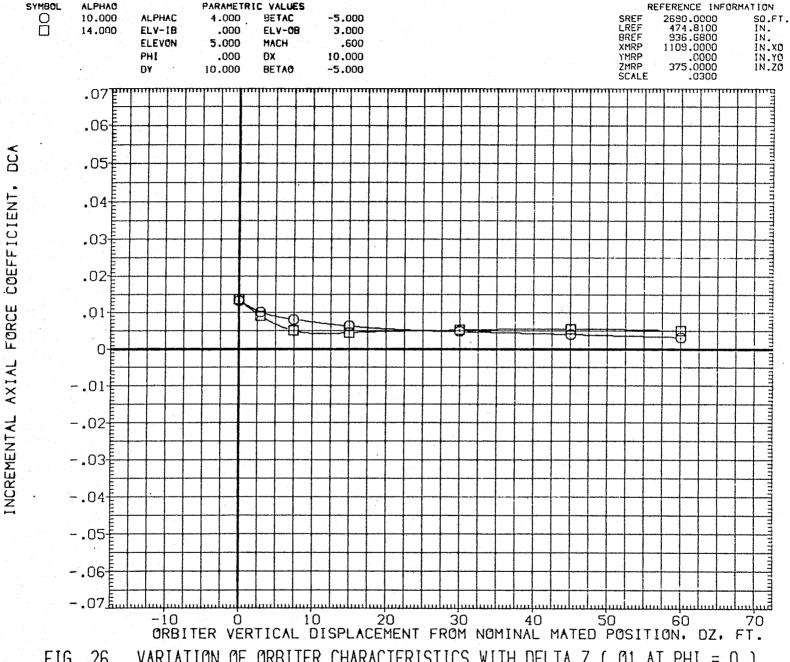


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 927



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

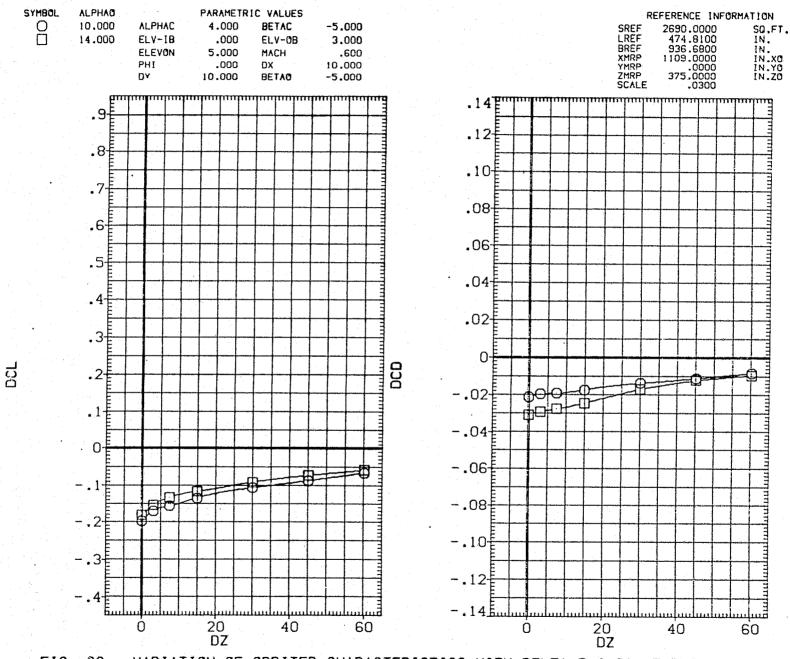


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 929

FIG PAGE 930

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ORBITER DATA (NGN108)

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 932

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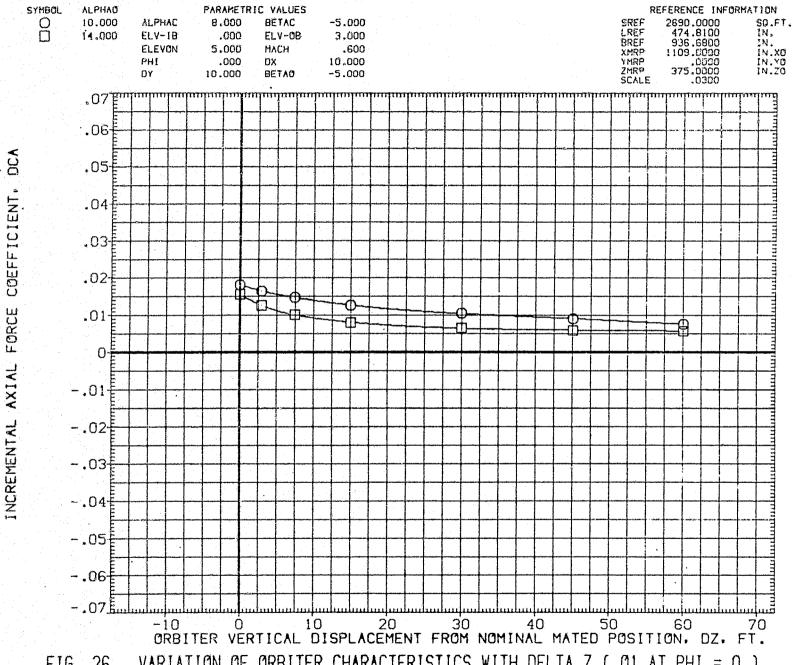
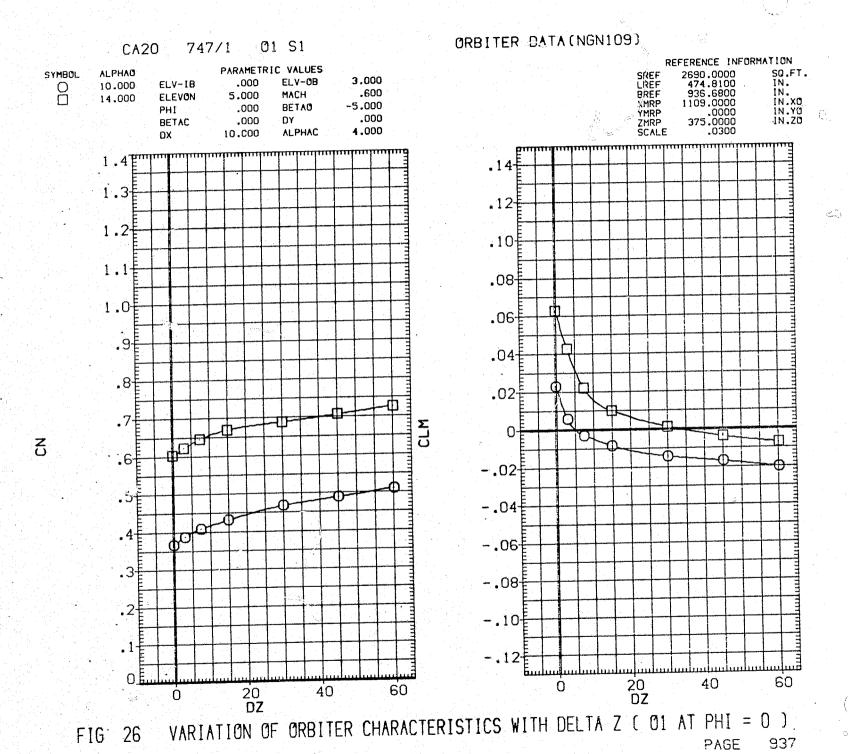
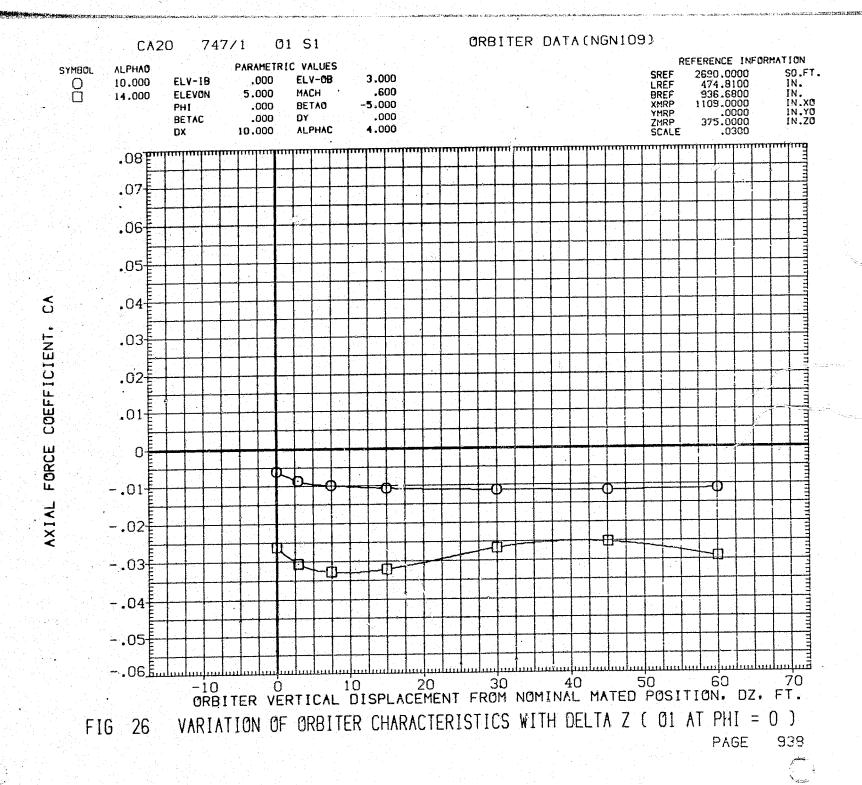


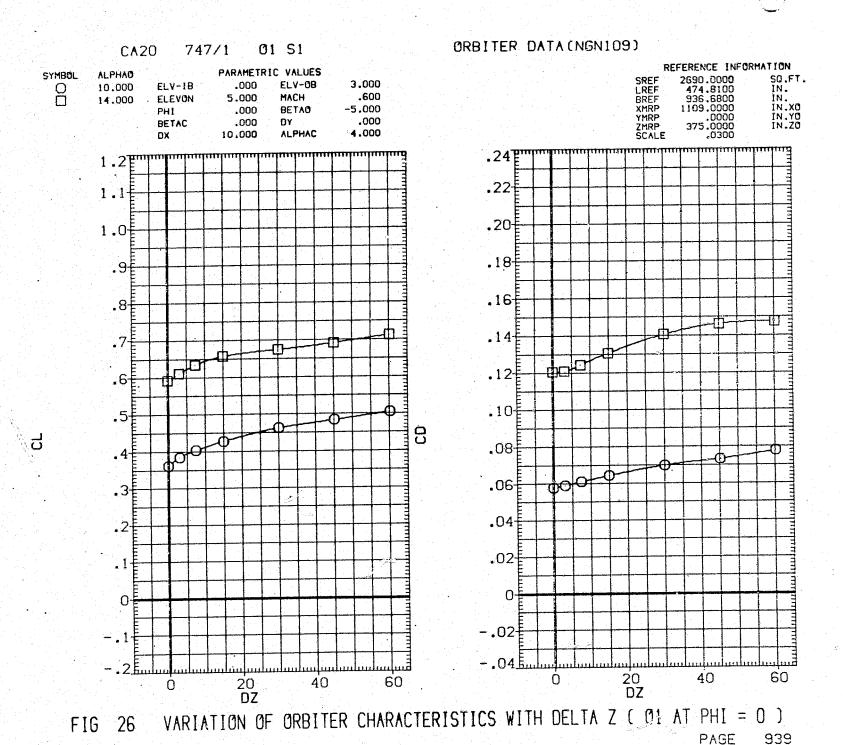
FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 936









PAGE 940

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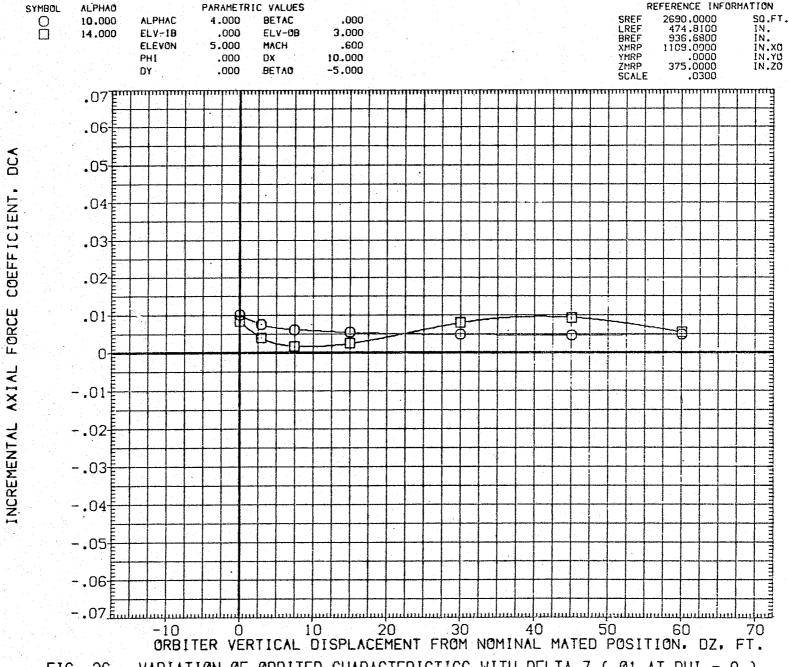
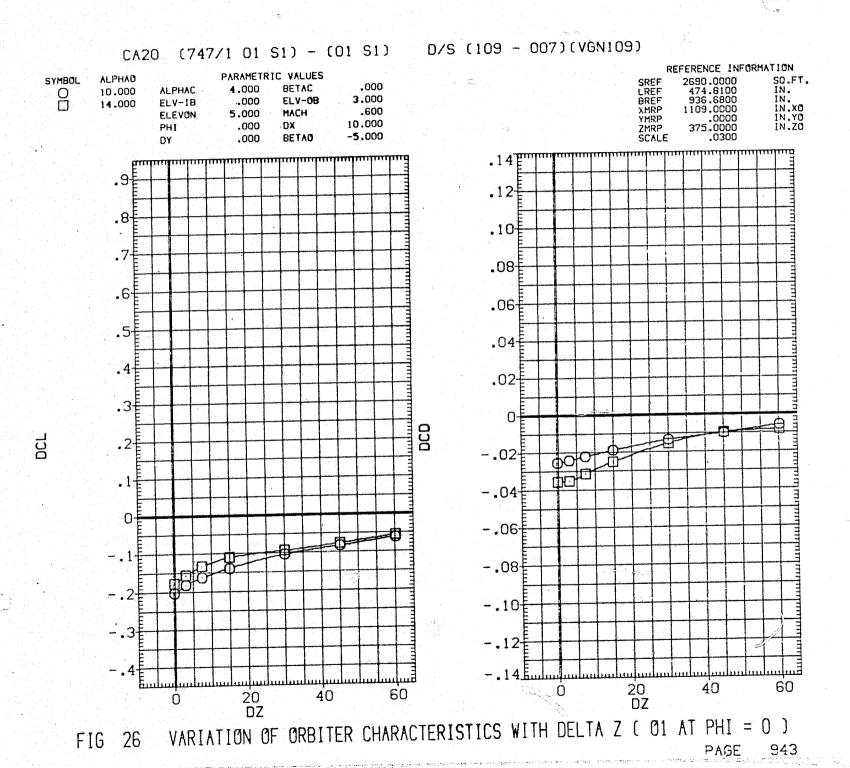


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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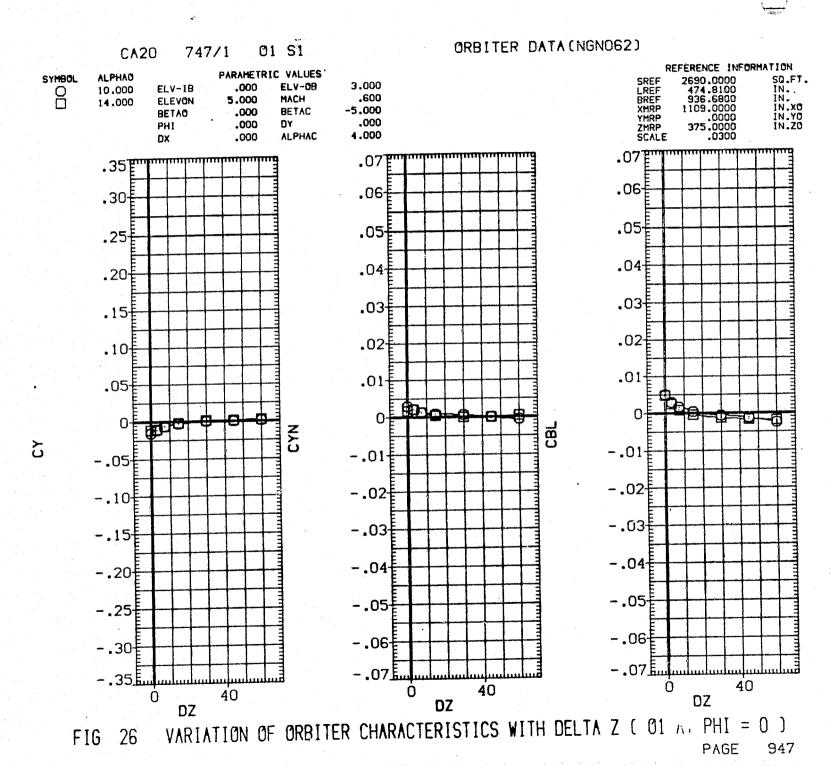
ORBITER DATA (NGNO62) CA20 747/1 01 S1 REFERENCE INFORMATION PARAMETRIC VALUES **ALPHAO** 2690.0000 474.8100 936.6800 1109.0000 SO.FT. IN. IN. .000 ELV-OB 3.000 00 10.000. LREF BREF XMRP .600 ELEVON 5.000 14.000 IN.X0 IN.Y0 IN.Z0 .000 BETAC -5.000 BETAD YMRP ZMRP SCALE .000 .030 DY 4.000 .000 **ALPHAC** .055 բարարարարարարարարարարարարա .050 .045 .040 Š .035 CREFF! CIENT. .030 .025 .020<del>[</del> AXIAL FORCE .015<del>[</del> .010 .005 0 -.005-.010<del>[</del> سطِّ 015. – 50 60 30 40 20 10 -100 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG

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PAGE





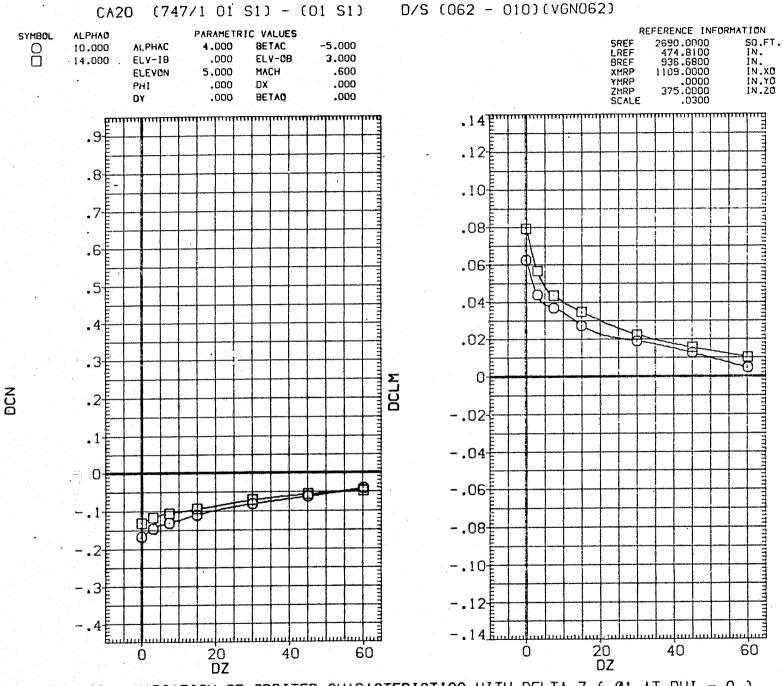


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )



CA20 (747/1 01 S1) - (01 S1) D/S (062 - 010)(VGN062) SYMBOL **ALPHAO** PARAMETRIC VALUES REFERENCE INFORMATION 4.000 0 10.000 ALPHAC BETAC -5.000 SO.FT. 2690.0000 LREF 474.8100 IN. 14.000 ELV-IB .000 ELV-0B 3.000 936.6800 1109.0000 .0000 375.0000 BREF IN.XO IN.YO IN.ZO ELEVON 5.000 MACH .600 XMRP PHI .000 .000 DX YMRP ZMRP DY .000 BETAO .000 SCALE .06 DCA .05 COEFFICIENT, .04 .03 .02<del>-</del> FORCE .01± φ 0 0 AXIAL -.01+INCREMENTAL -.02<del>[</del> -.03-.04 -.05 -.06<del>-</del> -.07<u>£</u>... -10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26

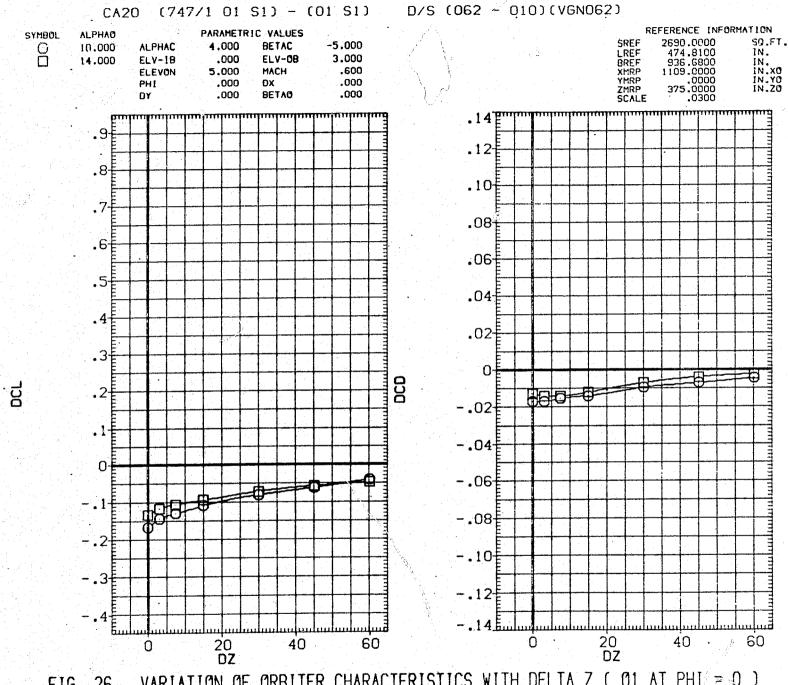
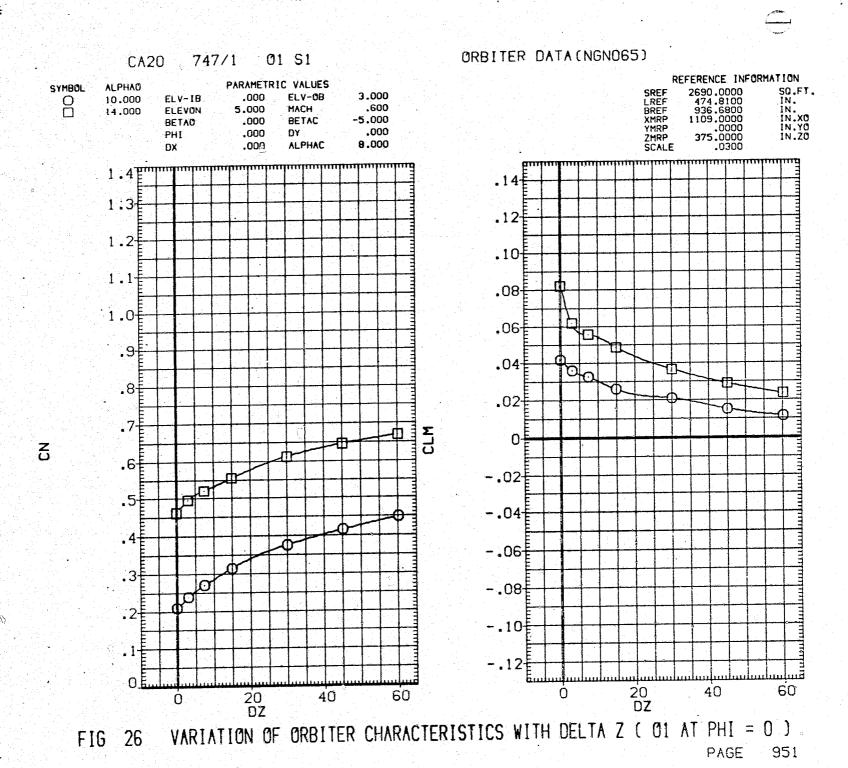
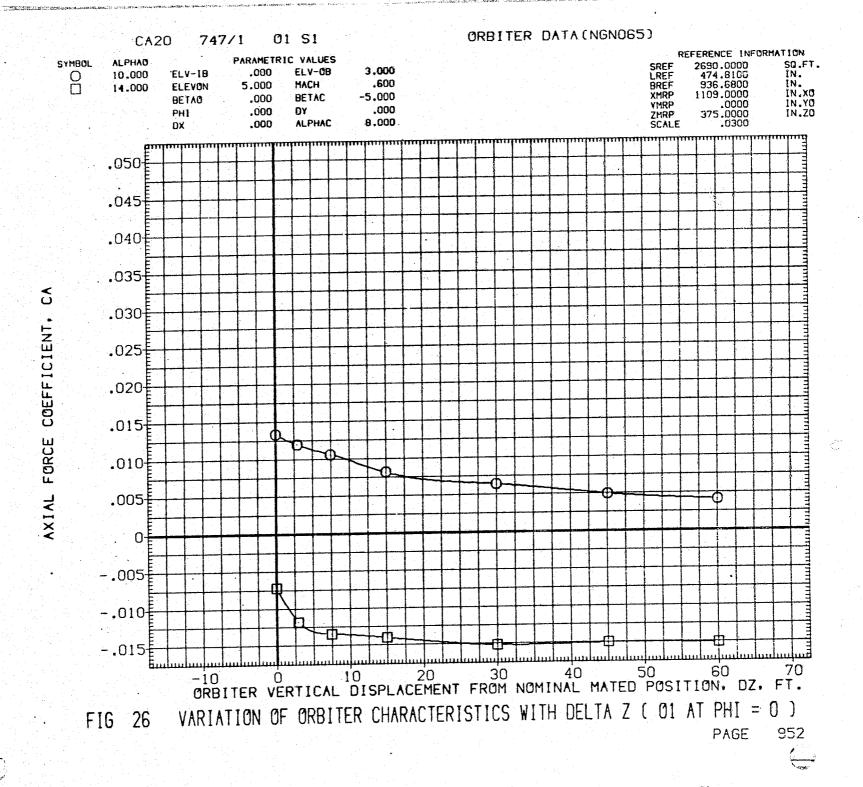
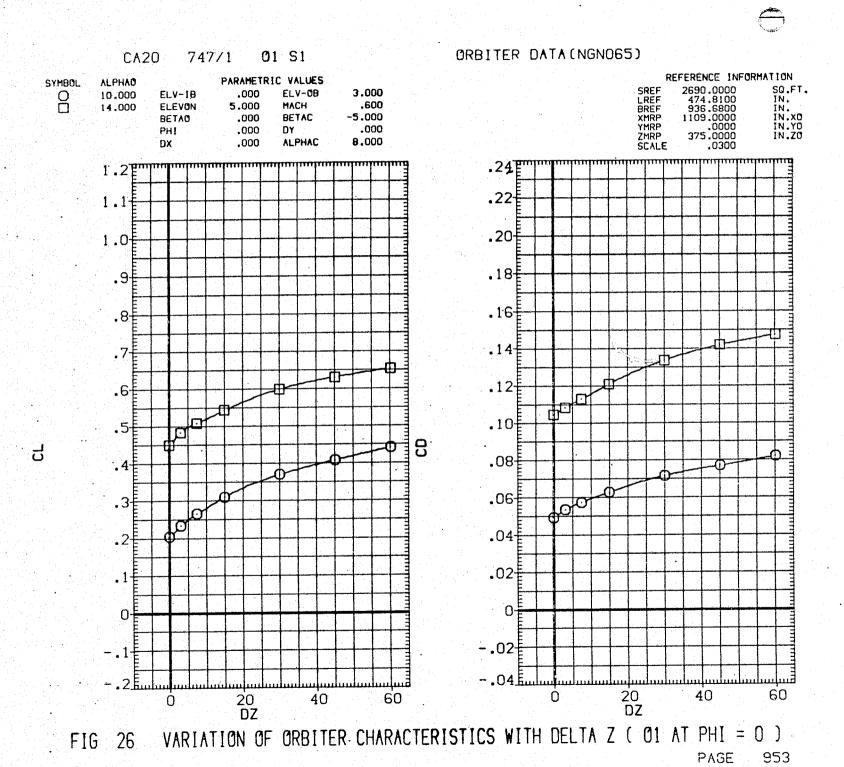


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 950









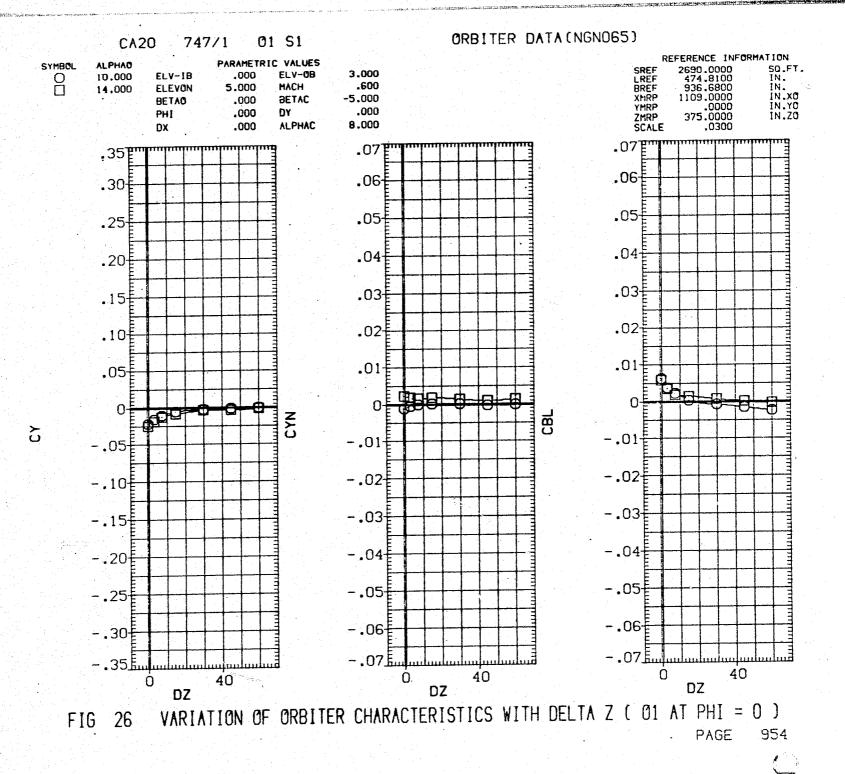
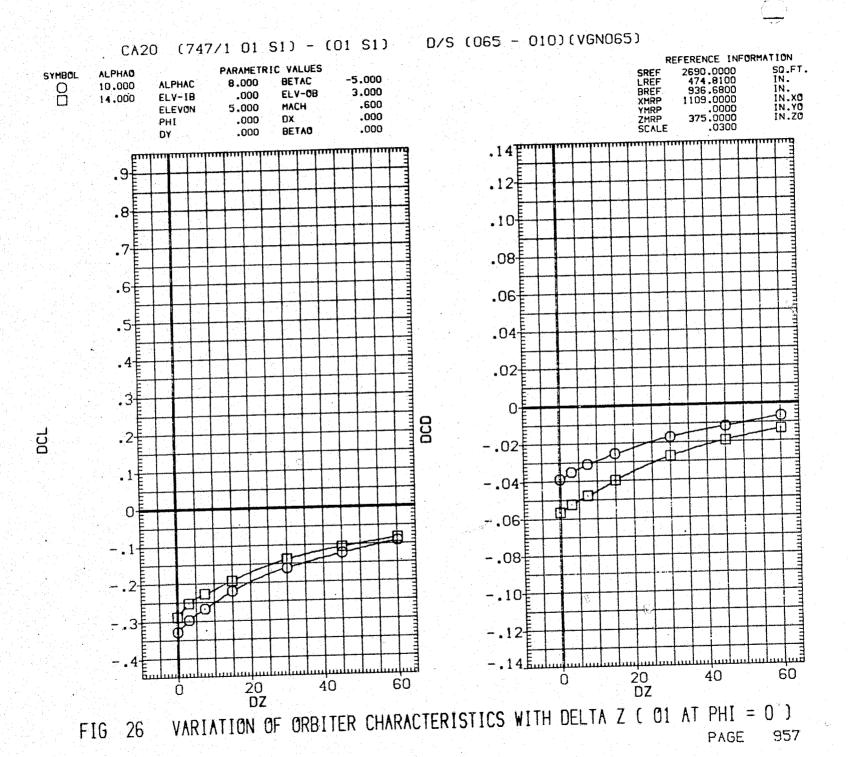
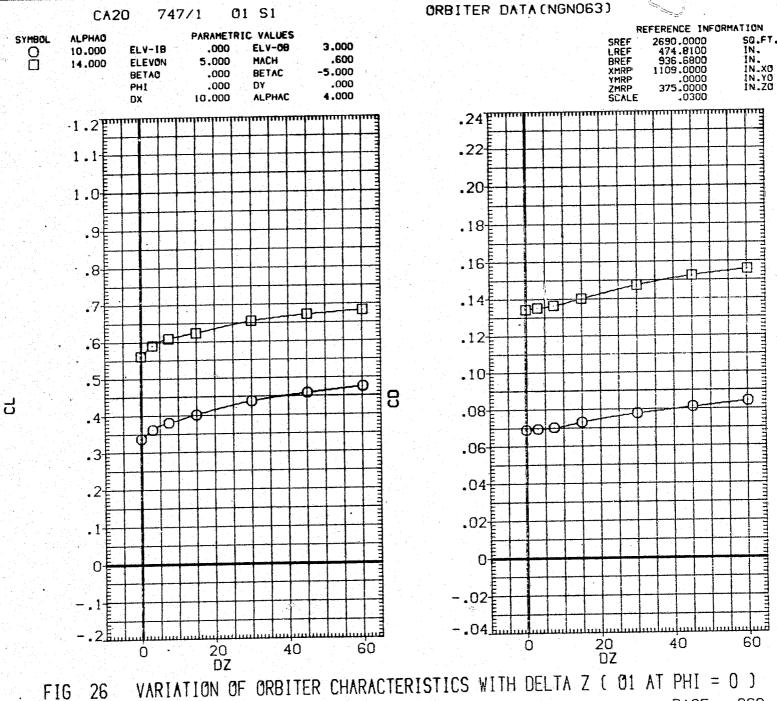


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 955



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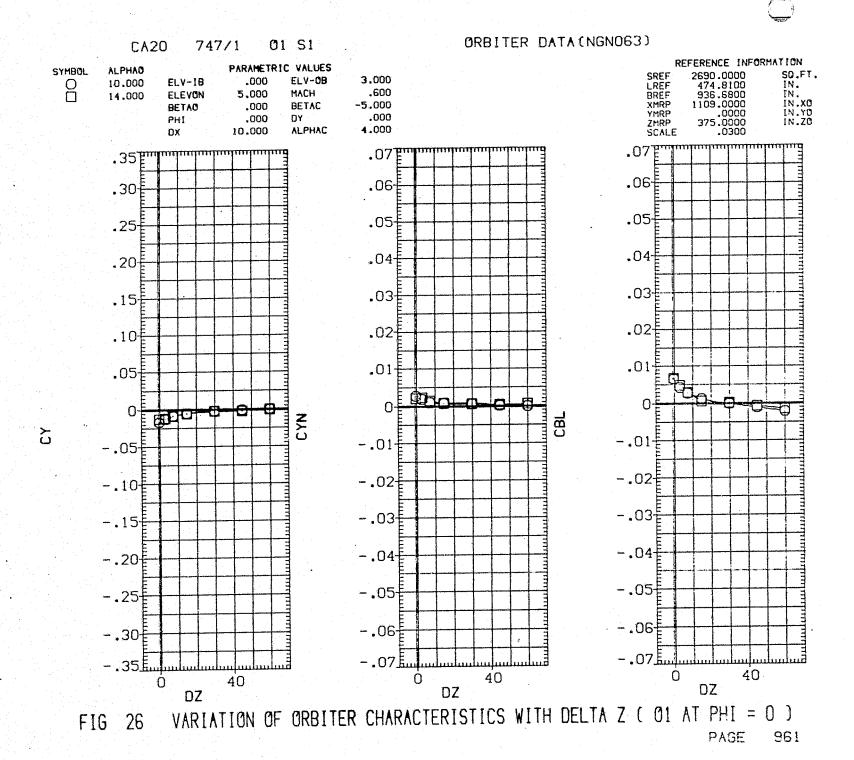
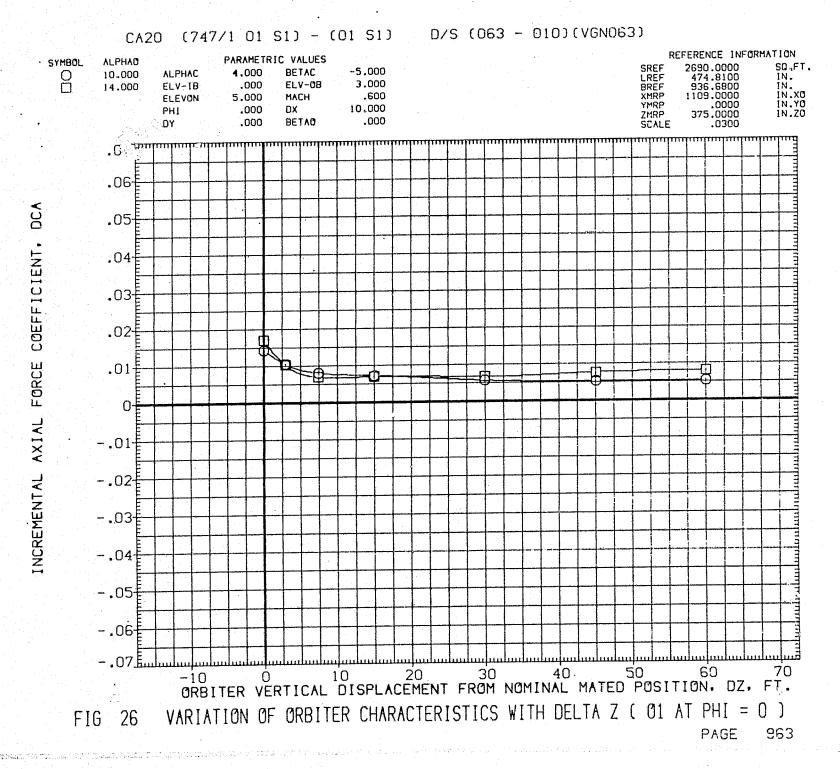


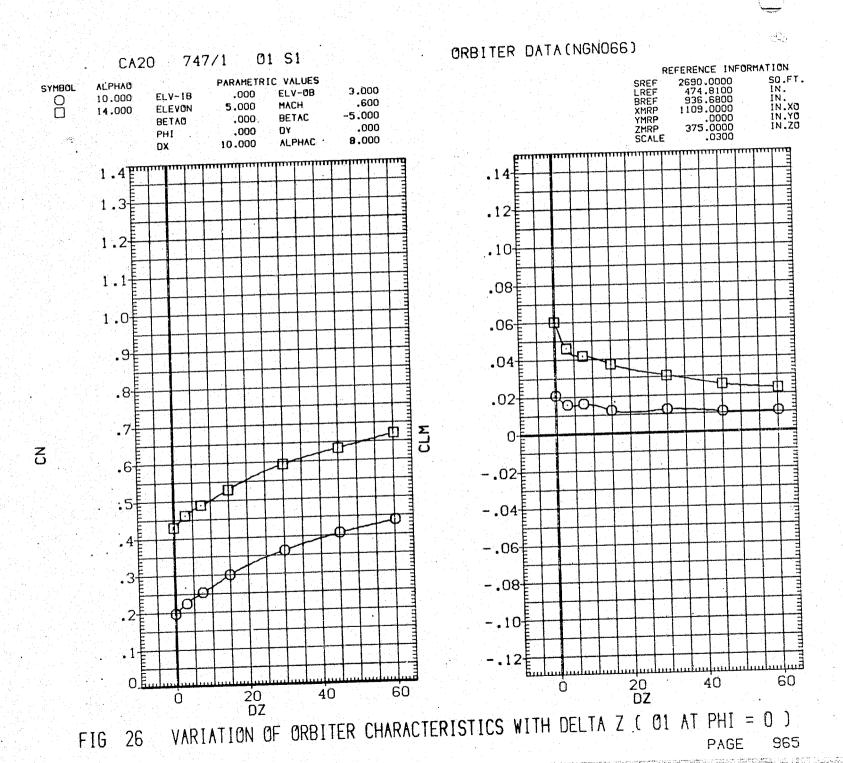
FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 96:

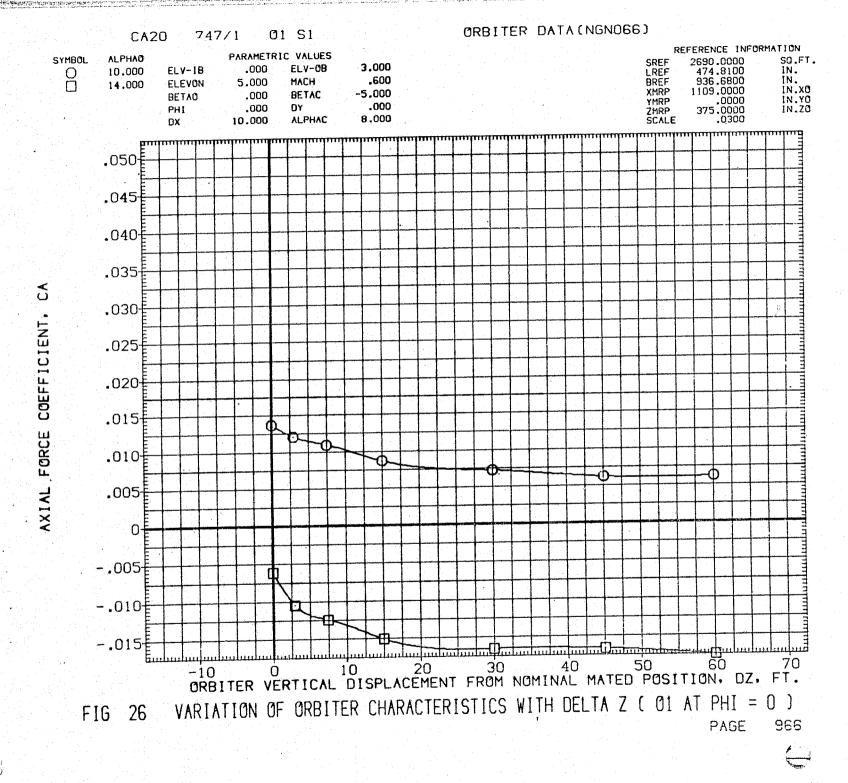




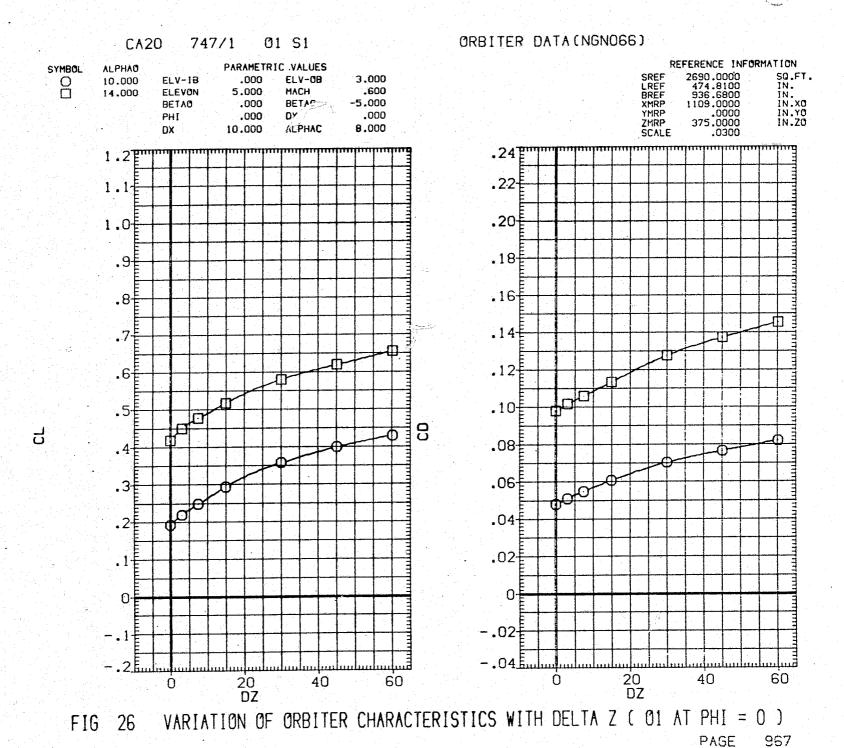
26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) PAGE



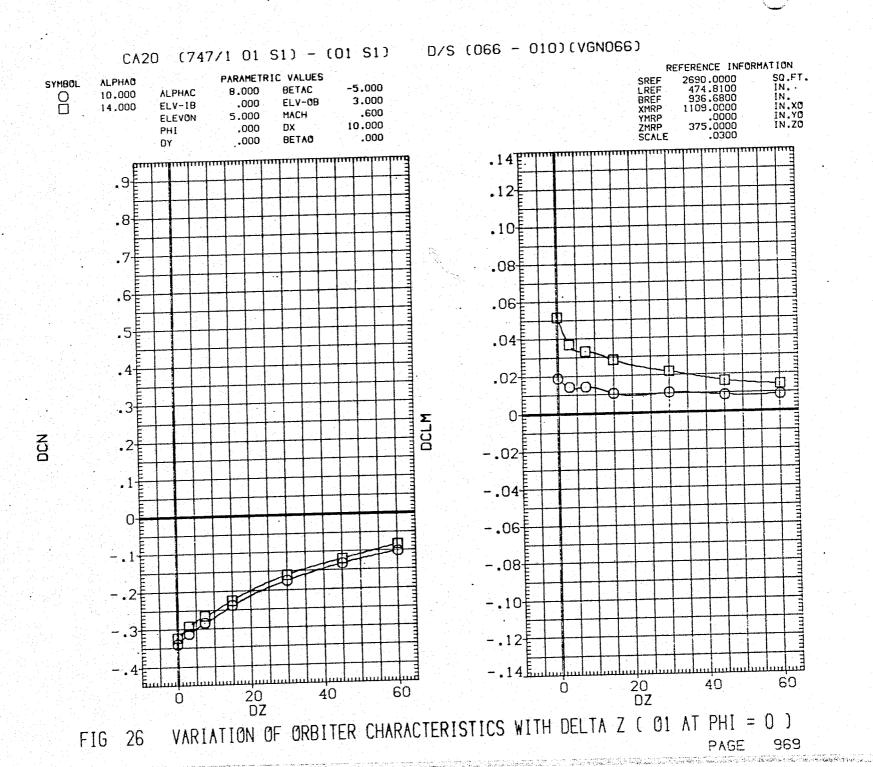




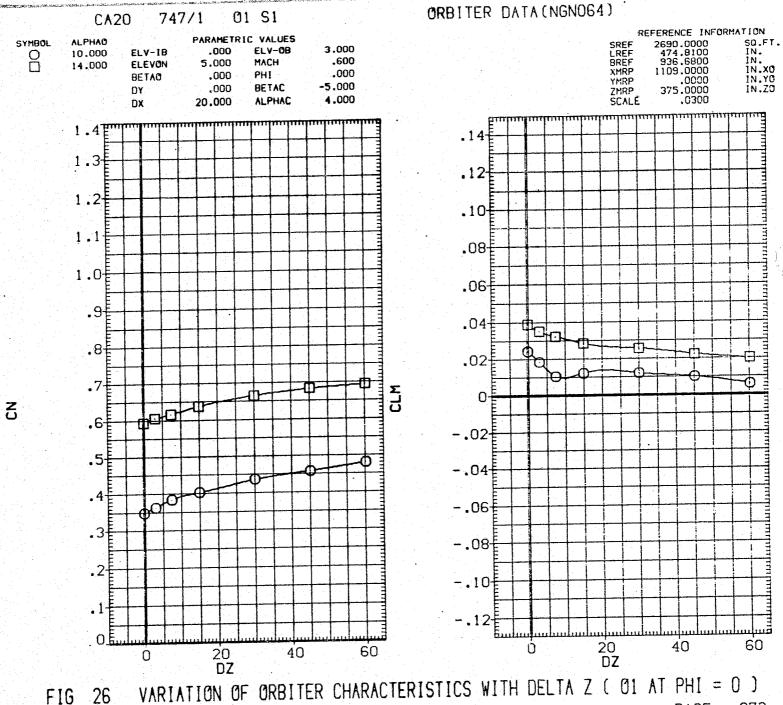








PAGE



26 FIG 972 PAGE

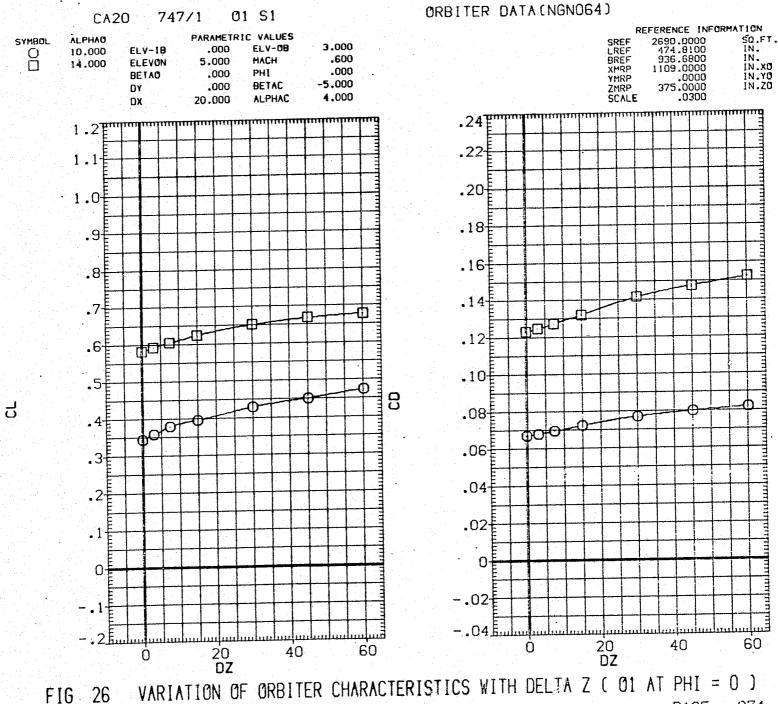


FIG 26 PAGE 974

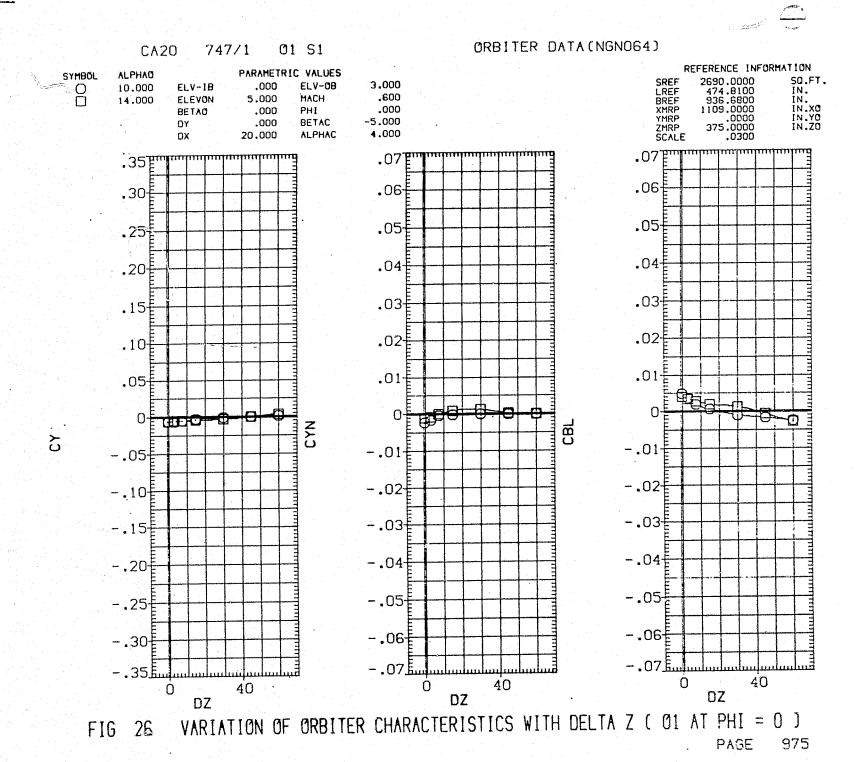


FIG 976 PAGE



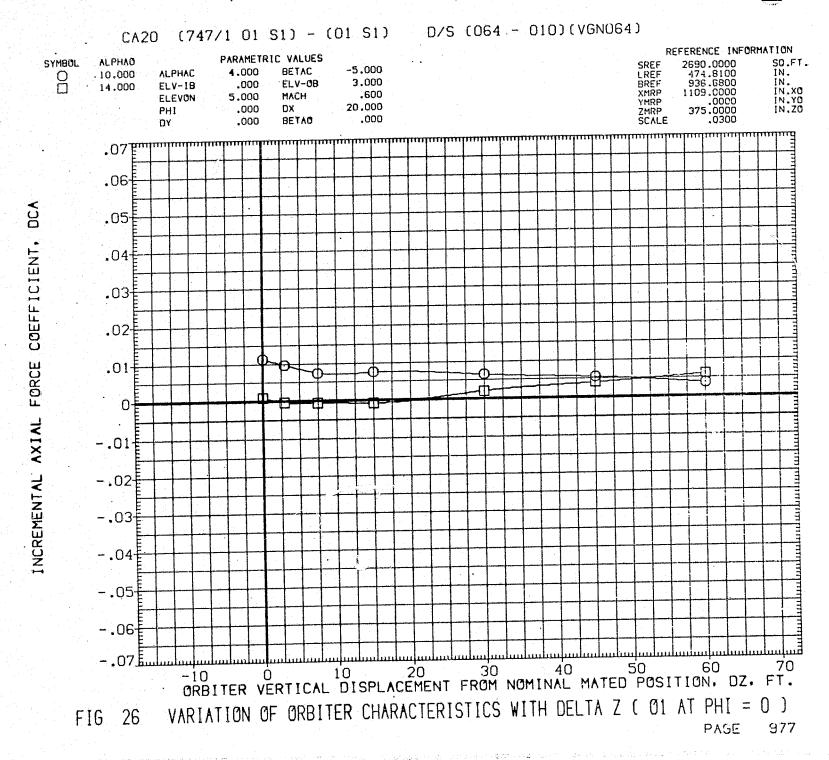


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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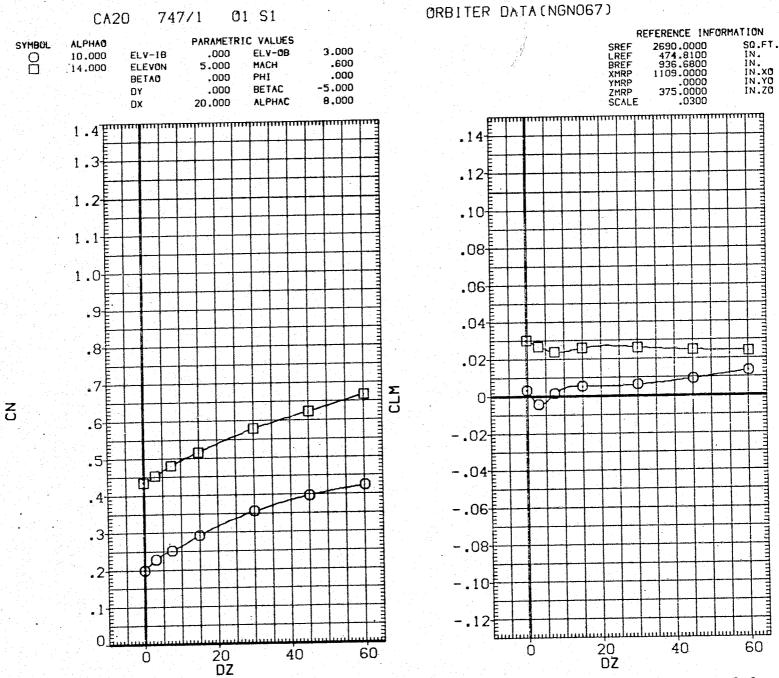
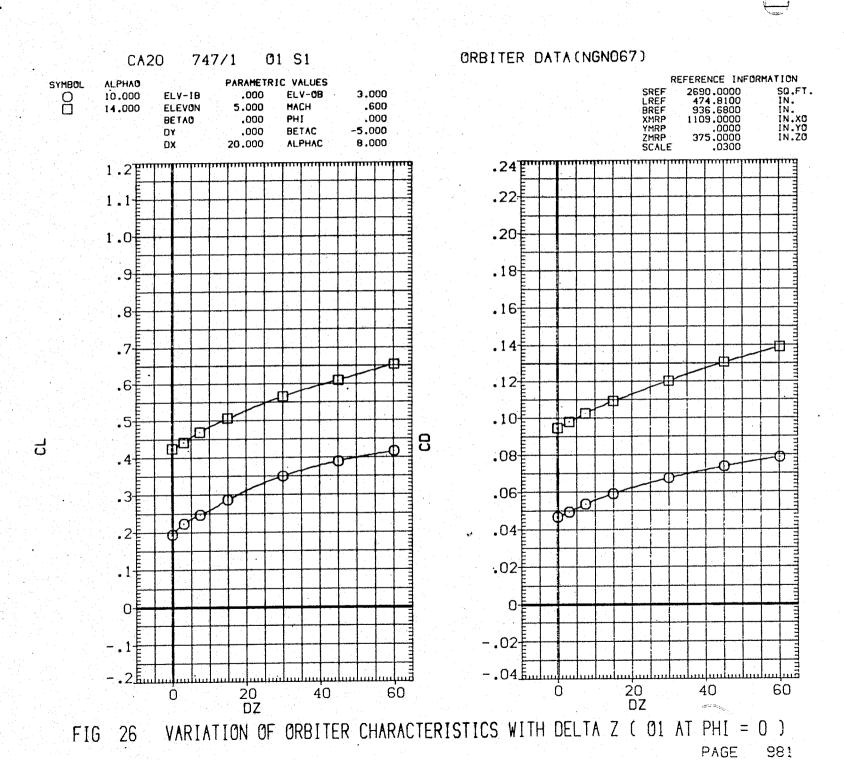
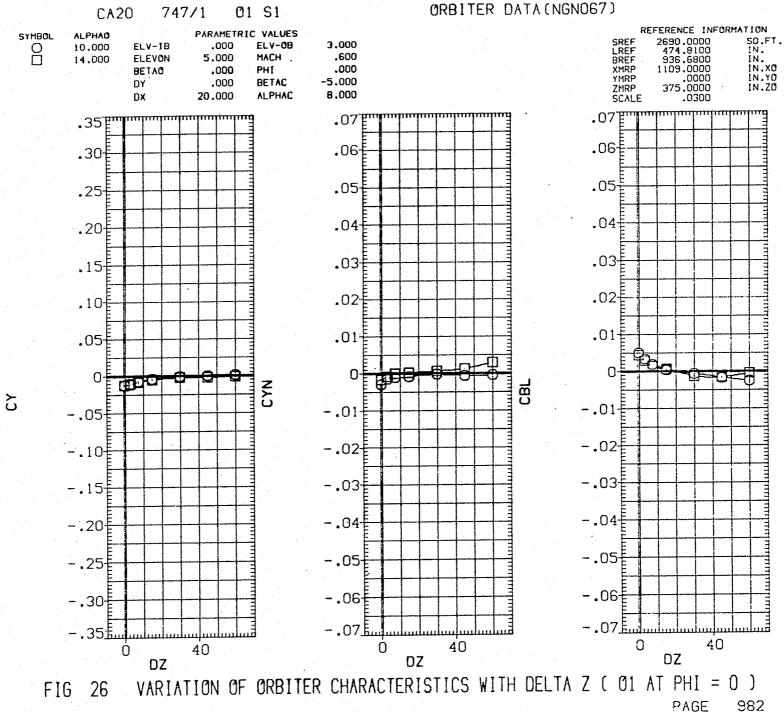


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )



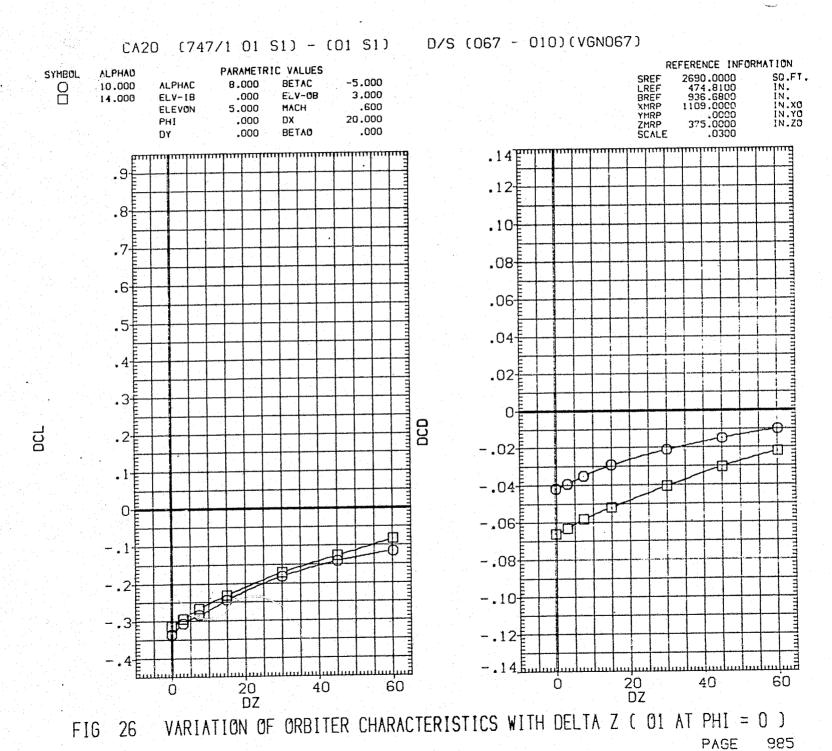


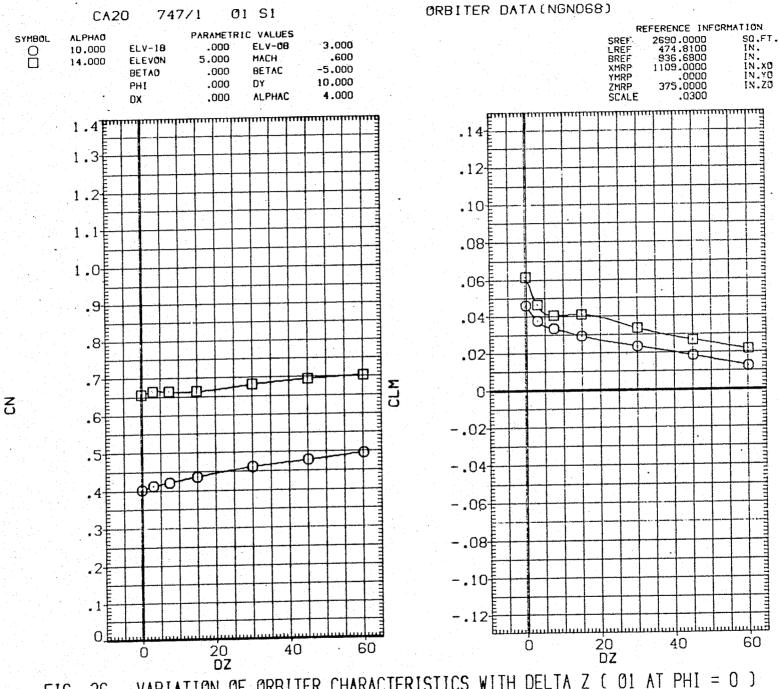
PAGE

D/S (067 - 010)(VGN067) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES **ALPHAO** SYMBOL SREF LREF BREF XMRP YMRP ZMRP SCALE 2690.0000 474.8100 936.6800 1109.0000 SO.FT. IN. IN. IN.XO IN.YO IN.ZO -5.000 8.000 BETAC 0 10.000 **ALPHAC** .000 ELV-0B 3.000 14,000 ELV-1B .600 5.000 MACH **ELEVON** .0000 20.000 .000 DX. PHI 375.0000 .000 DΥ .000 BETAO .0300 .14 Em .9 .12 .8 .10 .08 .6<del>-</del> .06<del>[</del> .5 .04 .02 DCLM 'n DCN DCN .2<del>[</del> -.02 .1卡 -.04 O -.06 -.08<del>[</del> -.10<del>[</del> -.12<del>[</del> -.14 Em 20 DZ 40 60 0 20 DZ 60 40 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) 26 FIG

PAGE

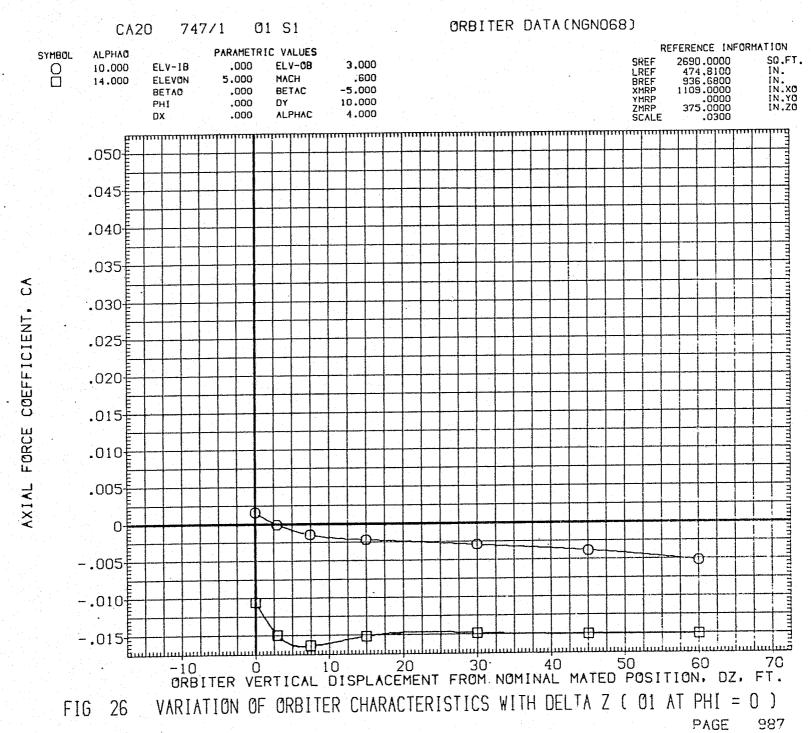






VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG PAGE 986





VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26

PAGE 983

FIG PAGE 990

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-10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) 26 FIG PAGE 991

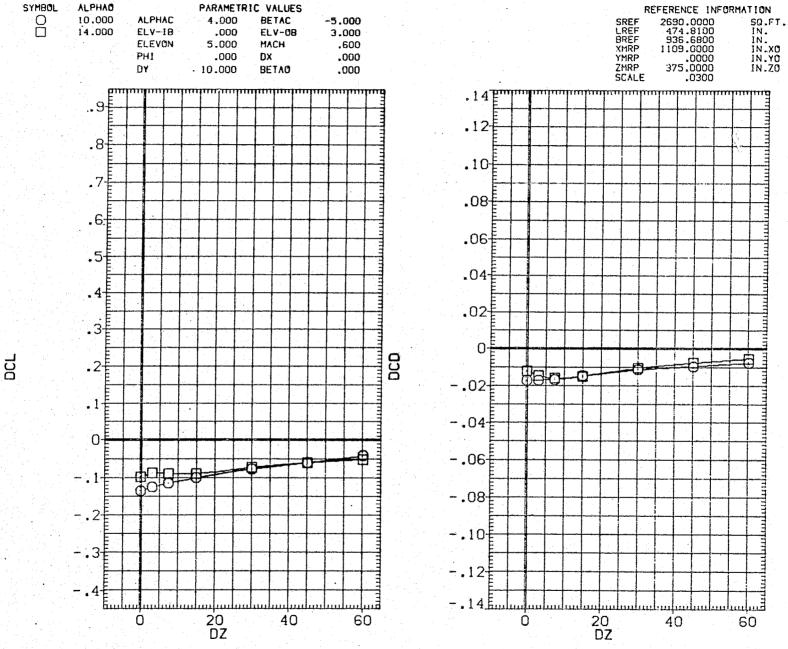
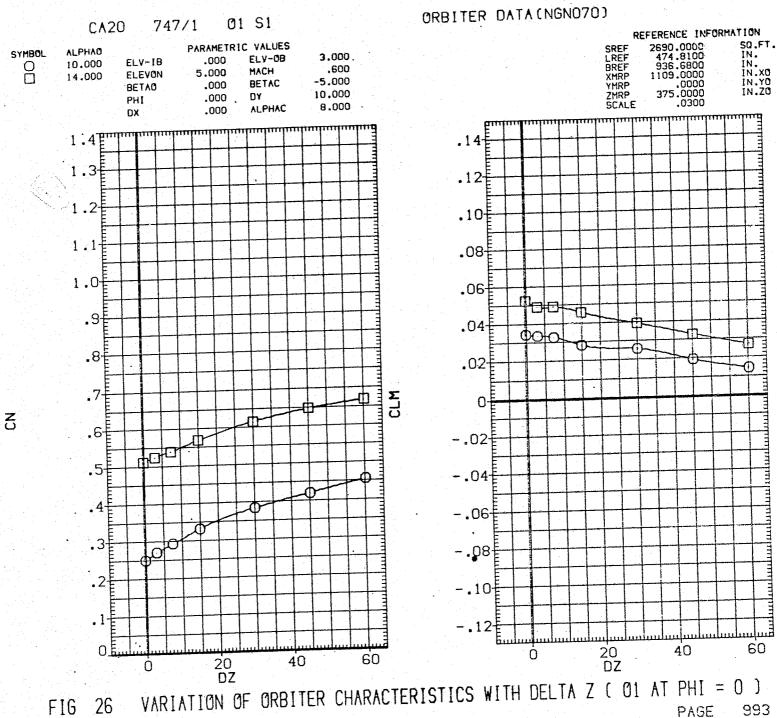


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 992





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PAGE

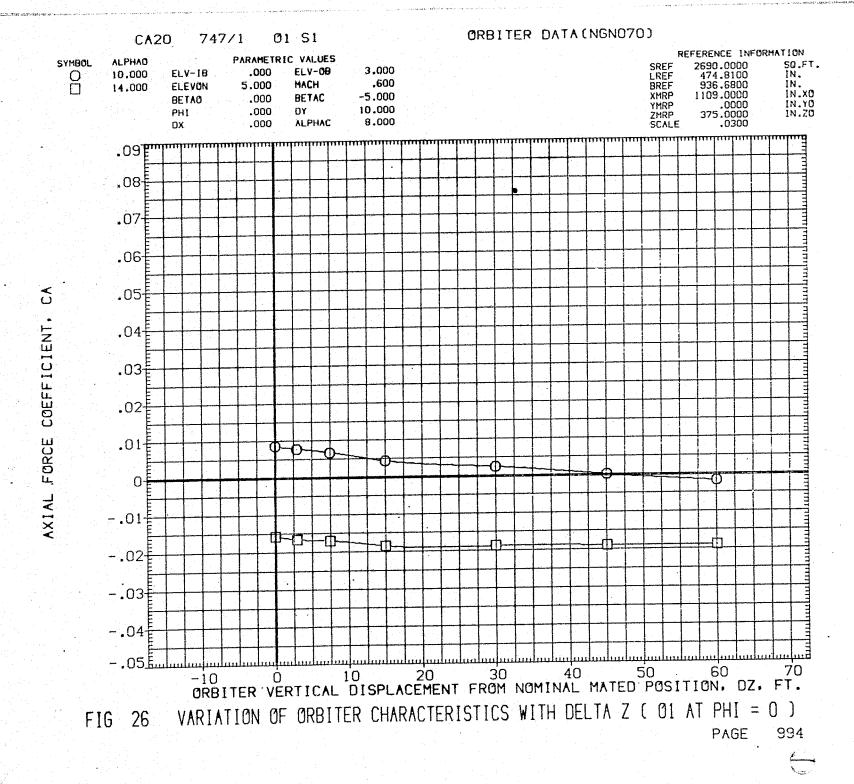
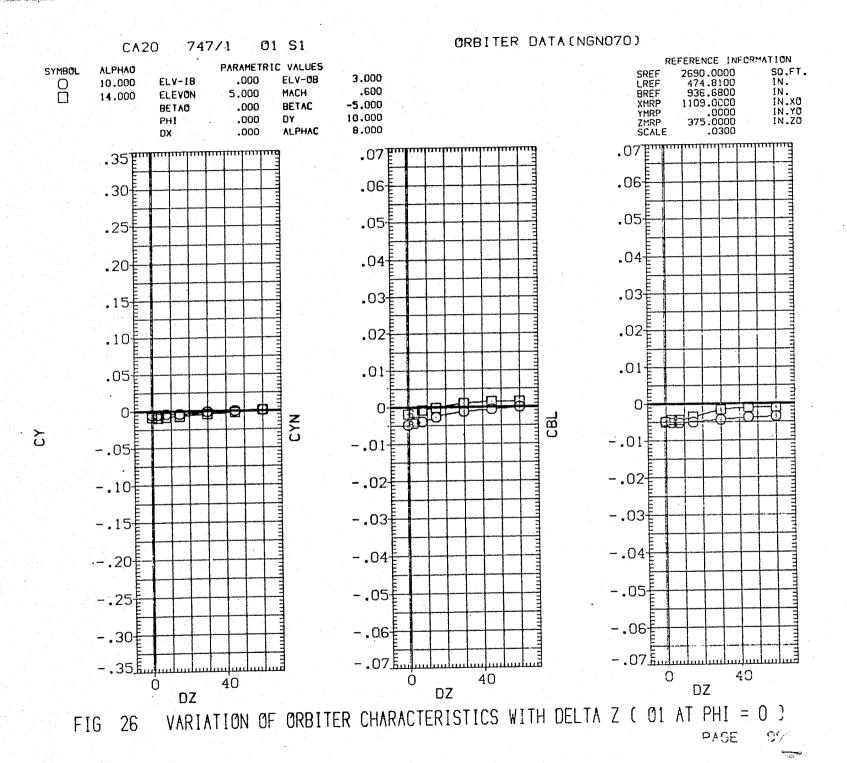


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 995



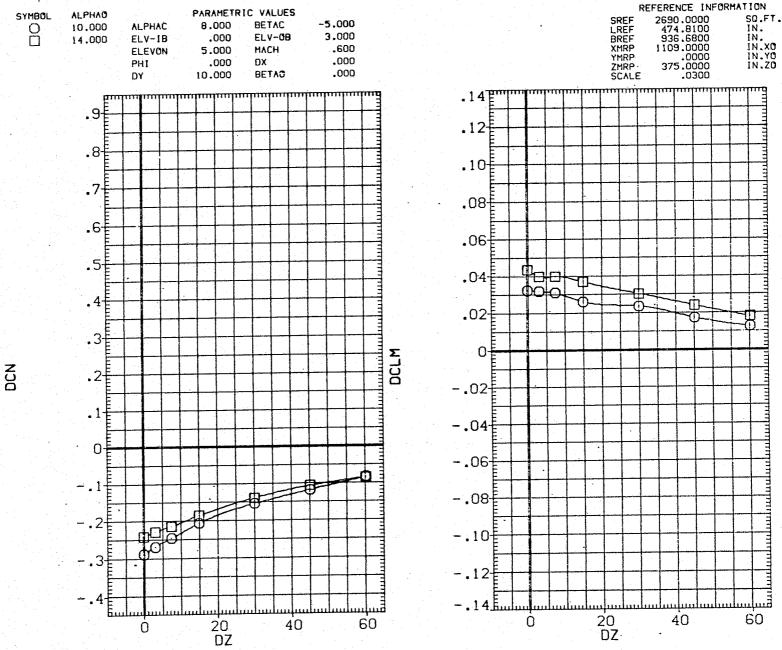


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

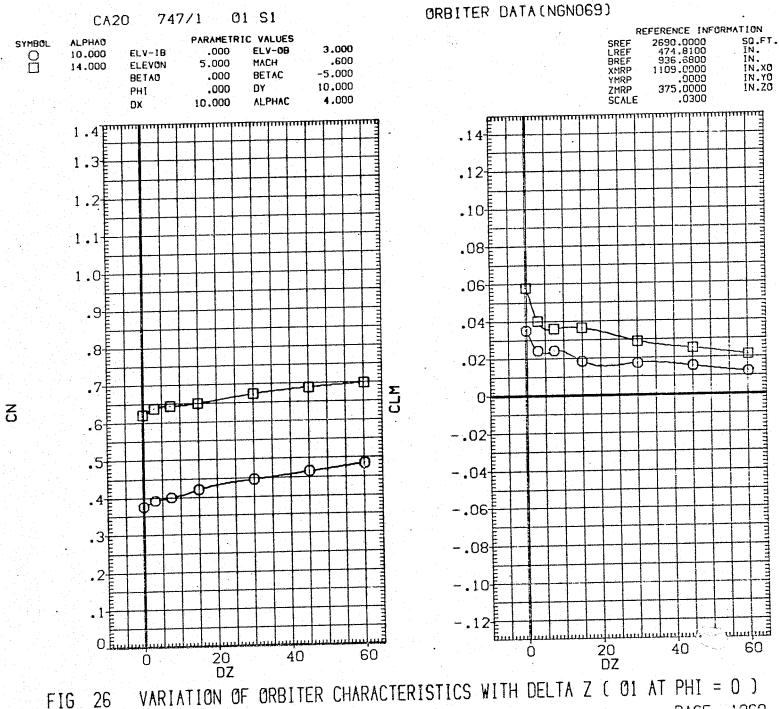
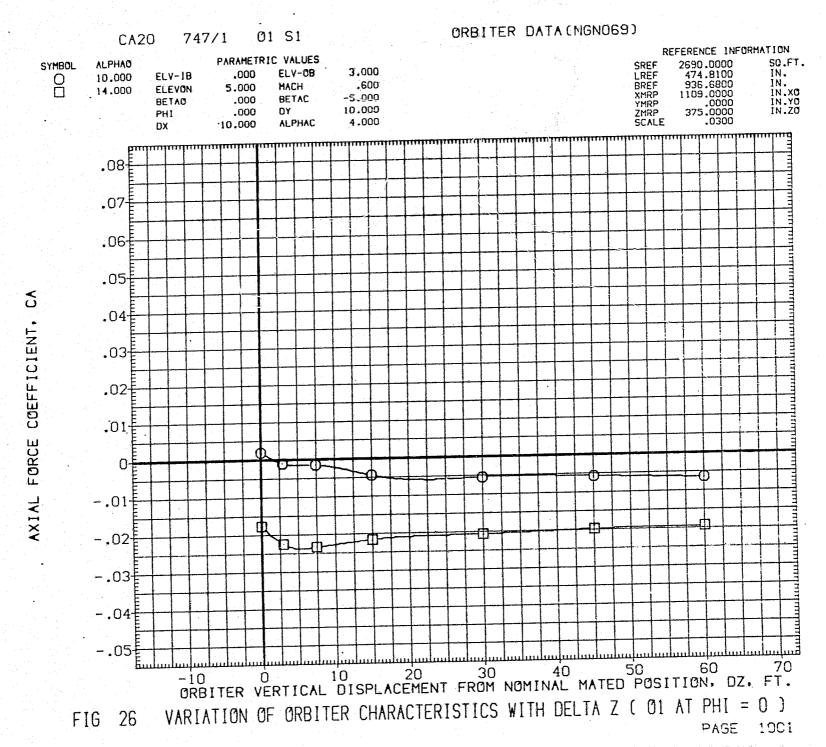
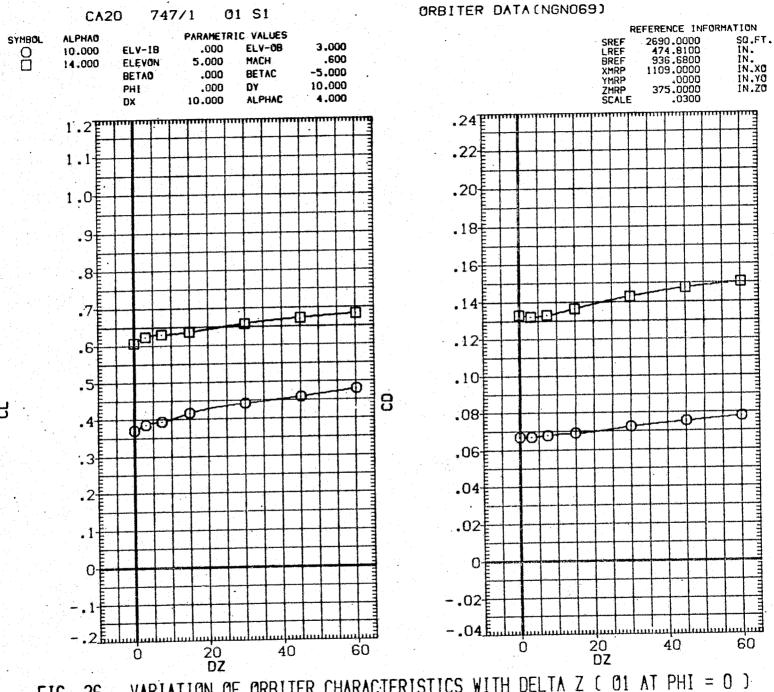
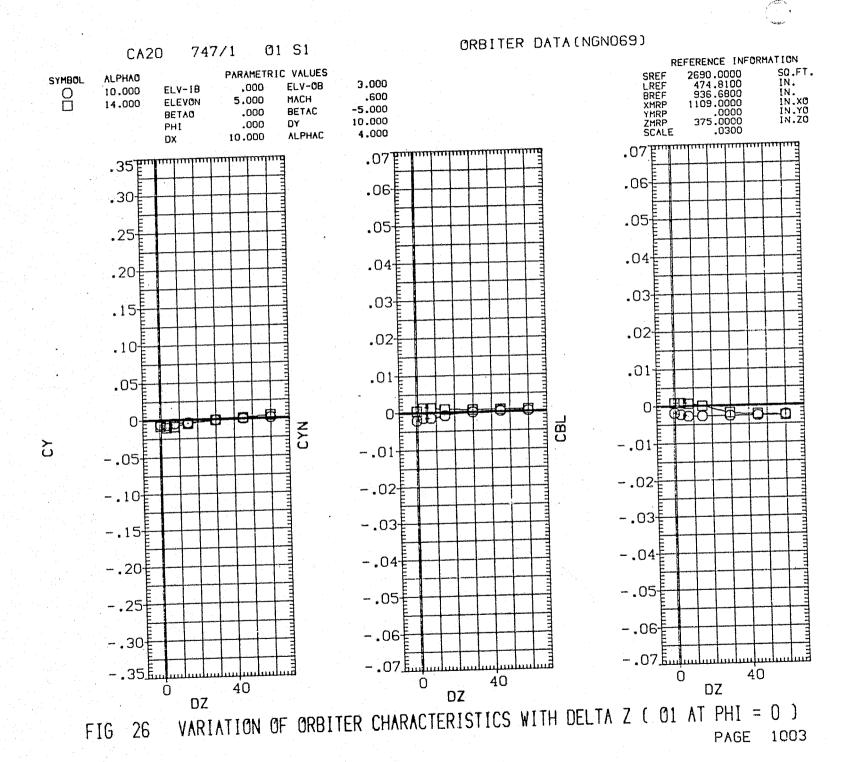


FIG 26 PAGE 1000





VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 1002 PAGE



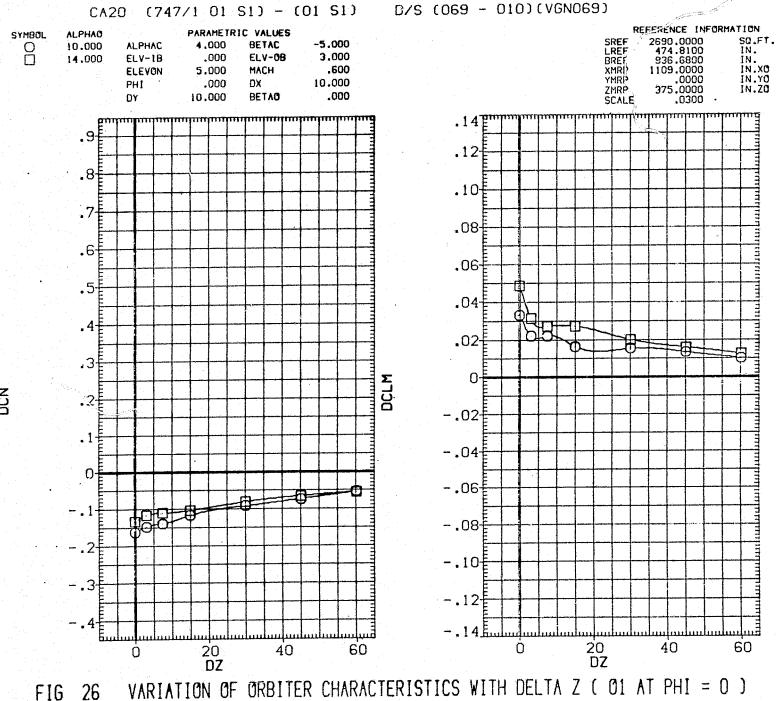
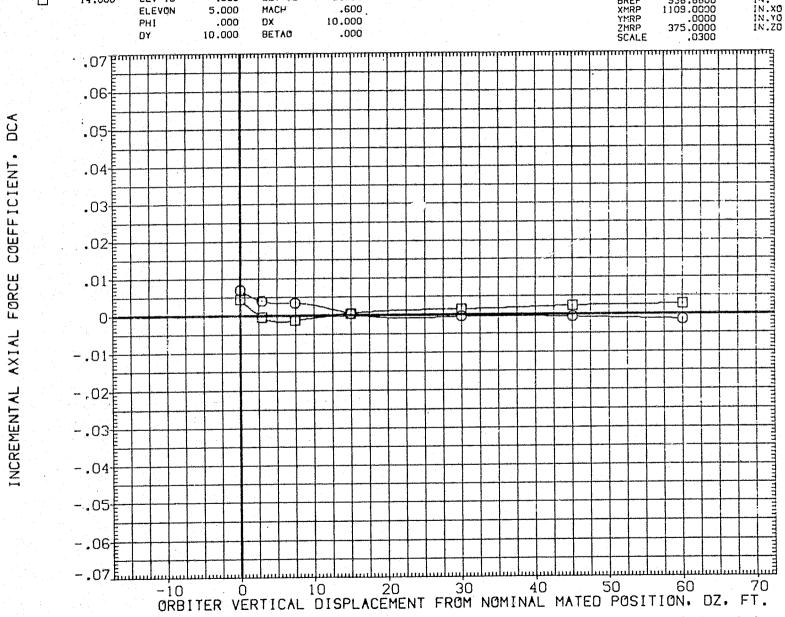


FIG PAGE 1004

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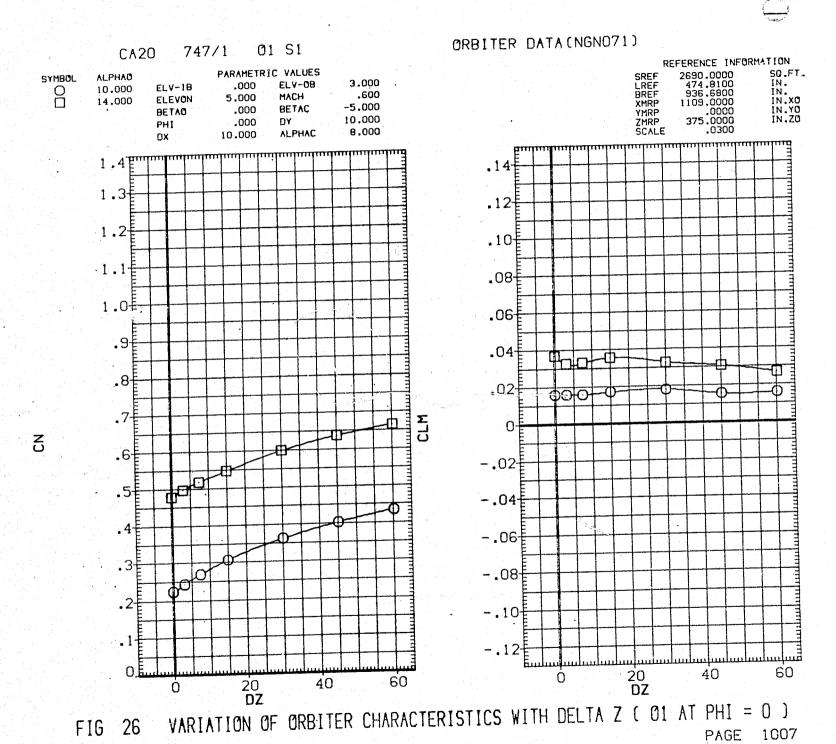
SYMBOL

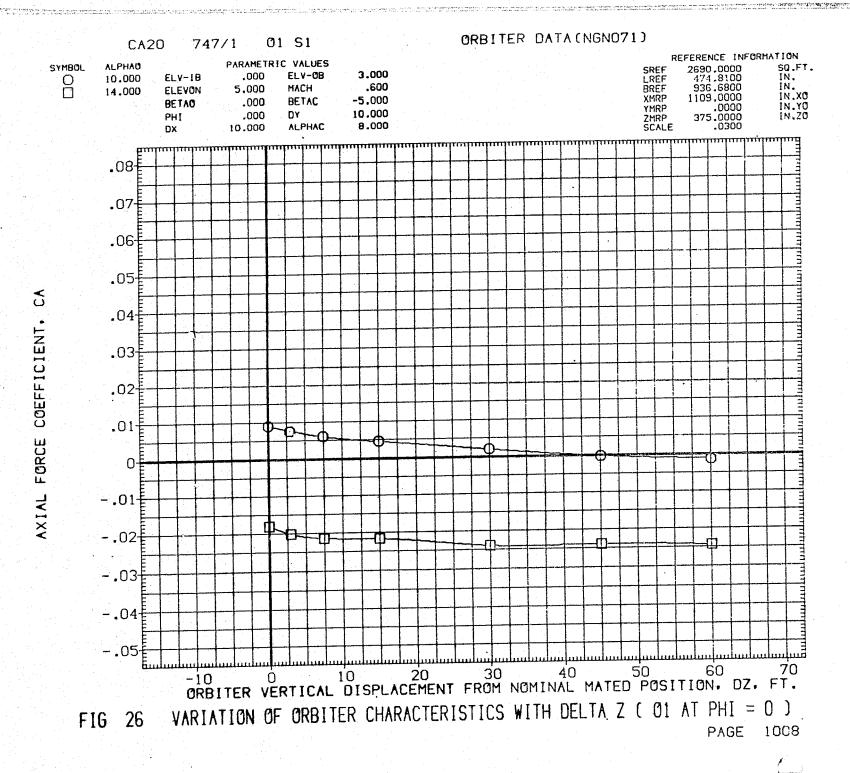


VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1005

1006 PAGE







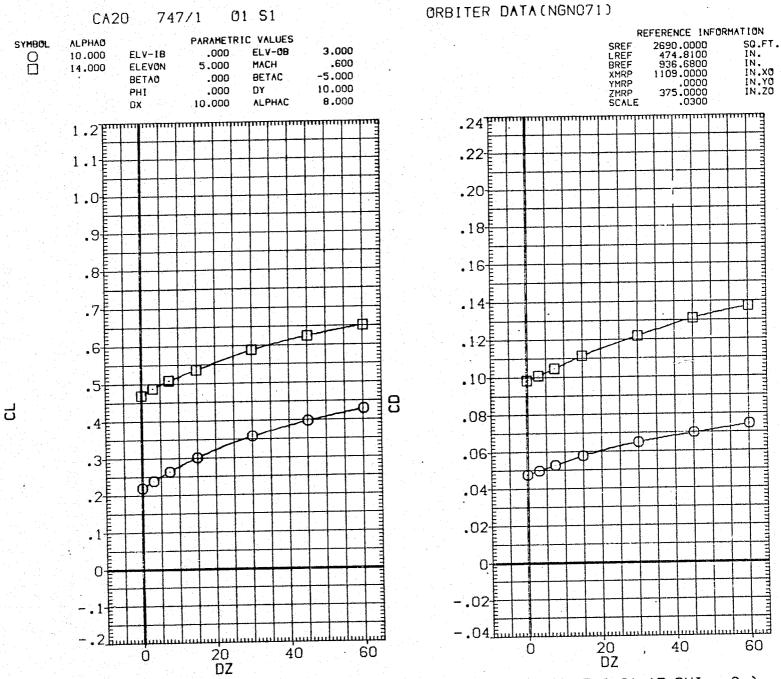
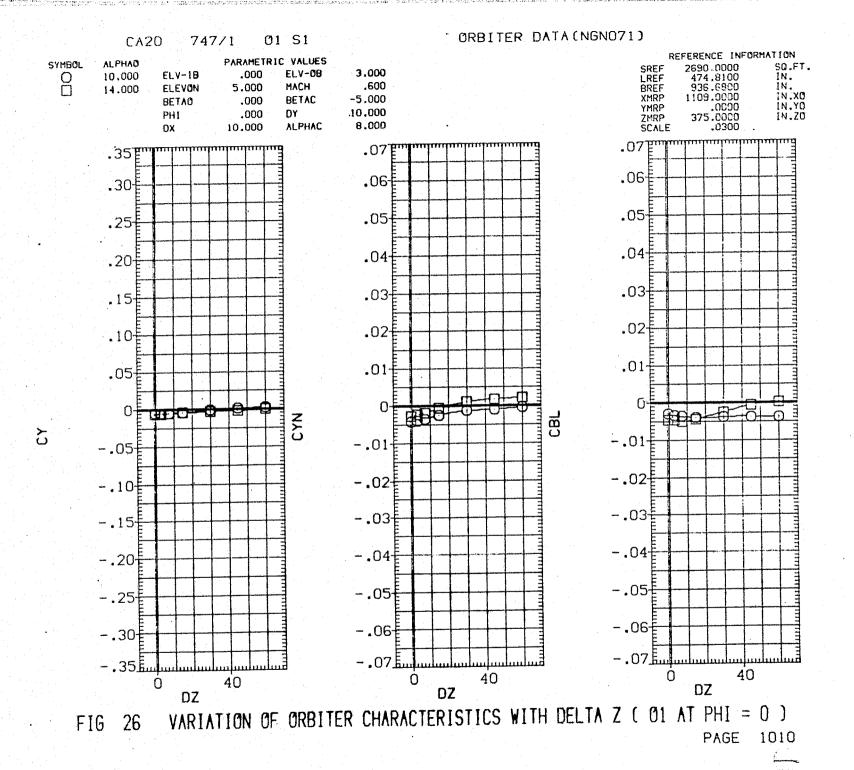
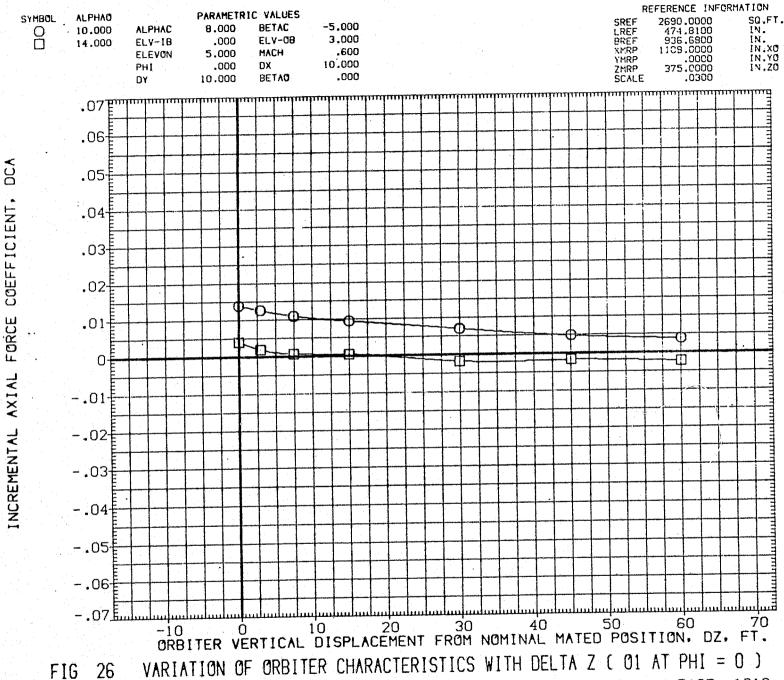


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )



D/S (071 - 010) (VGN071) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL **ALPHAO** 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 SREF LREF BREF XMRP YMRP SQ.FT. -5.000 8.000 BETAC 0 **ALPHAC** IN. 10.000 3.000 ELV-0B .000 IN. 14.000 ELV-IB IN.XO IN.YO IN.ZO .600 ELEVON 5.000 MACH 10.000 .000 DX PHI ZMRP SCALE .000 DY 10.000 BETAO .0300 .12-.8ŧ .10 .08 .6÷ .06 .04-.4± DCLM 0-CN -.02 -.04<del>+</del>  $-.06 \pm$ -.1E ~.08 -.10 -.12<del>[</del> -.14 Embadan 20 DZ 40 60 60 20 DZ 40 0

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1011



PAGE 1012

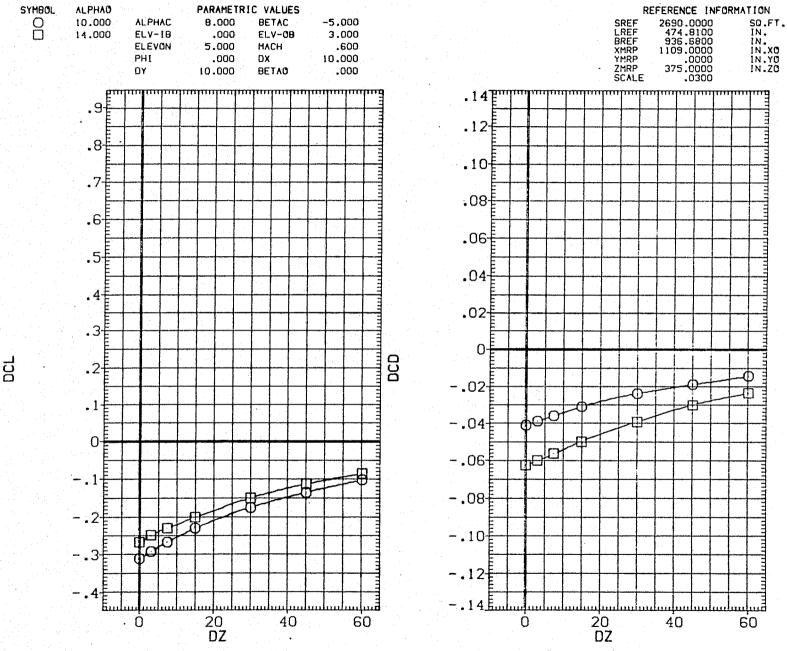


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1013

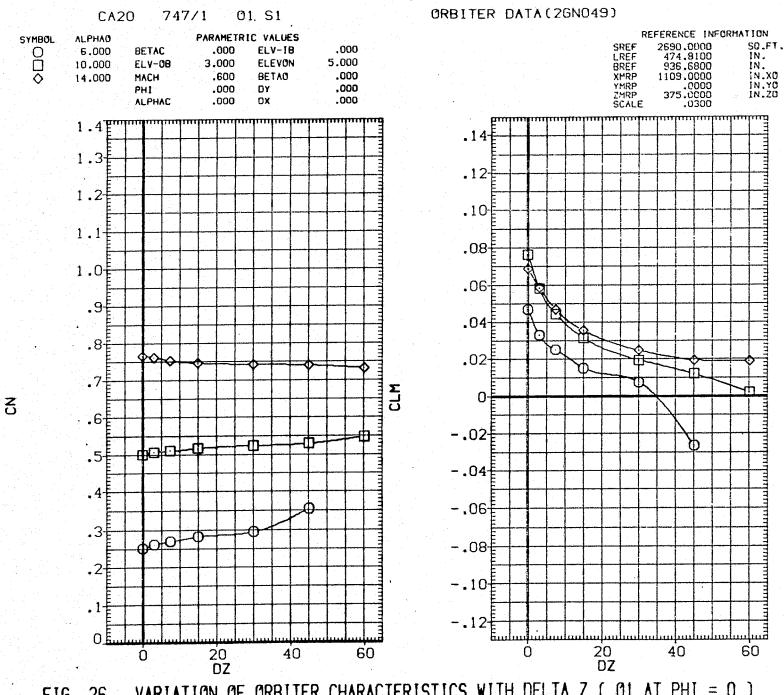


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1014

ORBITER DATA (2GNO49) 01 S1 747/1 CA20 REFERENCE INFORMATION **ALPHAO** PARAMETRIC VALUES SYMBOL SQ.FT. SREF 2690.0000 .000 BETAC .000 ELV-IB 000 6.000 IN. LREF 474.8100 ELEVON 5.000 3.000 936.6800 ELV-0B 10.000 BREF IN. IN.XO IN.YO IN.ZO .000 XMRP 14.000 MACH .600 .0000 YMRP DY .000 .000 PHI ZMRP SCALE .000 **ALPHAC** .000 DΧ .0300 .055Tm .050 .045 .040 **∀**∪ .035 COEFFICIENT. .030 .025 .020 Ф AXIAL FORCE .015 .010 .005-0--Ф -.005 -.010 سطِ 015. – -10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. 70 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26

PAGE

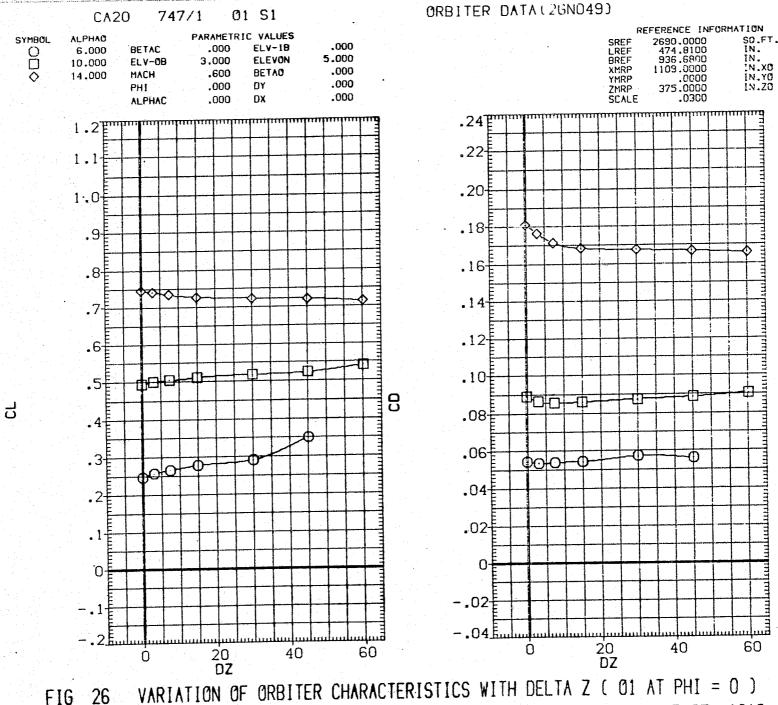


FIG PAGE 1016

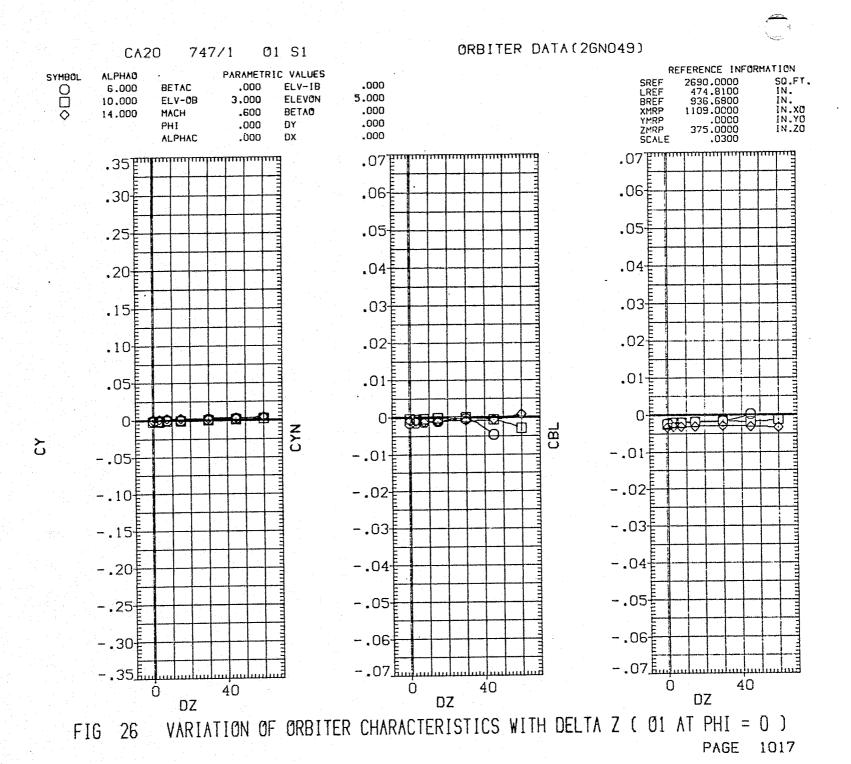


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1018

26

FIG

CA20 (747/1 01 S1) - (01 S1) D/S (049 - 010)(4GN049) REFERENCE INFORMATION PARAMETRIC VALUES **ALPHAO** SYMBOL SO.FT. 2630.0000 SREF BETAC .000 LREF BREF XMRP YMRP .000 474.8100 936.6800 1109.0000 .0000 375.0000 000 6.000 **ALPHAC** 3.000 IN. .000 ELV-0B 10.000 ELV-18 IN.XO IN.YO IN.ZO .600 5.000 MACH ELEVON 14.000 .000 PHI .000 DX ZMRP SCALE .000 .000 BETAC .0300 DY .07Fm .06 DCA .05 COEFFICIENT, .04 .03 .02 FORCE .014 0-AXIAL  $-.01 \pm$ INCREMENTAL -.02 -.03 -.04 -.05 -.06+ - .07 <u>L</u>... -10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. 70 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

PAGE

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1020

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20 DZ

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20 DZ 40

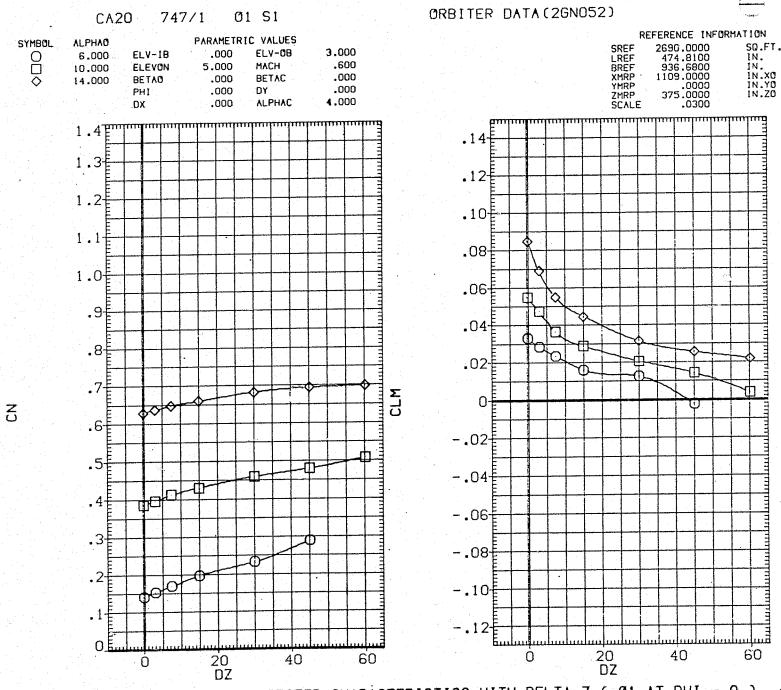


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1021

PAGE 1022

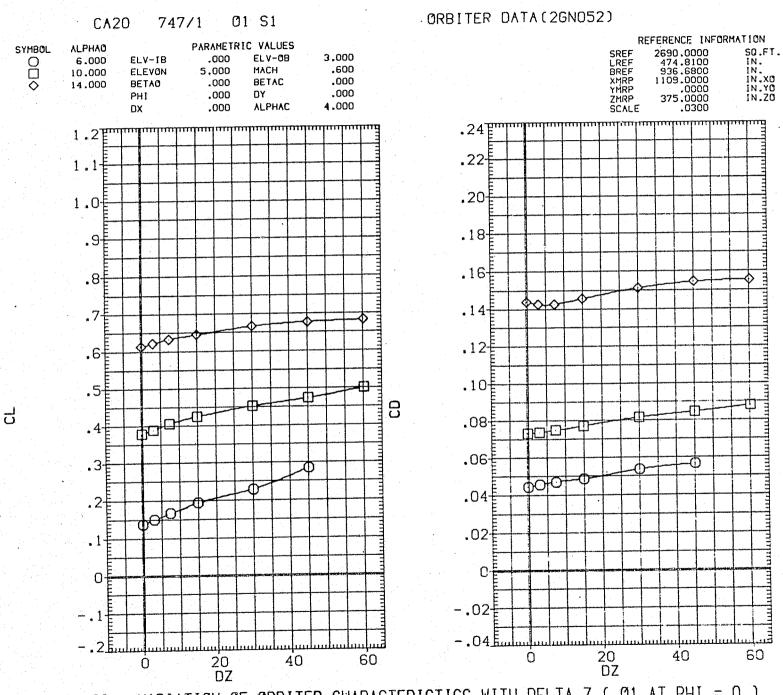
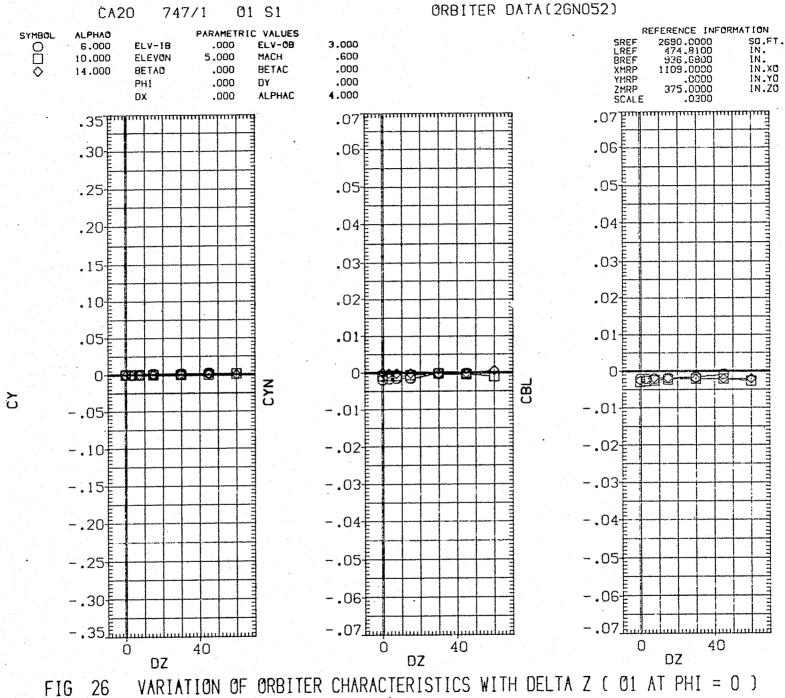
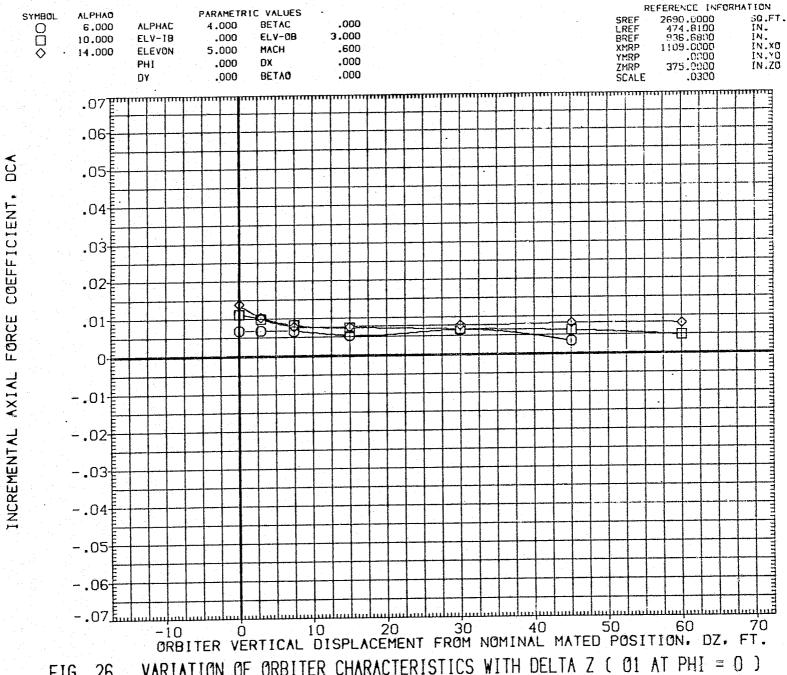


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1023



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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1025



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 1026 PAGE



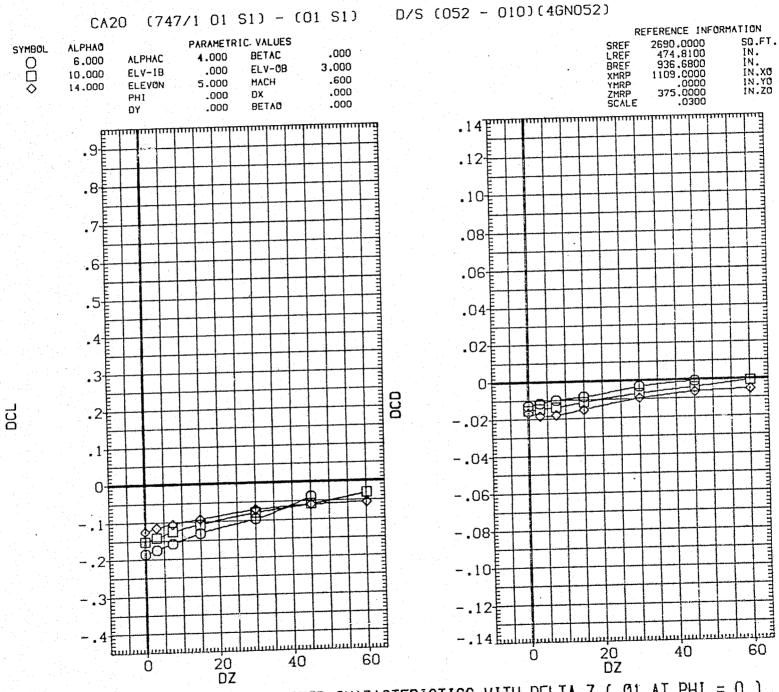


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1027

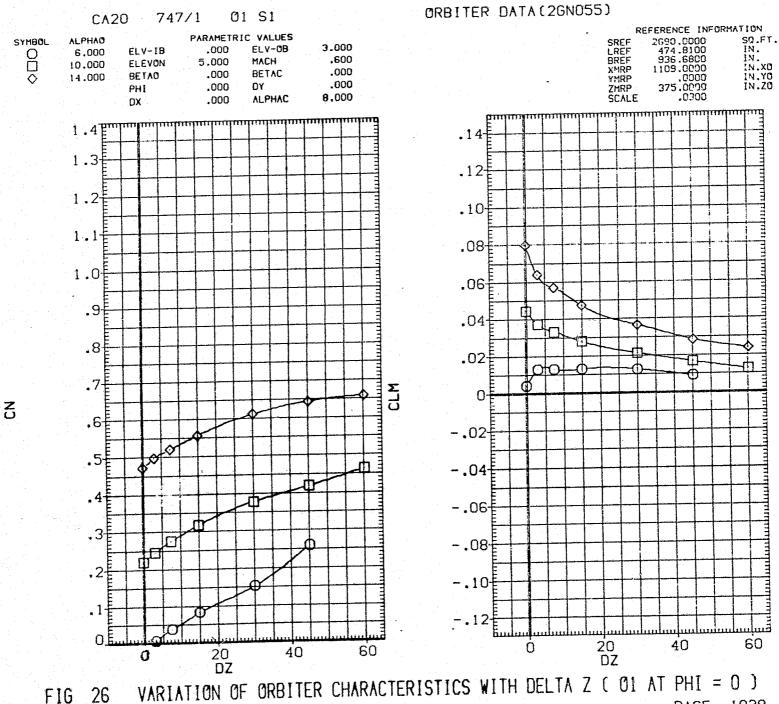


FIG 26 PAGE 1028

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

PAGE

FIG

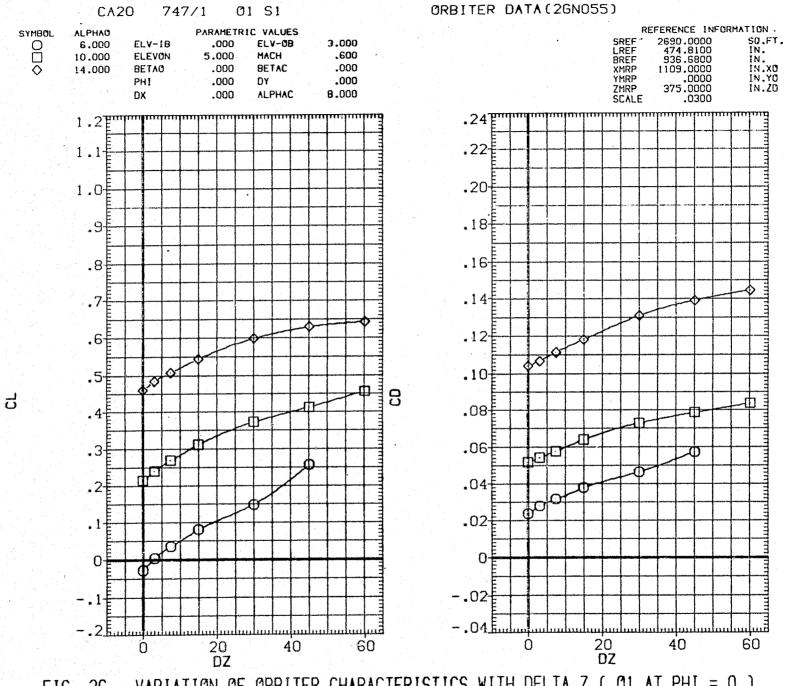
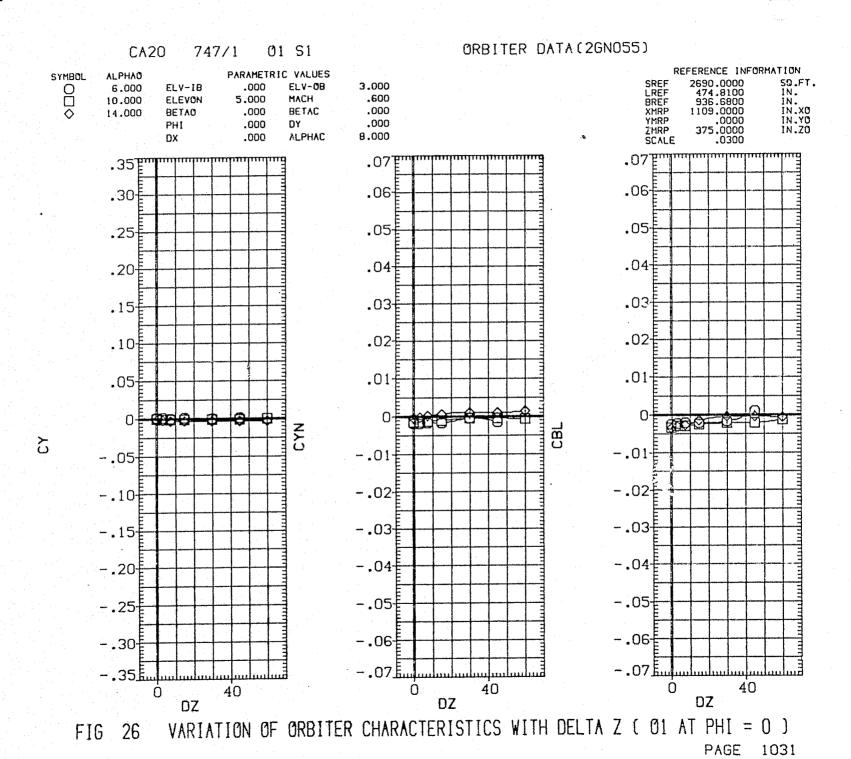
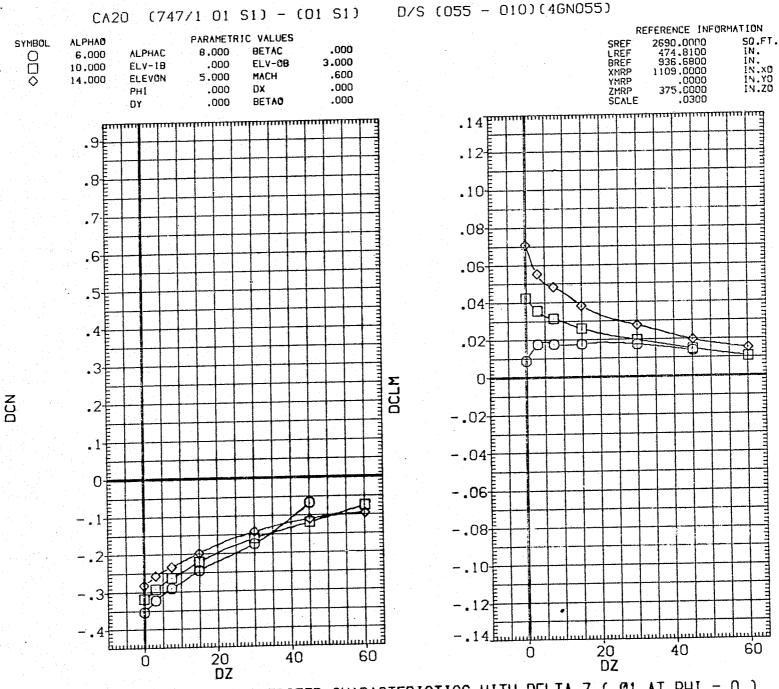


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1030



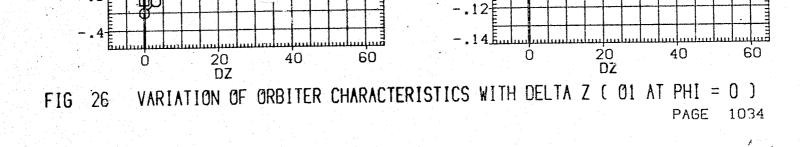


VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 1032 PAGE

PAGE

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O.F

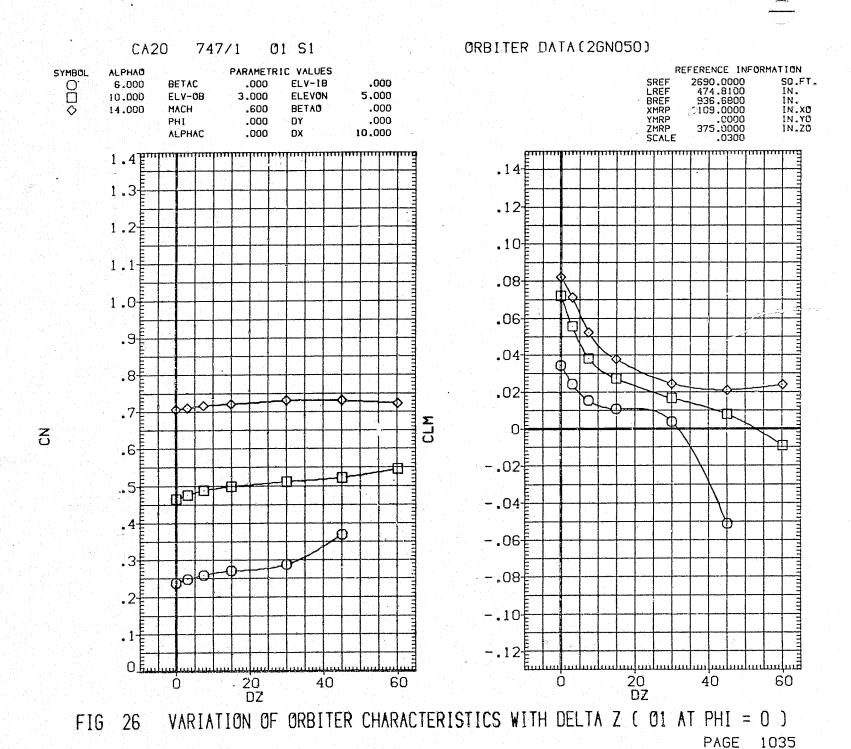


-.04

-.06-

-.08

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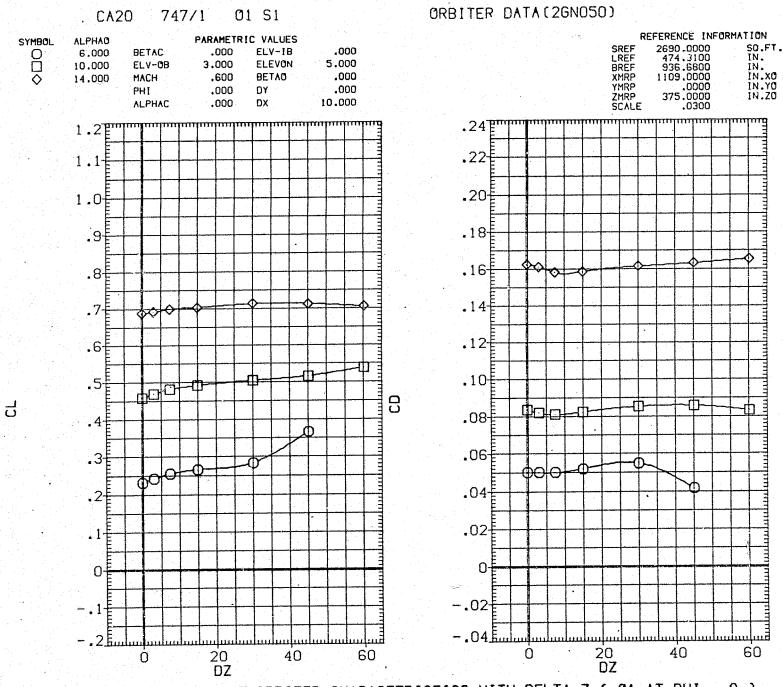
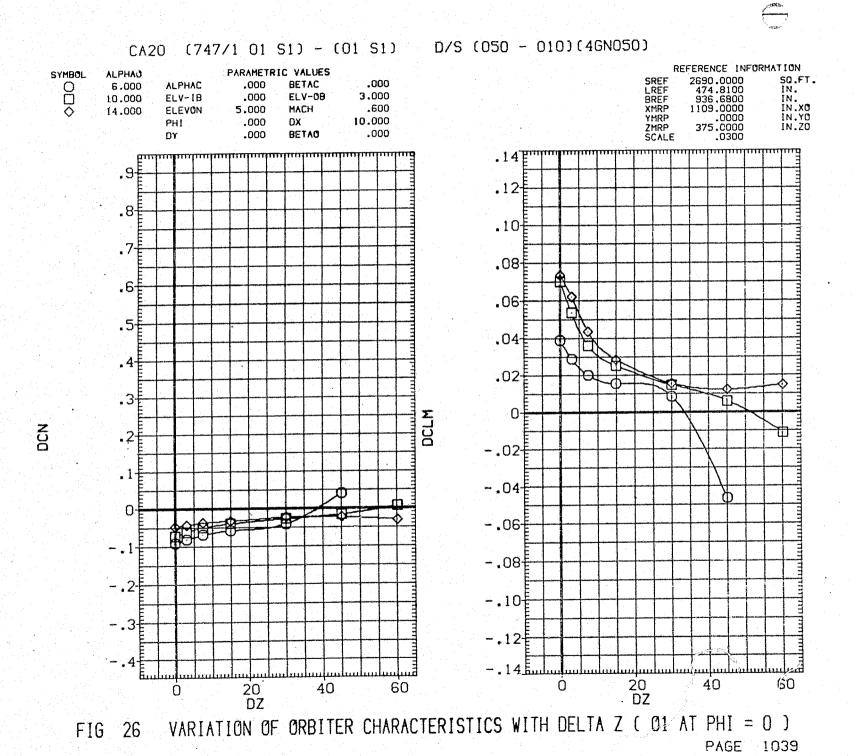
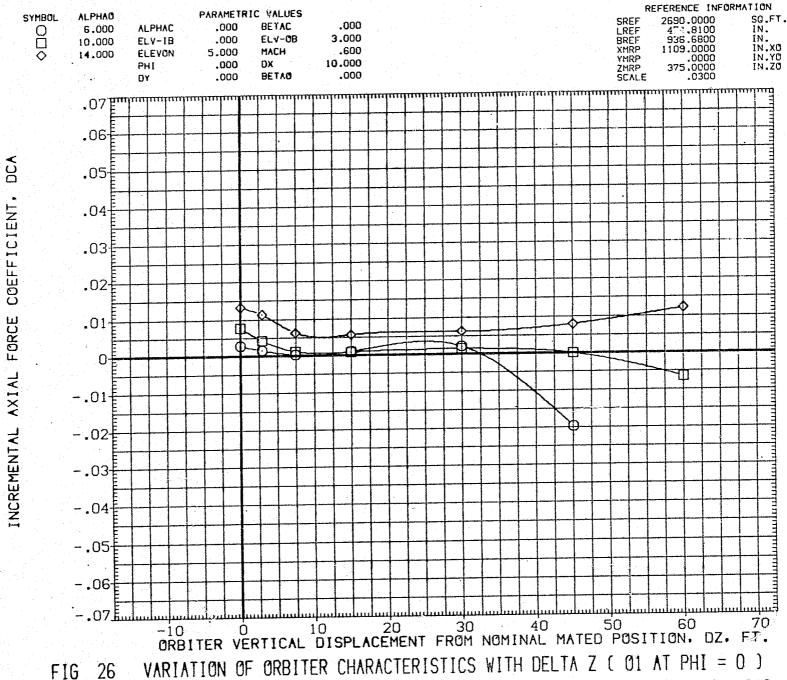


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1037

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1038





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FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1041

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1042

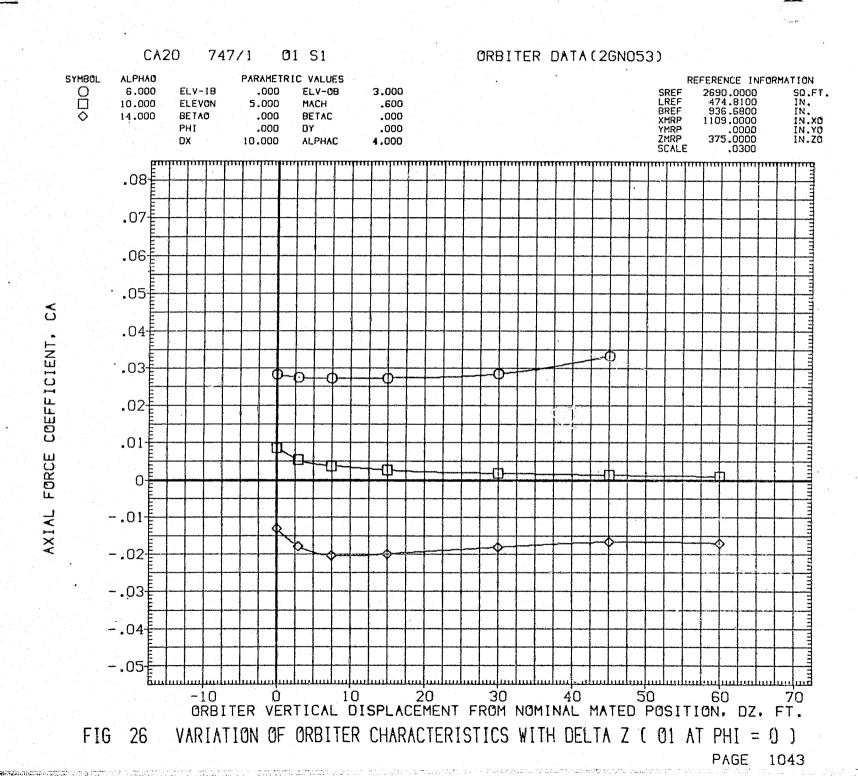
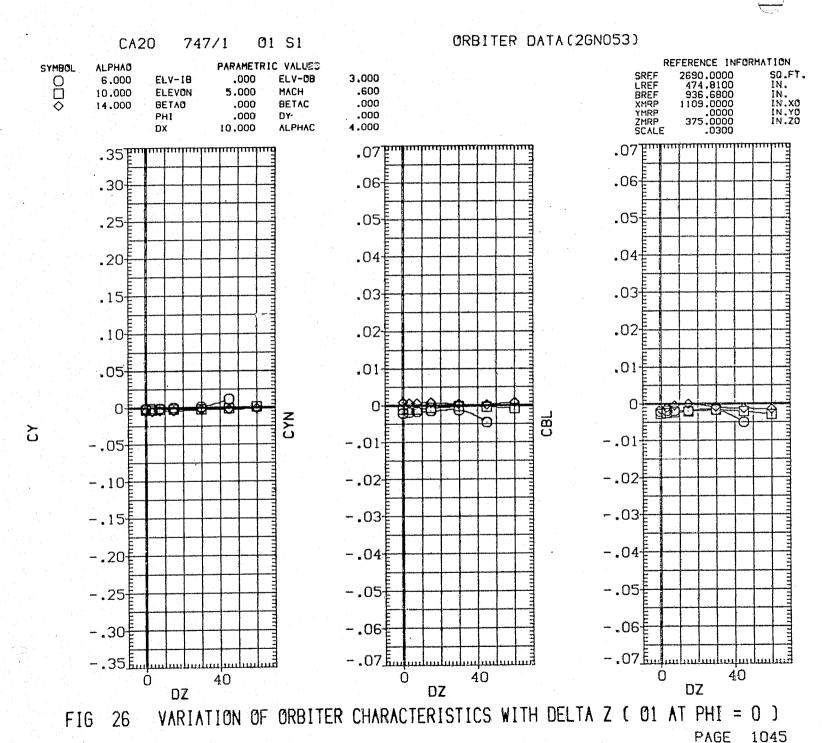


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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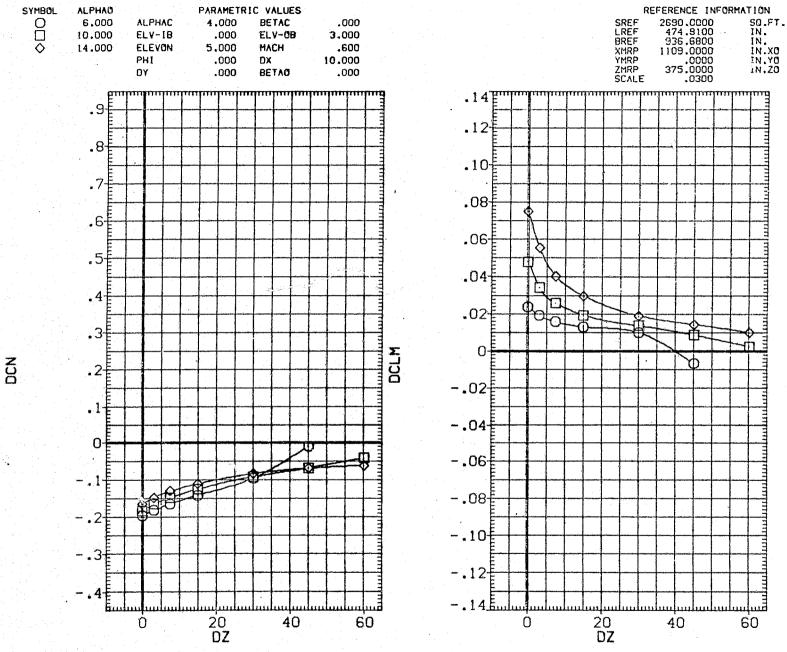


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1046



053 - 010)(4GN053)  $(747/1 \ 01 \ S1) - (01 \ S1)$  $\mathbf{D}\lambda$ CA20 REFERENCE INFORMATION PARAMETRIC VALUES ALPHAO SYMBOL 2690.0000 474.8100 SREF LREF BREF SO.FT. .000 BETAC ALPHAC 4.000 6.000 IN. 936.6800 ELV-IB .000 ELV-0B 3.000 IN. 10.000 IN.XO IN.YO IN.ZO .600 XMRP MACH 5.000 14.000 ELEVON YMRP ZMRP 375.0000 10.000 .000 DX .000 BETAO DY .000 SCALE .06-DCA .05 COEFFICIENT .04 .03<del>E</del> .02<del>-</del> FORCE .01 -.01 INCREMENTAL -.02<del>-</del> -.03 -.04 -.05 -.06 - .07 <u>E</u>...l... -10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1047

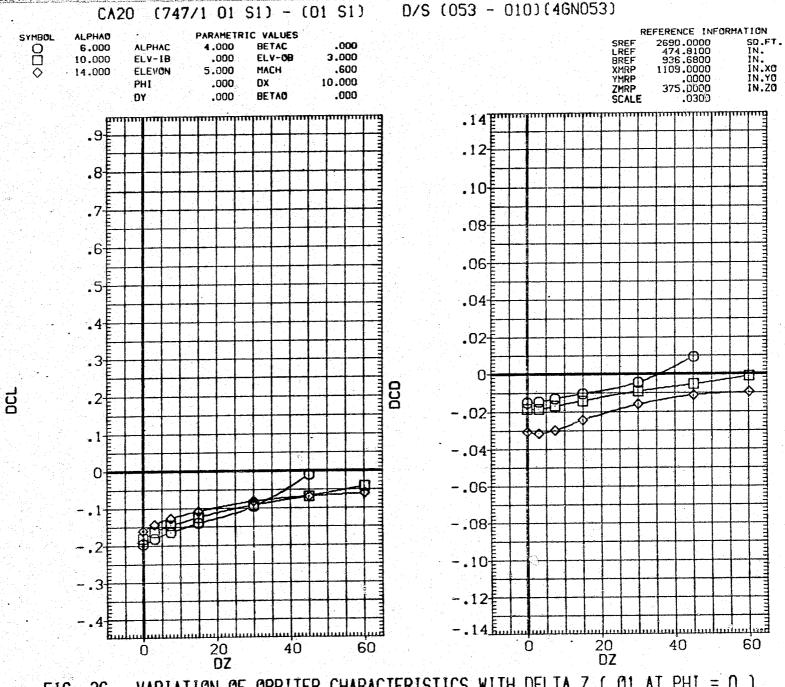


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1048

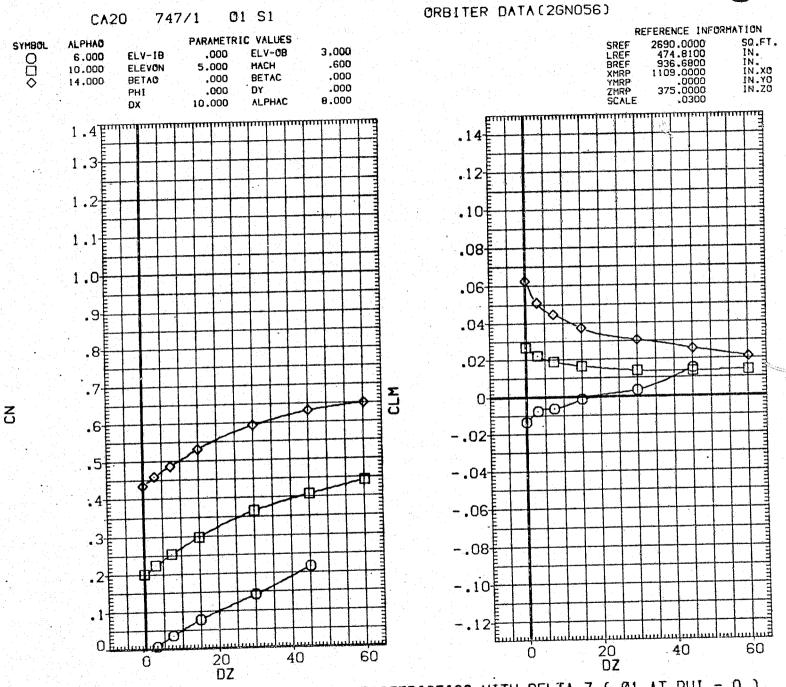


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( .01 AT PHI = 0 )
PAGE 1049

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1051

TO DESCRIPTION OF THE PROPERTY CONTROL OF THE PROPERTY OF THE

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1052

CA20 (747/1 01 S1) - (01 S1) . D/S (056 - 010)(4GN056) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL ALPHAO SQ.FT. SREF 2690.0000 **ALPHAC** 8.000 BETAC .000 000 6.000 LREF 474.8100 3.000 IN. IN.XO IN.YO IN.ZO .000 ELV-08 BREF XMRP YMRP ELV-18 10.000 936,6800 1109.0000 .0000 375.0000 .600 ELEVON 5.000 MACH 14.000 10.000 .000 DX PHI ZMRP SCALE .000 .000 BETAO DY .0300 .14 Em .9 .12 .8E .10 .08 .6+ .06 .04 .02 DCLM OO -.02 -.04 -.06<del>[</del> -.08<del>[</del> -.10<del>+</del> -.12<del>[</del> - . 14 起 25 DZ 40 60 20 DZ 60 40 0

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1053

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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) 26 PAGE 1055

DZ

FIG

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1056

CA20 747/1 01 S1

## ORBITER DATA (2GN051)

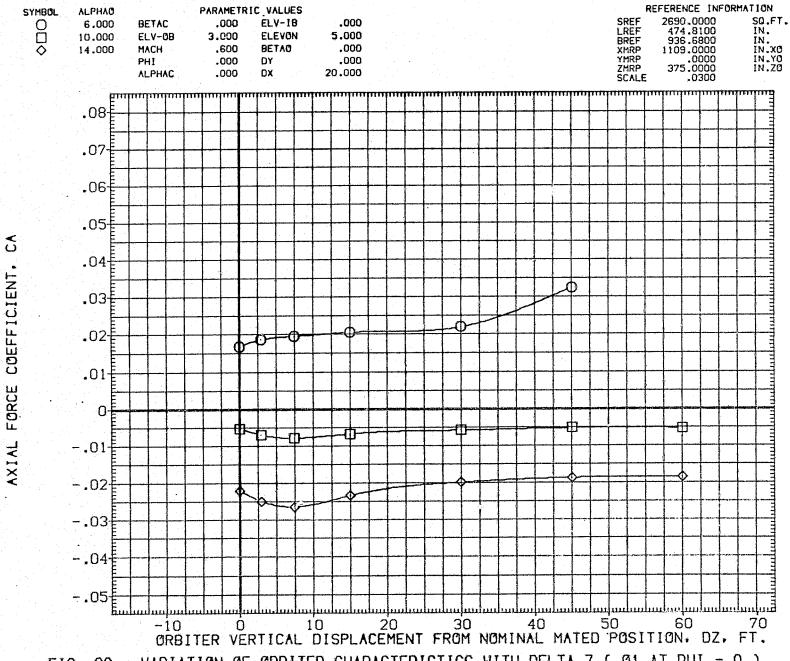
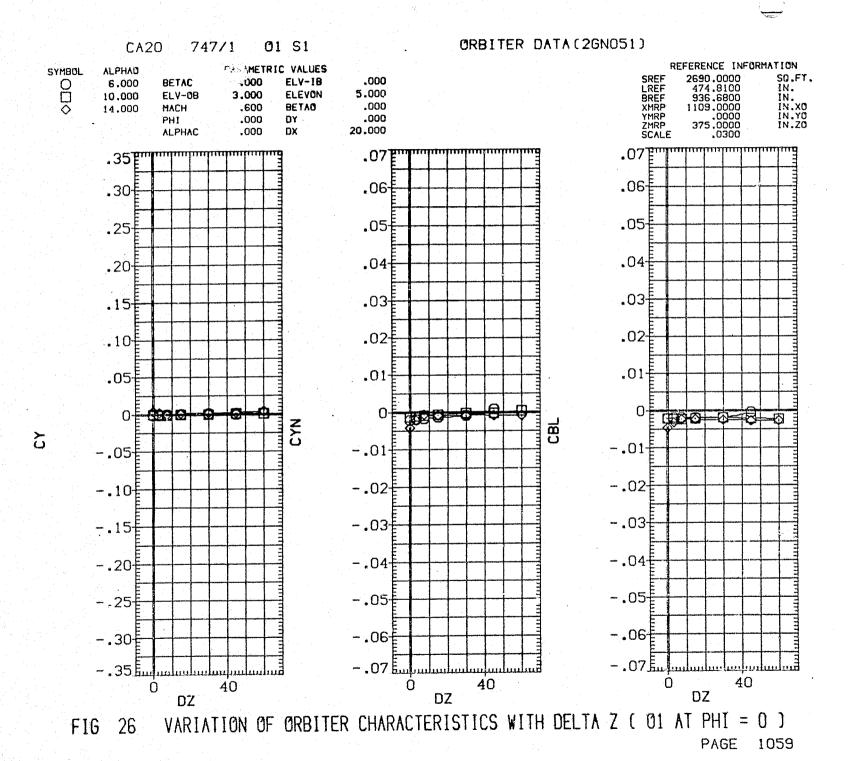


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1057

FIG PAGE 1058

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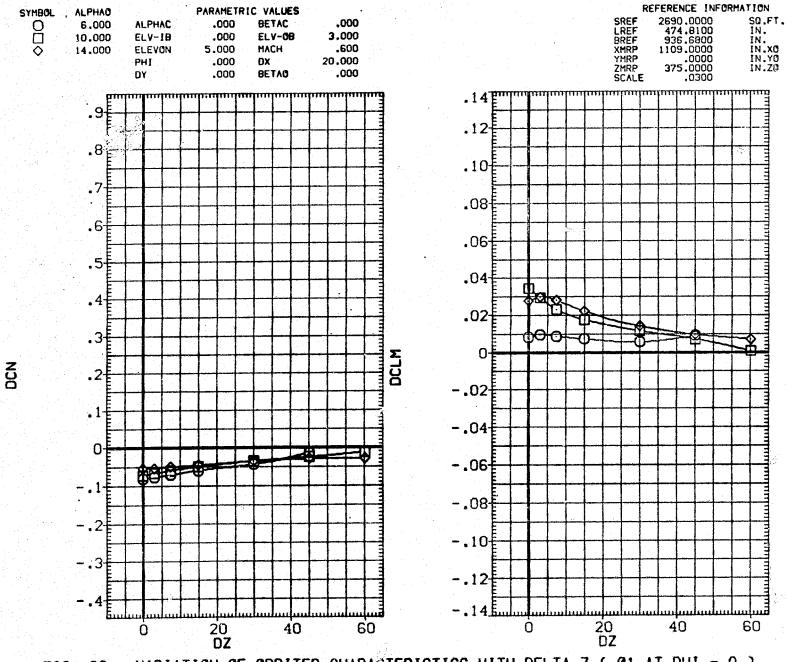
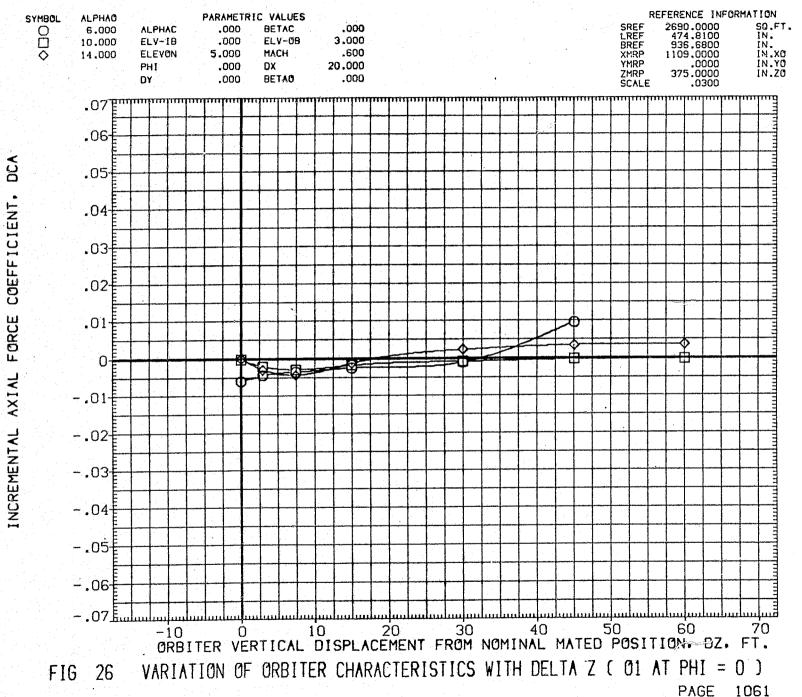


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1060



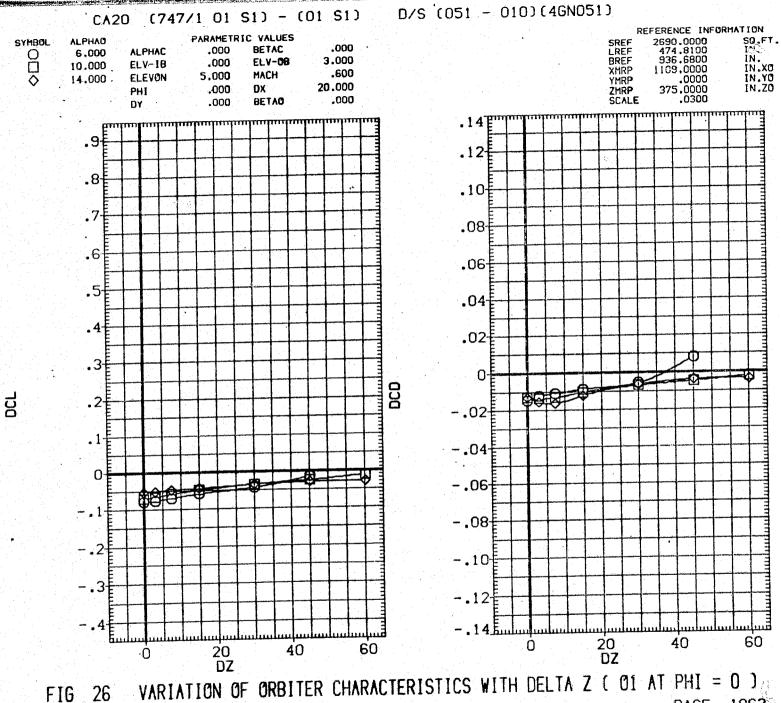
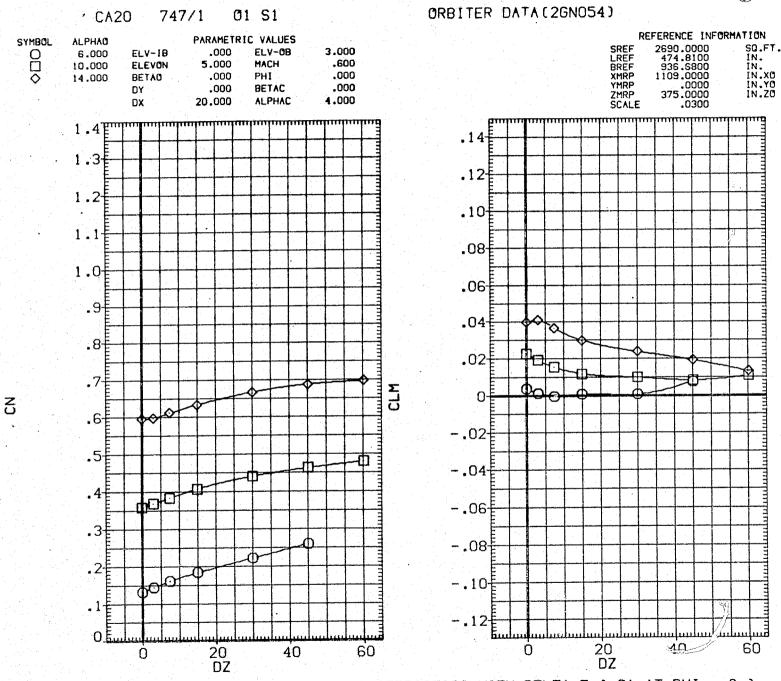
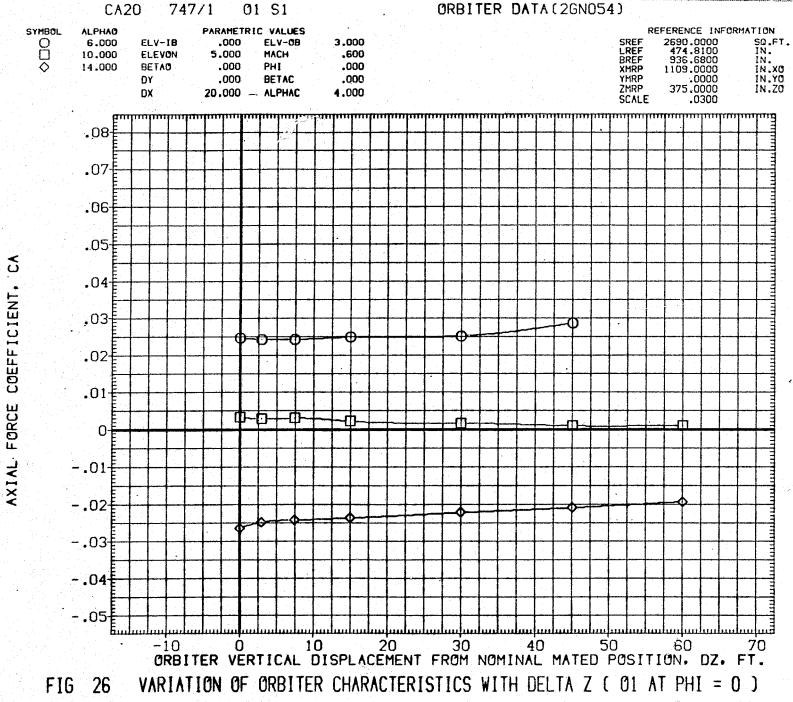


FIG 26 1062 PAGE

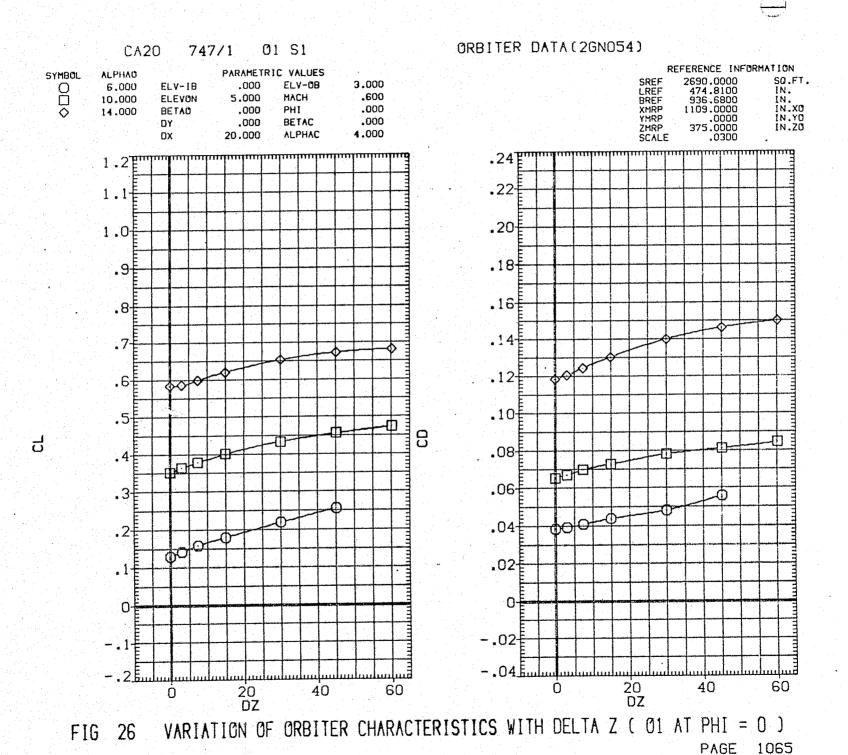


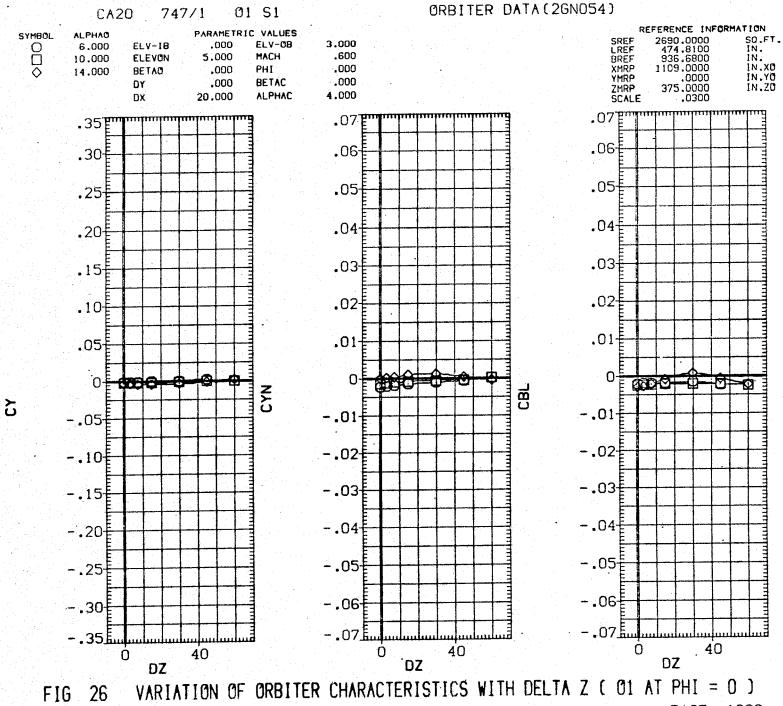
26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
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FIG



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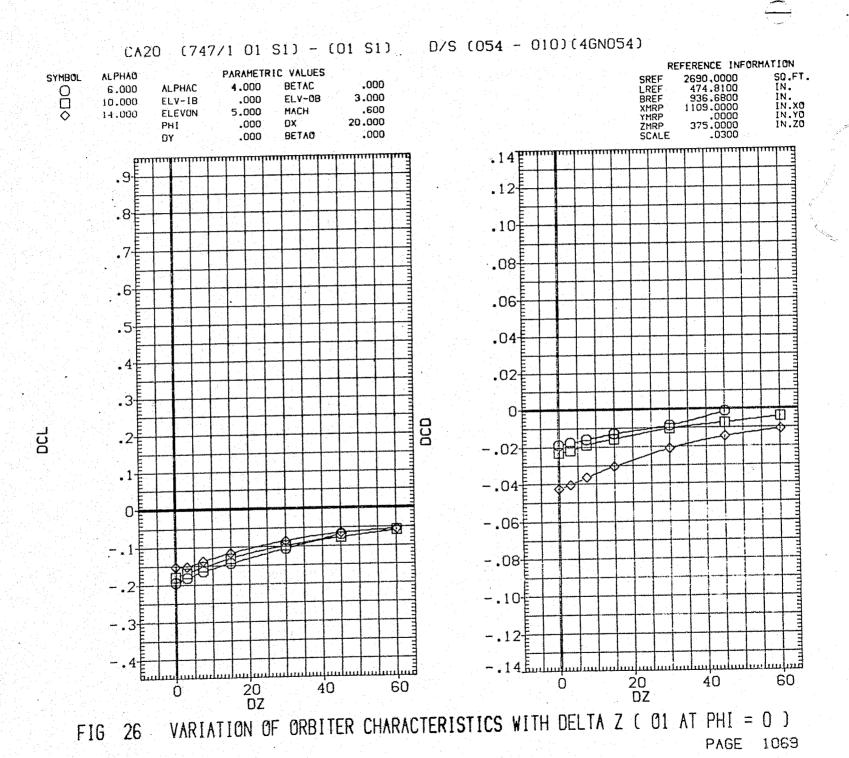


PAGE 1066

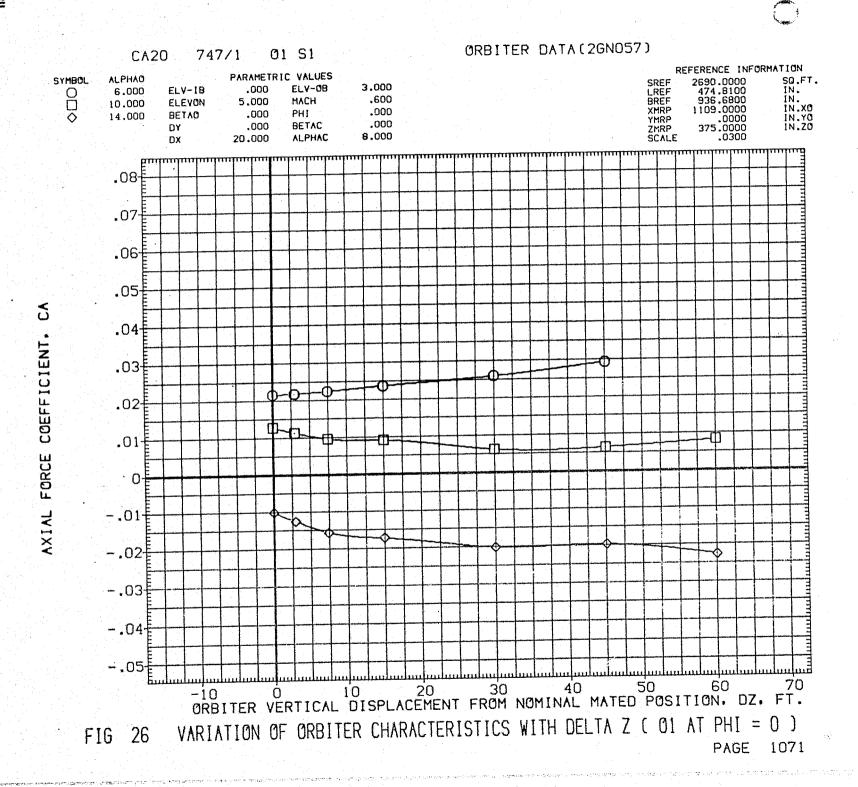
D/S (054 - 010) (4GN054)  $(747/1 \ 01 \ S1) - (01 \ S1)$ REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL **ALPHAO** 2690.0000 474.8100 936.6800 SO.FT. SREF 4.000 BETAC .000 6.000 **ALPHAC** LREF BREF XMRP YMRP ZMRP 0 IN. 3.000 ELV-OB ELV-1B .000 IN. 10.000 IN.XO IN.YO IN.ZO 1109.0000 .0000 375.0000 .600 ELEVON 5.000 MACH 14.000 20,000 .000 DX PHI .000 BETAO .000 SCALE DY .14 Emmunum .9 .12 .8 .10 .08-.6<del>-</del> .06 .04 .02 DCLM 0 -.02 -.04 0 = -.06 -.08 -.10 -.12<del>F</del> - . i 4 hala 20 DZ 40 60 20 DZ 0 40 60 0

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1067

FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1068



26 FIG PAGE 1070



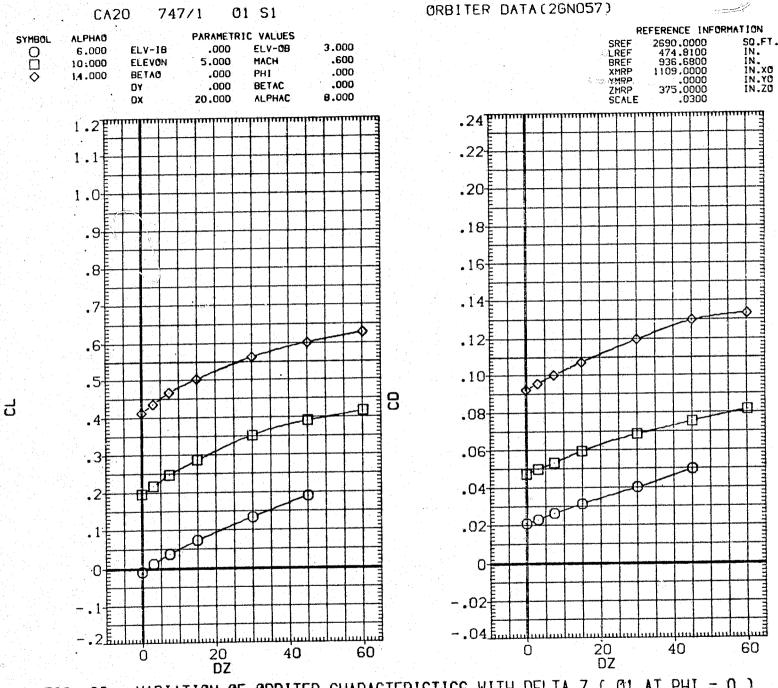


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1072

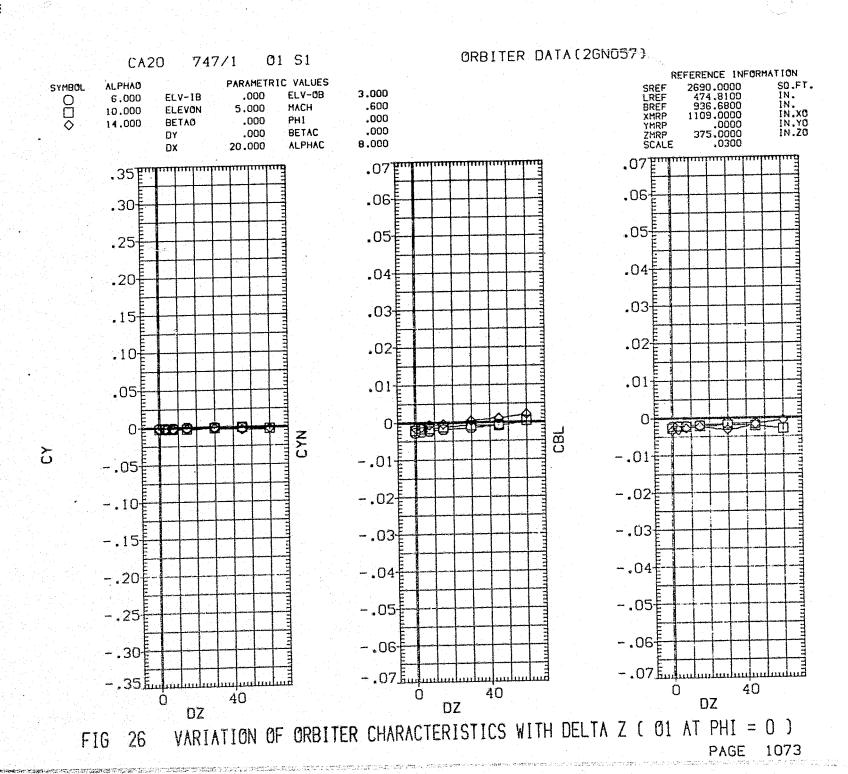
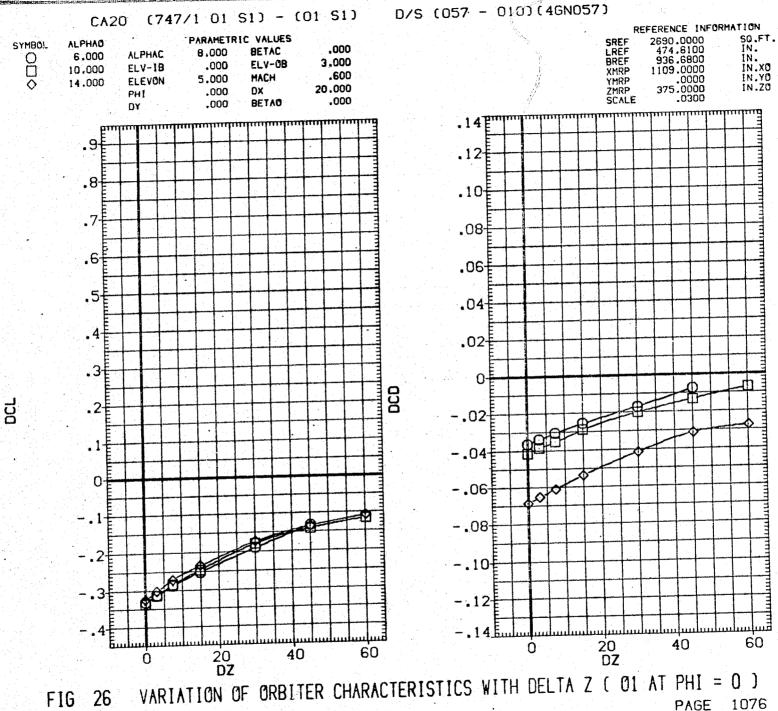


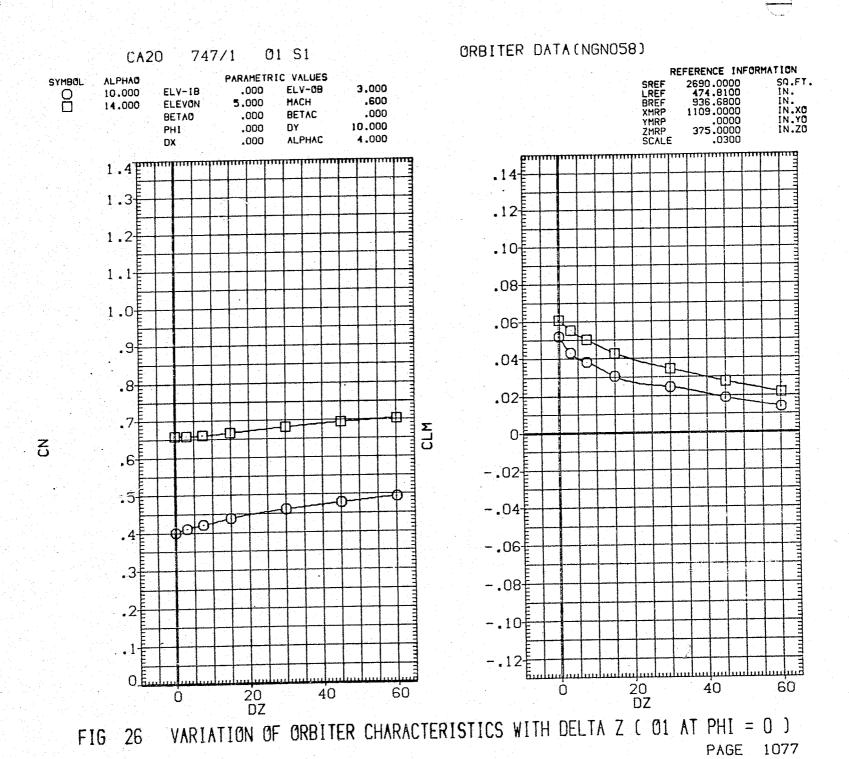
FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1074

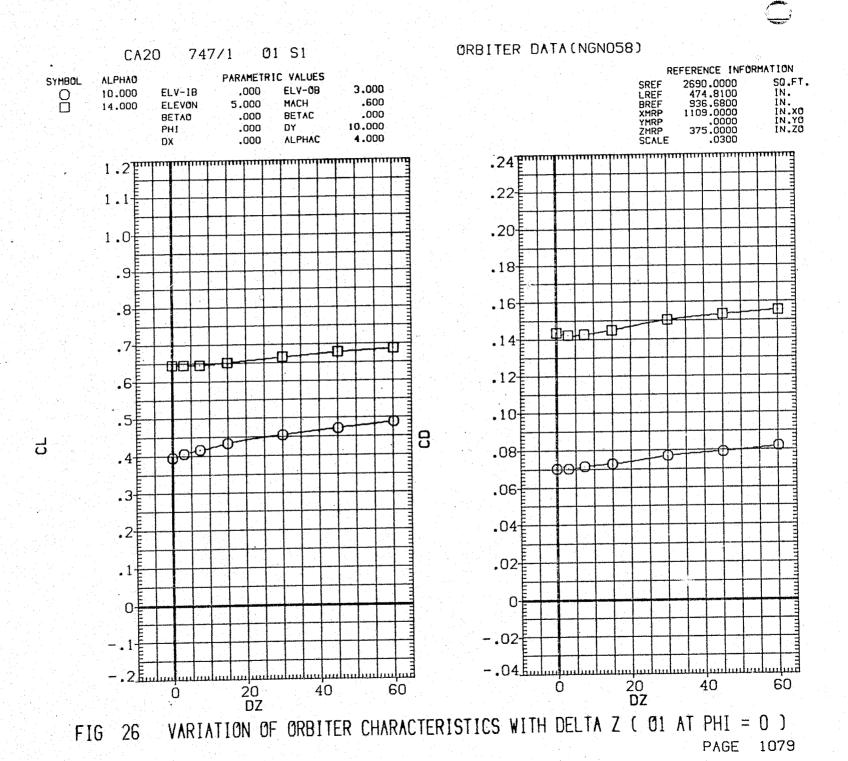
D/S (057 - 010)(4GN057) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES **ALPHAO** SYMBOL 2690.0000 474.8100 936.6800 SO.FT. 8.000 BETAC .000 6.000 ALPHAC IN. IN. IN.XO IN.YO IN.ZO 3.000 ELV-0B .000 10.000 ELV-IB XMRP YMRP ZMRP SCALE 1109.0000 .0000 375.0000 .600 5.000 14.000 20.000 .000 DX. PHI .000 BETAO .000 .0300 .07E''' .06 .05<del>[</del> COEFFICIENT.  $.04\frac{1}{2}$ .03<del>[</del> FORCE .01 AXIAL -.01<del></del> INCREMENTAL -.02<del>[</del> -.03<del>[</del> -.04<del>[</del> -.05 -.06--10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. FIG 26

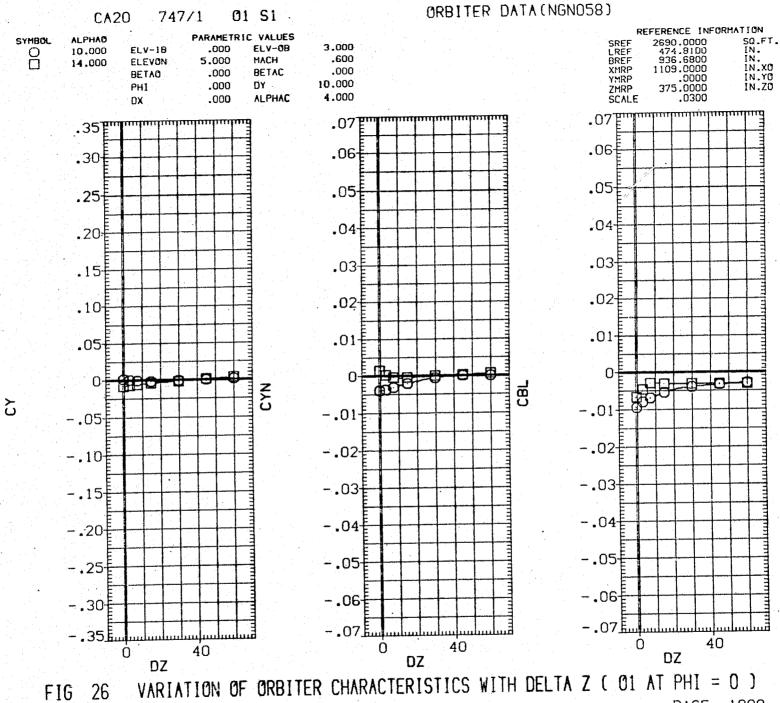
VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) PAGE 1075



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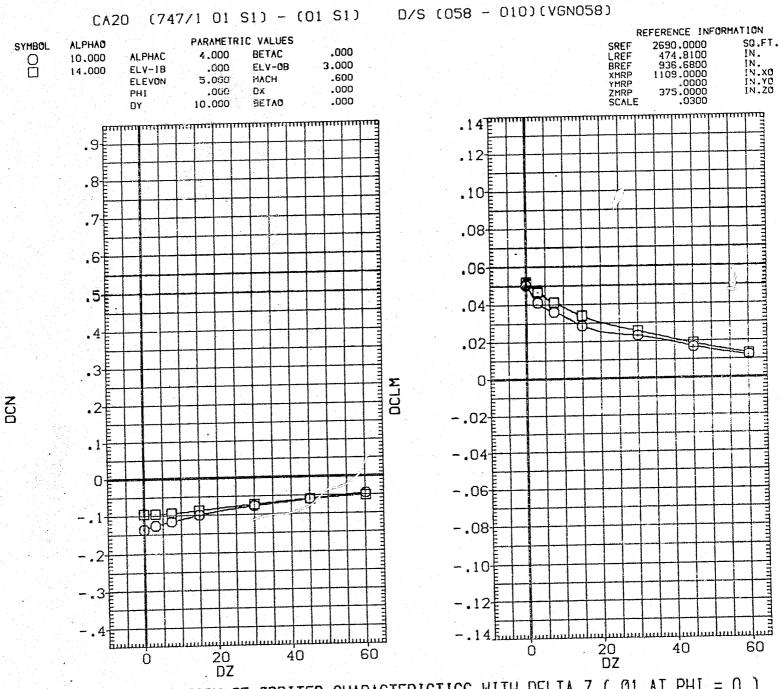


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

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D/S (058 - 010)(VGN058) (747/1 01 S1) - (01 S1) CA20 REFERENCE INFORMATION PARAMETRIC VALUES SO.FT. IN. IN. **ALPHAO** SYMBOL 2690.0000 474.8100 BETAC .000 4.000 0 **ALPHAC** 10.000 936.6800 1109.0000 .0000 375.0000 .0300 3.000 ELV-0B .000 14.000 ELV-IB IN.XO IN.YO IN.ZO .600 5.000 ELEVON YMRP .000 .000 PHI ZMRP .000 BETAO 10.000 .14 Emilia .12 .8<del>[</del> .10 .08 .6<del>-</del> .06 .04 .02 .3<del>[</del> 0-000 DCL -.02 -.04 0--.06 -.08 -.10 -.12€ 20 DZ 40 60 60 40 20 DZ 0 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )

PAGE

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FIG

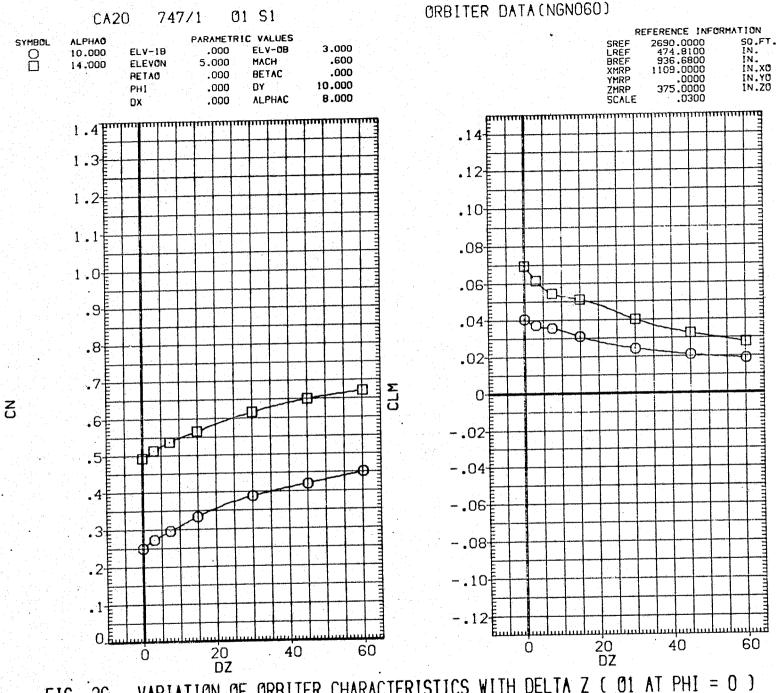
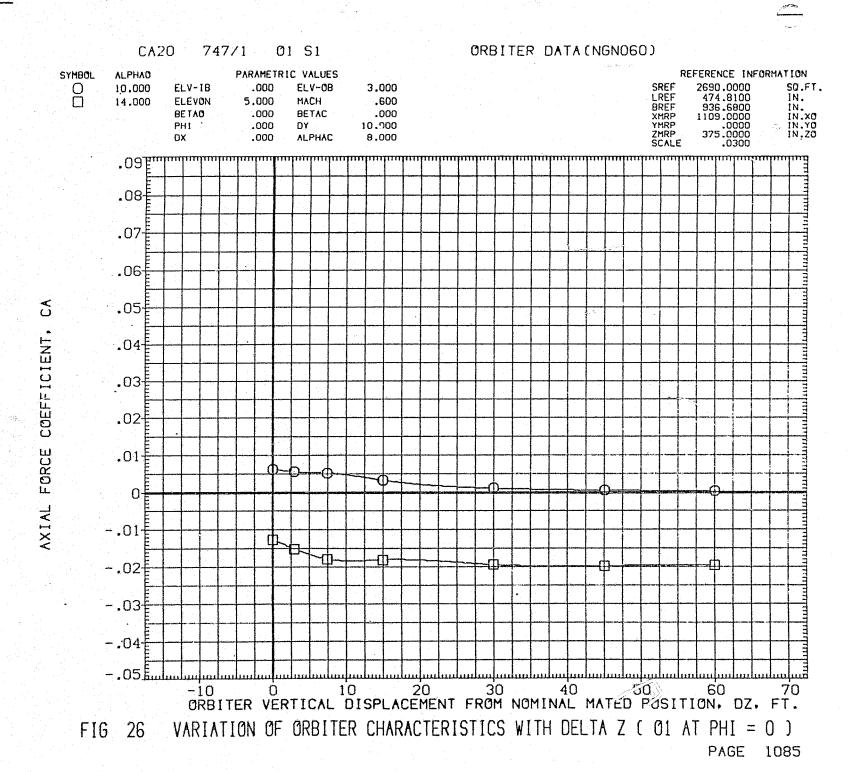
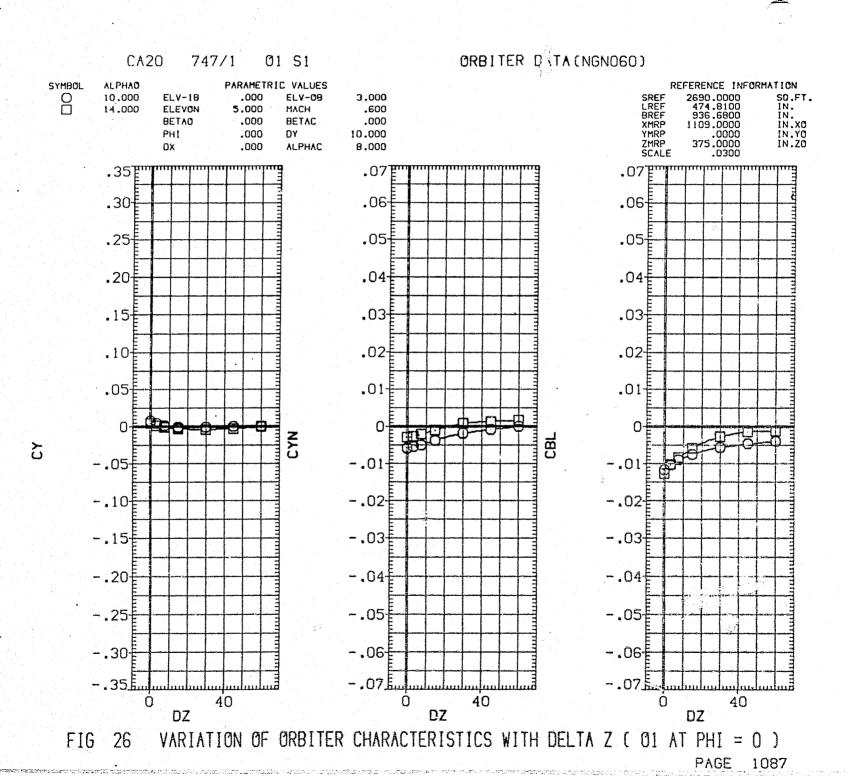


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1084



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) 26 FIG PAGE 1086



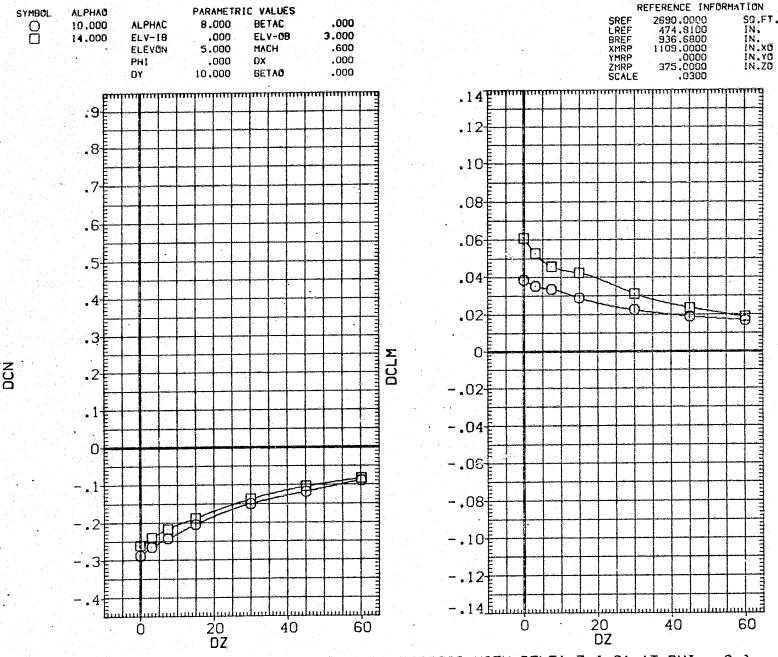


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1088

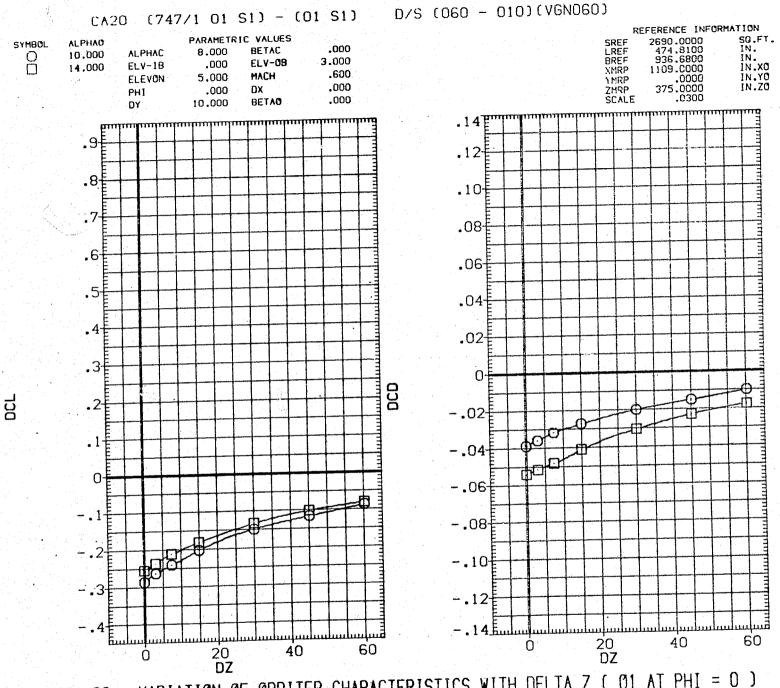


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1090

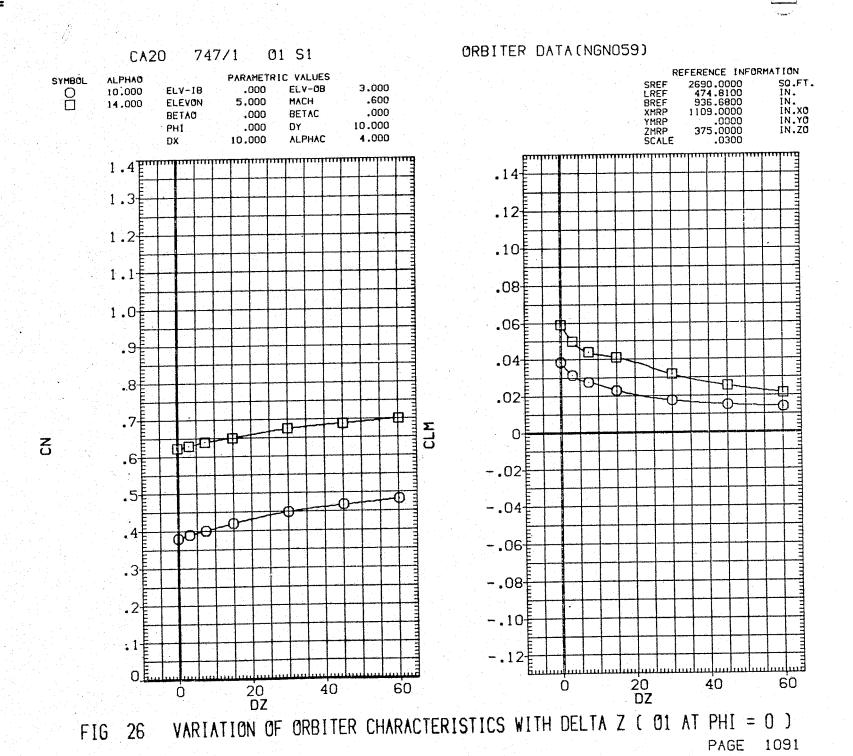
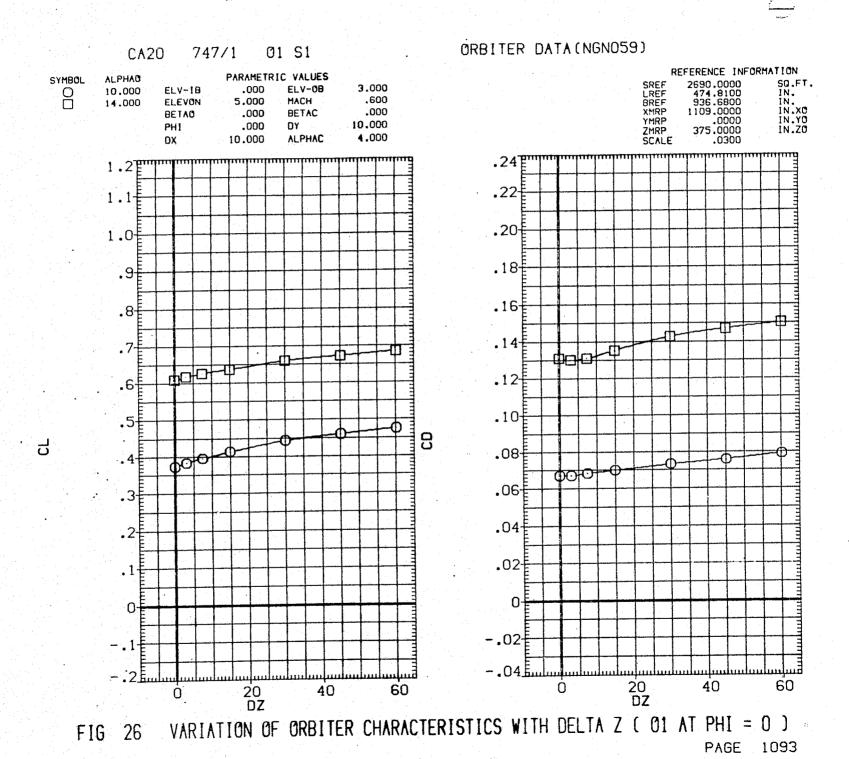
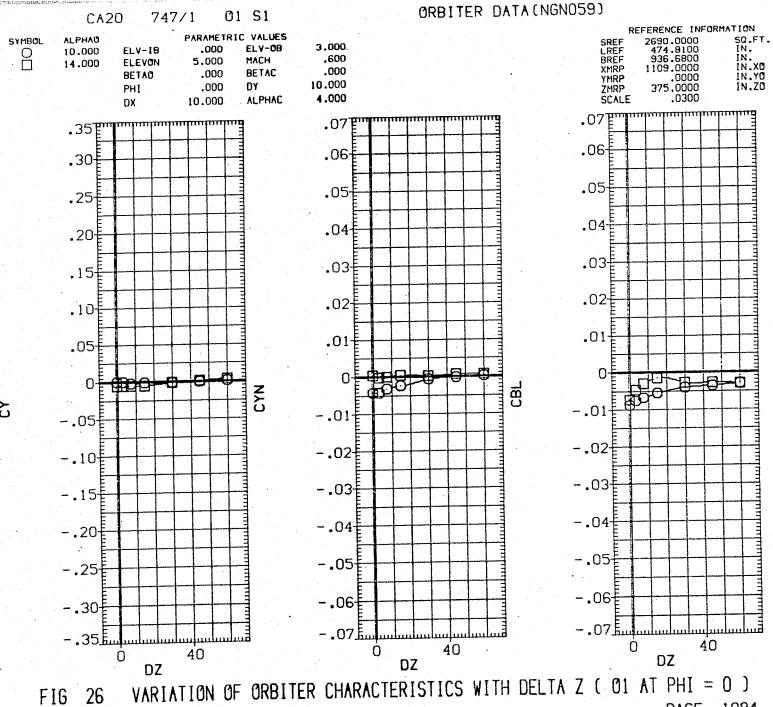


FIG 26 PAGE 1092





PAGE 1094

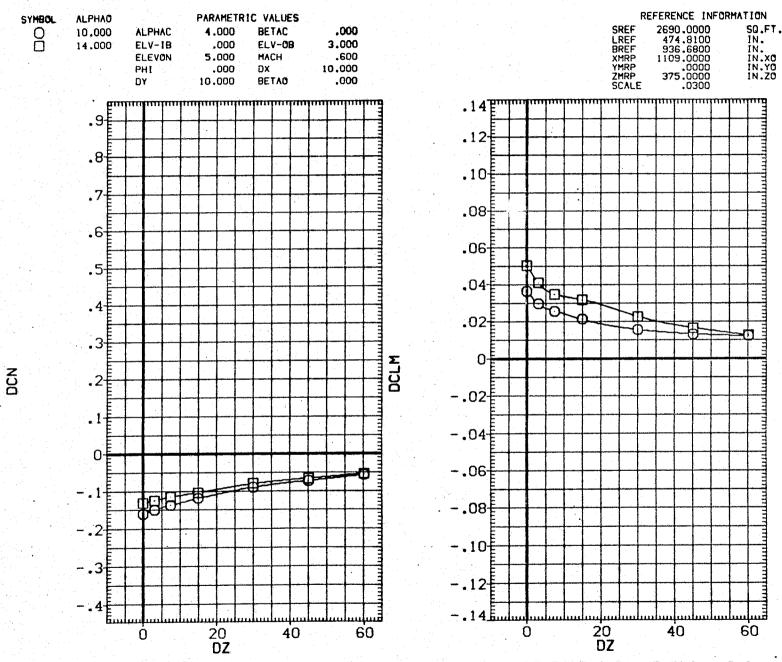


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1095

FIG 26 PAGE 1096

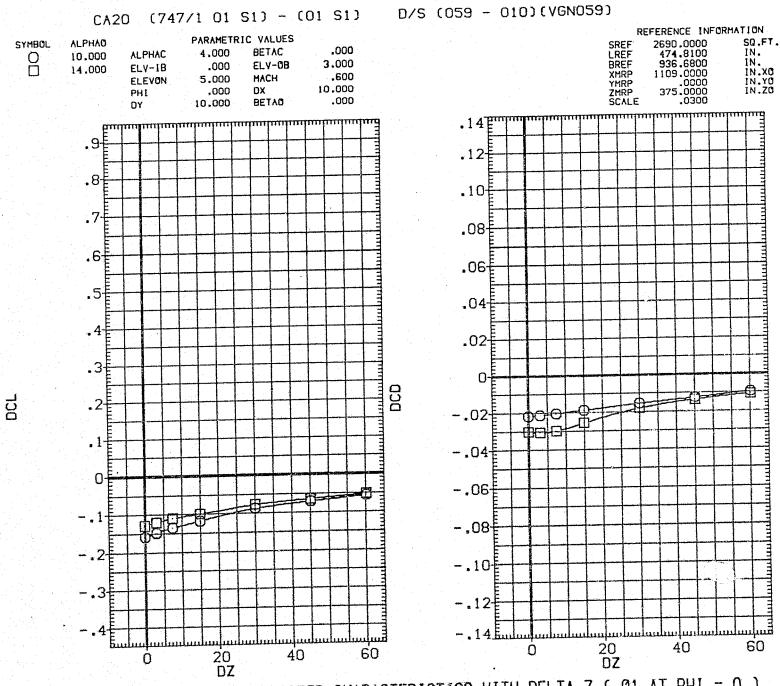
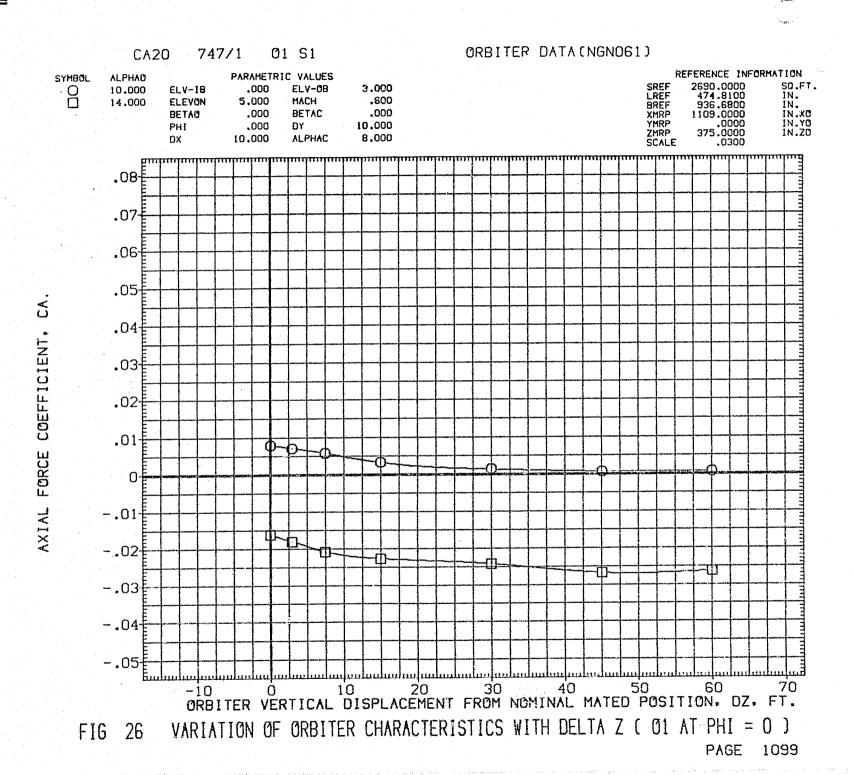


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1097

26 FIG 1098 PAGE



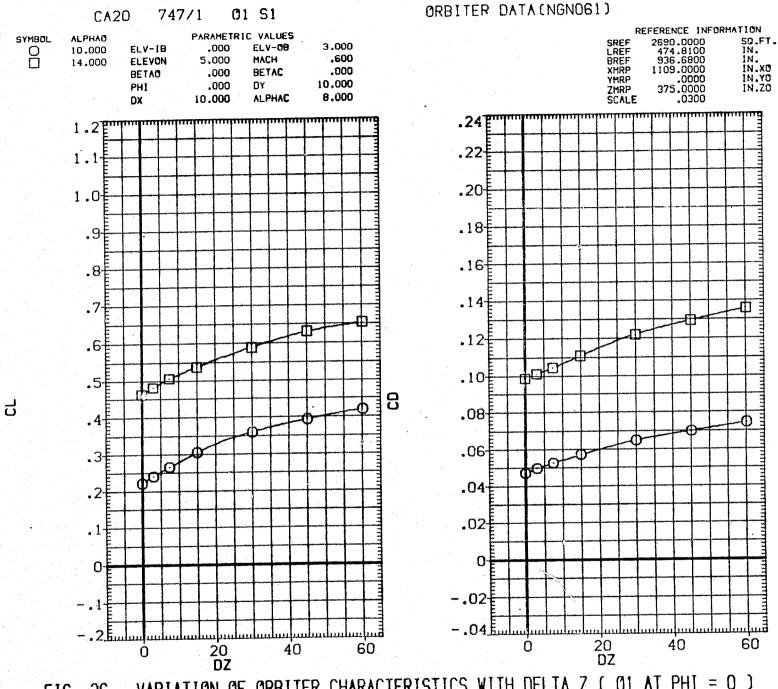
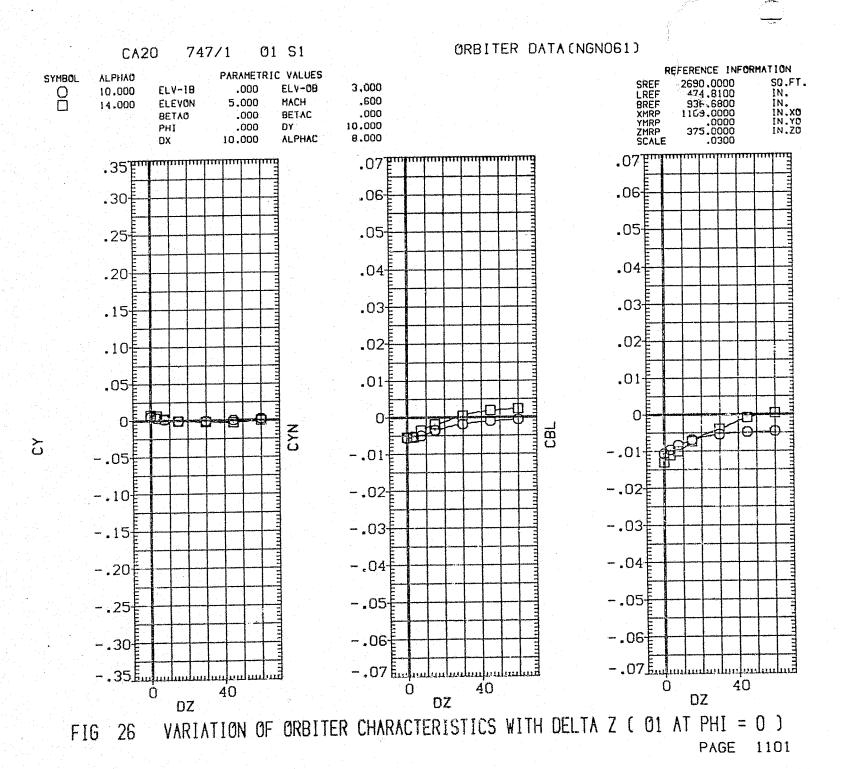


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1100



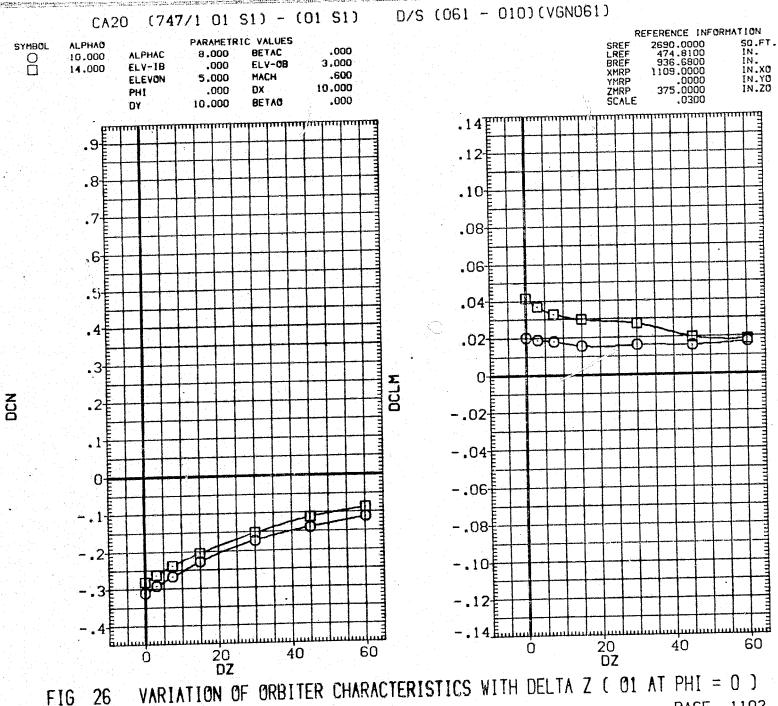
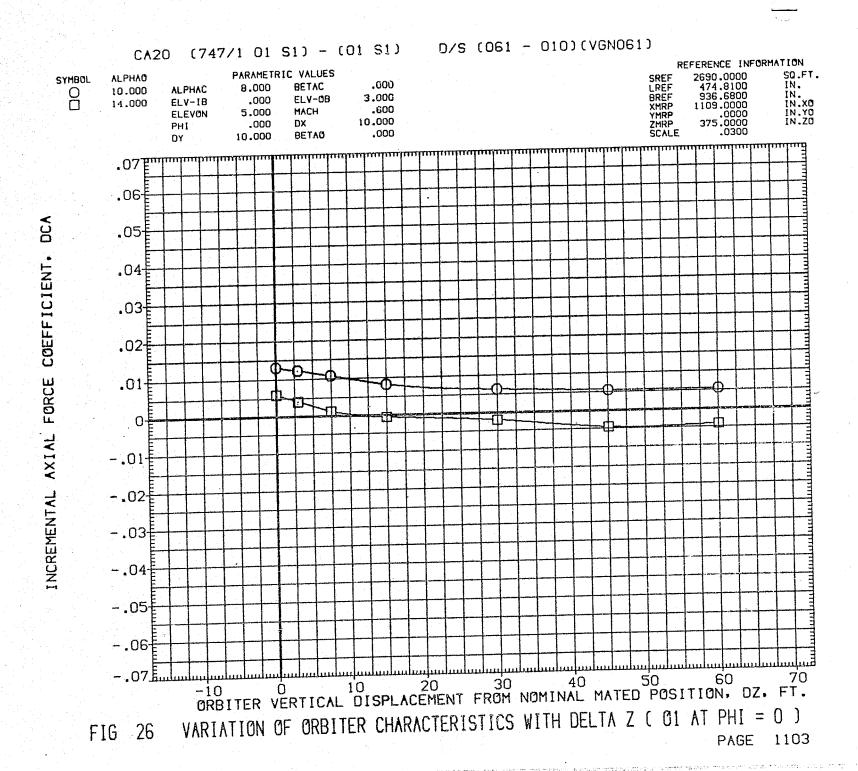


FIG 26 PAGE 1102



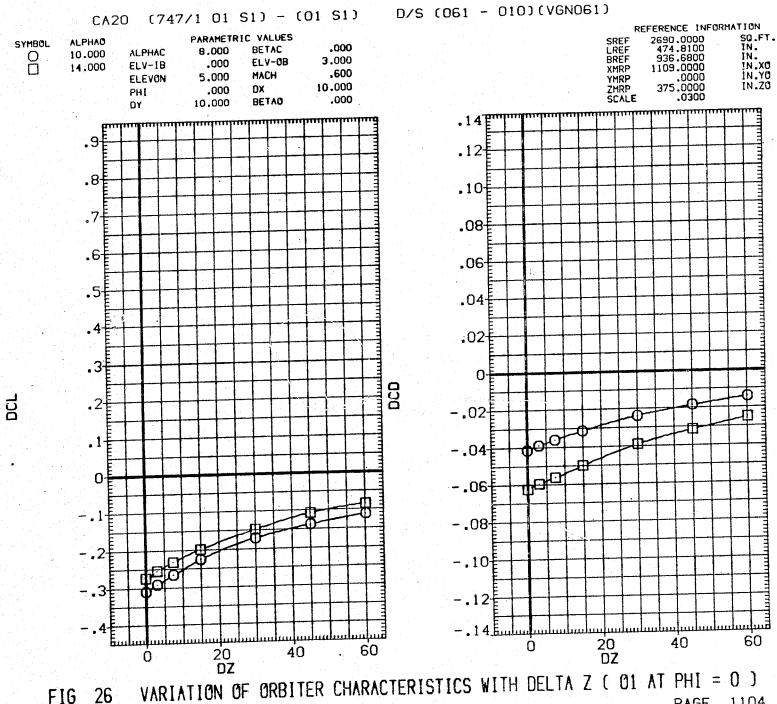
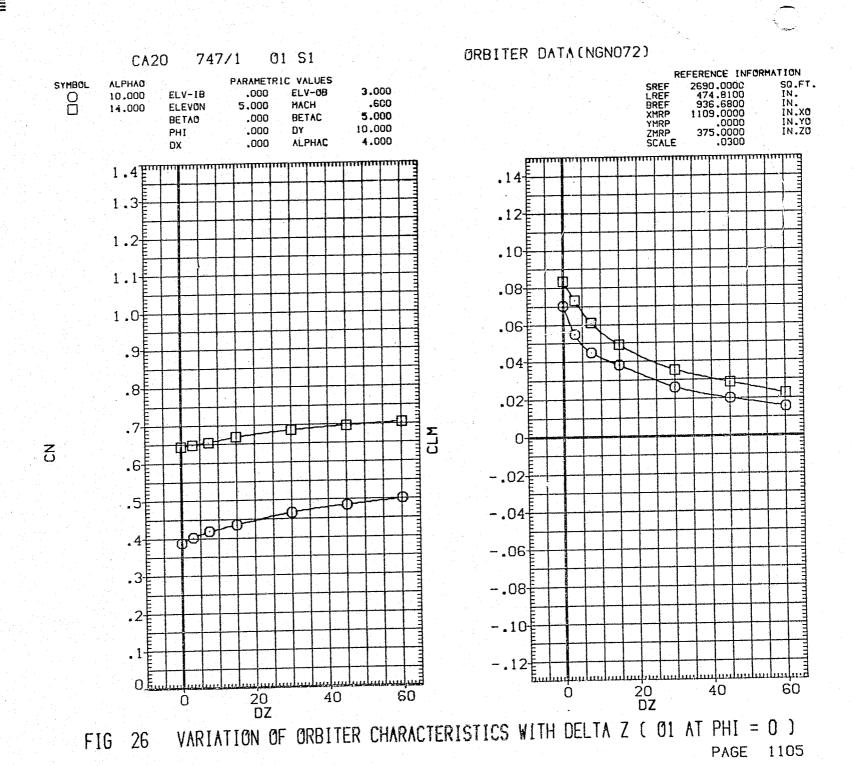


FIG 26 PAGE 1104



PAGE 1106

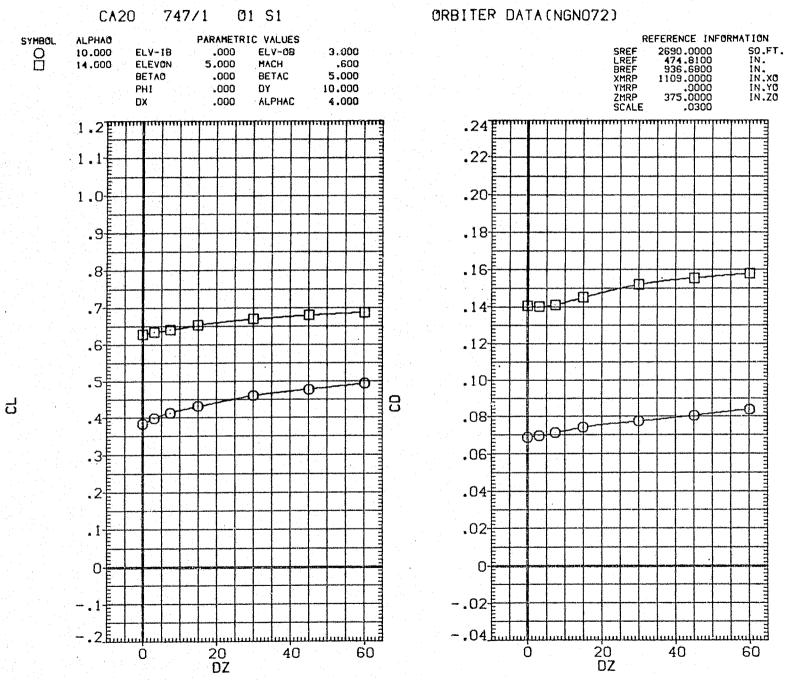


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1107

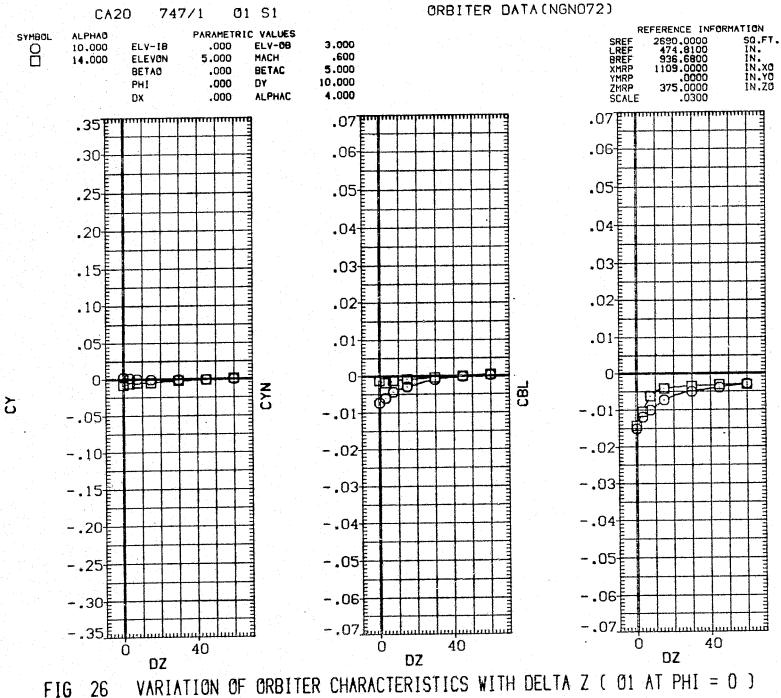


FIG PAGE 1108

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1109

PAGE 1110

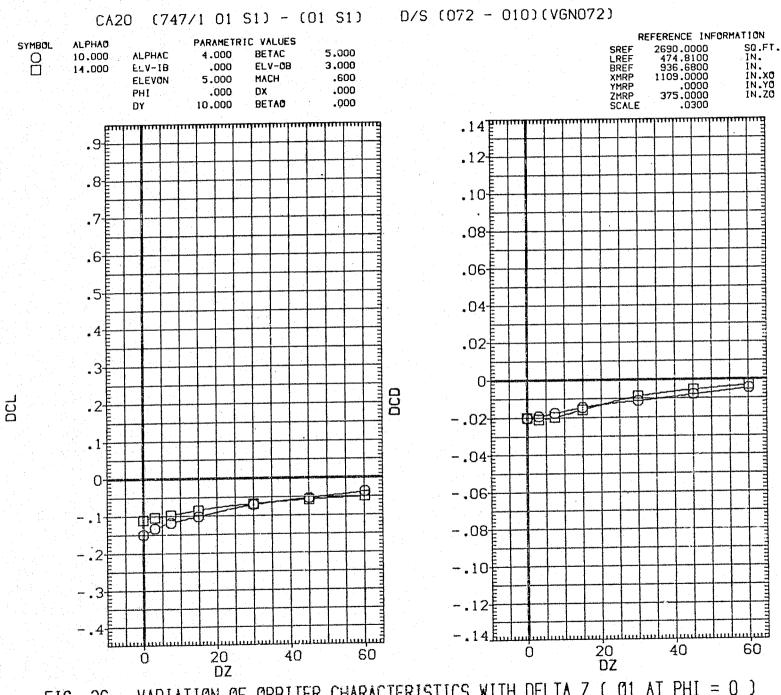


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1111

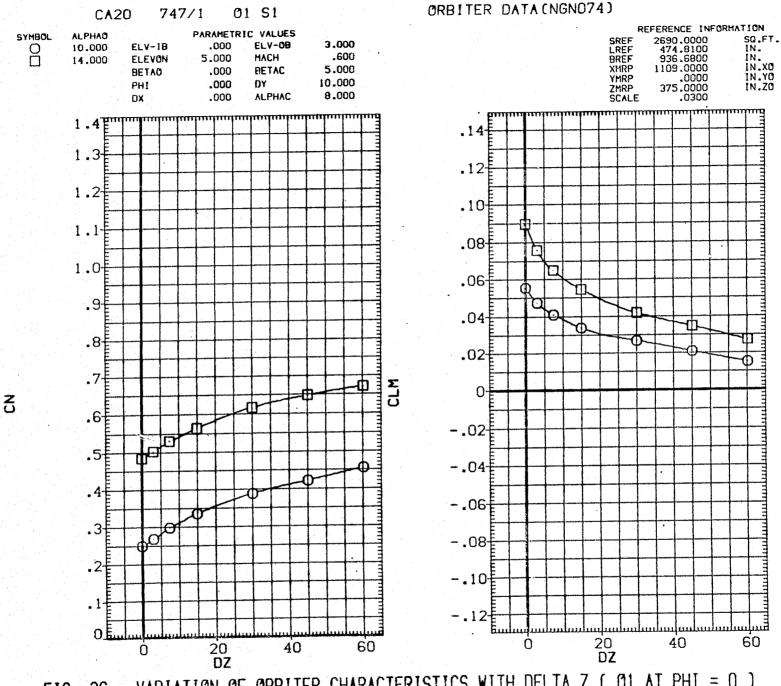


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1112

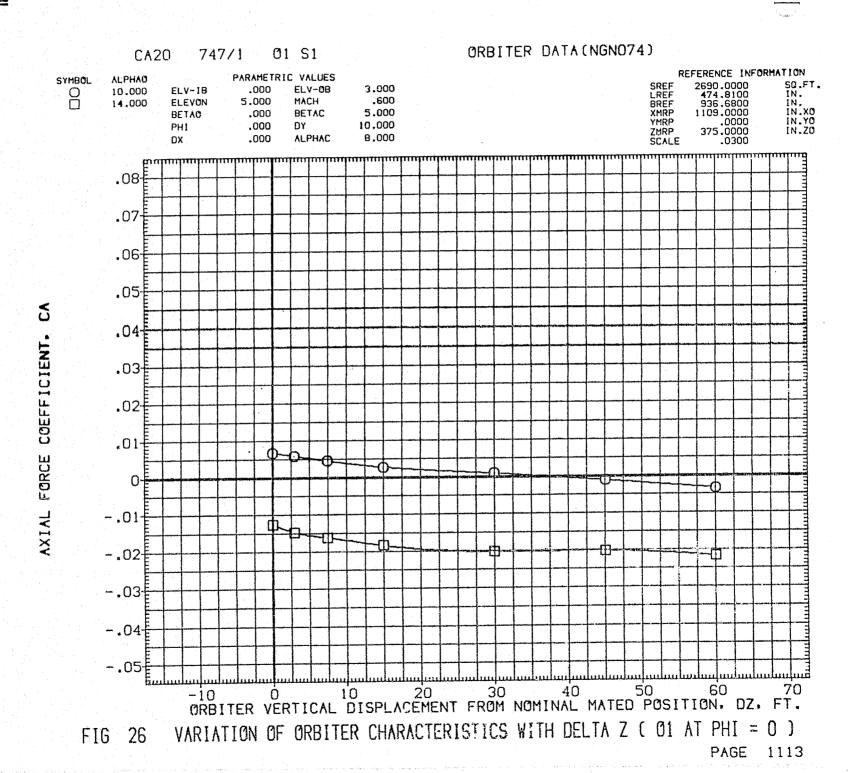


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1114

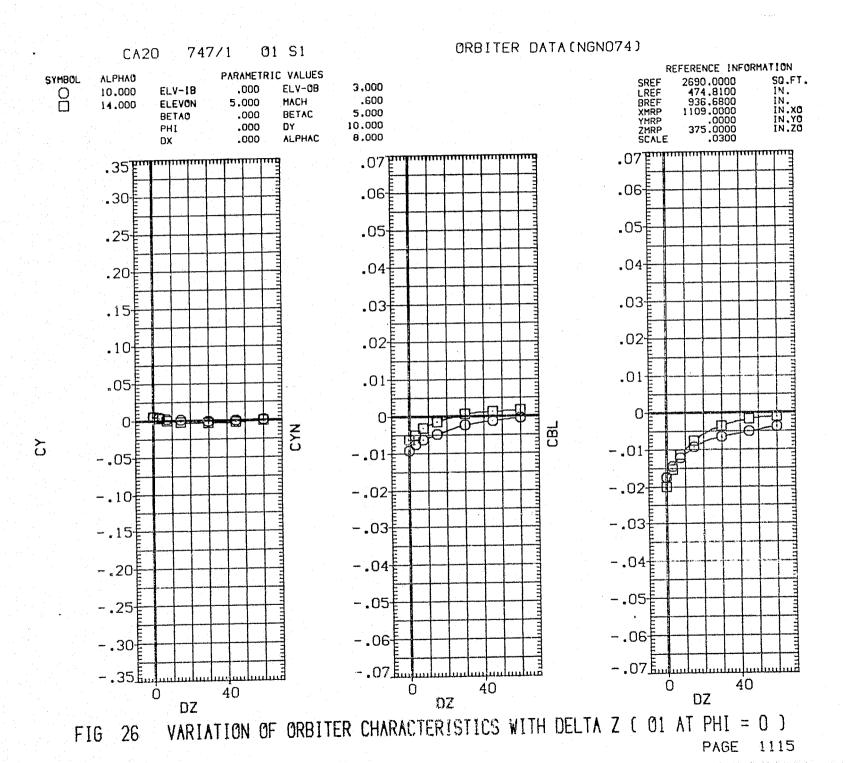
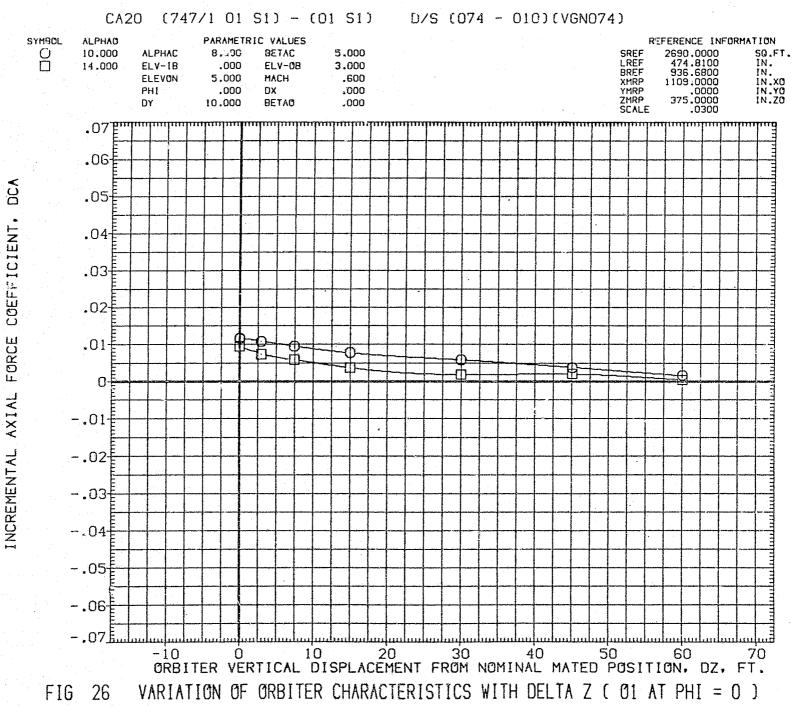


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1116



PAGE 1117

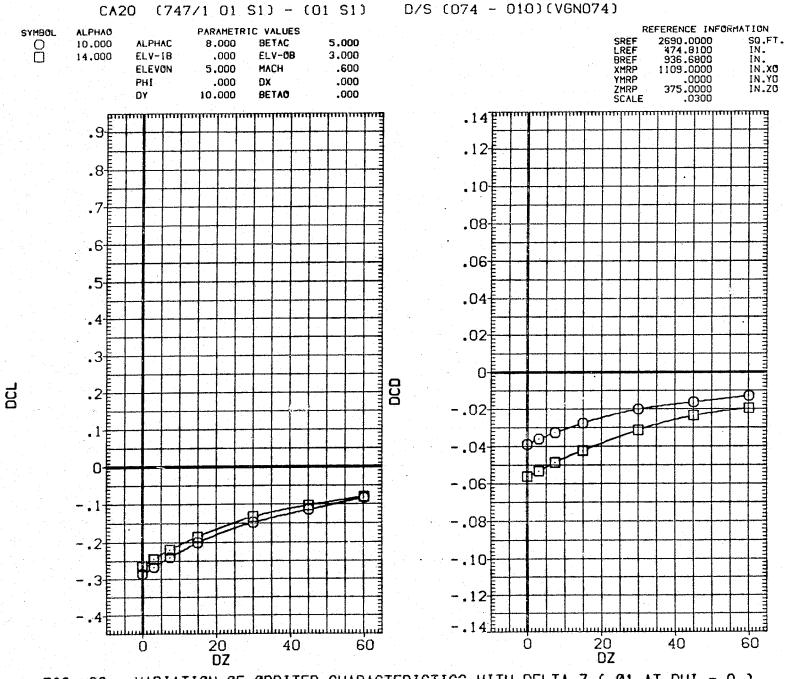
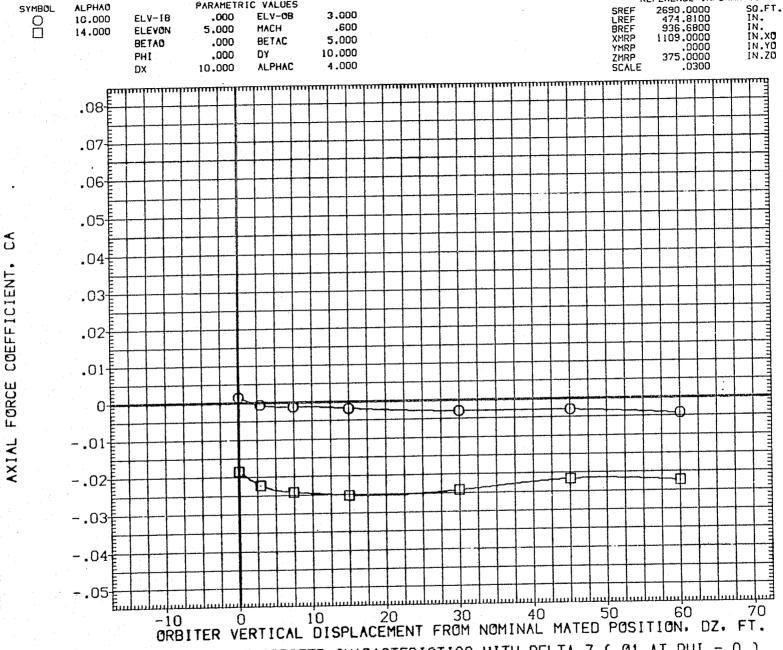


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1118

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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1120

REFERENCE INFORMATION

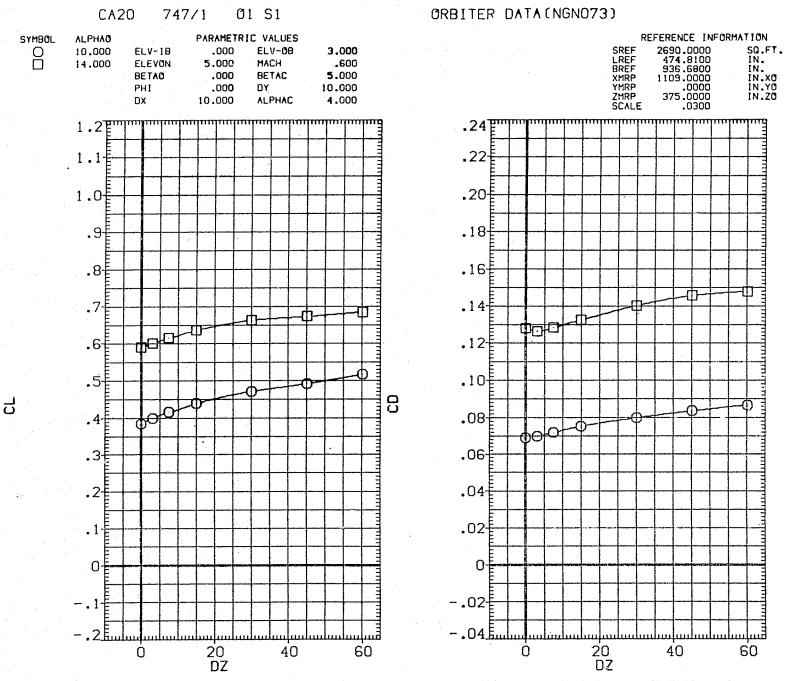


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1121

FIG 26 PAGE 1122

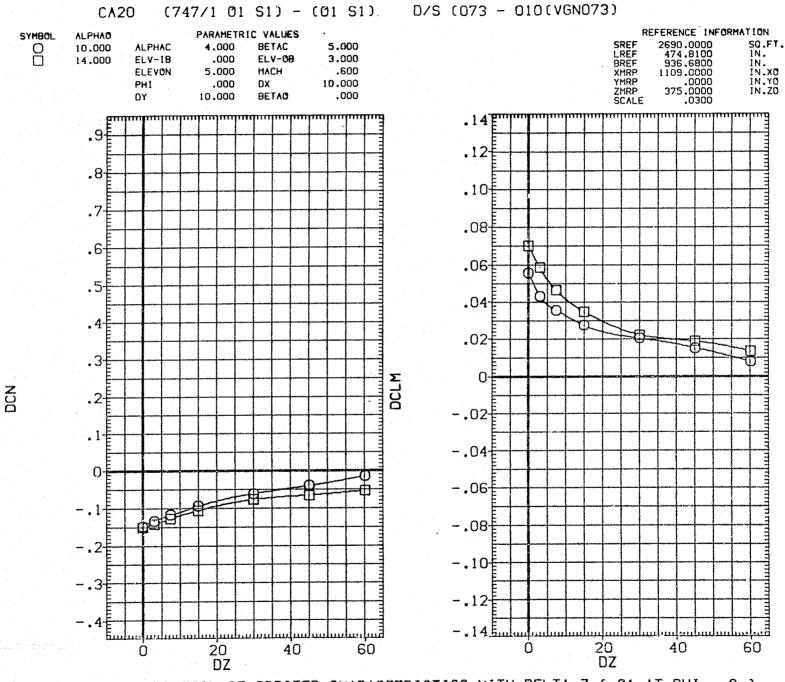
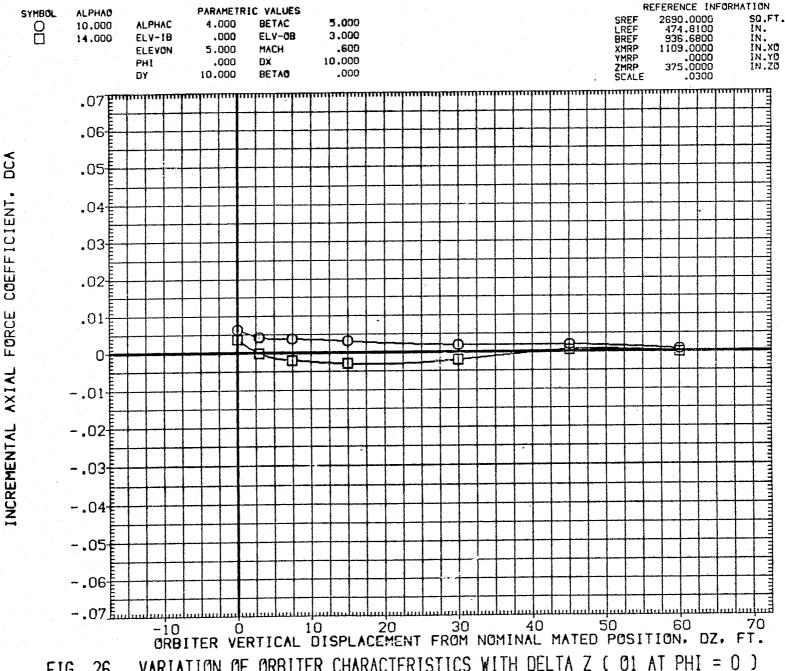


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1123



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG PAGE 1124

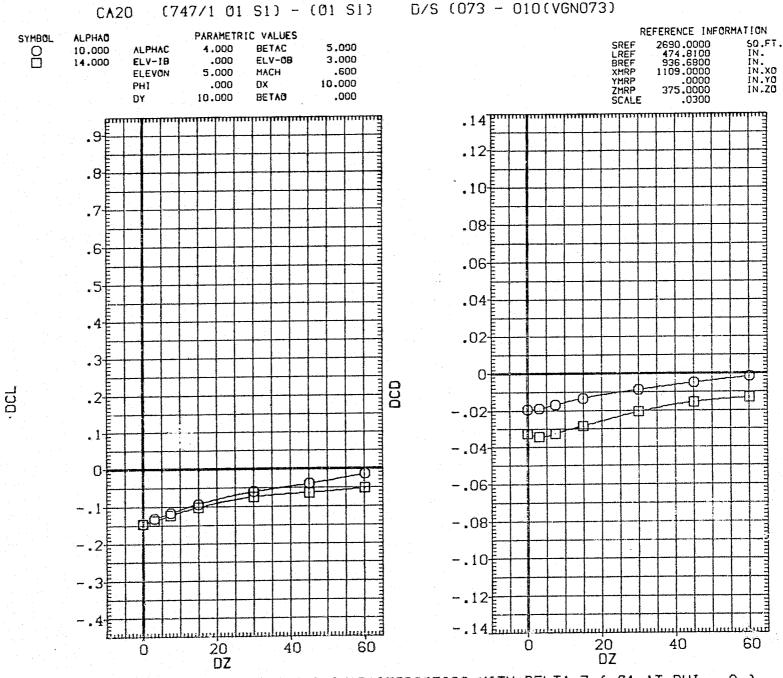
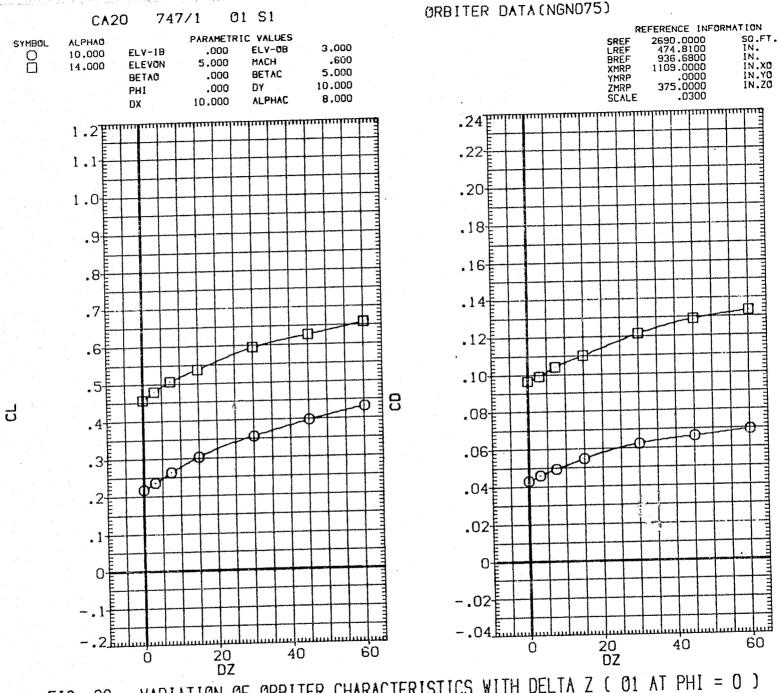


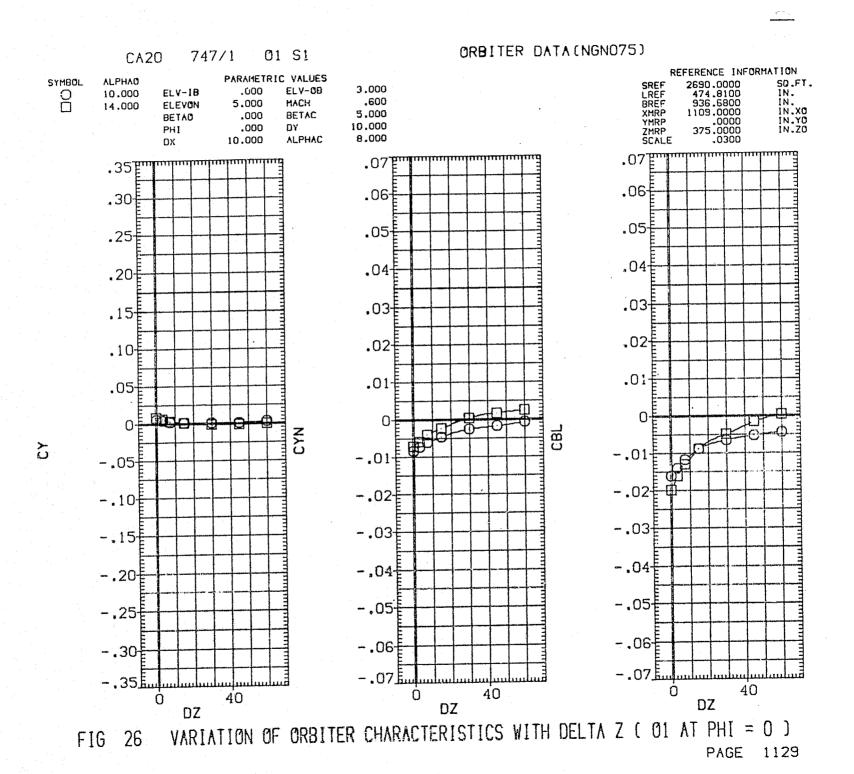
FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1125

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PAGE 1127



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 ) FIG 26 PAGE 1128



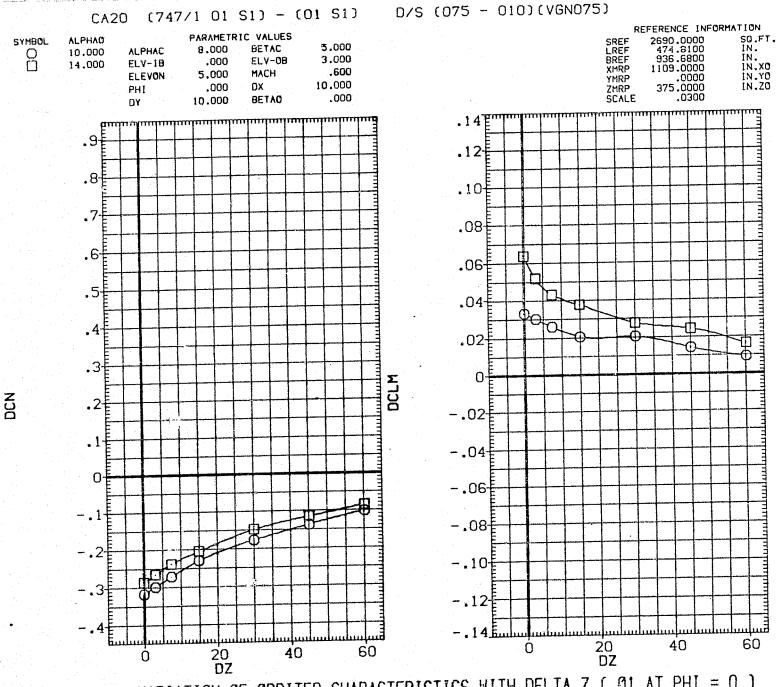


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1130

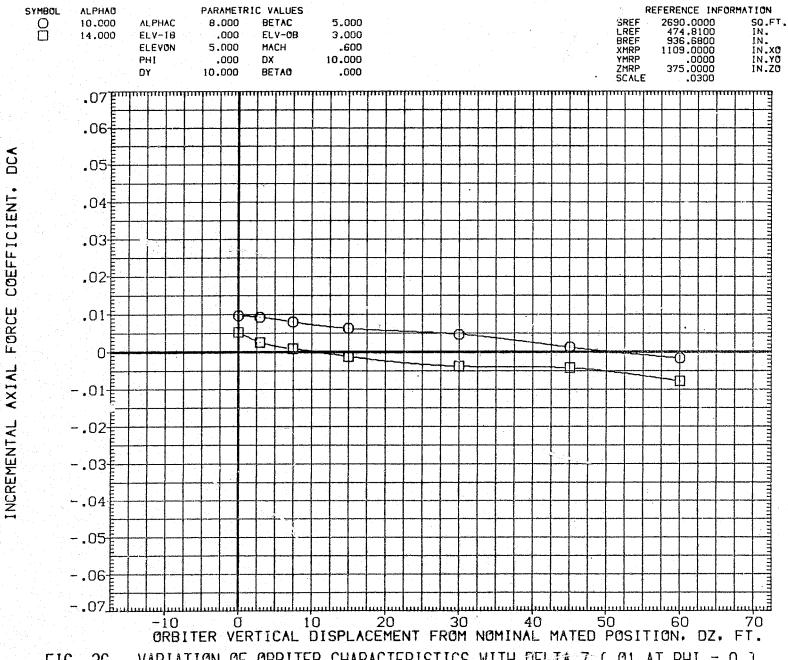


FIG 26 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 0 )
PAGE 1131

26 FIG PAGE 1132

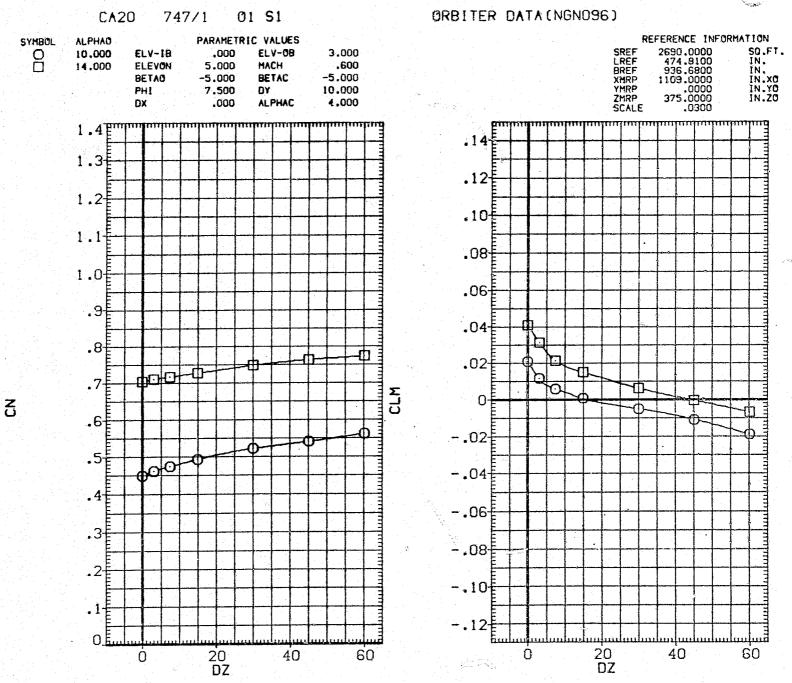
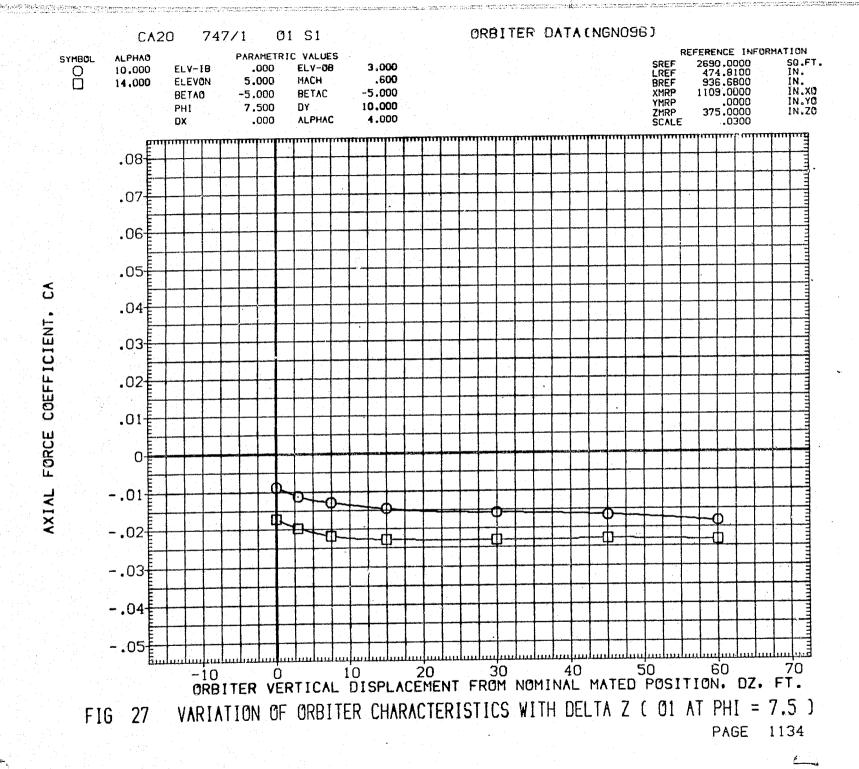


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1133



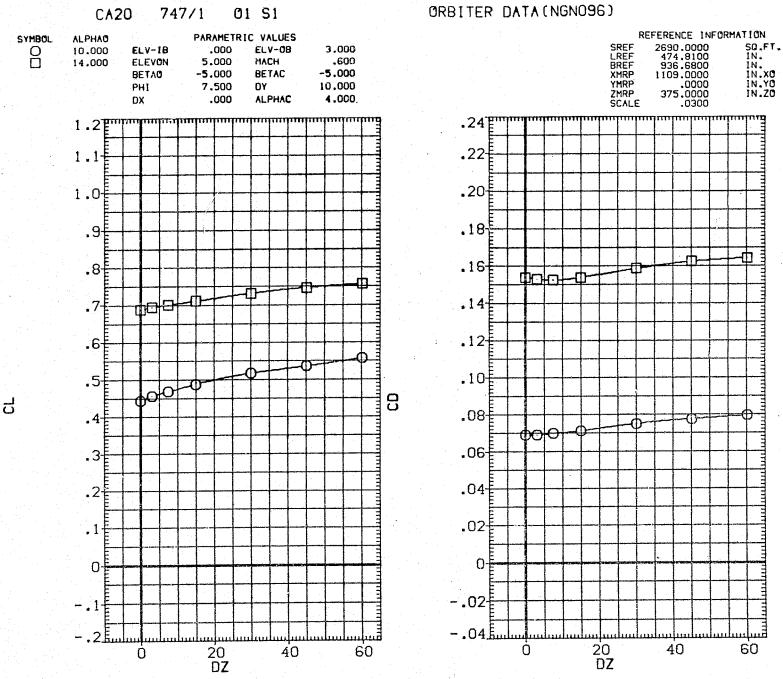


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1135

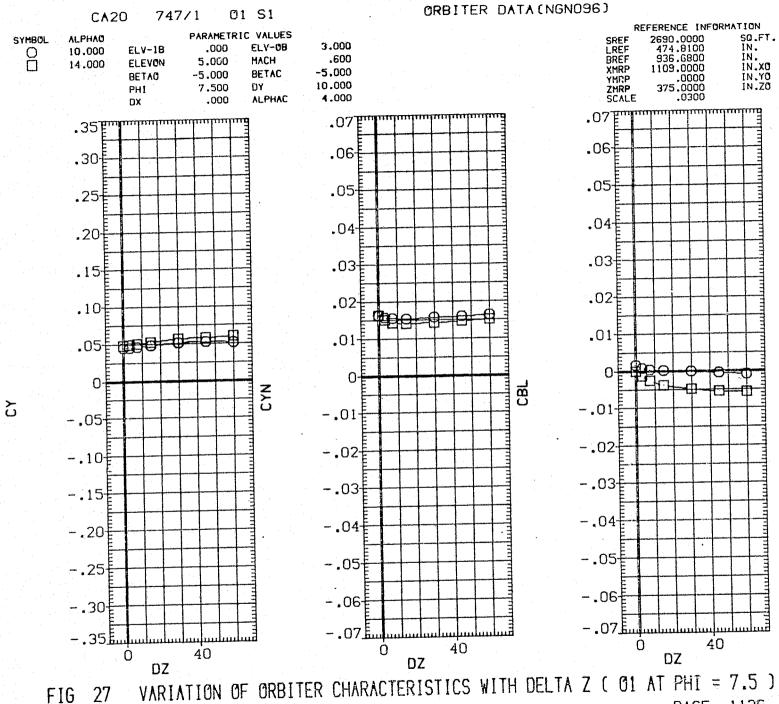
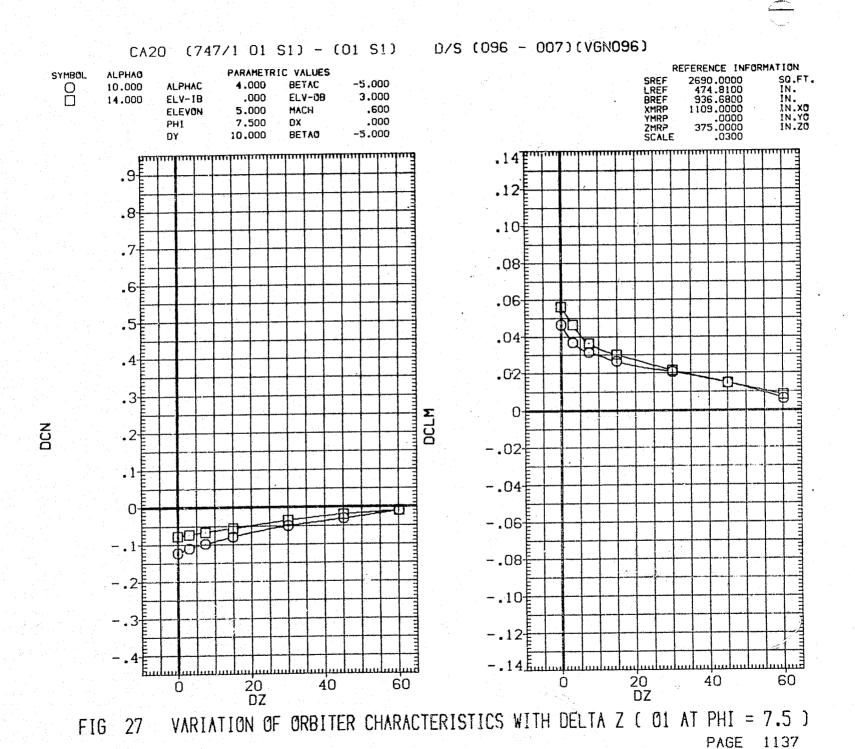


FIG 27 PAGE 1136



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1138

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1139

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1140

PAGE 1141

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1142

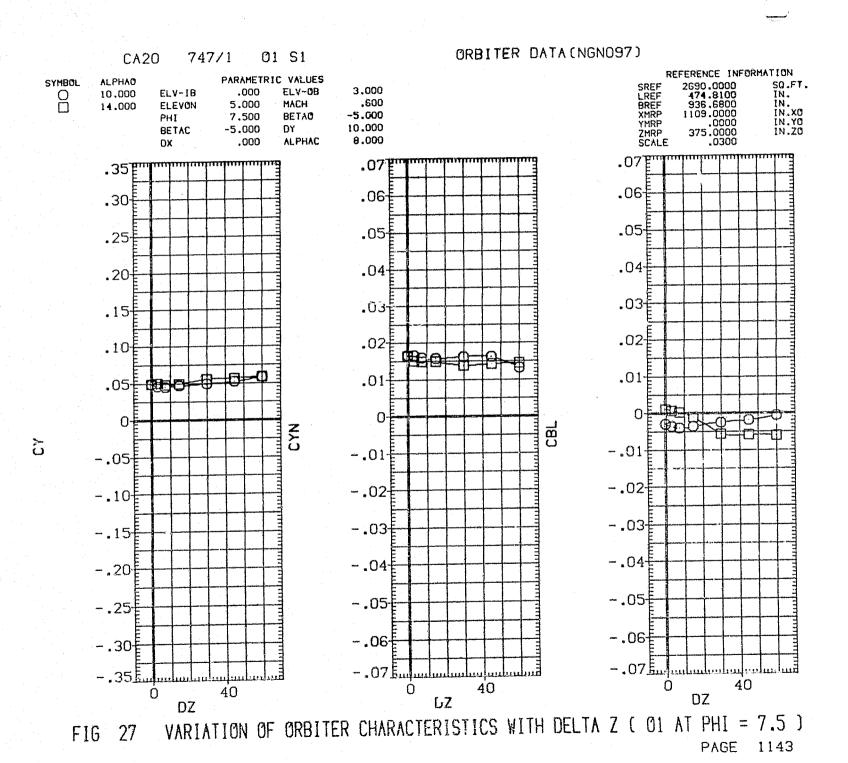


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1144

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1145

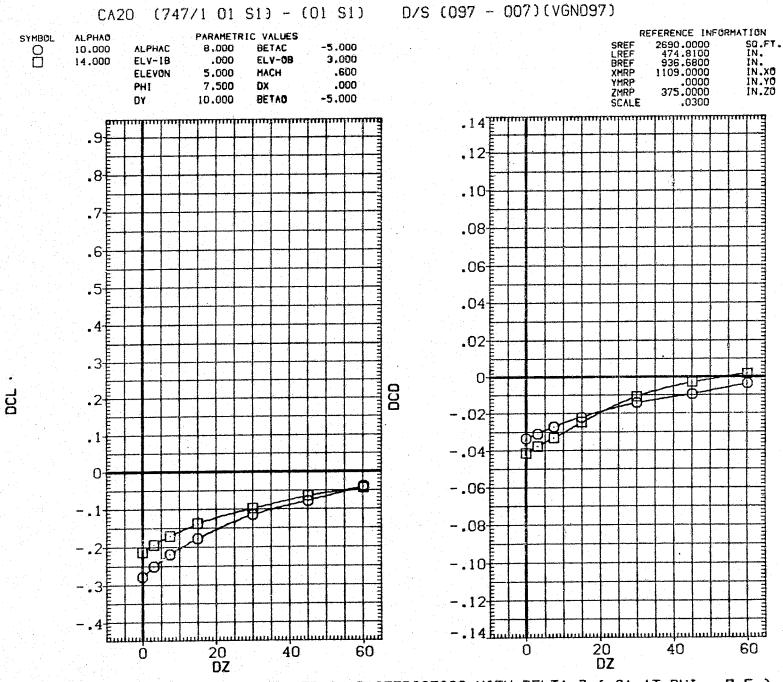


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1146

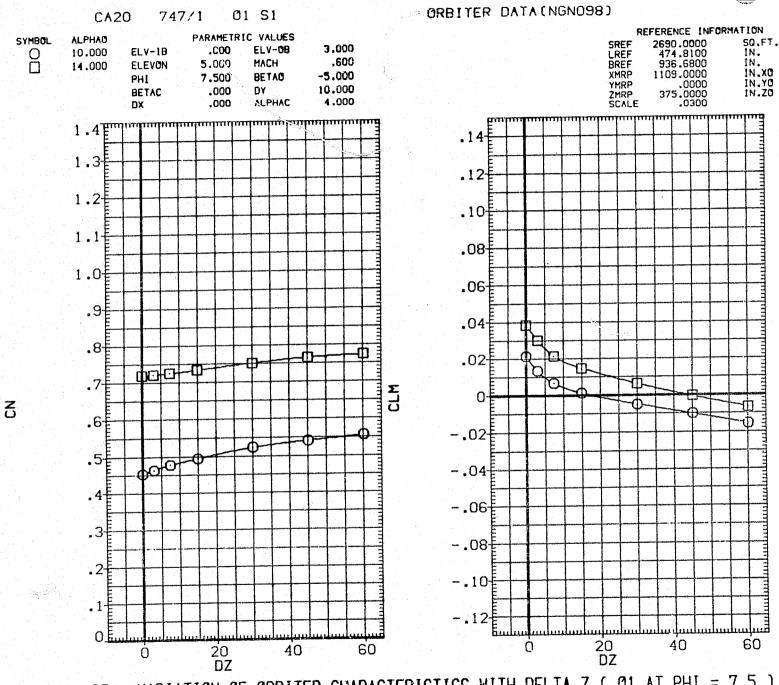


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1147

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1148

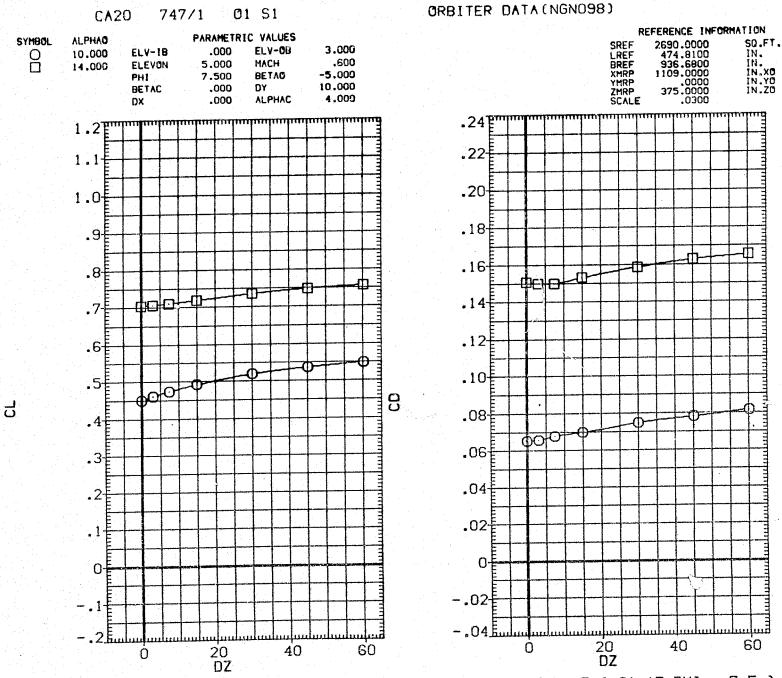


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1149

PAGE

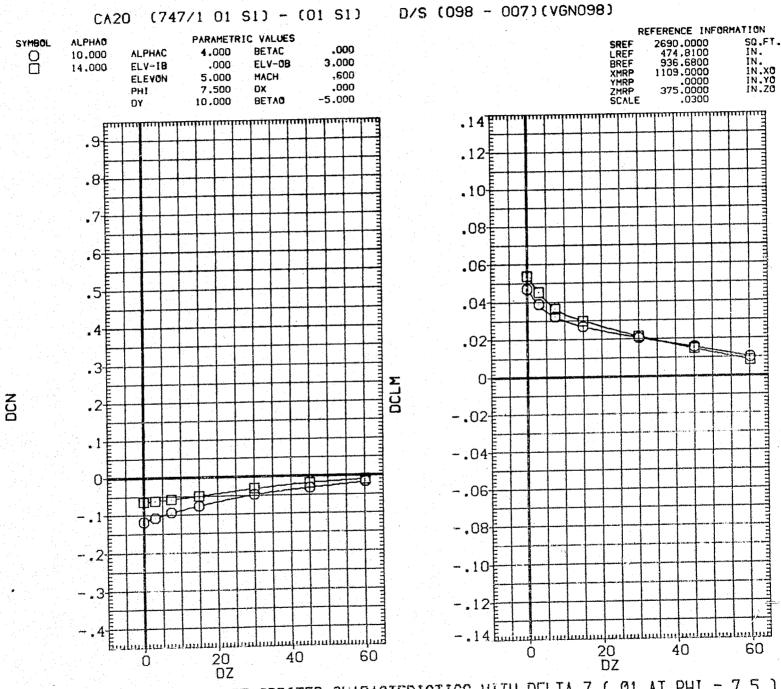
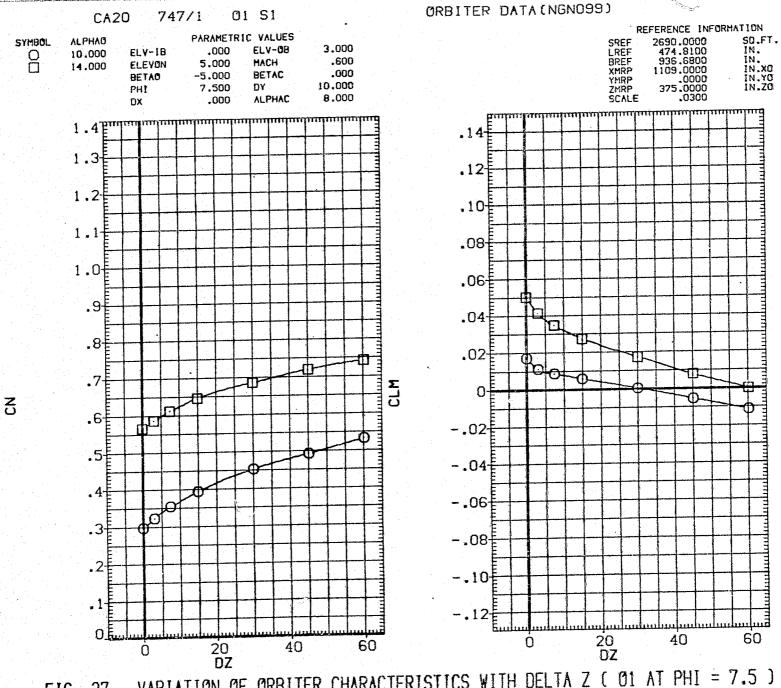


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1151

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1152

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1153



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) 27 FIG PAGE 1154

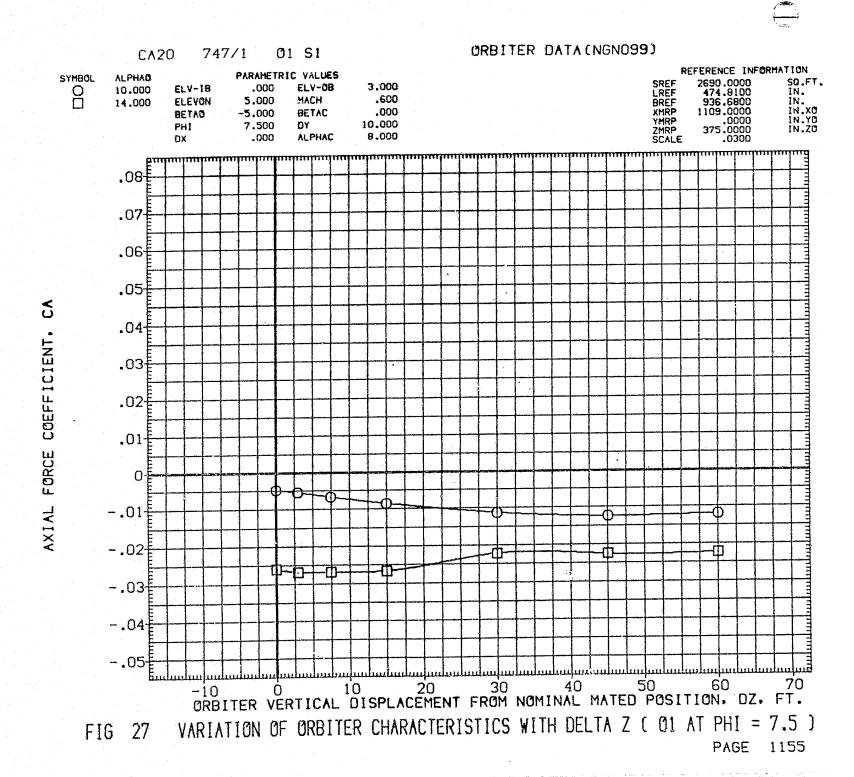
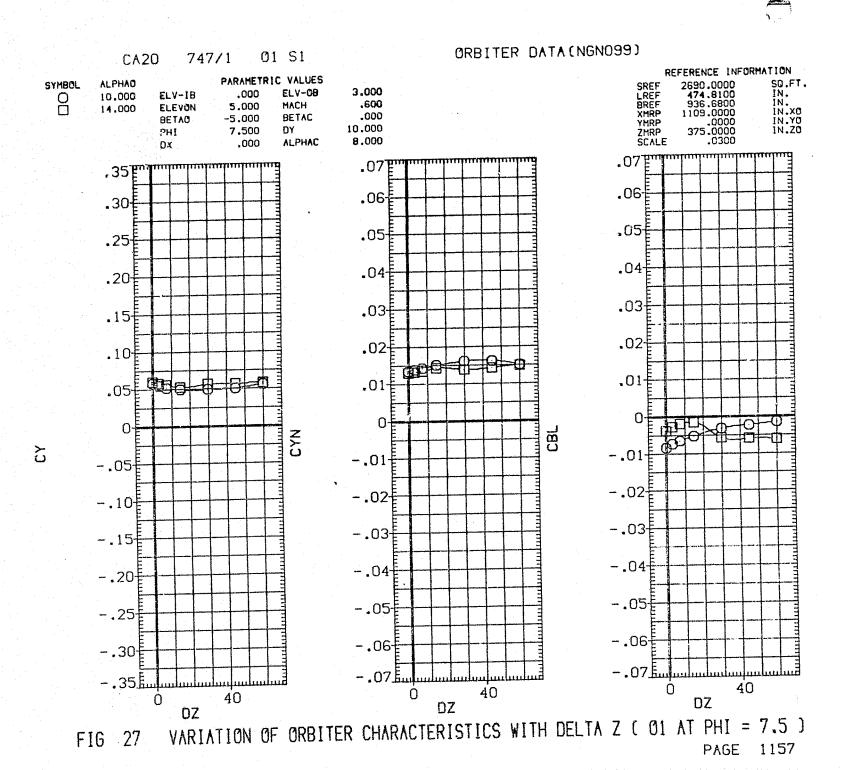
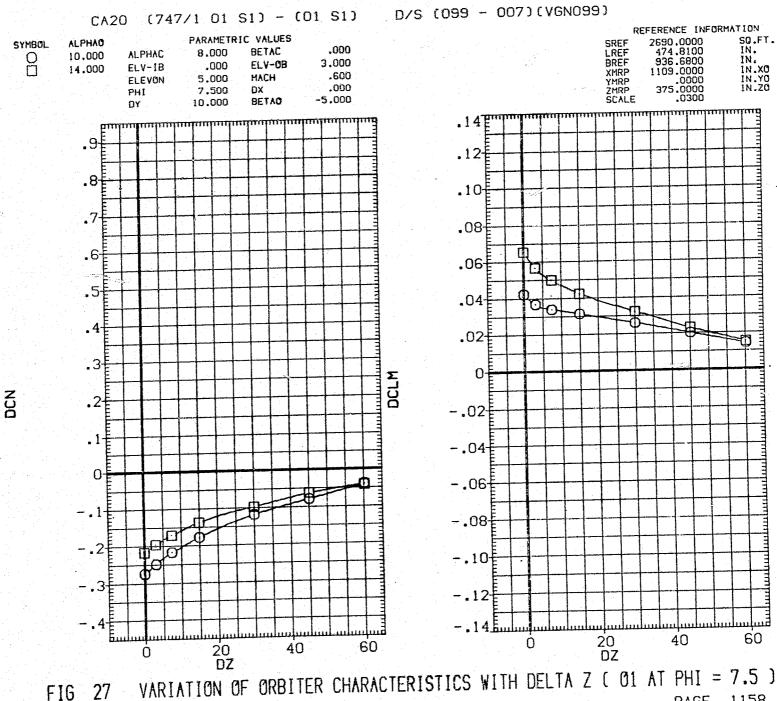


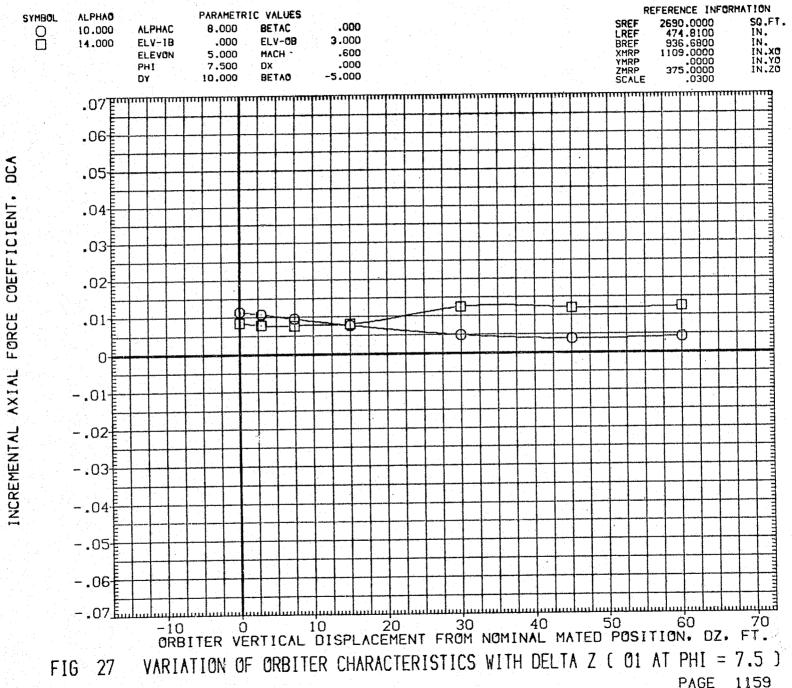
FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1156

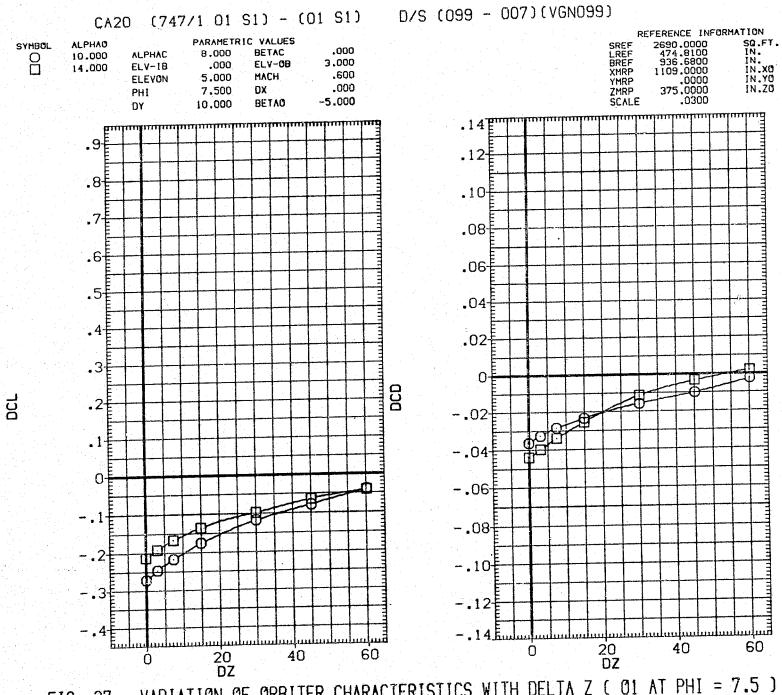






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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG PAGE 1160

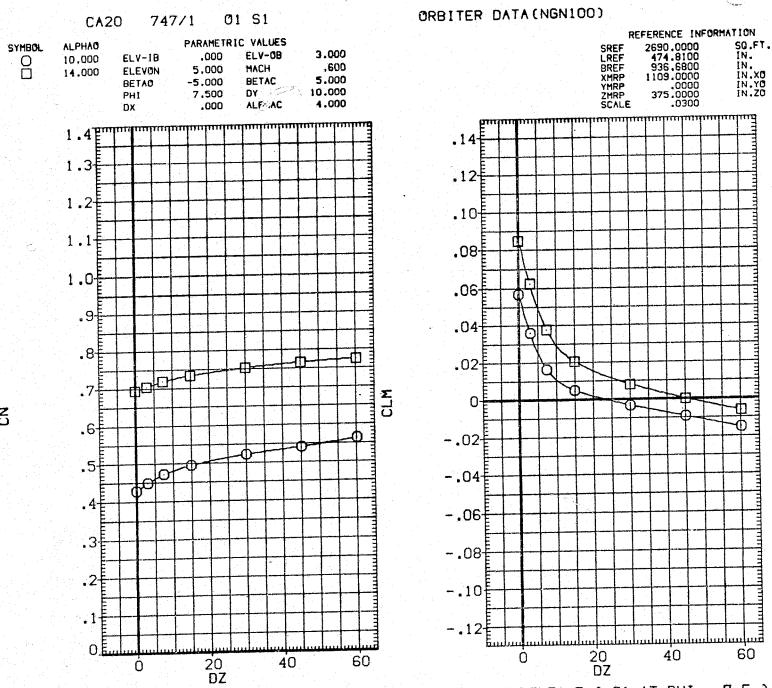


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1161

(Carrier of Children Ships of Control of Con

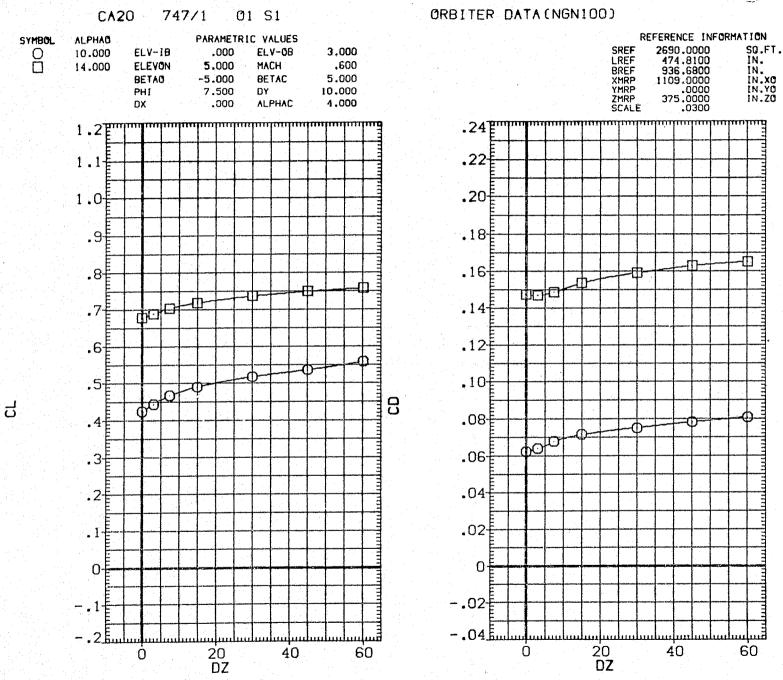
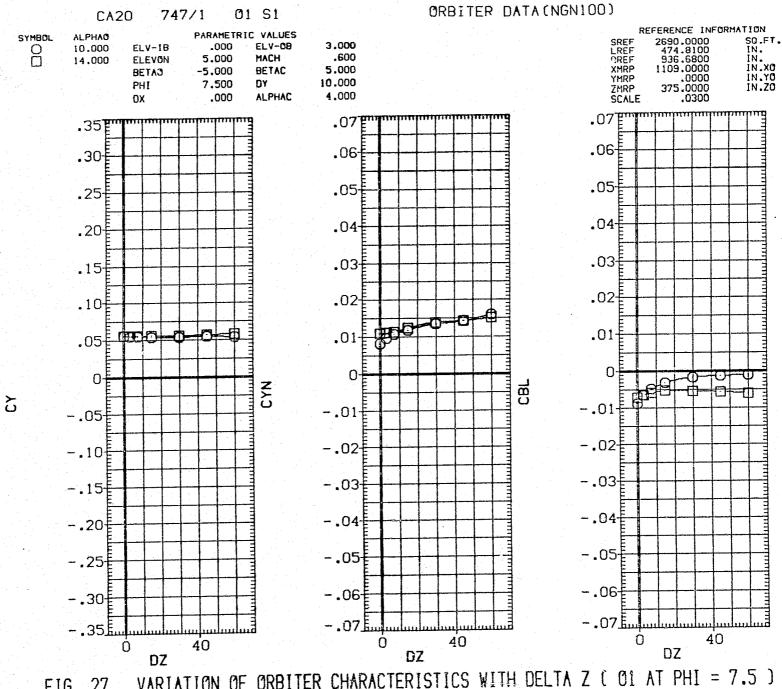


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1163



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG PAGE 1164

D/S (100 - 007) (VGN100)  $(747/1 \ 01 \ S1) - (01 \ S1)$ REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL **ALPHAO** 2690.0000 474.8100 936.6800 1109.0000 0000 375.0000 SQ.FT. SREF LREF BREF 5.000 4.000 BETAC 8 10.000 **ALPHAC** 3.000 IN. IN.XO IN.YO IN.ZO .000 ELV-08 14.000 ELV-IB XMRP .600 ELEVON 5,000 MACH .000 DX 7.500 PHI ZHRP -5.000 BETAO DY 10,000 .14E .9₽ .12 .8 .10 .08<sup>£</sup> .6<del>‡</del> .06 .04 .02 0+ DCLM DCN -.02 -.04

0-

-.3E

0

20 DZ VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 1165 PAGE

60

40

-.06

-.08<del>‡</del>

-.10

-.12

- 14<u> Fulmhaladada</u>

20 DZ

40

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1166

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1167

60

20 DZ

0

40

-.12

0

20 DZ 40

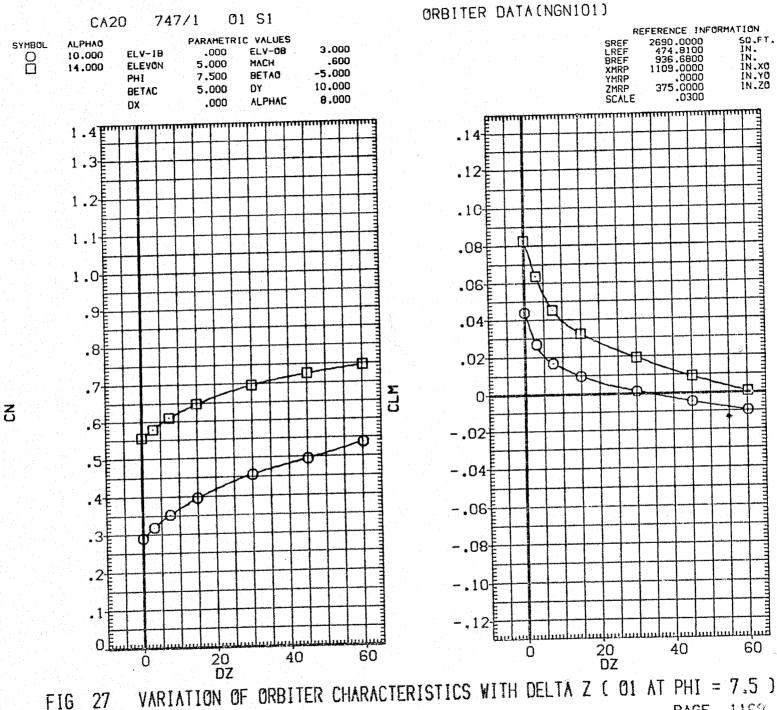
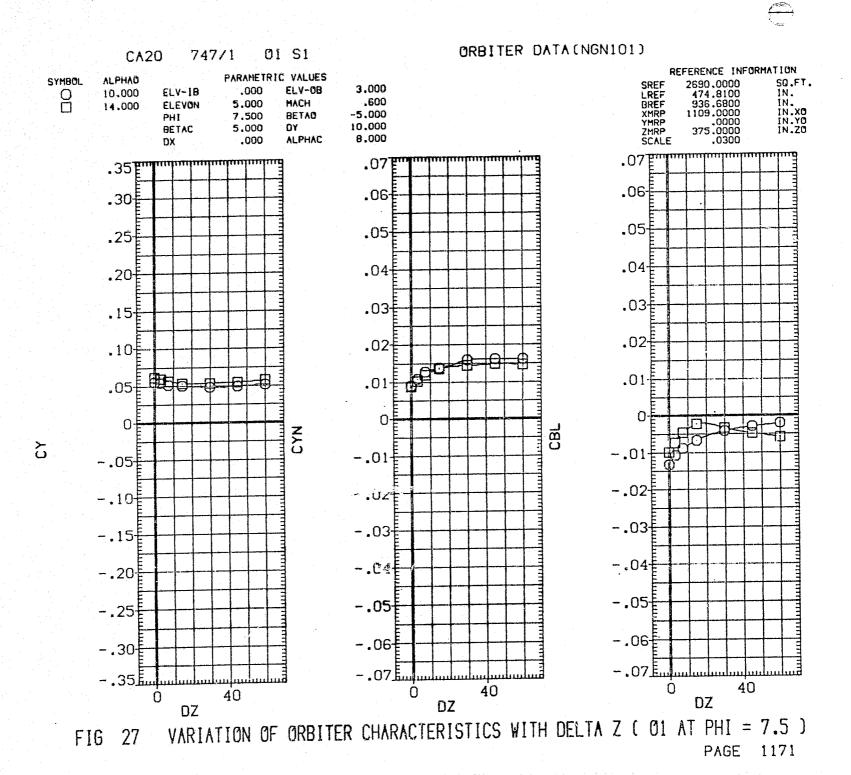


FIG 27 PAGE 1168

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FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1170



27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG PAGE 1172

Ó

60

40

60

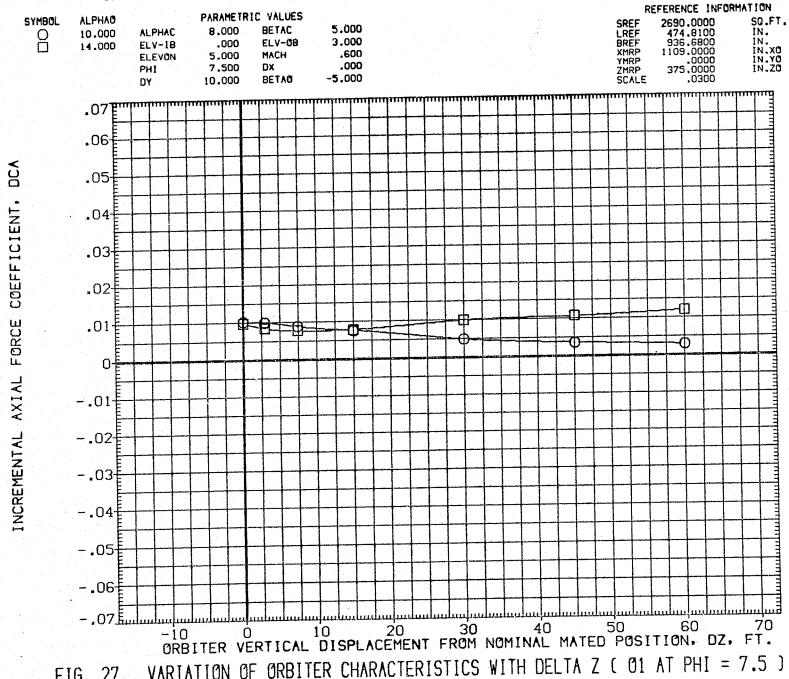
20 DZ

0





D/S (101 - 007) (V6N101) CA20 (747/1 01 S1) - (01 S1)



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1173

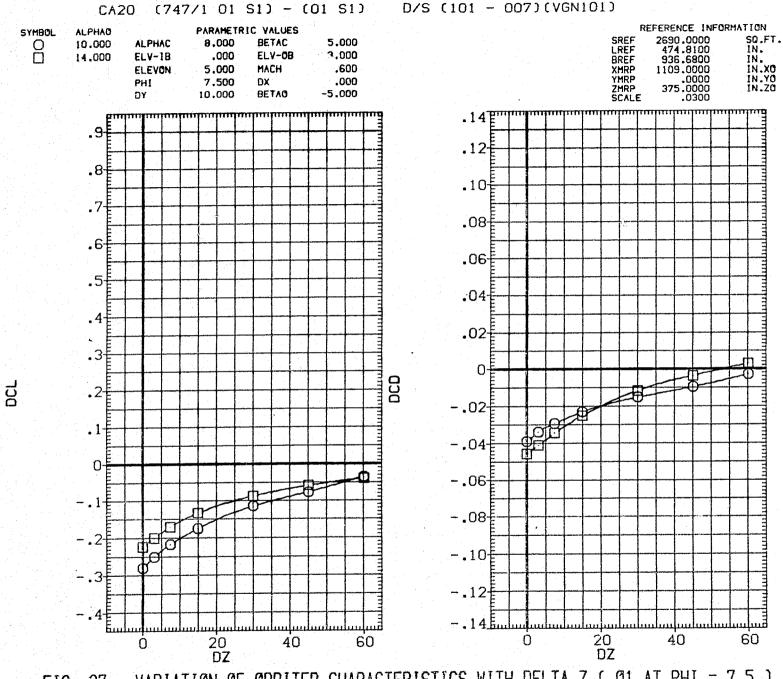
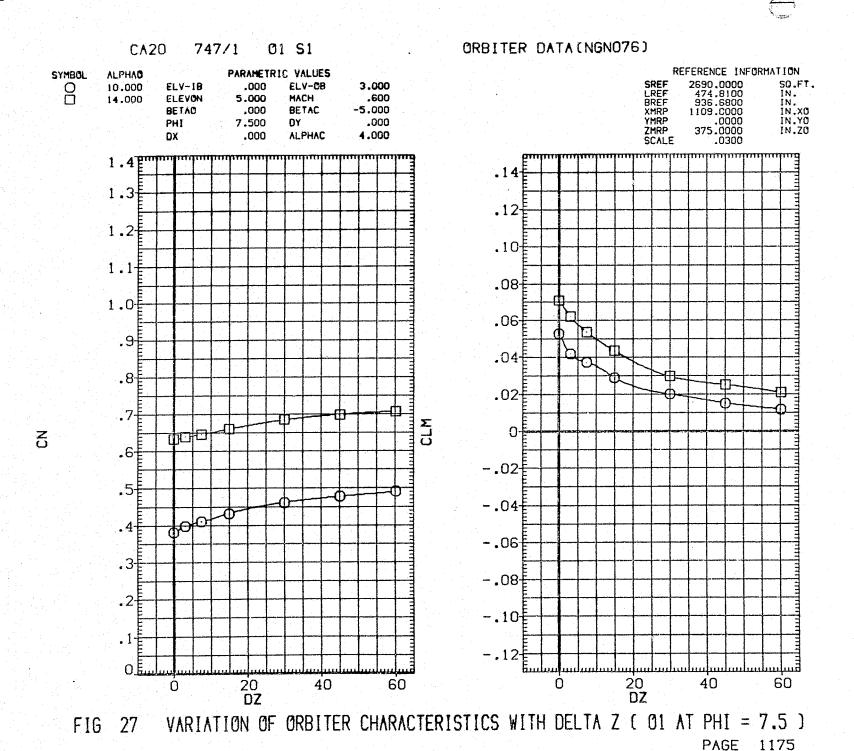


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1174



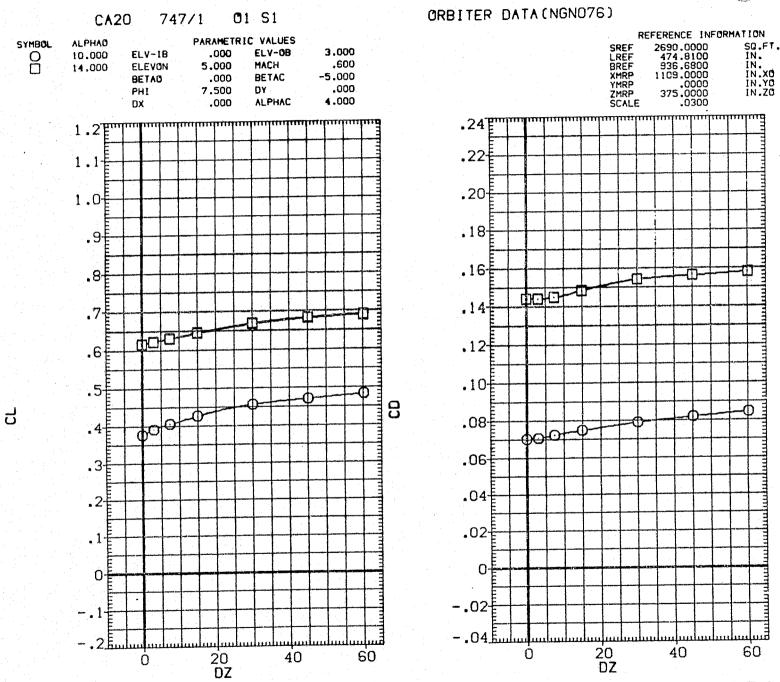


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1177

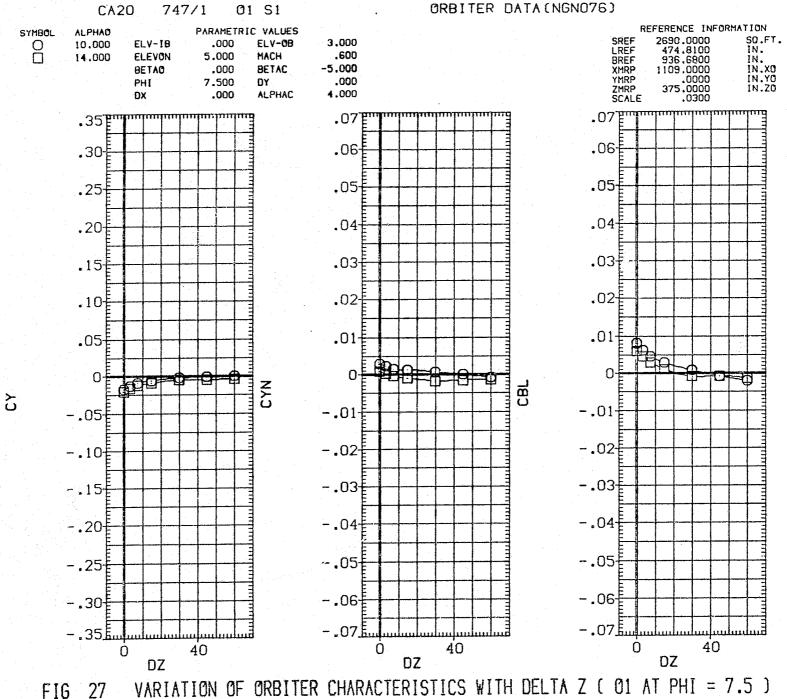


FIG PAGE 1178

CA20 (747/1 01 S1) - (01 S1) . D/S (076 - 010)(VGN076)

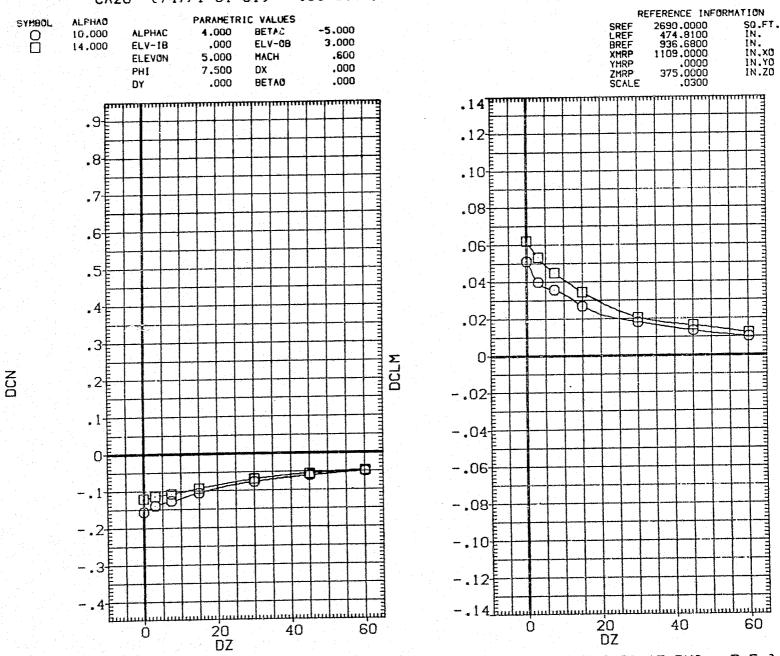
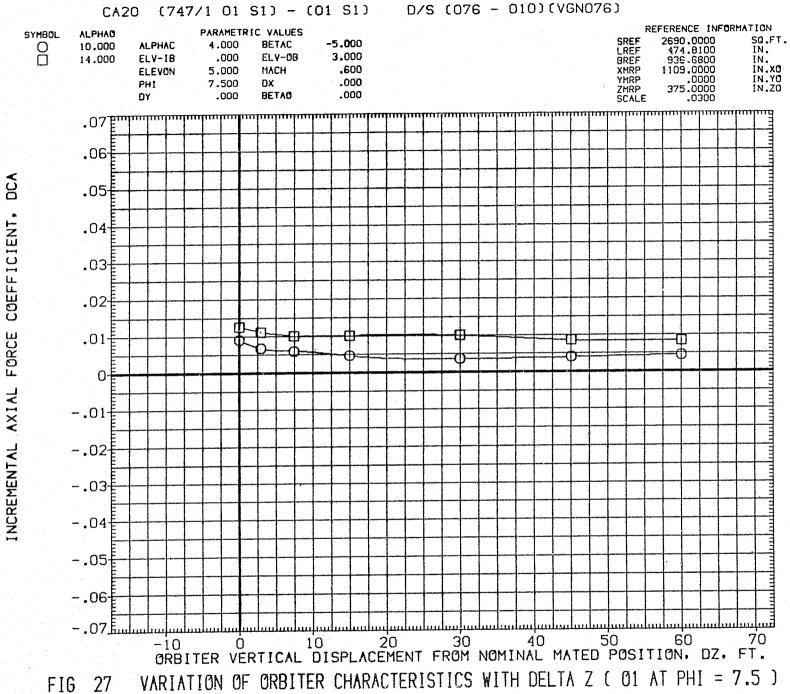


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
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D/S (076 - 010)(VGN076) (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES **ALPHAO** 2690.0000 474.8100 SO.FT. SYMBOL -5.000 IN. 0 4.000 BETAC **ALPHAC** 10.000 LREF 3.000 936.6800 .000 ELV-08 ELV-IB BREF 14.000 IN.XO IN.YO IN.ZO .600 XMRP HACH 5.000 **ELEVON** YMRP ZMRP .0000 375.0000 .0300 .000 7,500 DΧ PHI .000 BETAO .000 .14 Emilion .9 .12 .8<del>E</del> .10 .08 .6<del>[</del> .06 .04 .02‡ DCL -.02 -.04 0--.06 -.08 -.10<del>[</del> -.3<del>[</del> -.12 - . 14 <u>Eudu</u> ш|ш 20 DZ 20 DZ 60 40 40 60 0 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

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FIG

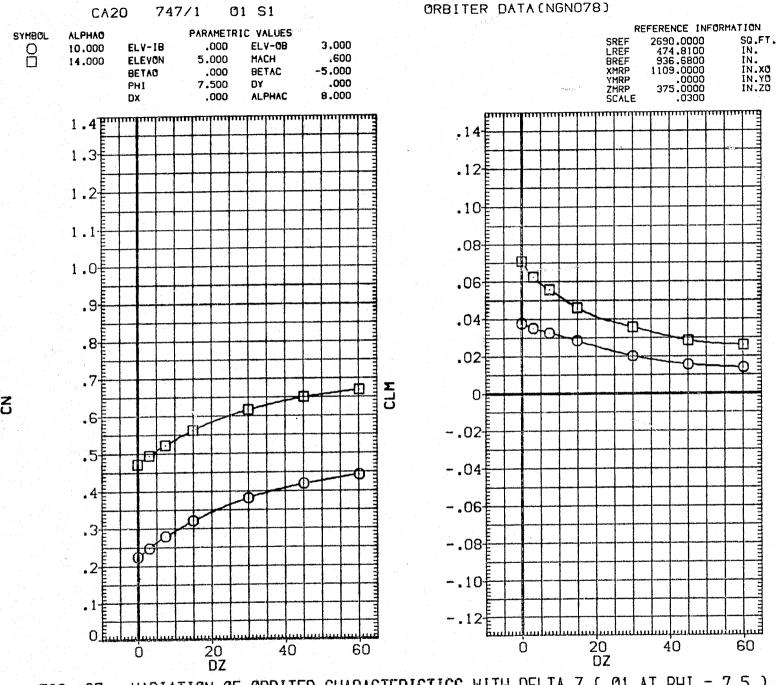


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1182

01 51 ORBITER DATA (NGNO78) CA20 747/1 PARAMETRIC VALUES REFERENCE INFORMATION ALPHAO 2690.0000 474.8100 3.000 SREF SQ.FT. 00 10.000 .000 ELY-OB IN. .600 14,000 **ELEVON** 5.000 MACH. BREF IN. -5.000 IN.XO IN.YO IN.ZO BETAO .000 BETAC XMRP 1109.0000 YMRP ZMRP SCALE PHI 7.500 DY .000 .000 8.000 DX **ALPHAC** .0300 .050 .045 .040 .035 .030 COEFFICIENT, .025 .020 .015<del>[</del> FORCE .010<del>[</del> AXIAL .005 -.005 -.010 -.015E 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

PAGE 1183

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )



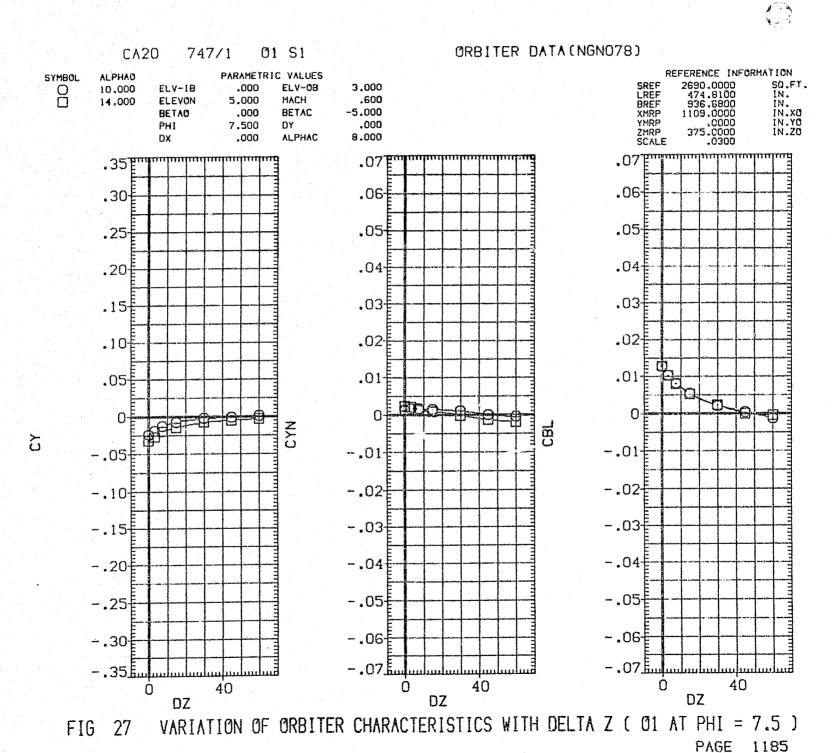


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1186

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1188

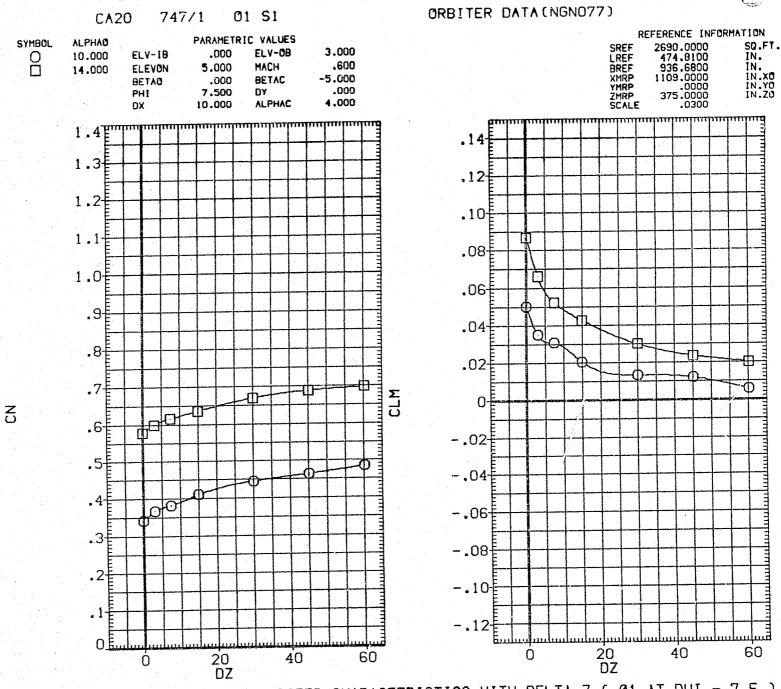


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1189

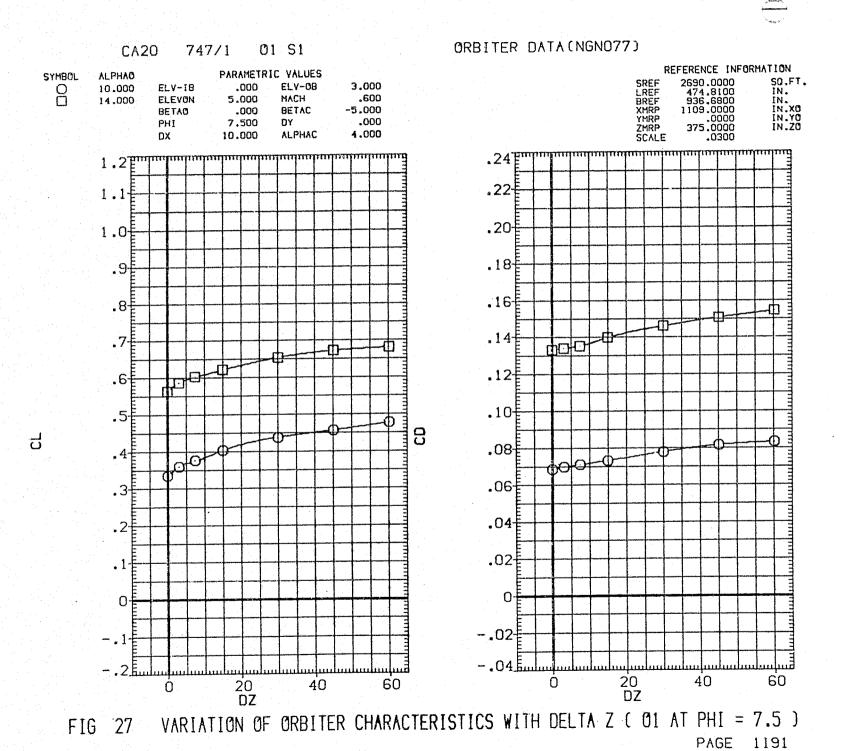
ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION. DZ. FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1190

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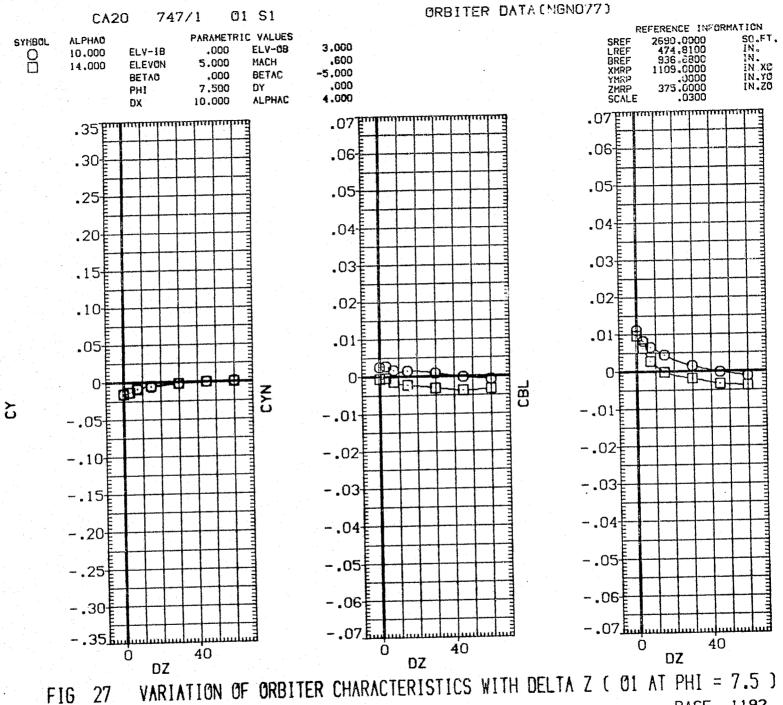


FIG 1192 PAGE

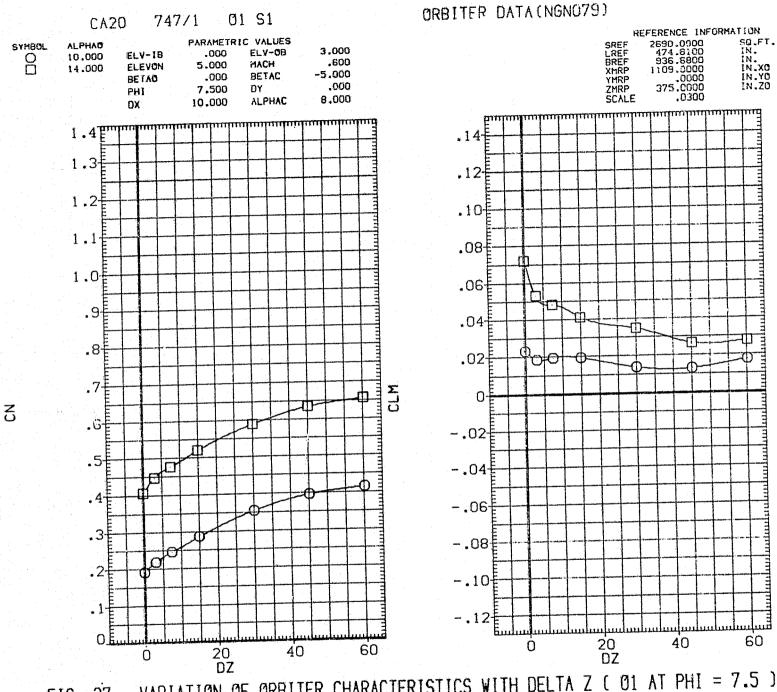
D/S (077 - 010)(VGN077) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES 2690.0600 474.8100 SO.FT. SYMBOL ALPHAO SREF LREF BREF XMRP -5.000 4,000 BETAC 0 10.000 ALPHAC 936.6800 1109.0000 .0000 IN. .000 ELV-08 3.000 14.000 ELV-IB IN.XO IN.YO IN.ZO .600 ELEVON 5,000 MACH YMRP 10,000 7.500 DX ZMRP SCALE 375.0000 PHI .000 BETAO .000 DY .14 Engon .12 .8+ .10 .08 .06<del>-</del> .04-.02 φ 0. DCLM DCN .2ŧ -.02 .1E -.04 0 -.06<del>[</del> -.08<del>[</del> - .2--.10<del>[</del> -.12<del>-</del> -.14<u>E</u>... 20 DZ 40 60 Ò 20 DZ 60 40 0 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

PAGE 1193

FIG

D/S (077 - 010)(VGN077) CA20 (747/1 01 S1) - (01 S1) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL **ALPHAO** 2690.0000 474.8100 936.6800 1109.0000 SQ.FT. -5.000 8 4.000 BETAC 10.000 **ALPHAC** IN. LREF ELV-IB .000 ELV-0B 3.000 IN. IN.XO IN.YO IN.ZO 14.000 BREF XMRP YMRP .600 5.000 MACH **ELEVON** 10.000 PHI 7.500 DX 375,0000 ZMRP .000 .000 BETAO ĎΥ SCALE .14E .9-.12= .8<del>-</del> .10 .08-.6<del>-</del> .06 .5<del>[</del> .04 .02 •3<del>-</del> 0 020 DCL -.02 .1E -.04-0--.06 -.08 -.10+ -.3<del>[</del> -.12<del>-</del> 20 DZ 60 Ó 40 20 **DZ** 60 40 Ò

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1195



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) 27 FIG PAGE 1196

and the state of t

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1198

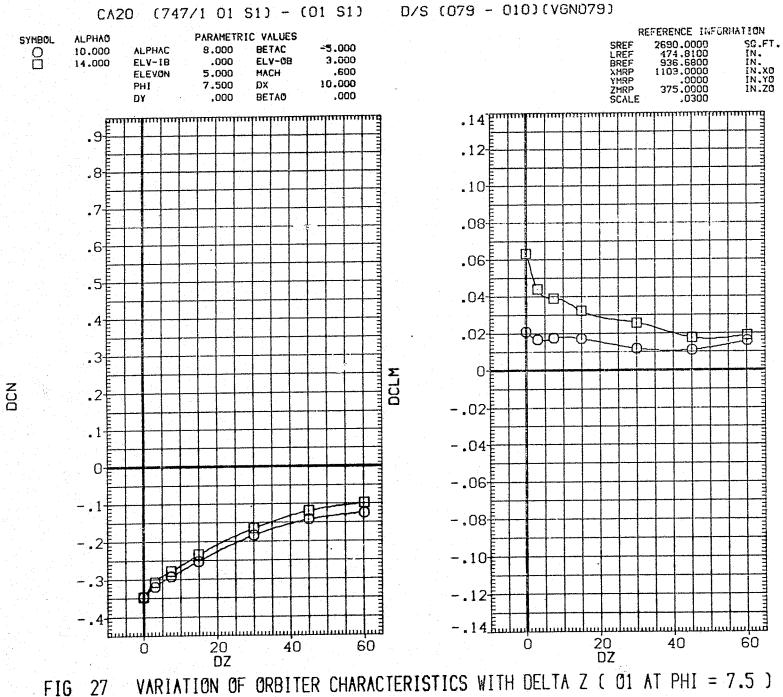


FIG PAGE 1200

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1201

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1202



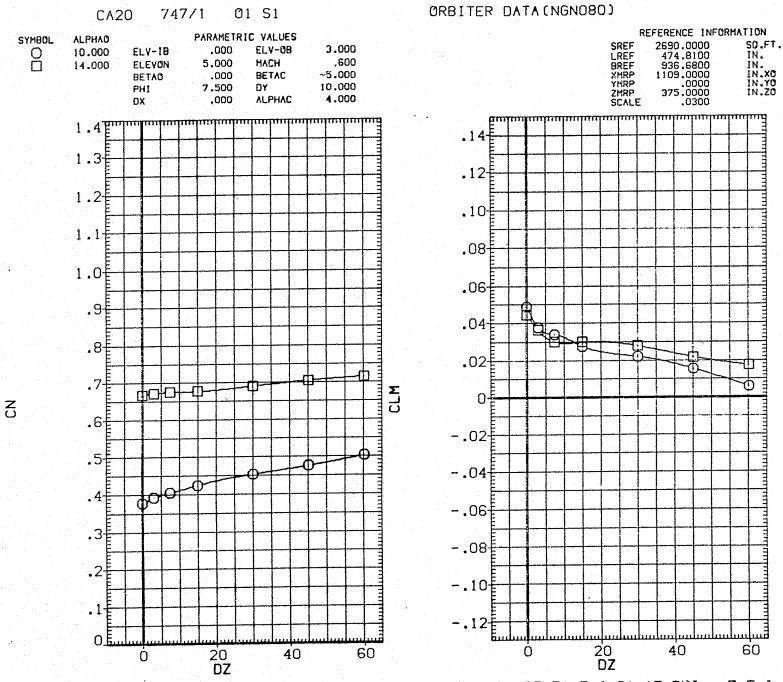


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1203

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

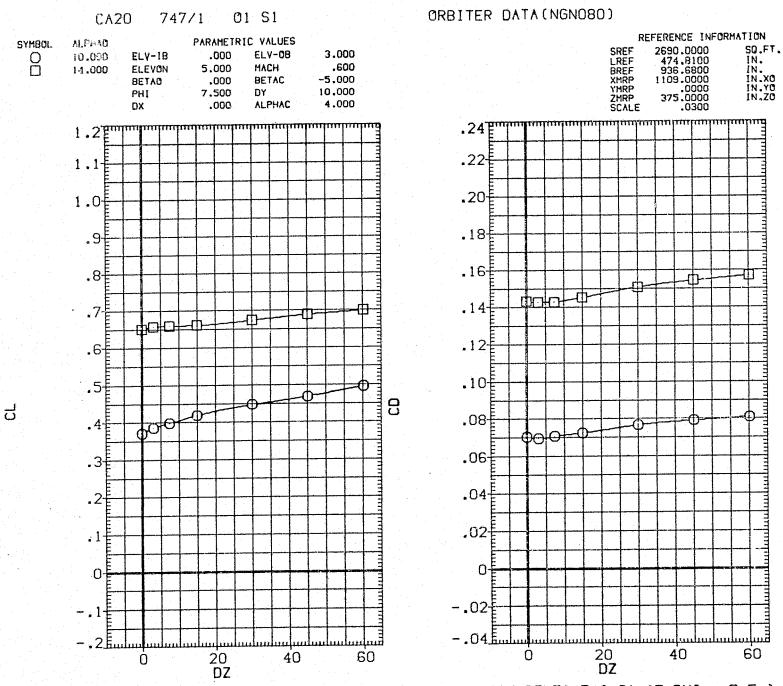


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1205

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1206

DZ

0

40

- **.**35 Lu

0

DZ

40

-.07<u>E</u>...

0

DZ

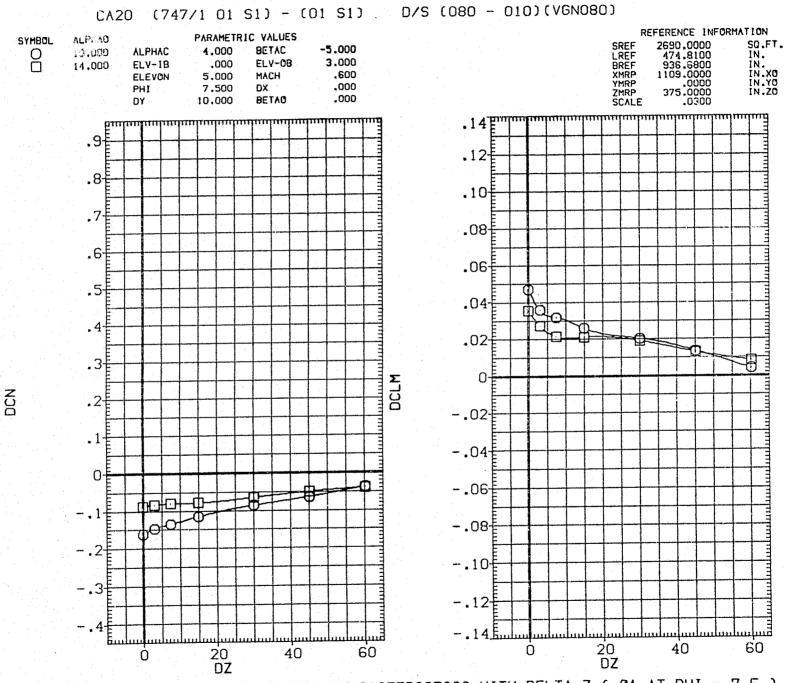


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1207

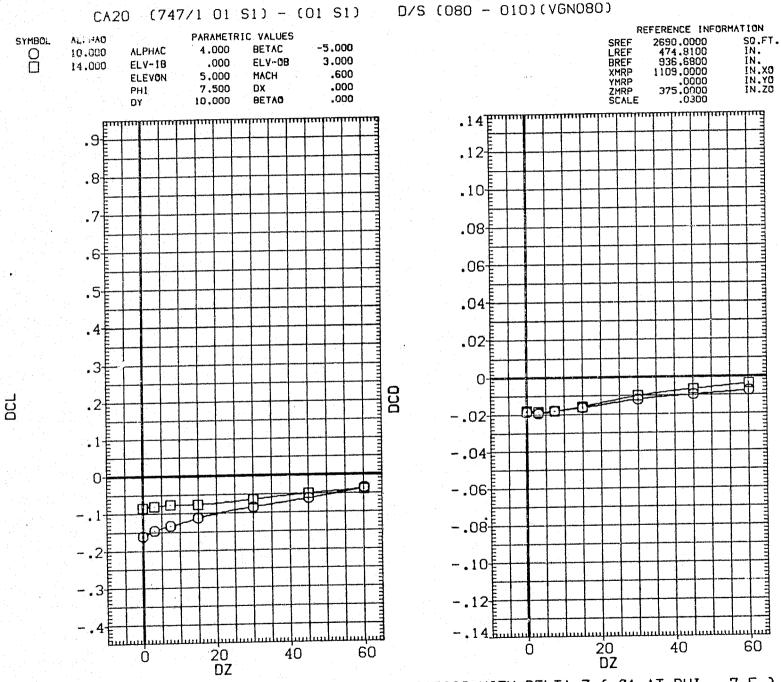


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1209

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

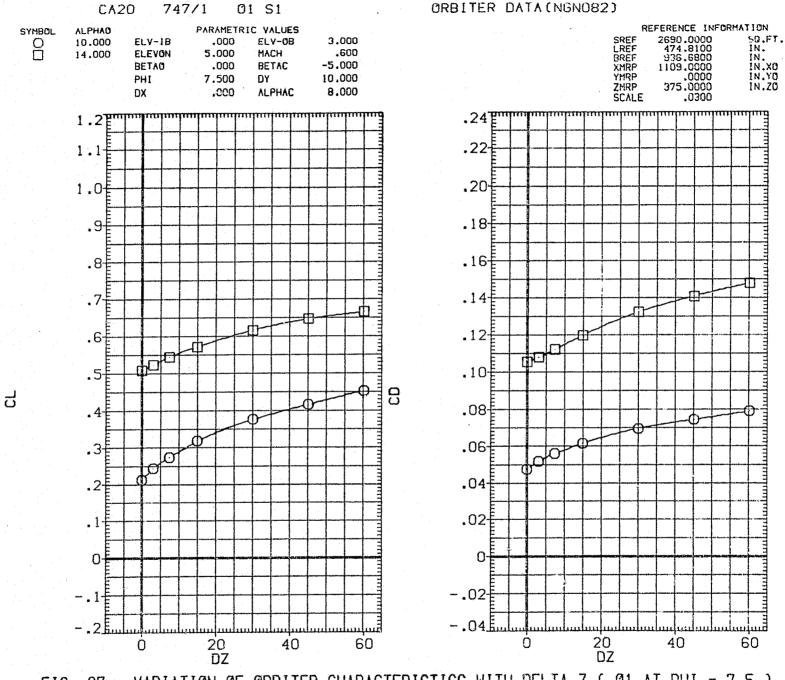
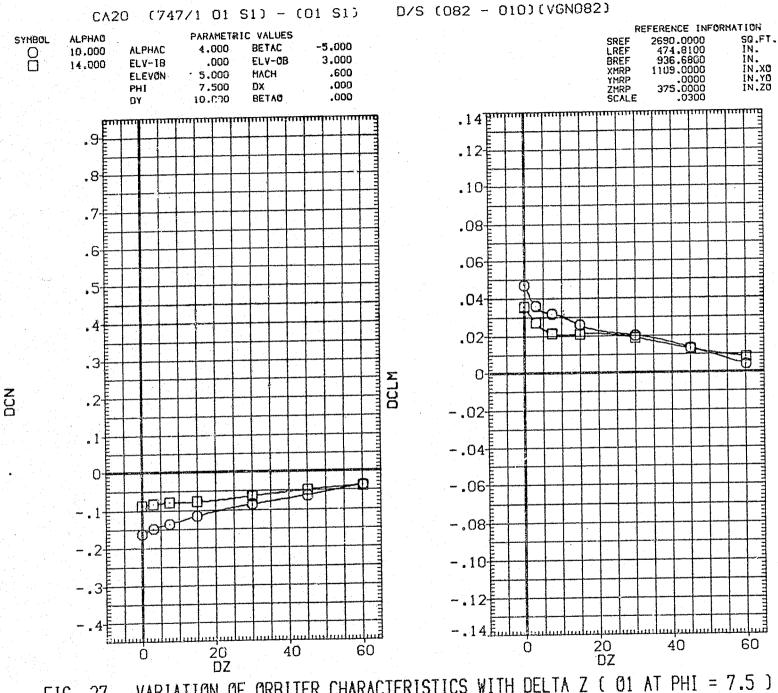
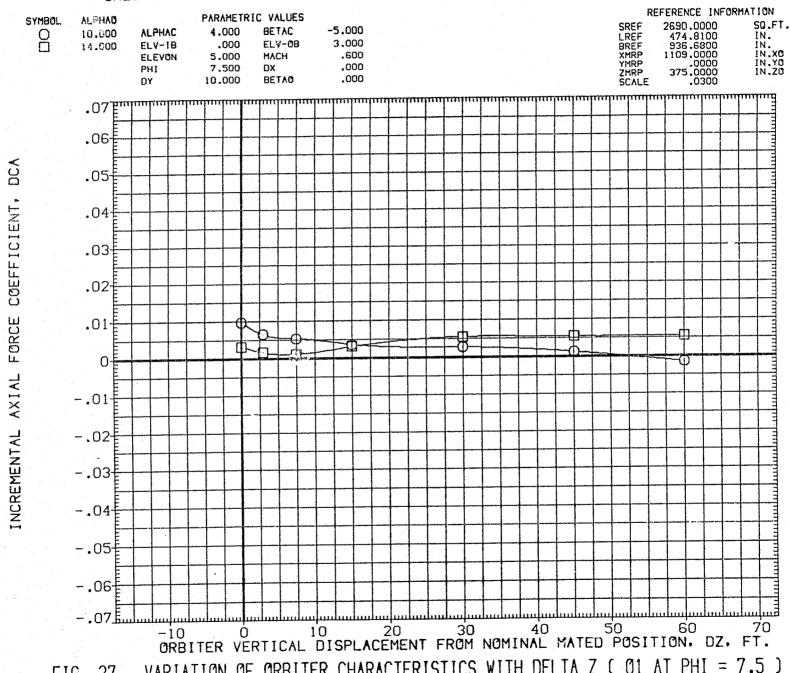


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1212

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1213



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) 27 FIG 1214 PAGE



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG PAGE 1215

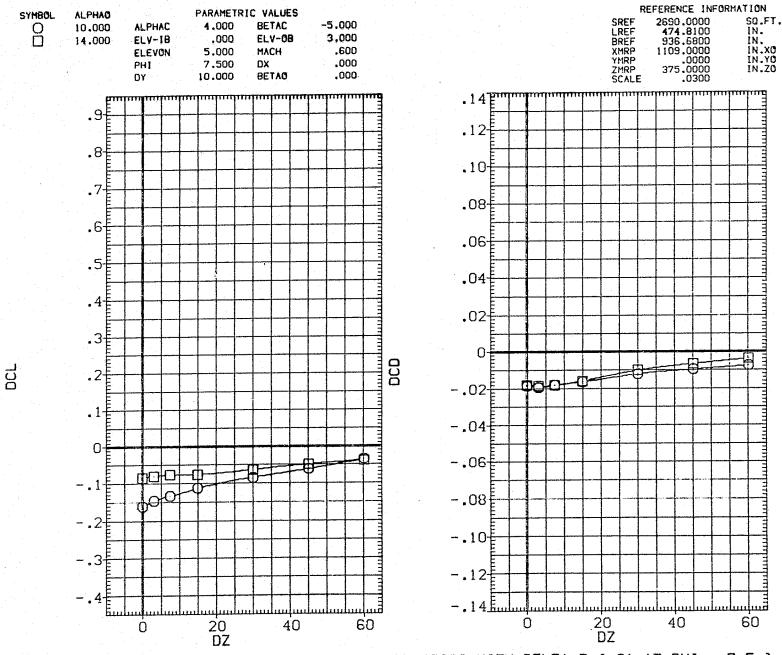


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1216

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1217

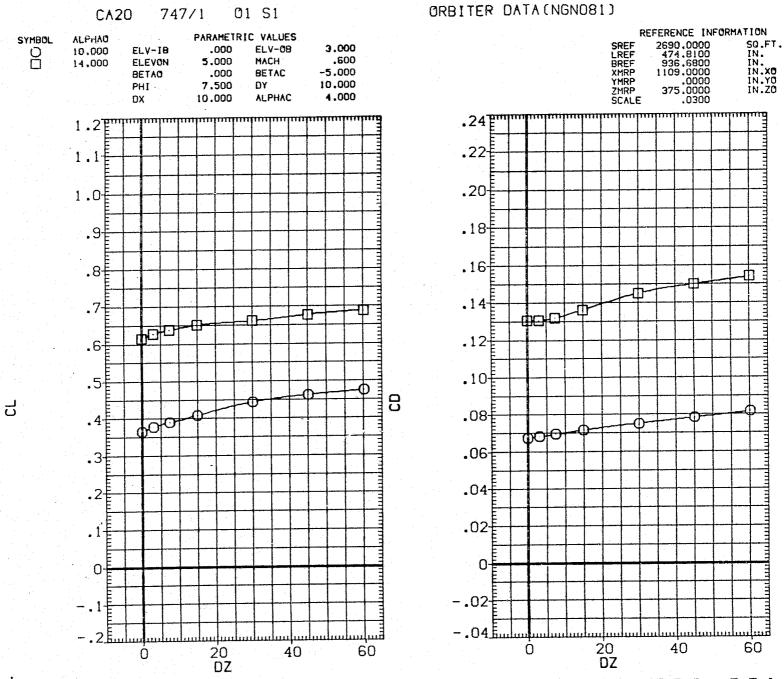


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1219

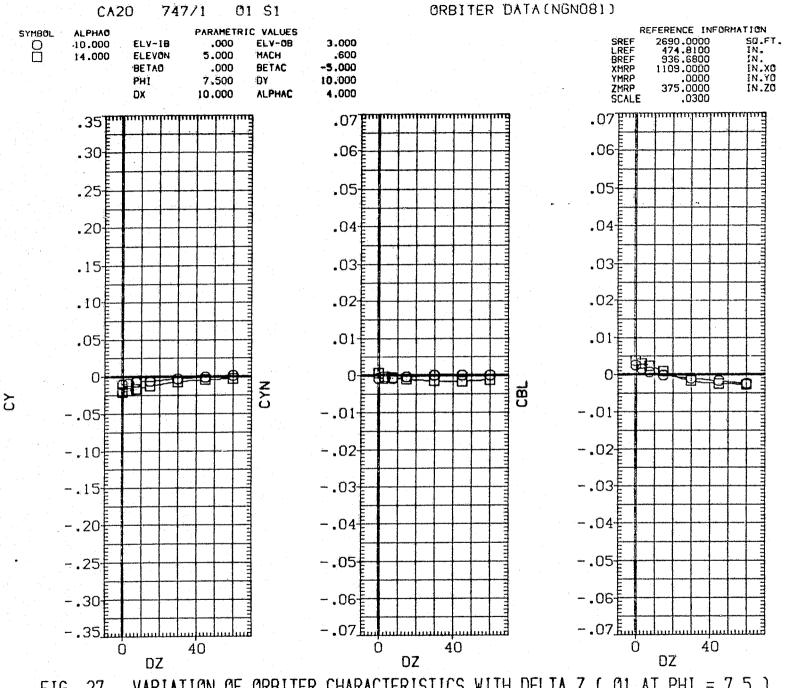


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

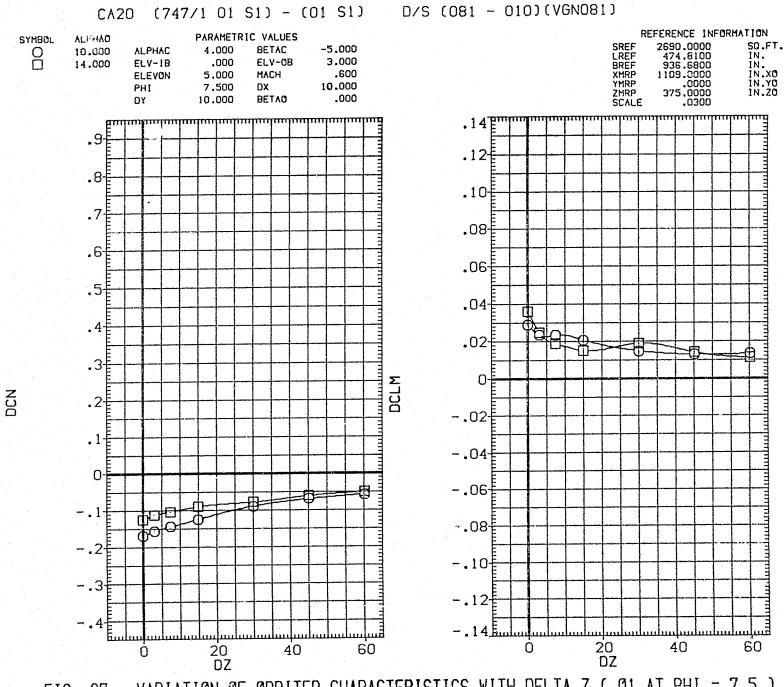


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1221

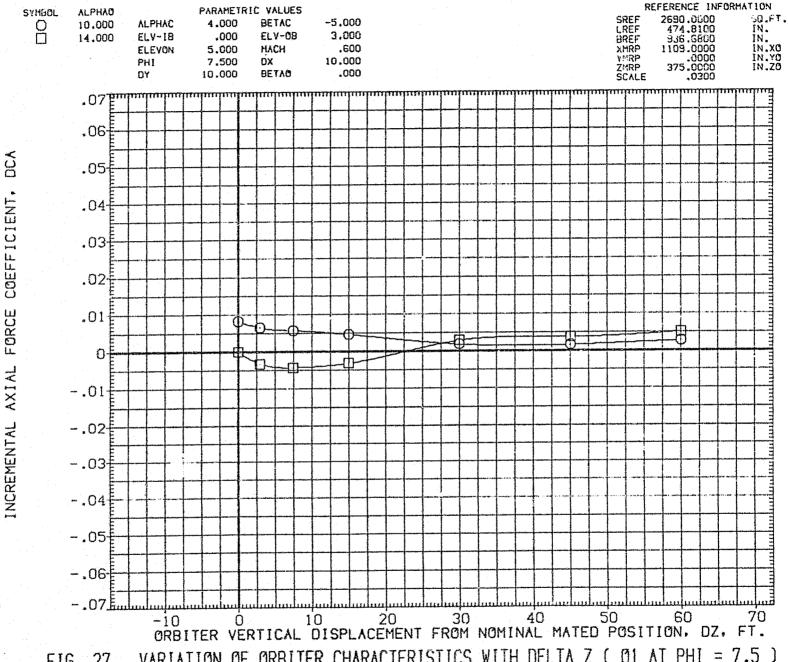


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1222



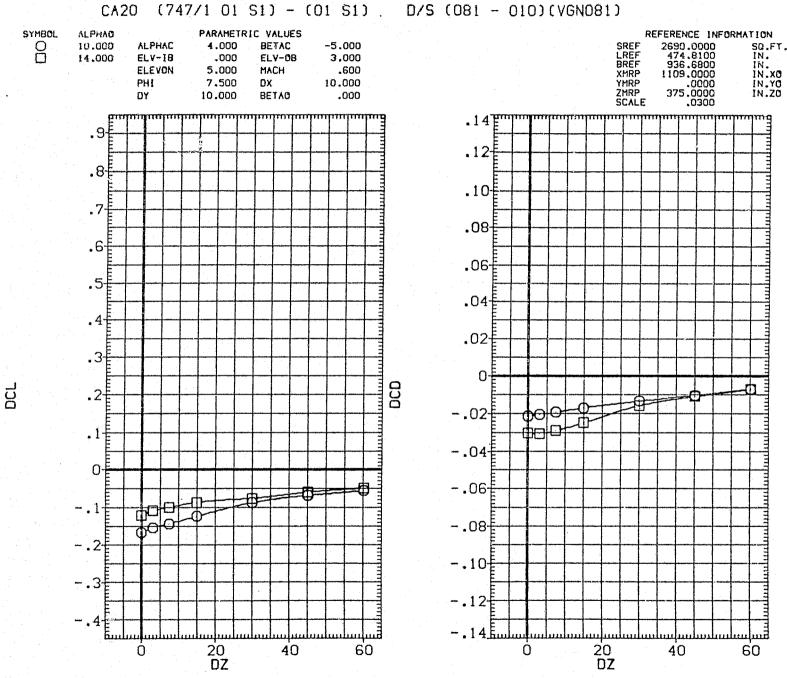


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1223

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1224

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1226

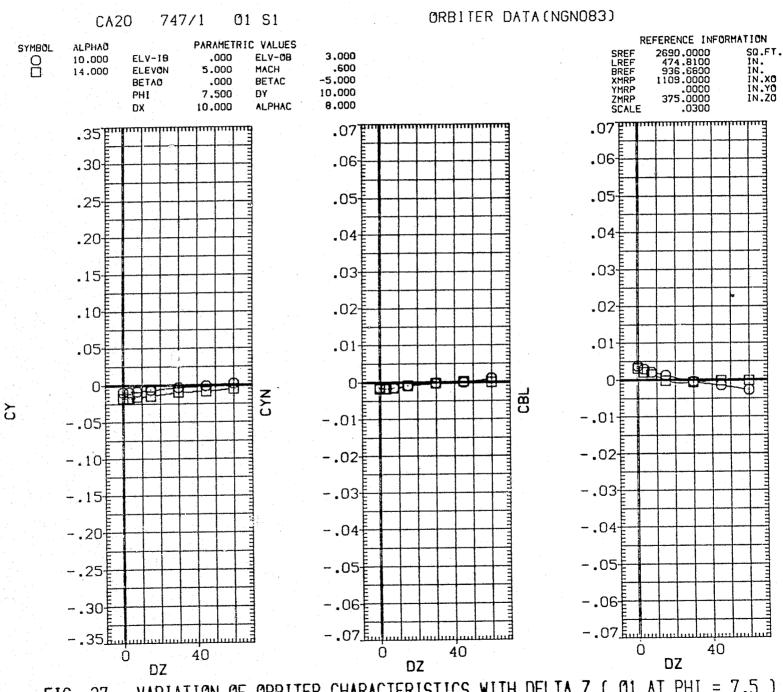


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1227

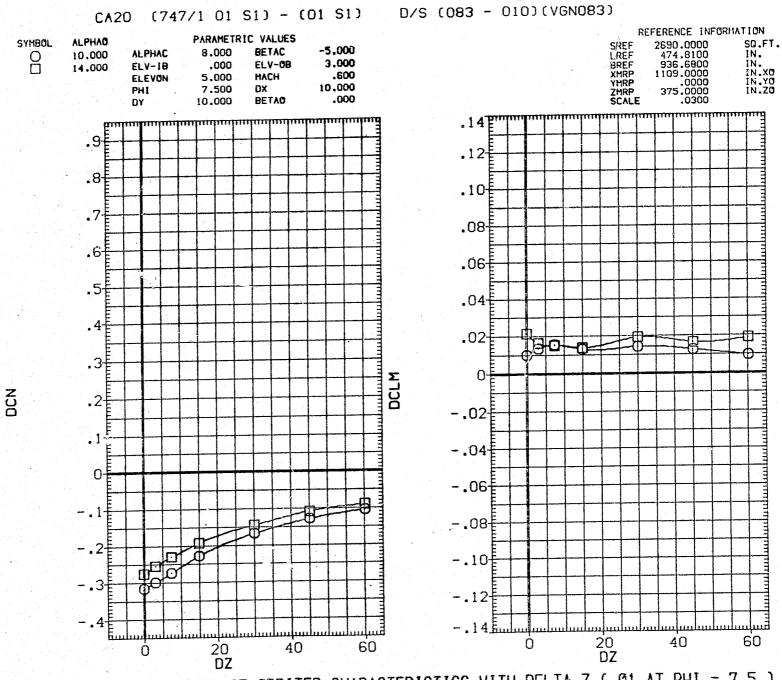


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1228

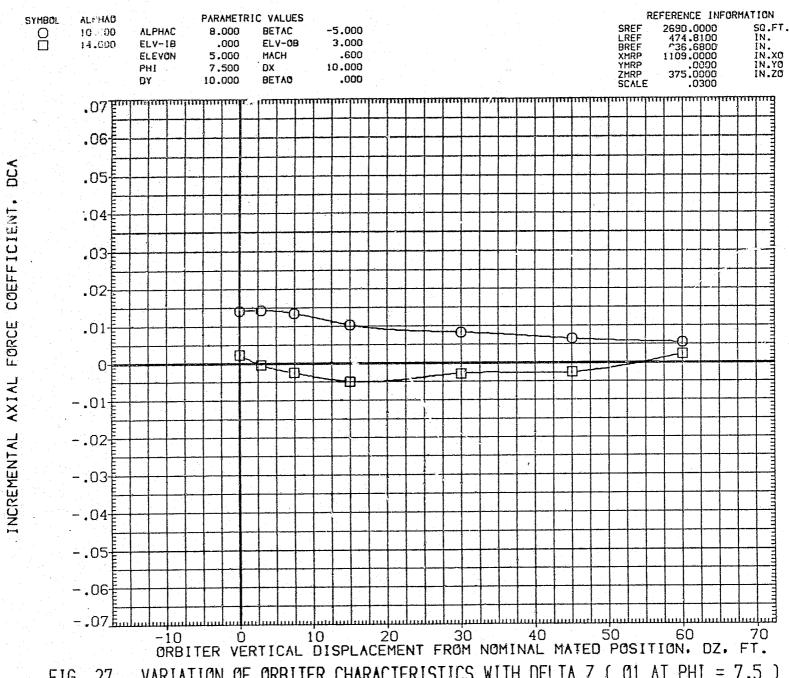


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1229

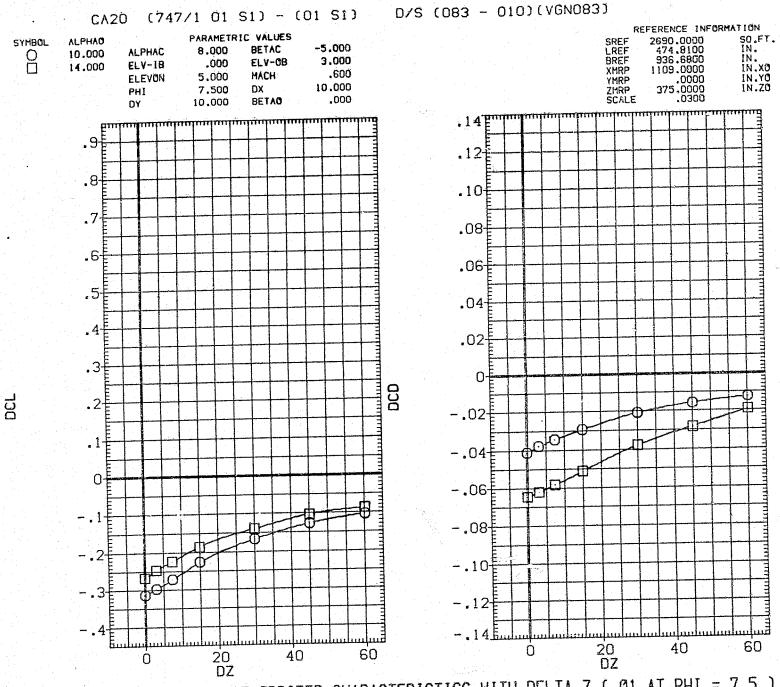


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1230

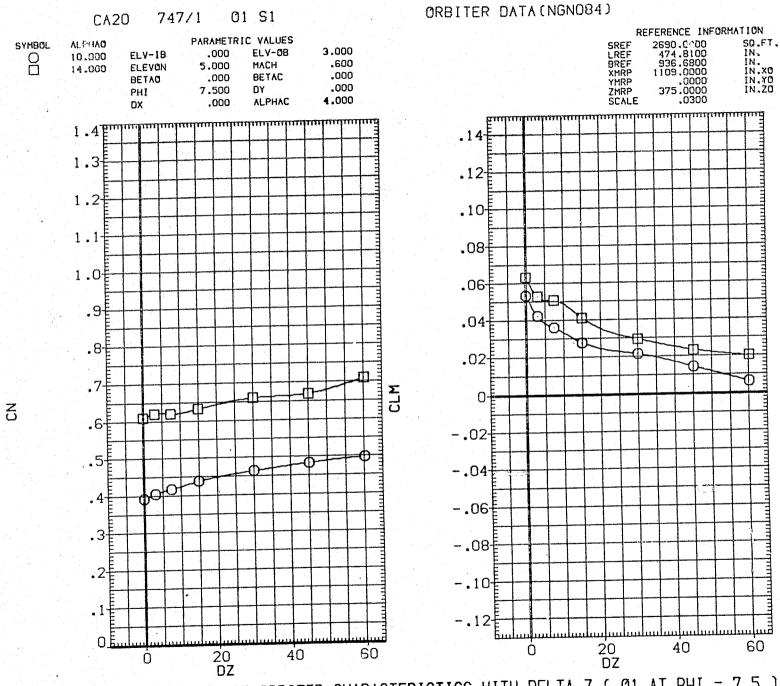


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1231



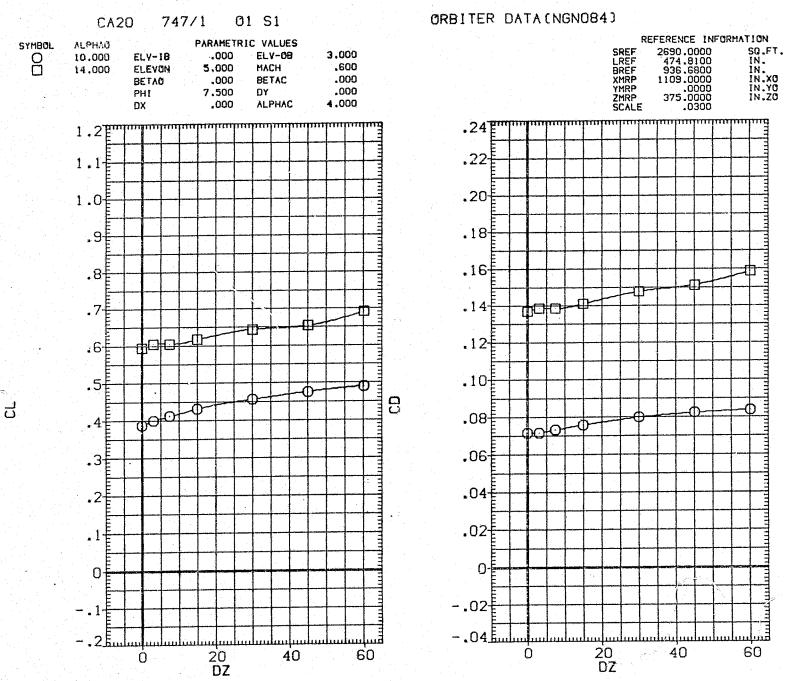
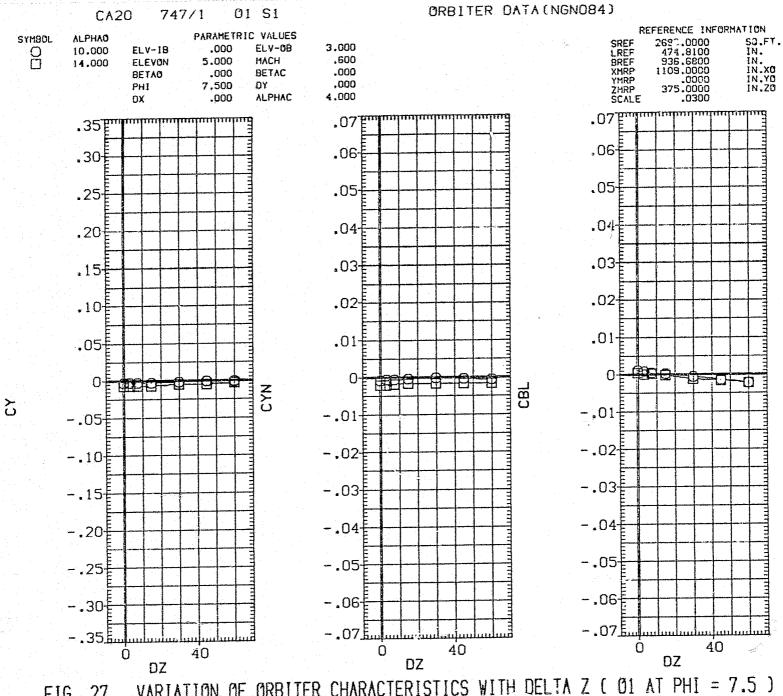


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1233



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) 27 FIG PAGE 1234

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1235

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT.

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

PAGE 1236

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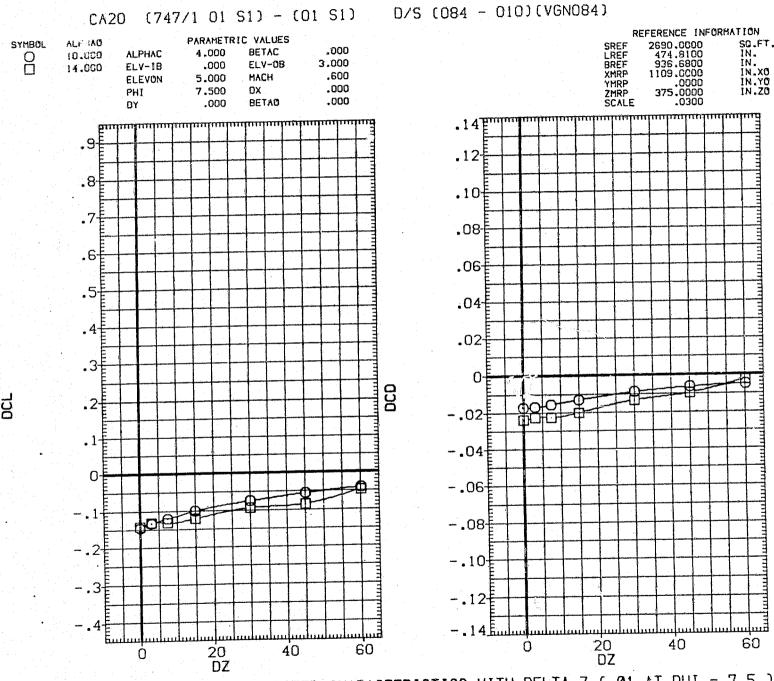


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1237

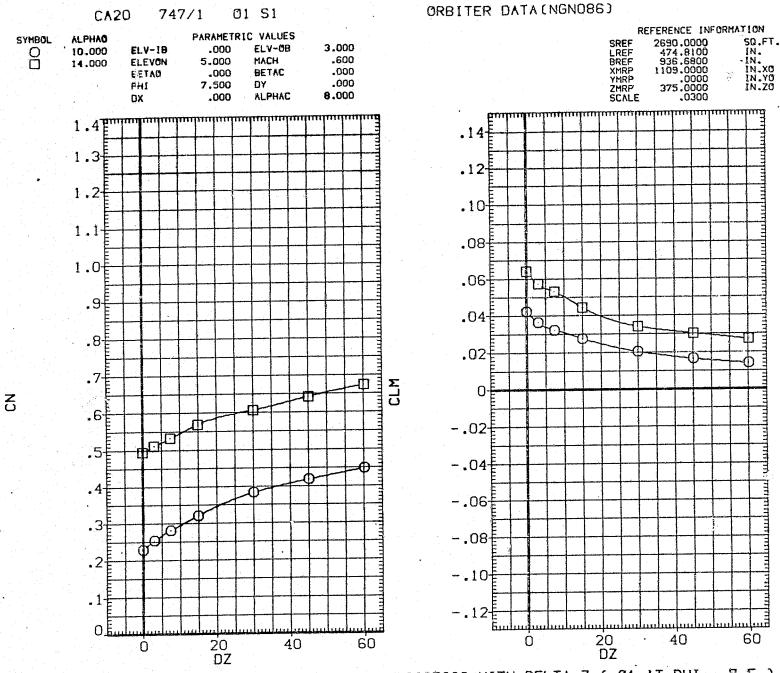


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1238

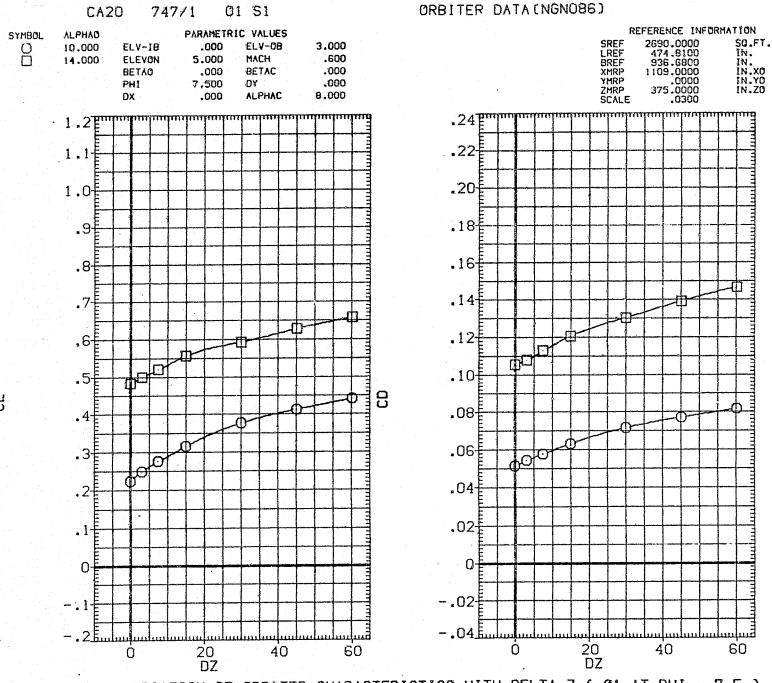
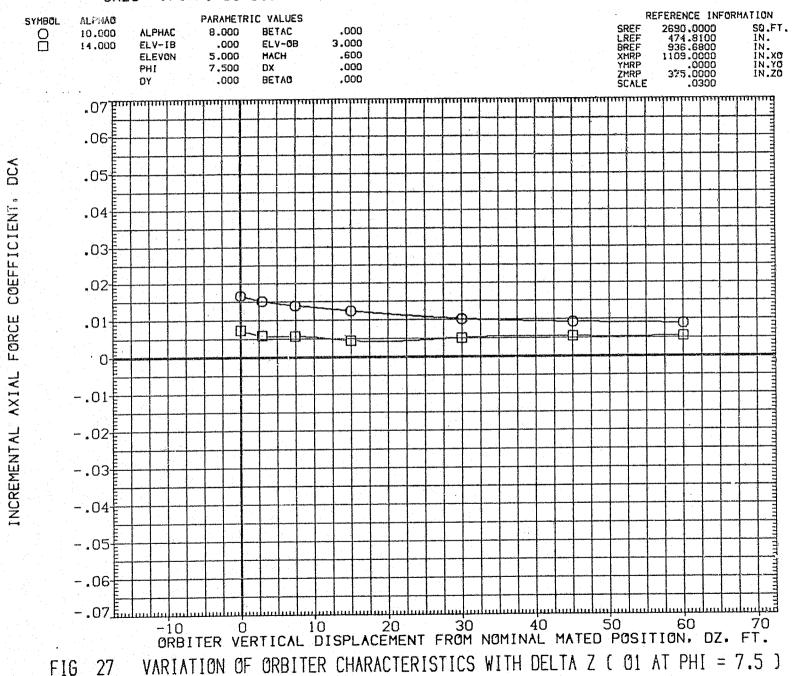


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1240

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1242

CA20 (747/1 01 S1) - (01 S1) . D/S (086 - 010)(VGN086)



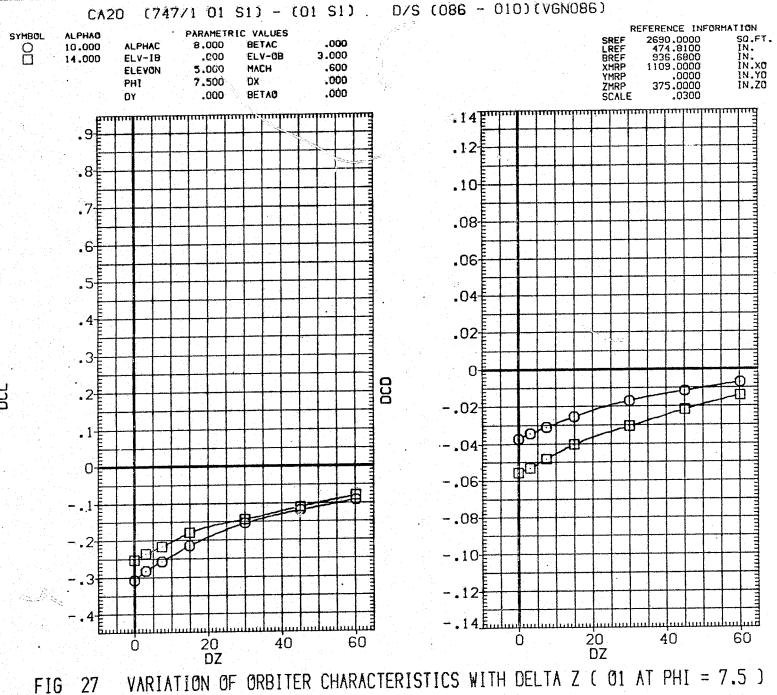
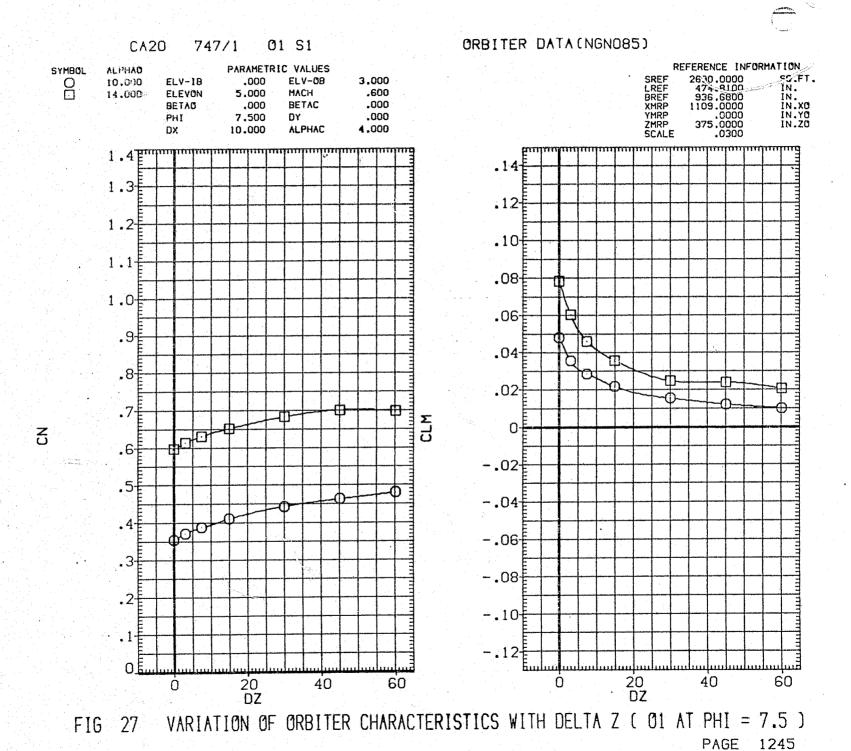
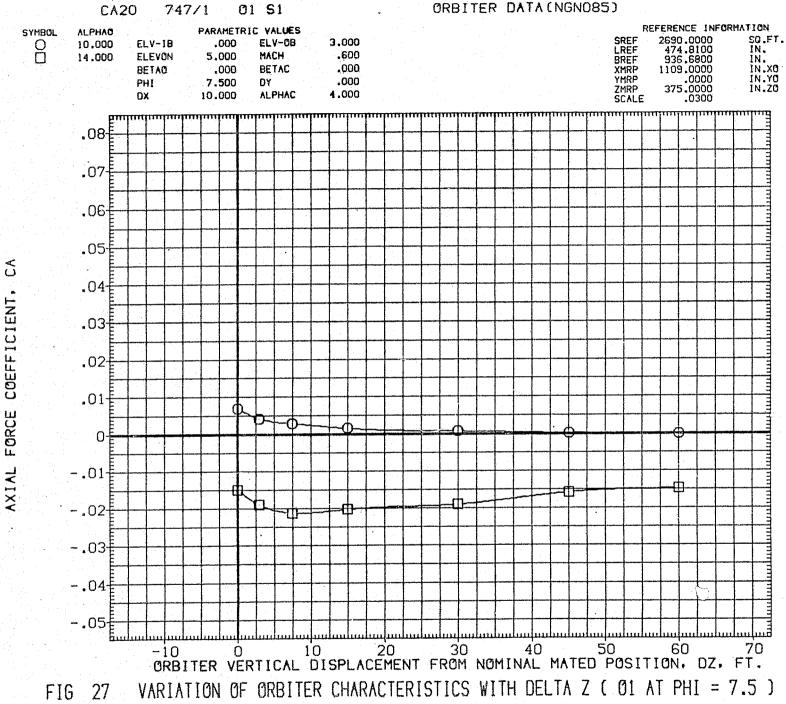


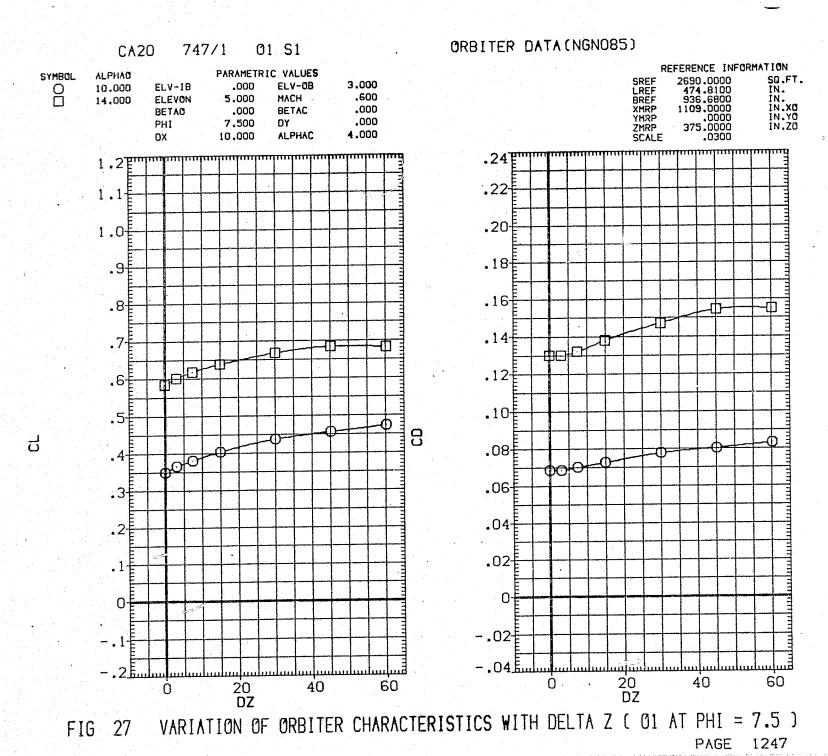
FIG PAGE 1244





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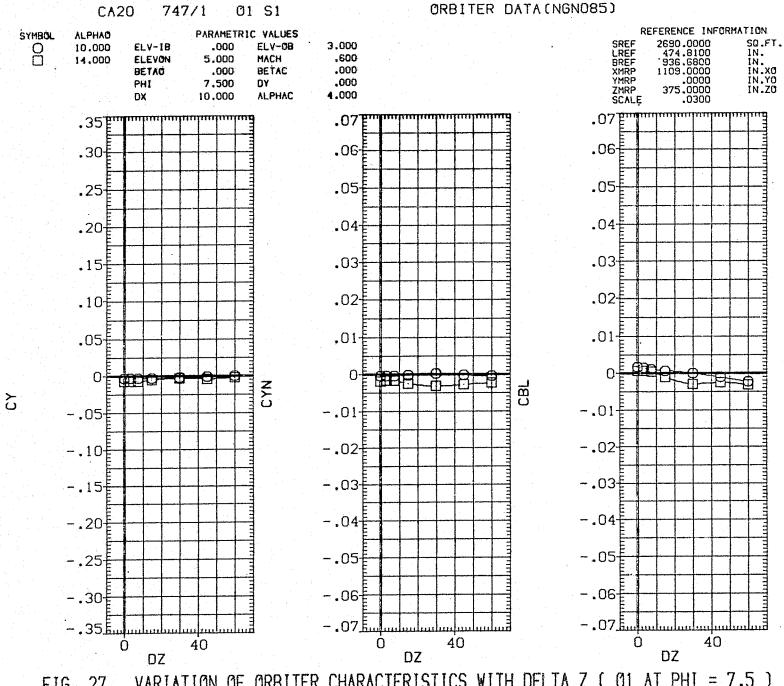


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1248

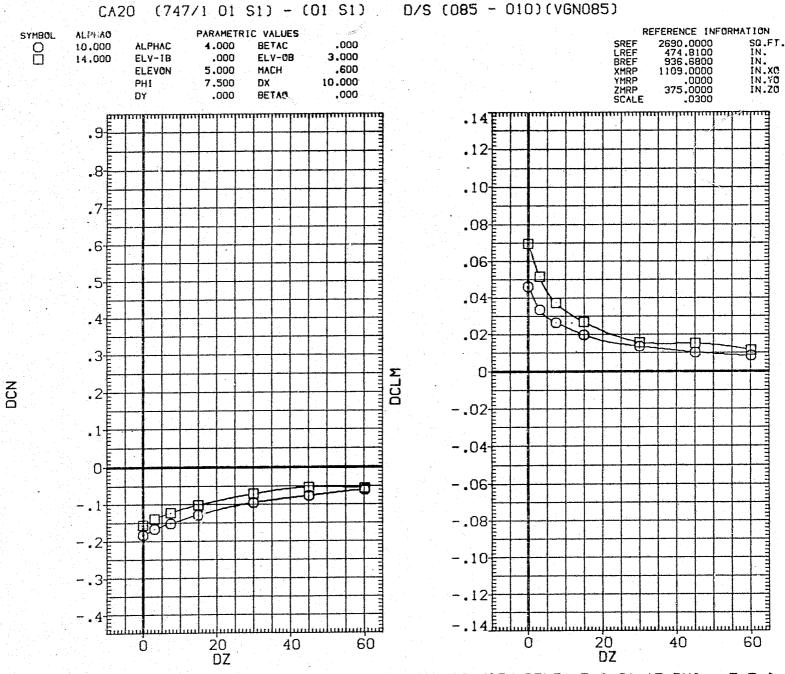


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1249

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1250

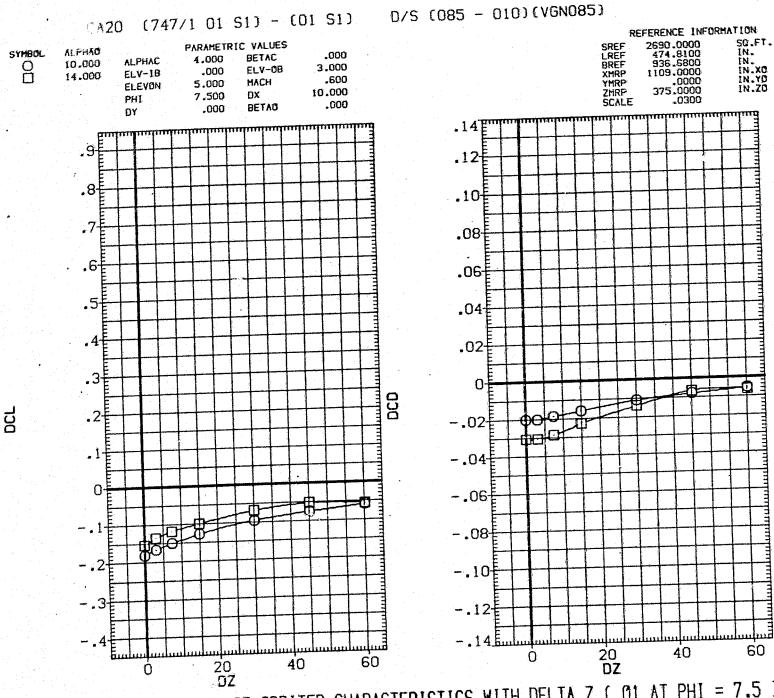
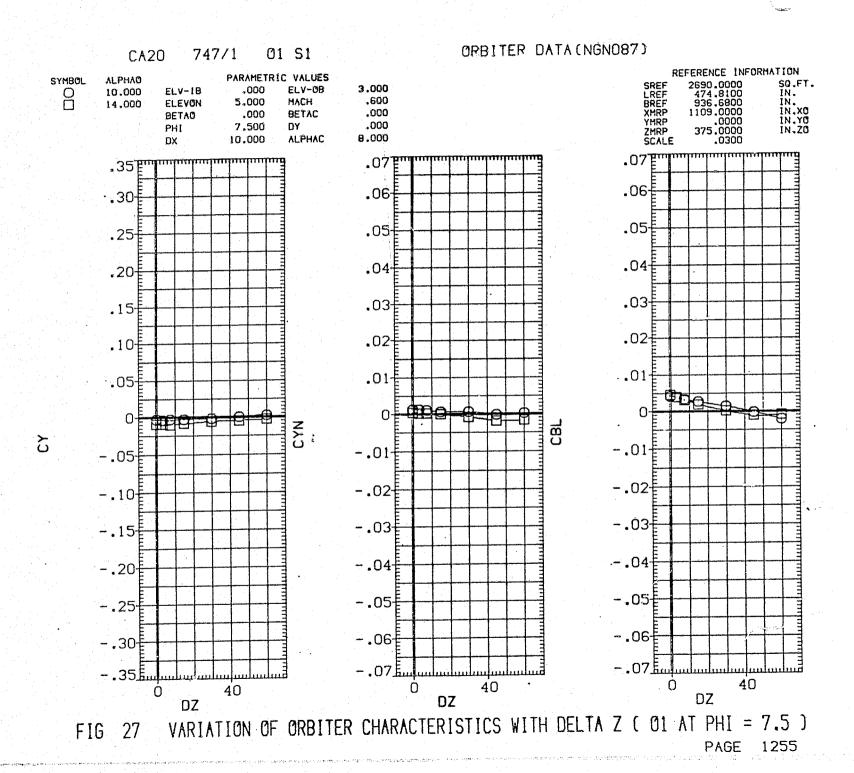


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1251

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1252

PAGE

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1254



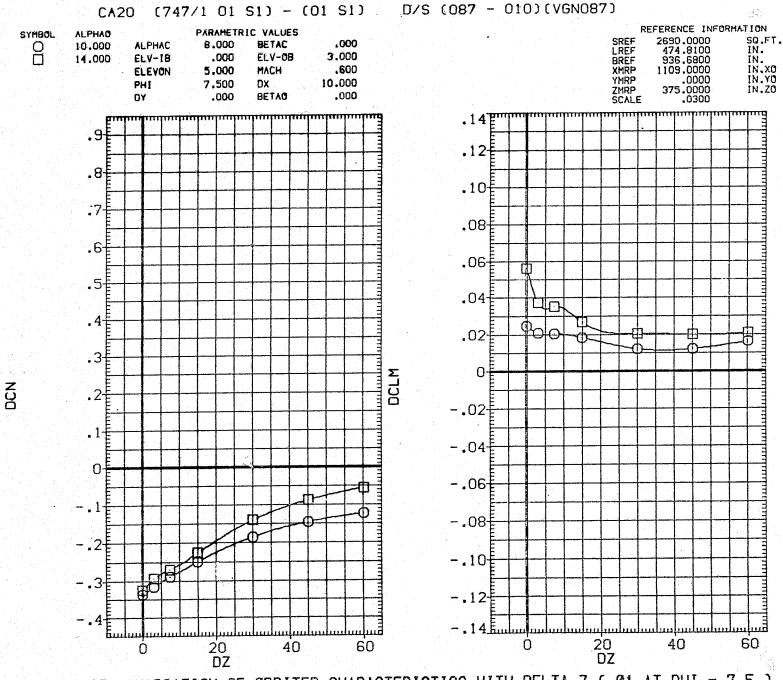


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )



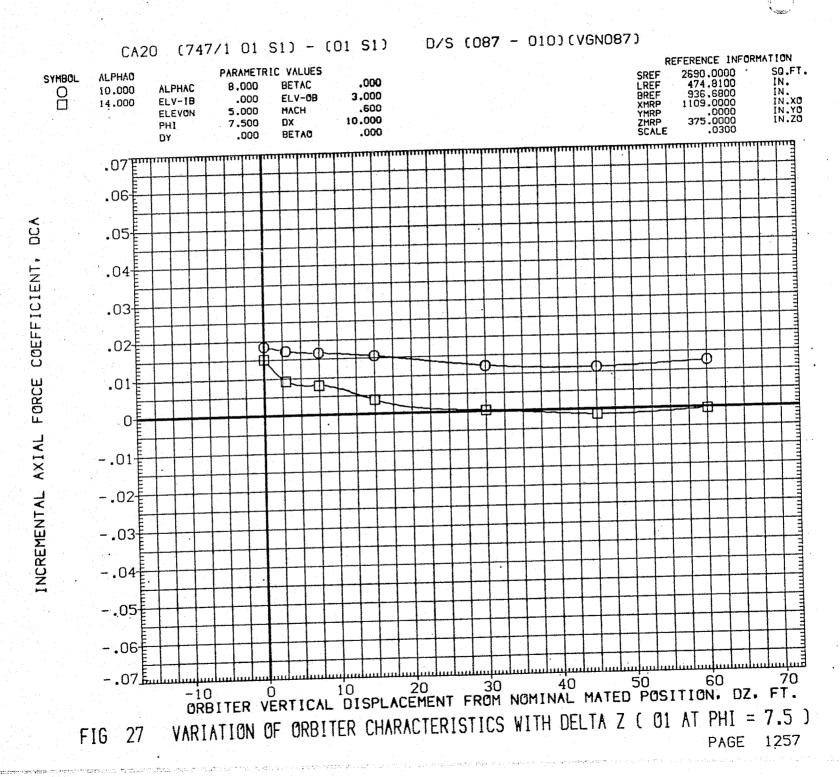
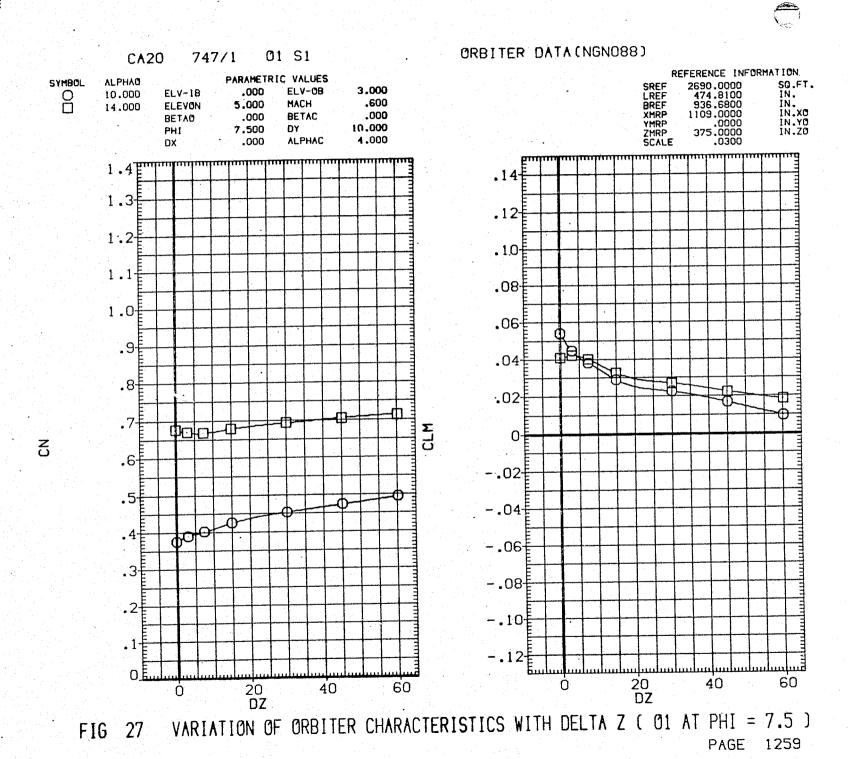
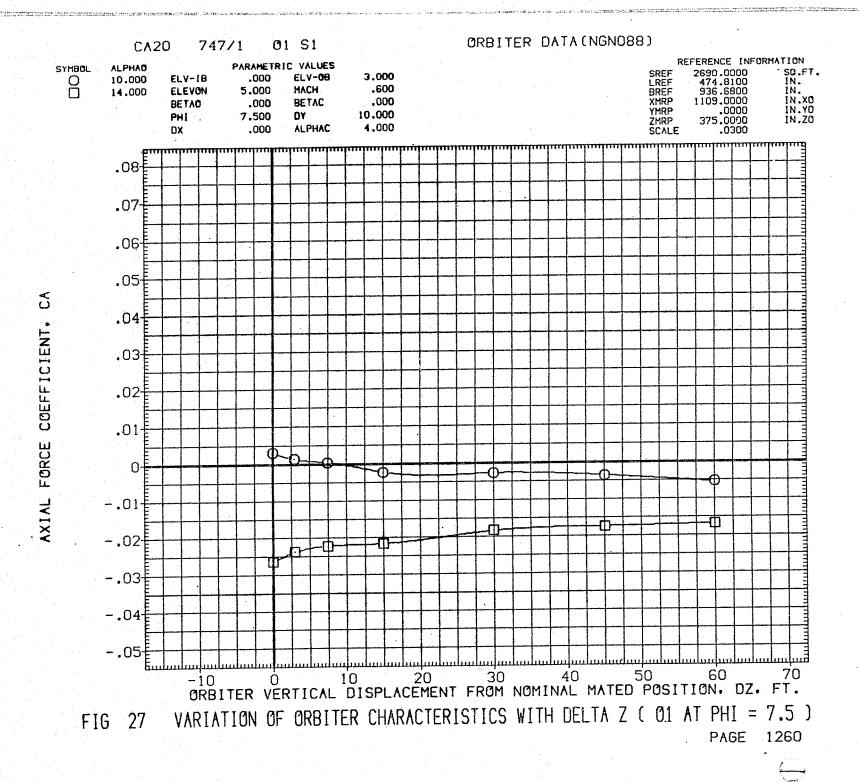
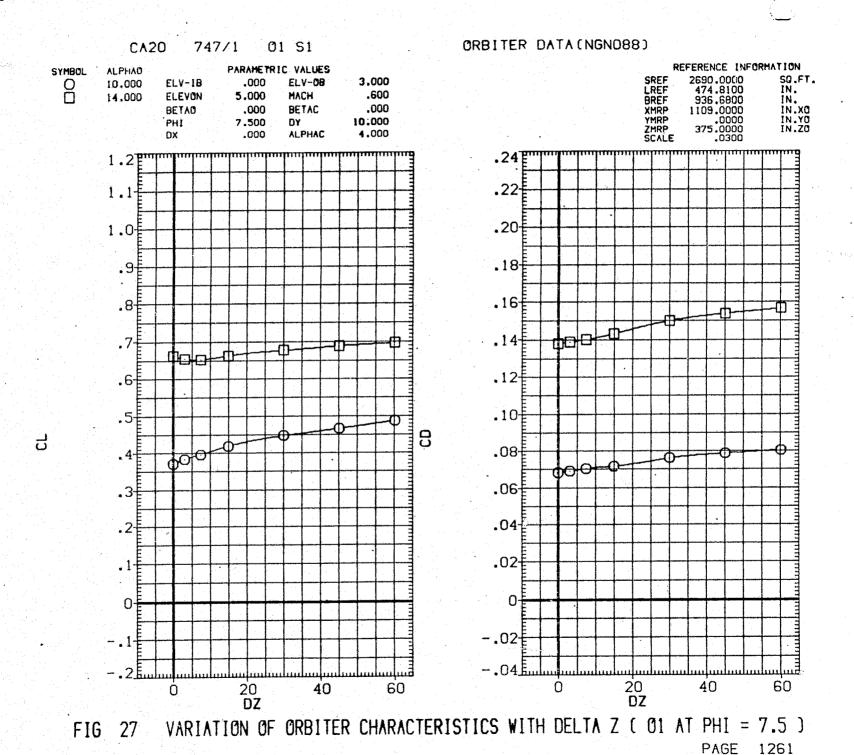
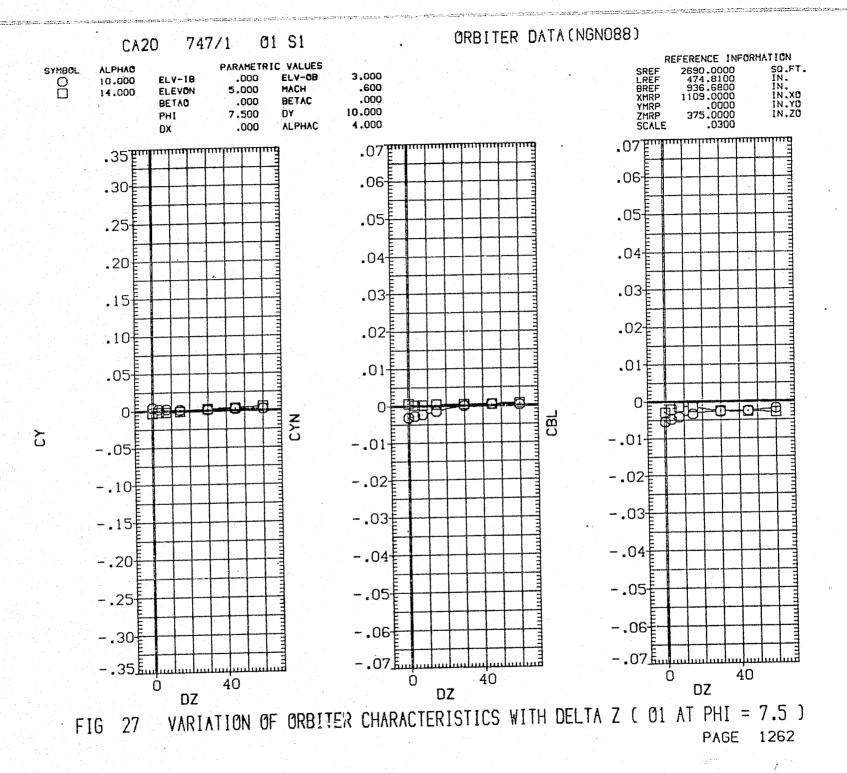


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1258









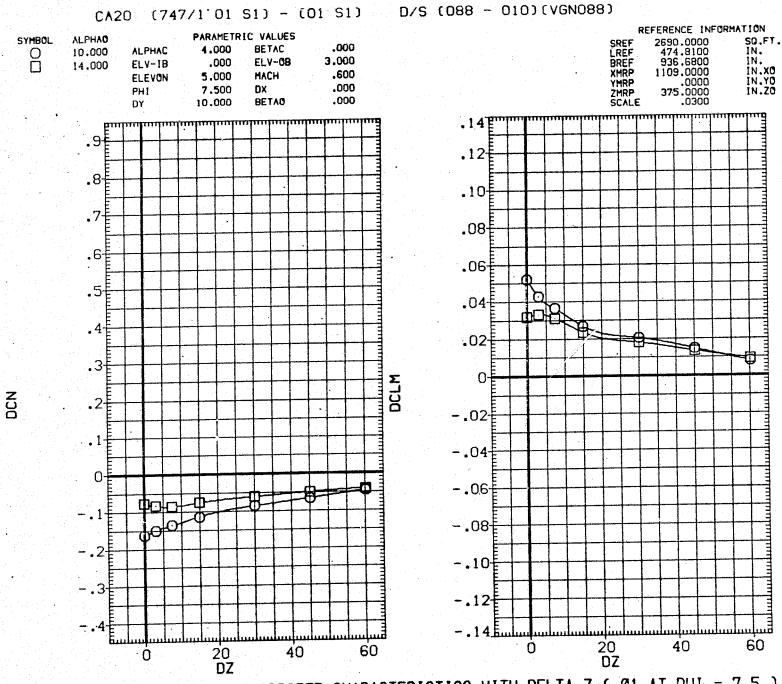
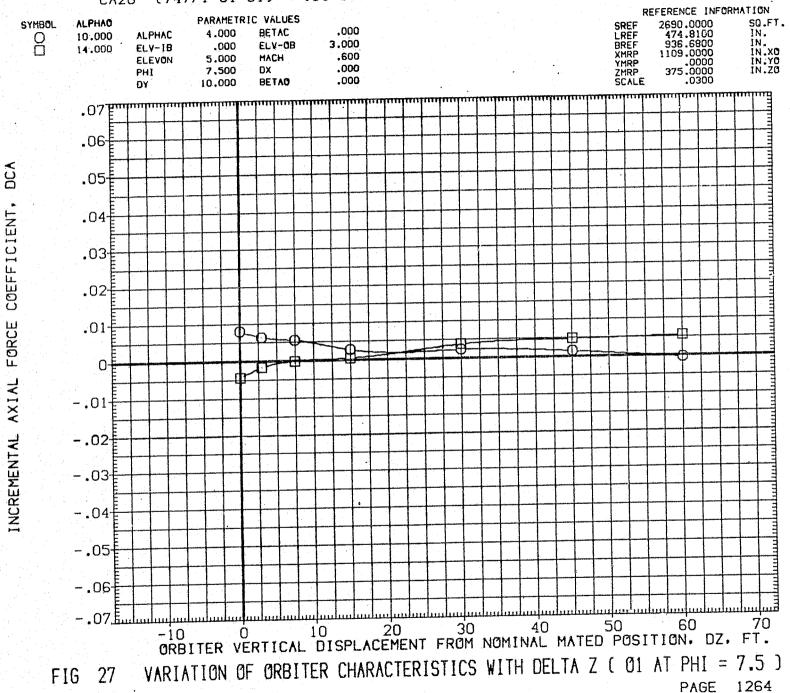


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1263



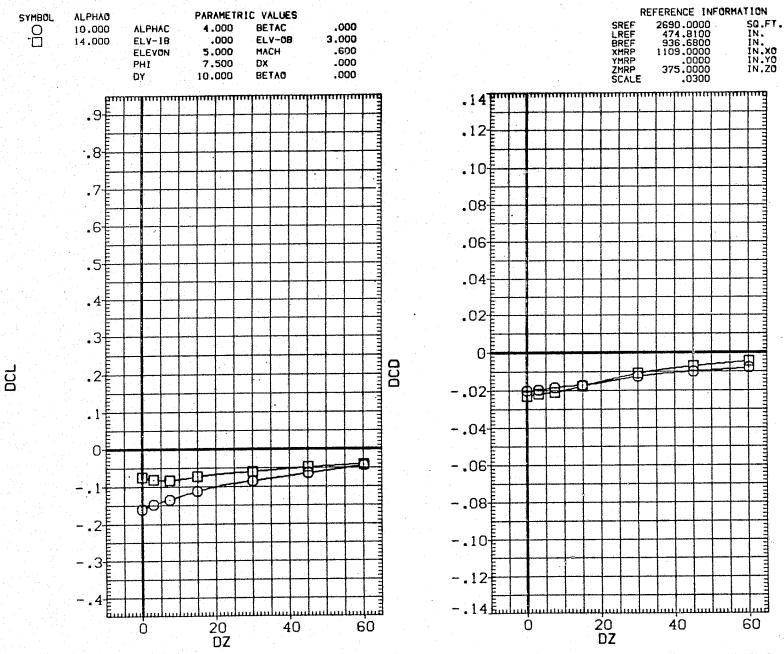
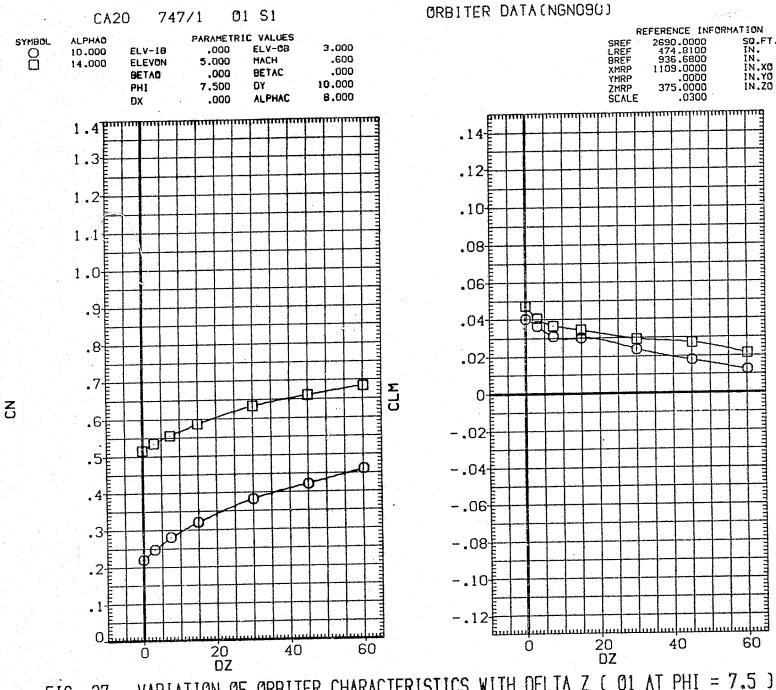


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG PAGE 1266

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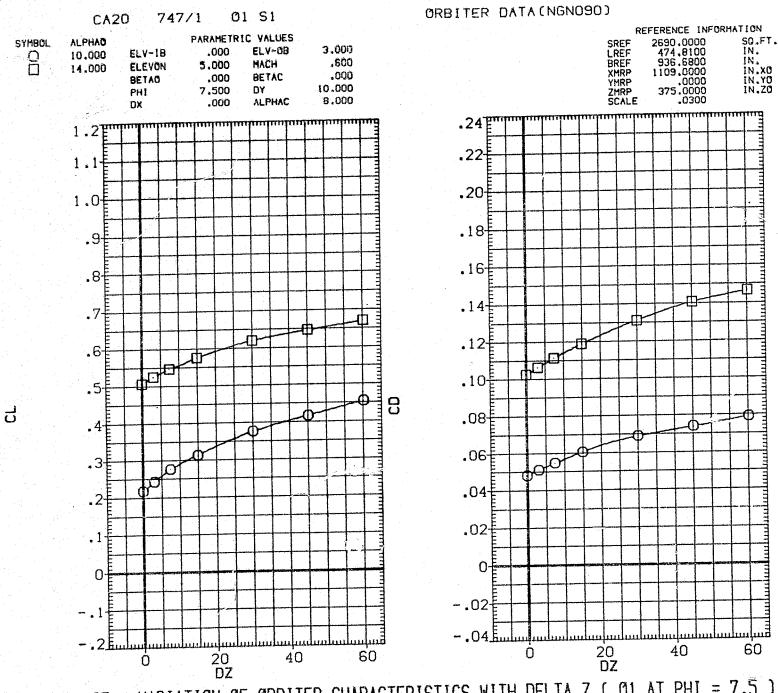


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

ORBITER DATA (NGN090) 01 51 **CA20** 747/1 REFERENCE INFORMATION PARAMETRIC VALUES 2690.0000 474.8100 936.6800 1109.0000 SYMBOL ALPHAO 3.000 ELV-0B .000 IN. IN. ELV-18 0 10.000 LREF ,600 MACH ELEVON 5.000 BREF 14.000 IN.XO XMRP ,000 BETAC .000 BETAO IN.YO 10.000 PHI 7.500 DY ZMRP SCALE 8.000 **ALPHAC** .000 .07F .07E .35E .06 .06<del>-</del> •30€ .05<del>-</del> .05 .25 .04 .04 .20 .03 .03 .15 .02 .02 .10 .01 .01 .05<del>[</del> 0  $\sim$ -.01 -.01 -.05<del>[</del> -.02<del>[</del> -.02 -.10<del>[</del> -.03 -.03 -.15<del>[</del> -.04 -.04 -.20<del>[</del> -.05 -.05 -.25<del>[</del> -.06 -.06 -.30<del>[</del>

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1269

DZ

0

40

-.35 E...

0

DZ

40

- .07.5...

0

DZ

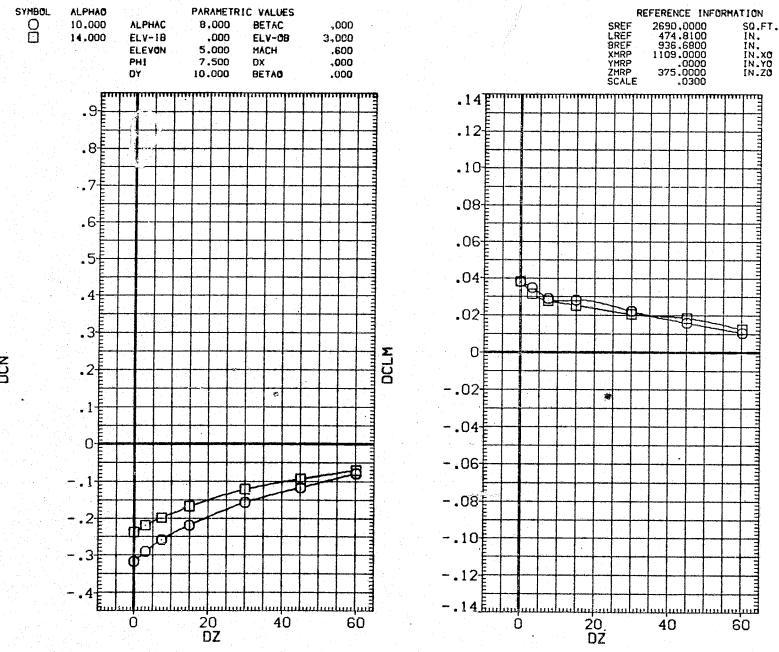
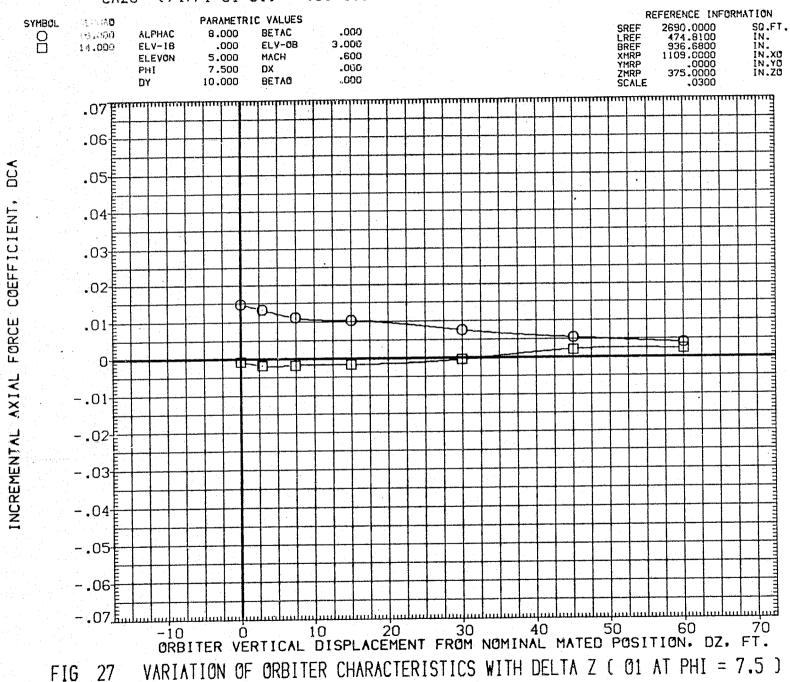


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1270



PAGE

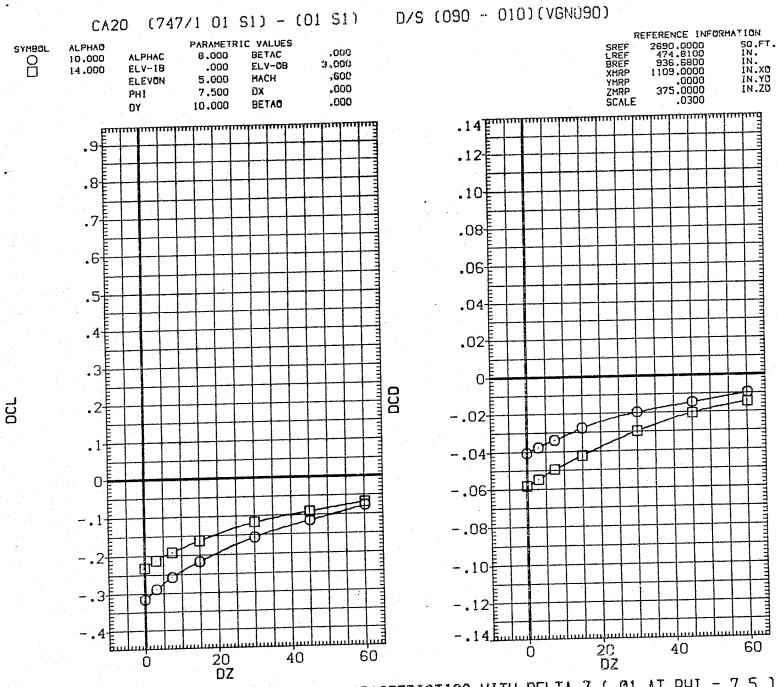


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z [ 01 AT PHI = 7.5 ]
PAGE 1272

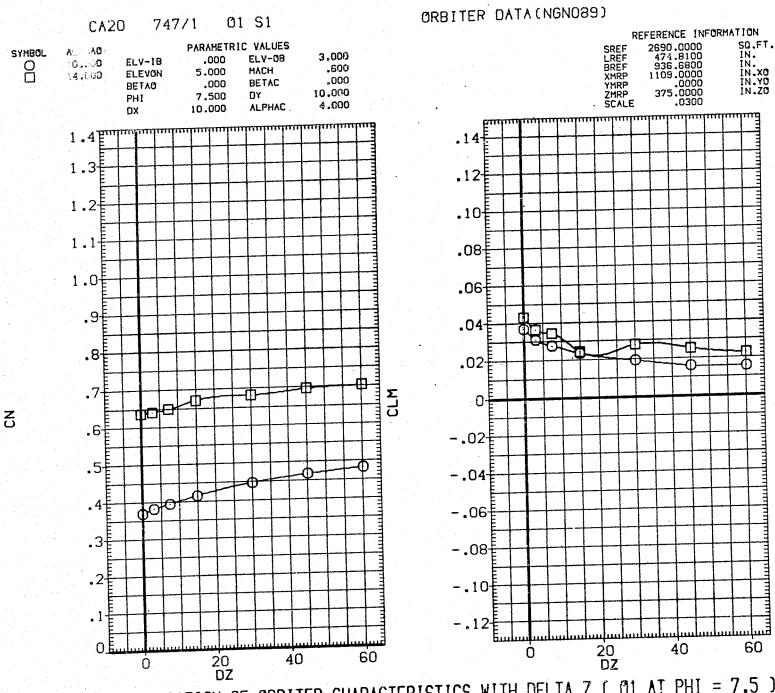


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1273

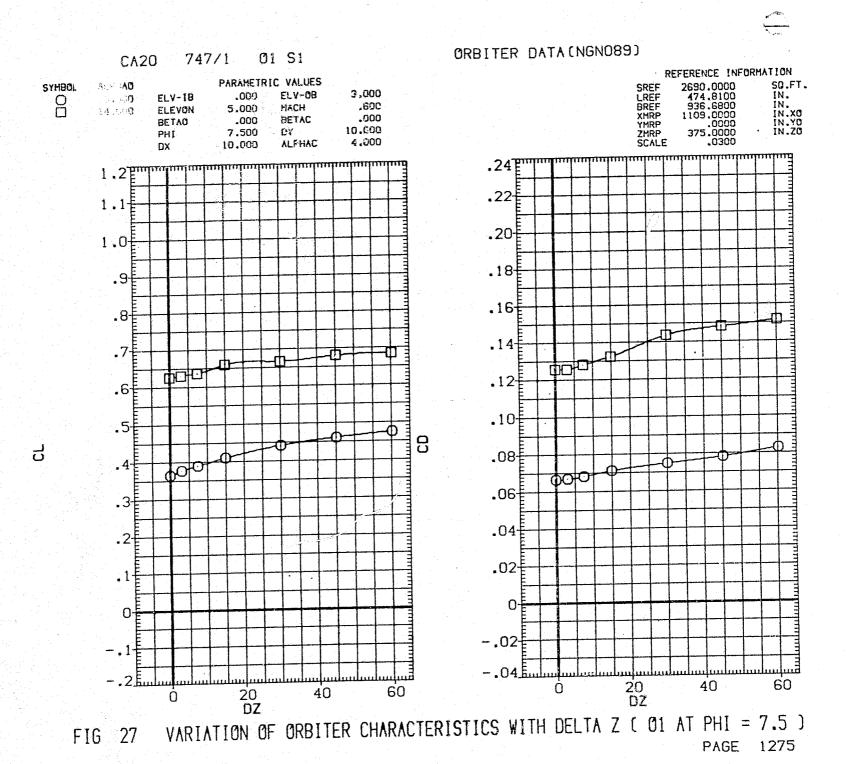


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1276





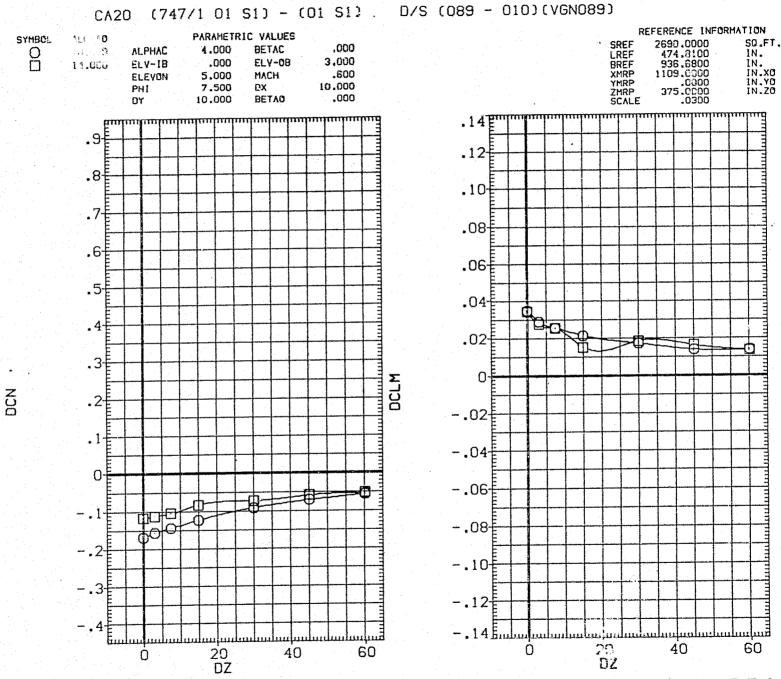


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1277

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1278

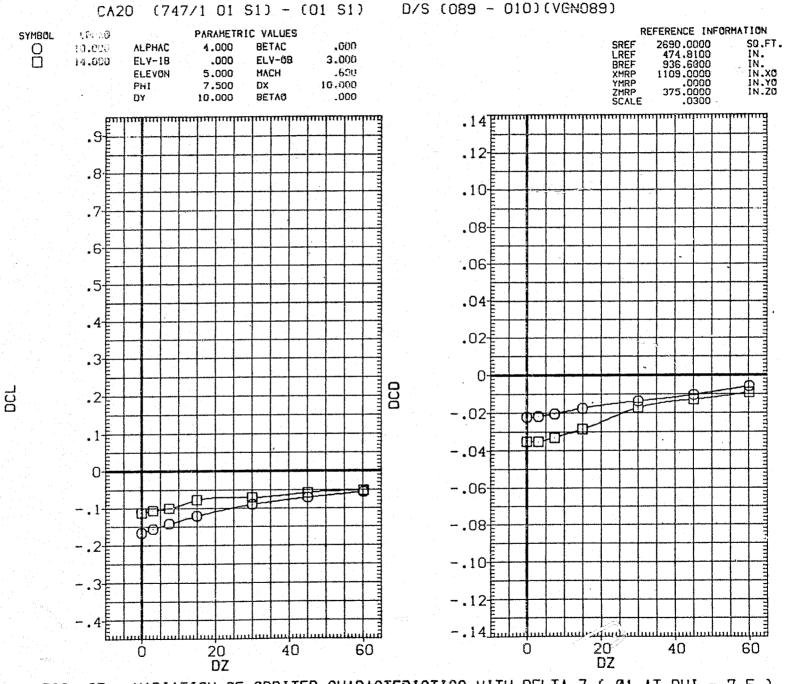


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1279

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1280

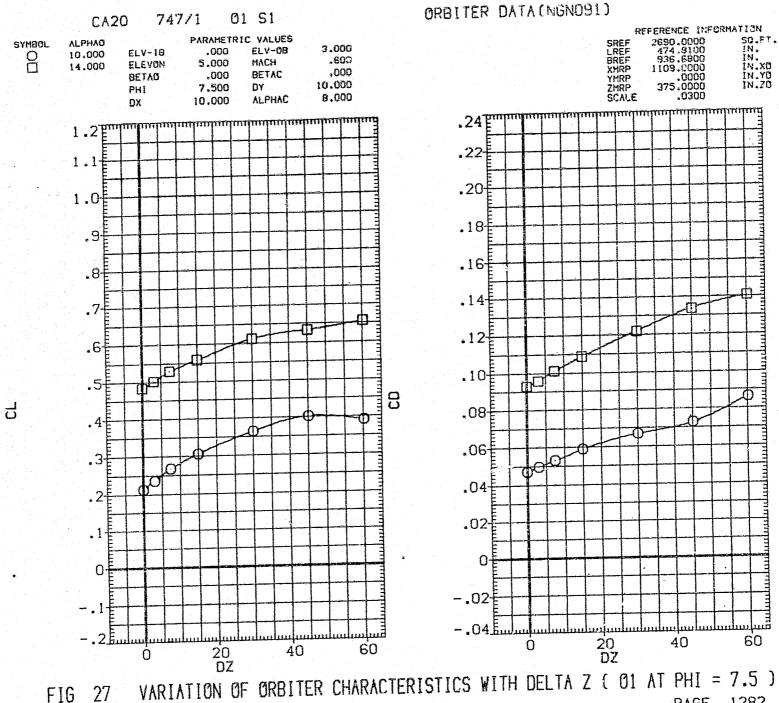
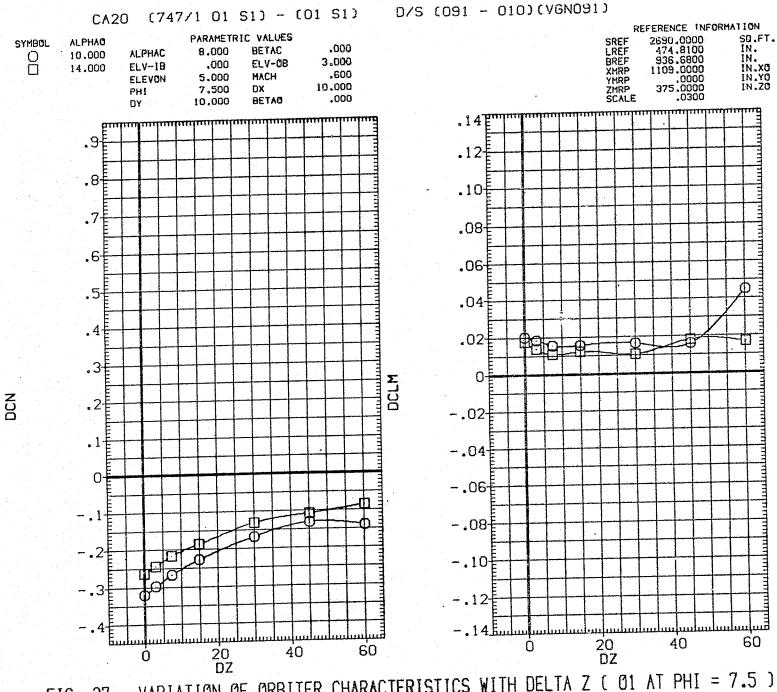


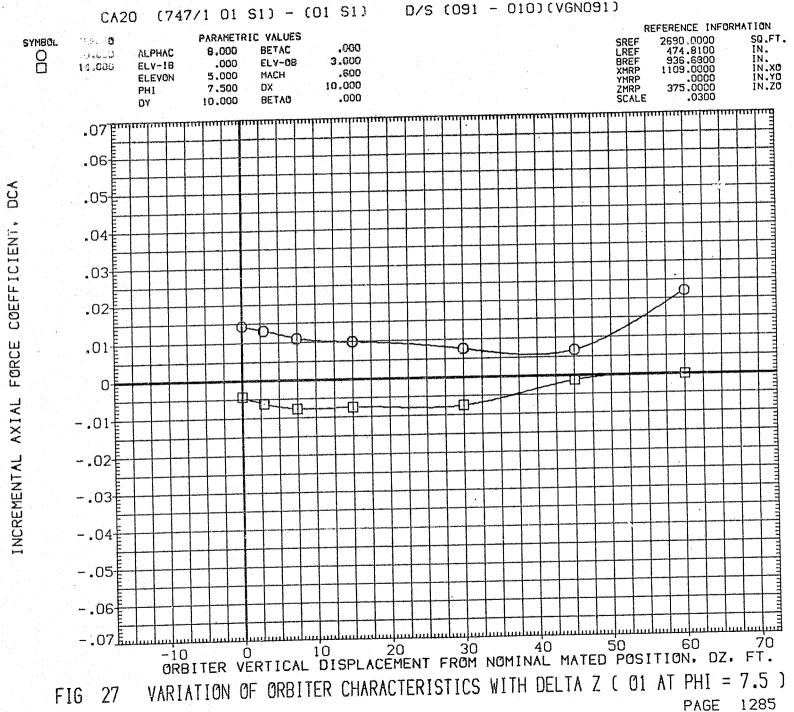
FIG PAGE 1282

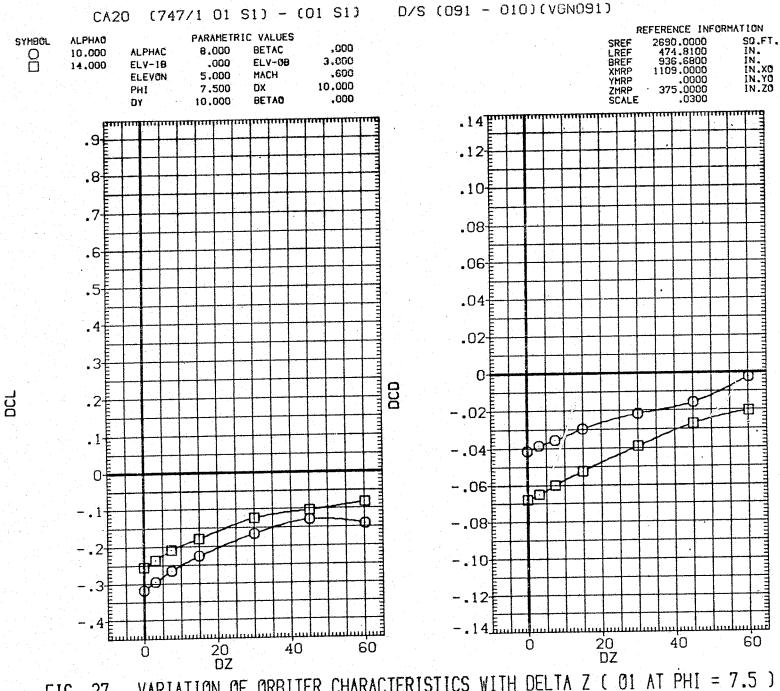
PAGE 1283



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE

D/S (091 - 010)(VGN091)





VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1286

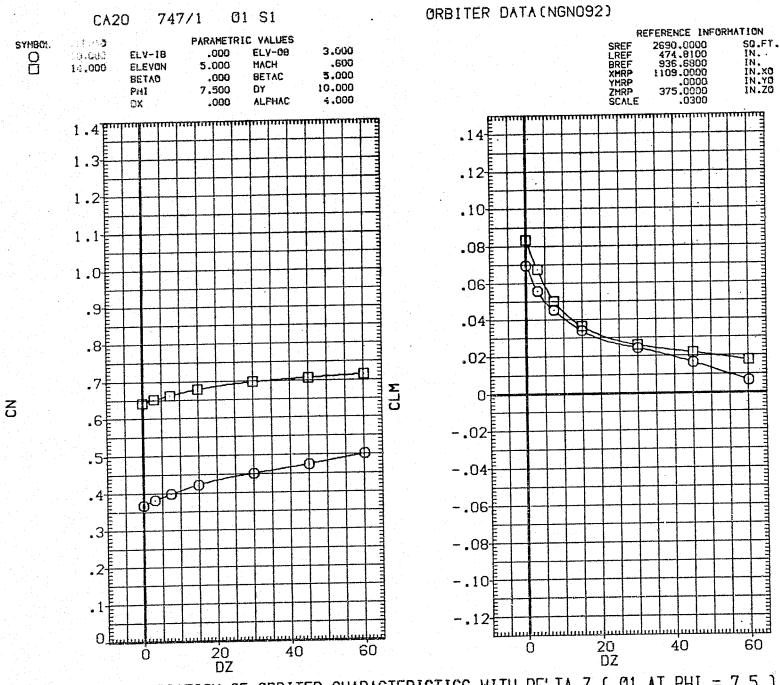


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1287

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FIG 27

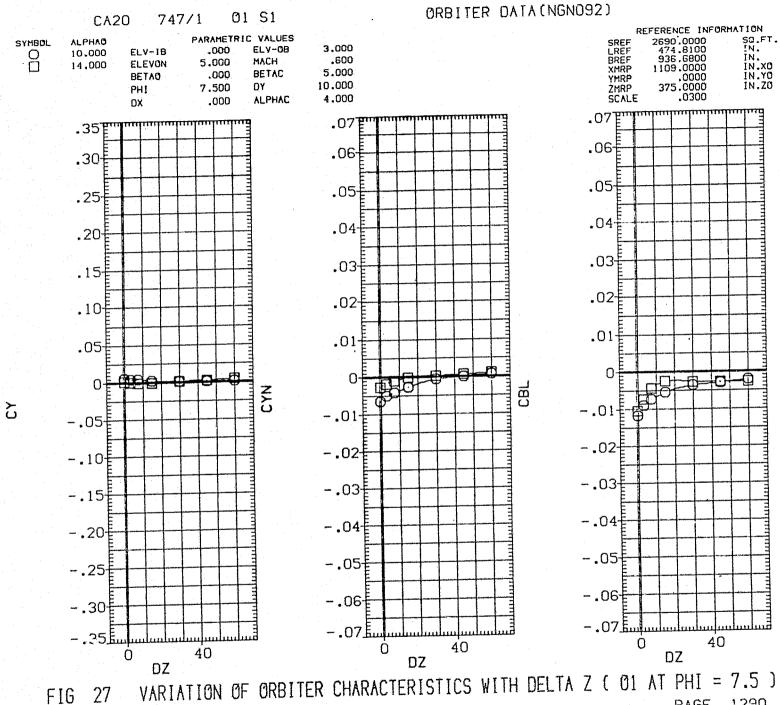


FIG 27 PAGE 1290

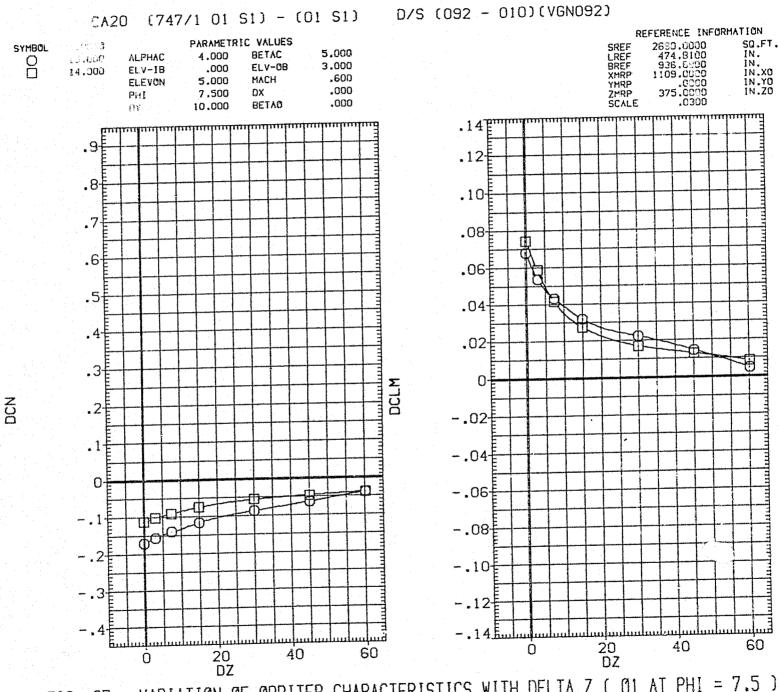
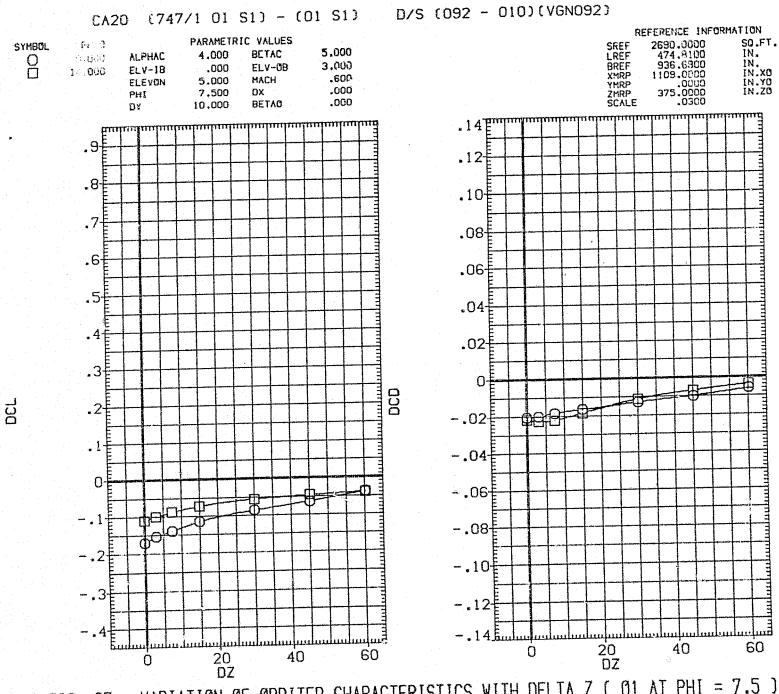


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )

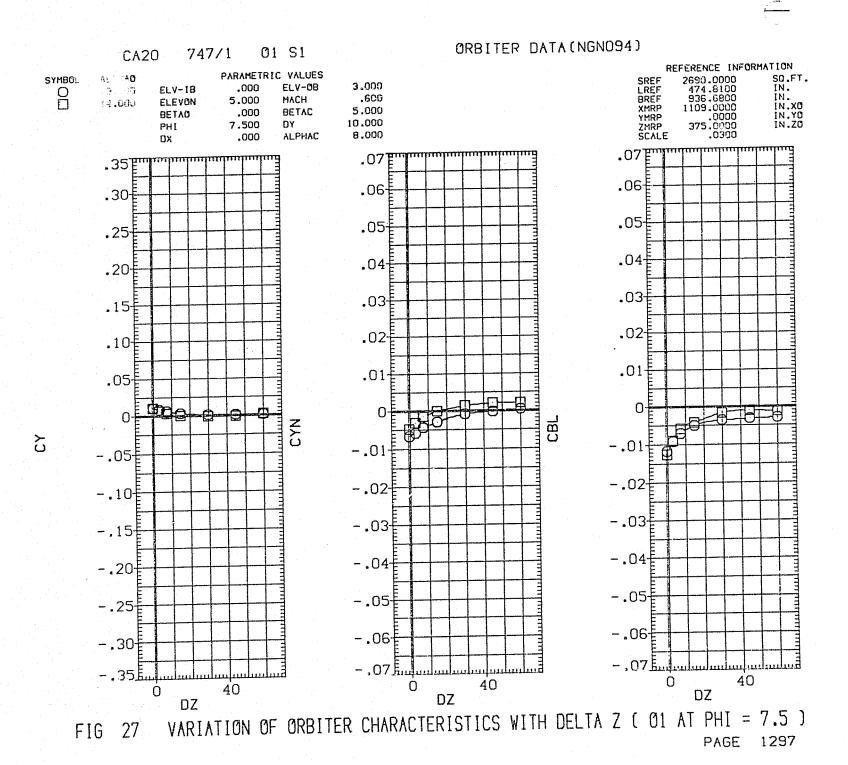


VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG PAGE 1293

or the following the complete control of the first control of the

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 1. PHI = 7.5 )
PAGE 1294

FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1296



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FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
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PAGE 1299

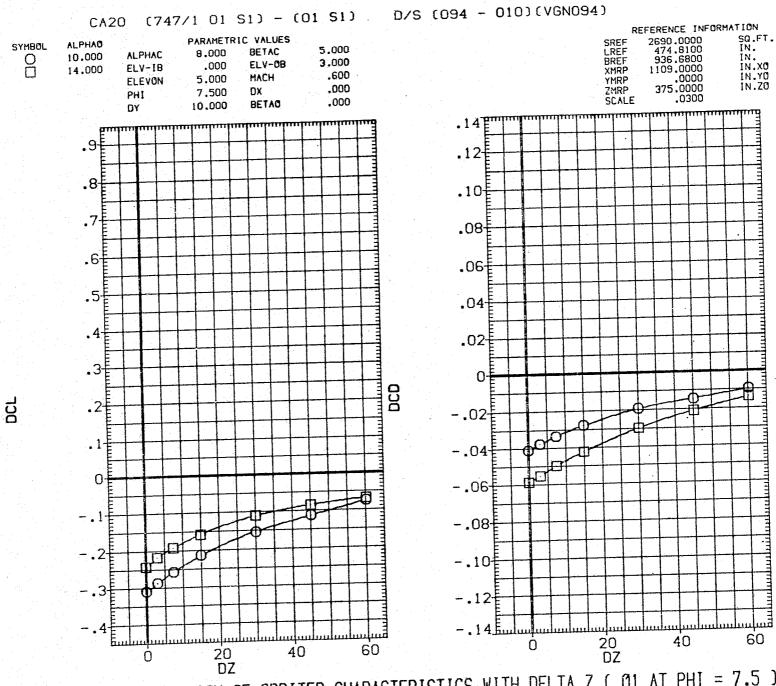


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )



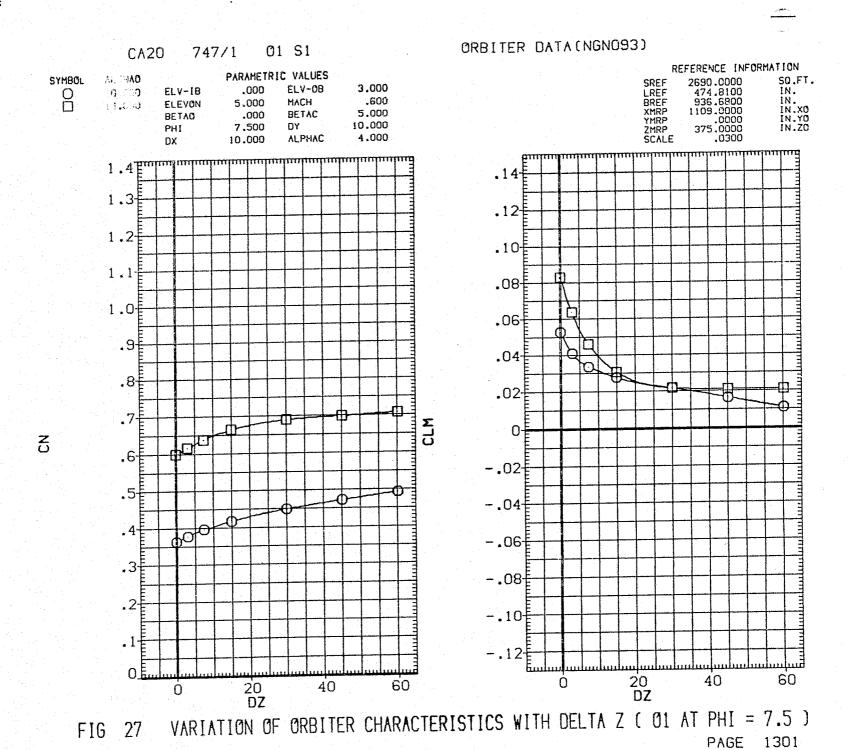


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )



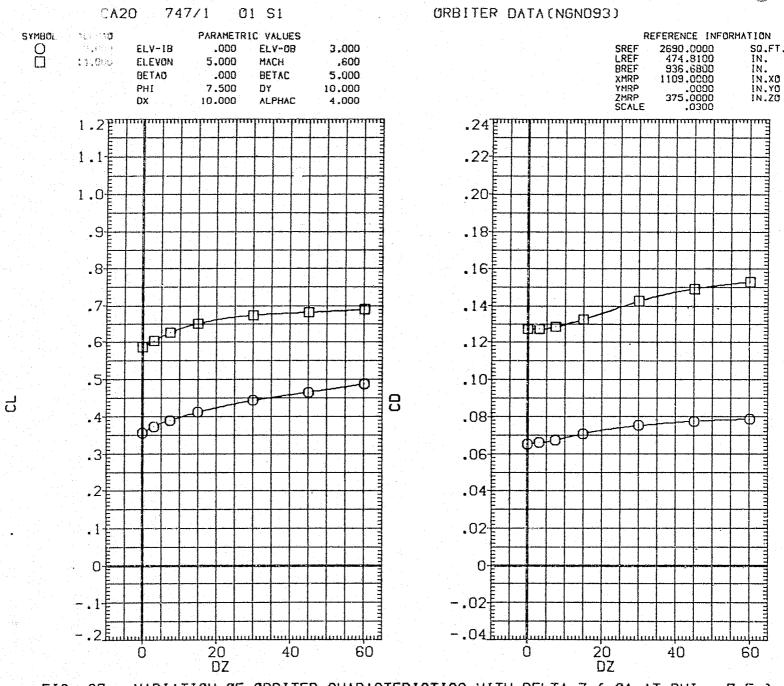
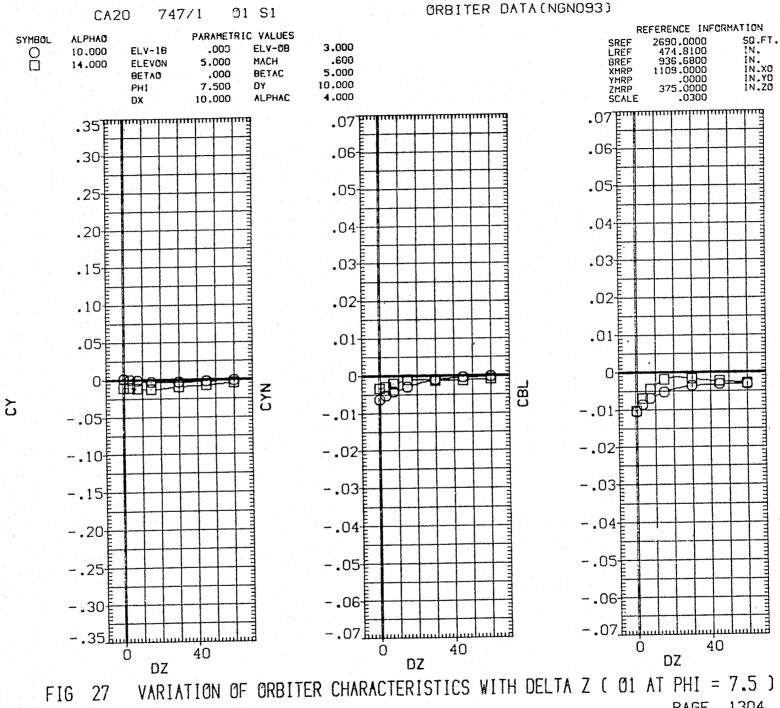


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1303



PAGE 1304

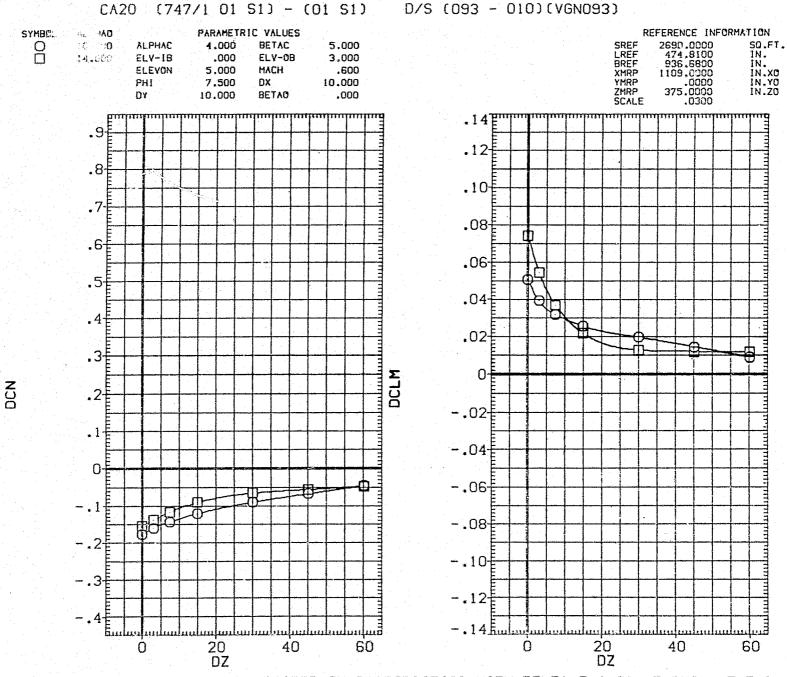
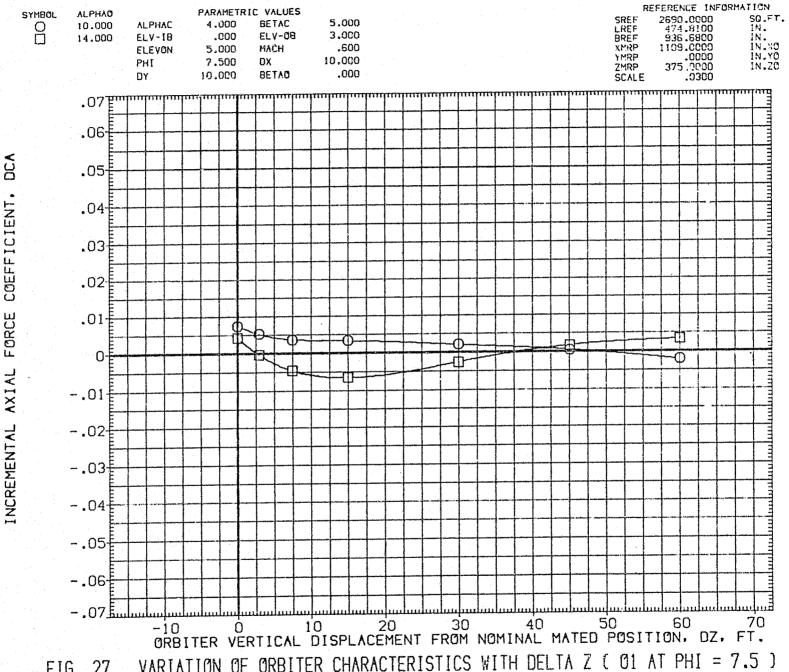


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1305

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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 ) FIG 27 PAGE 1306

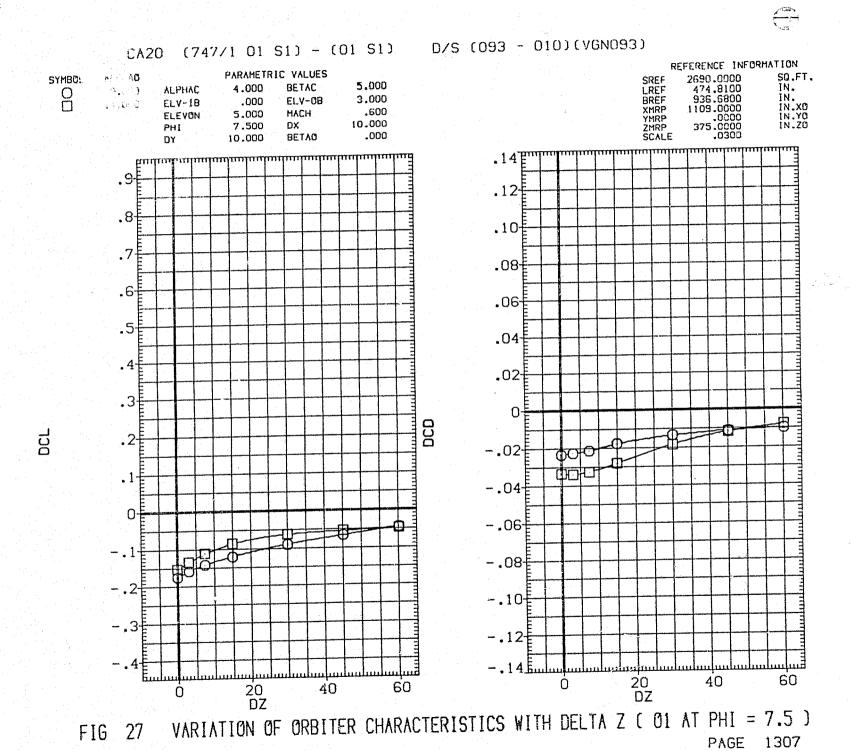


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1308

FIG PAGE 1310

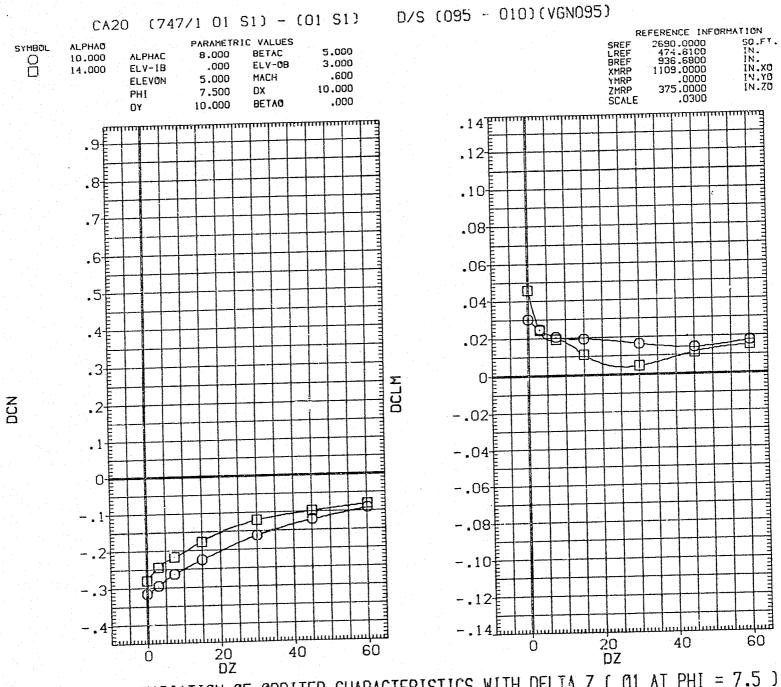


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1312

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1313

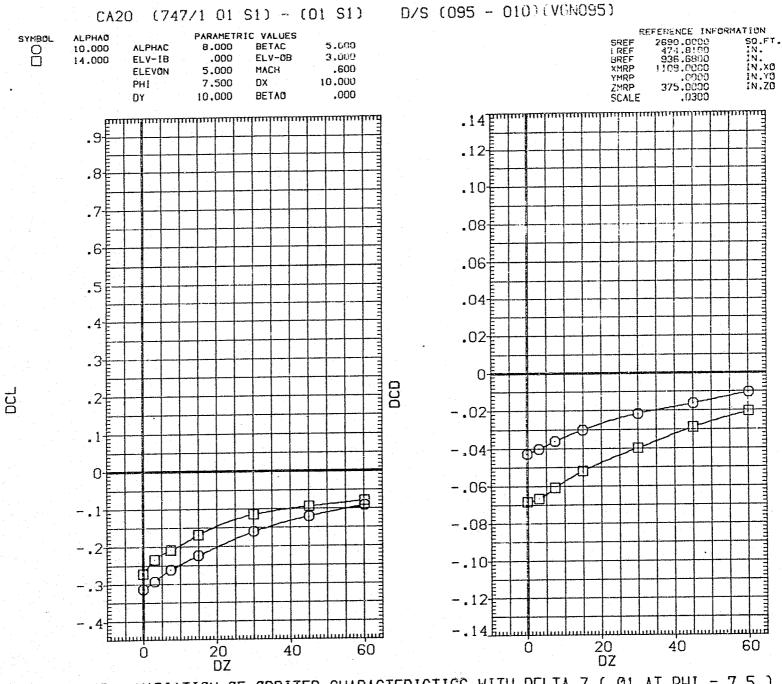


FIG 27 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 01 AT PHI = 7.5 )
PAGE 1314

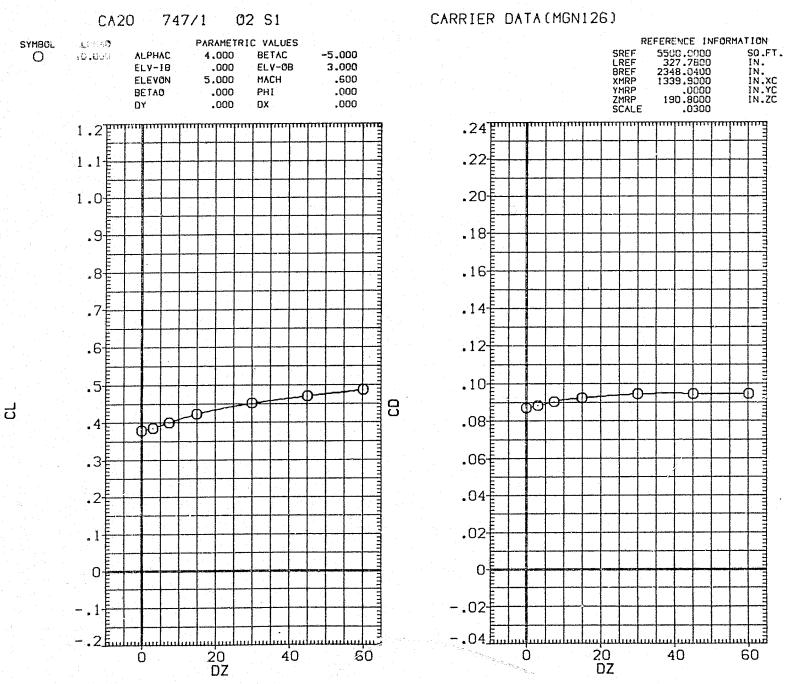
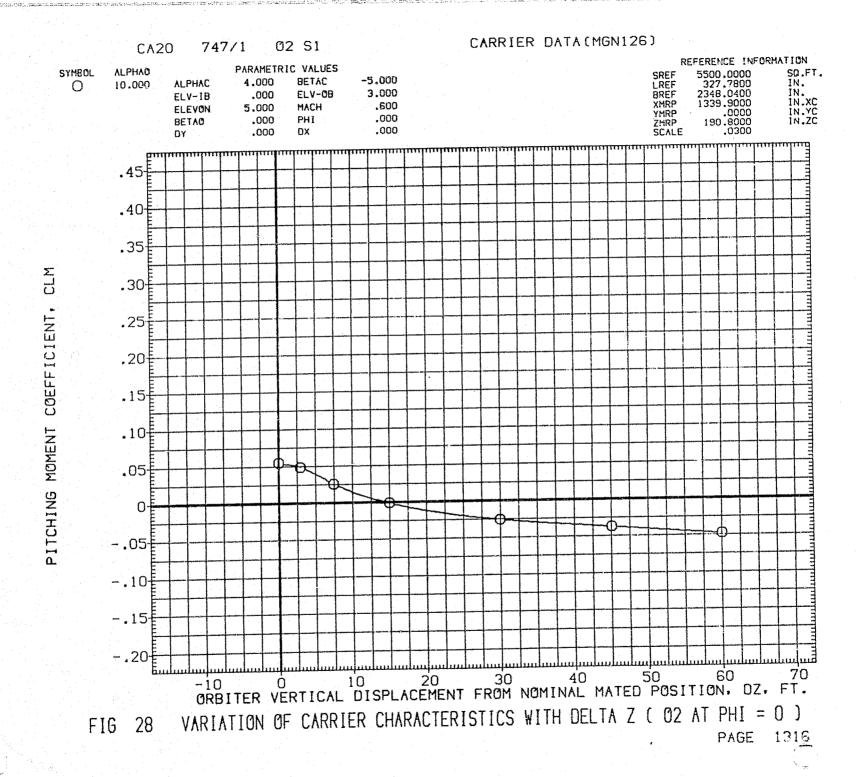
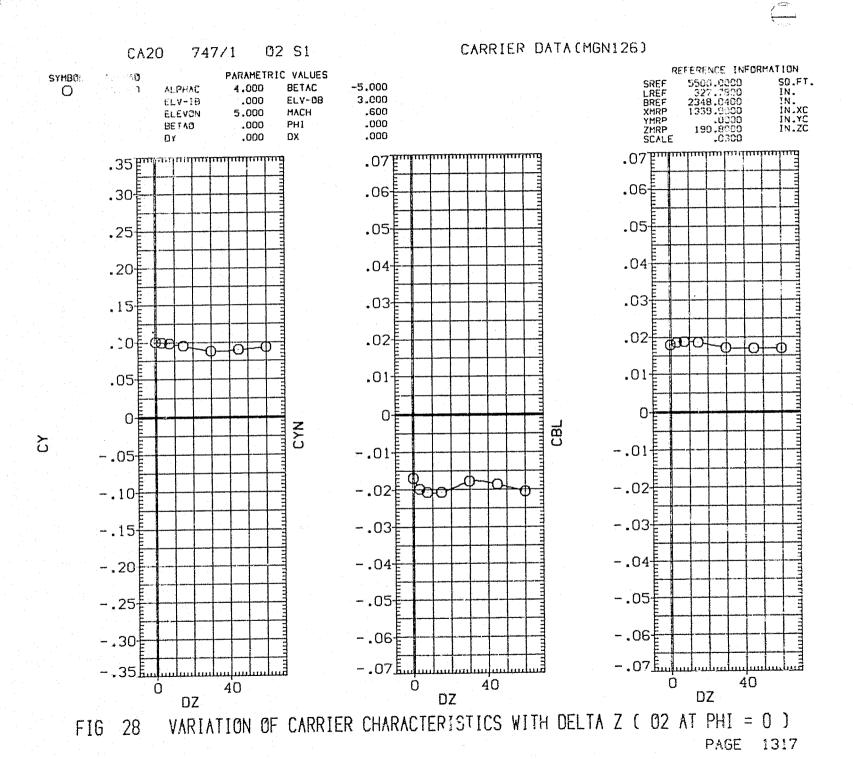


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1315





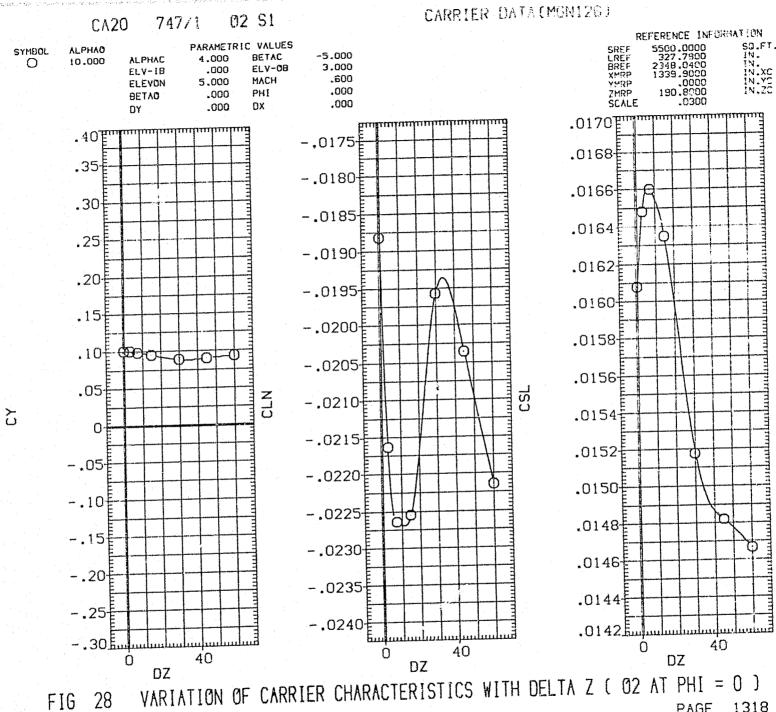


FIG 28 PAGE 1318

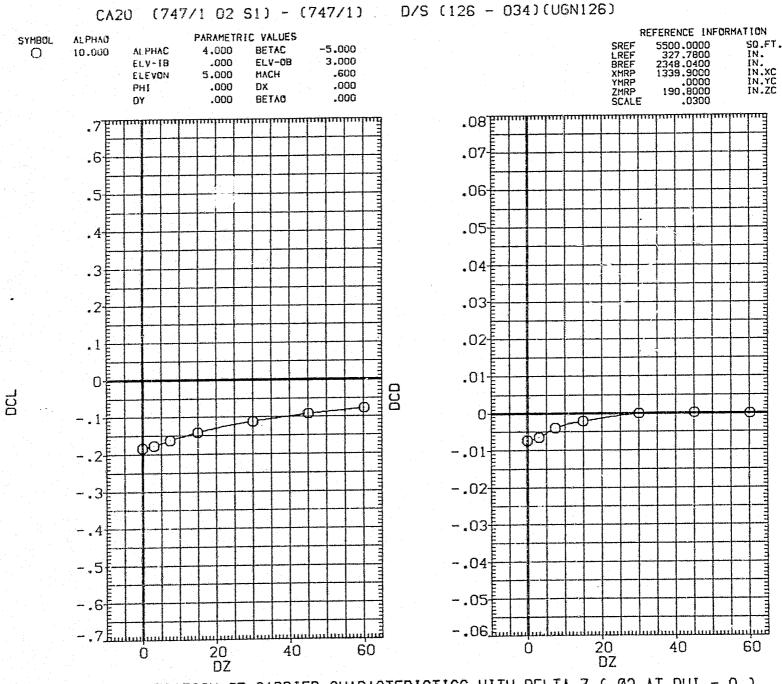
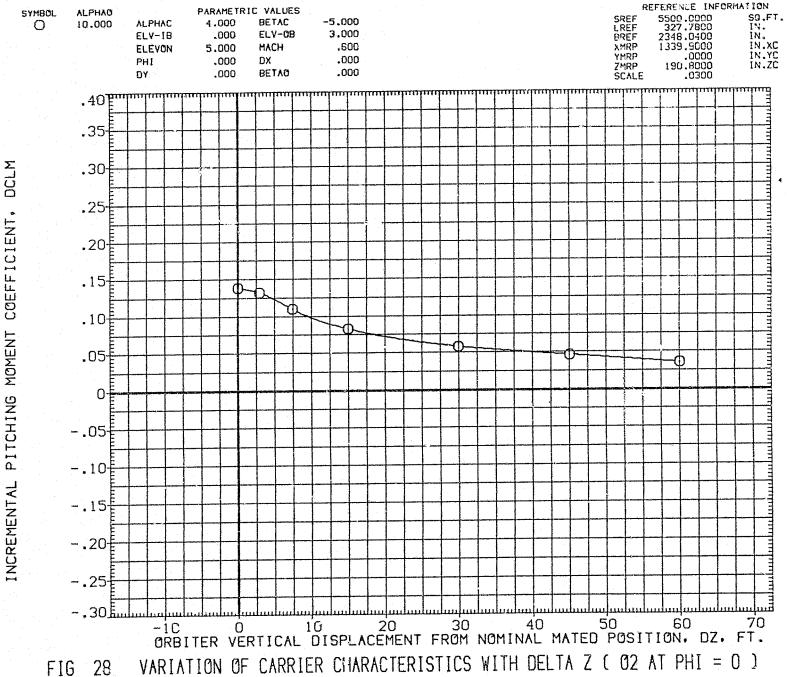


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )



PAGE 1320

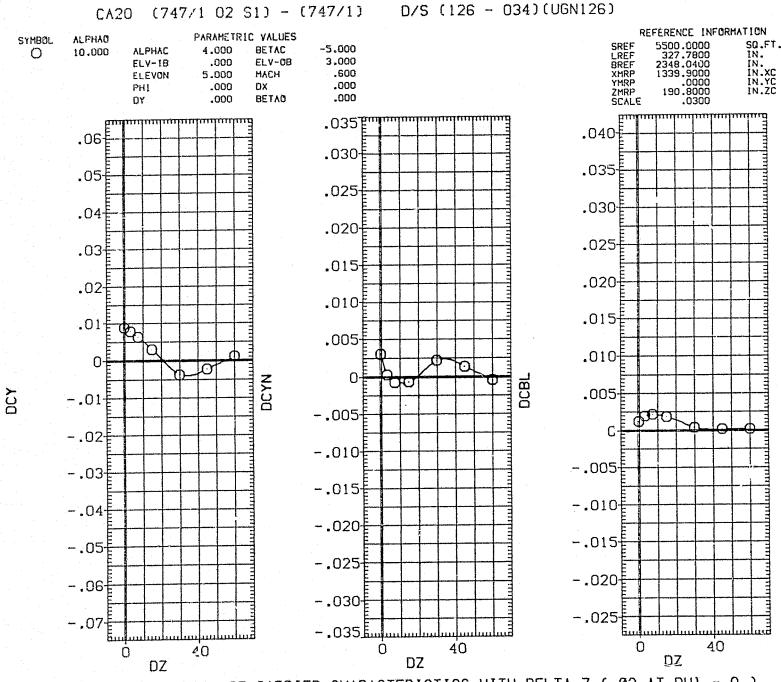
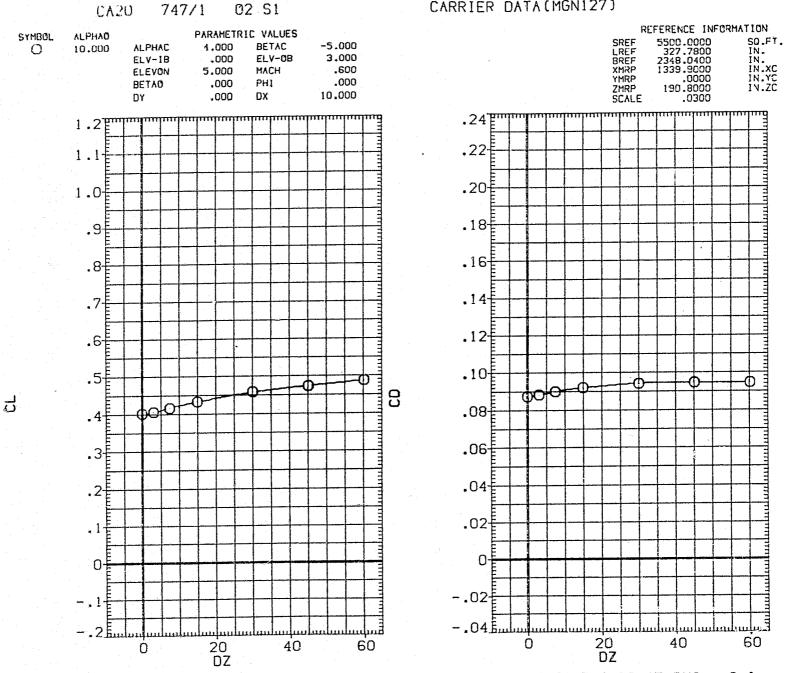


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PH) = 0 )

PAGE 1321

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG 28 PAGE



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE 1323

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1324

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CARRIER DATA (MGN127) **CA20** 747/1 02 S1 SYMBOL **ALPHAO** PARAMETRIC VALUES REFERENCE INFORMATION SO.FT.
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IN.YC
IN.ZC 5500.0000 327.7800 2348.0400 1339.9000 0 10.000 **ALPHAC** 4.000 BETAC -5.000 SREF .000 3.000 ELV-1B ELV-08 ELEVON 5.000 ,600 MACH XMRP BETAO ,000 PHI .000 0000. 0008.001 0000. ZMRP SCALE DY .000 DX 10.000 .30£ .06-.06 .25 .05 .05 .20£ .04 .04 .15<del>-</del> .03<del>[</del> .03<del>[</del> .10<del>[ @Qo</del> .02 .05 .01<del>=</del> .01-0-0-CBL  $\sim$ -.05£ -.01 -.01 -.02<del>[</del> -.10<del>[</del> -.02 100 -.15<del>-</del> -.03<del>E</del> -.03 -.20<del></del> -.04 -.04 -.25 -.05 -.05 -.30<del>[</del> -.06--.06<del>-</del>

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1325

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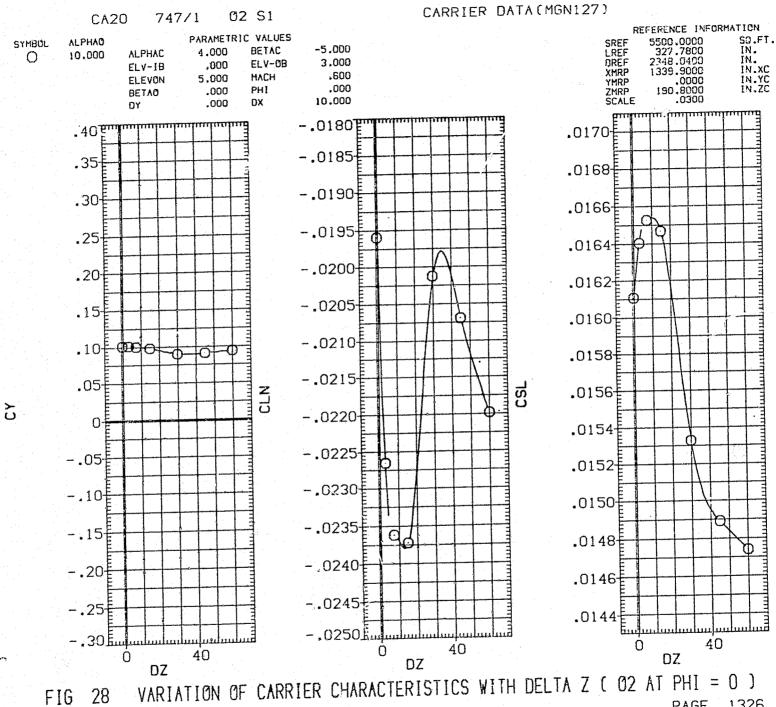
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PAGE 1326

CA20 (747/1 02 S1) - (747/1) D/S (127 - 034) (UGN127)

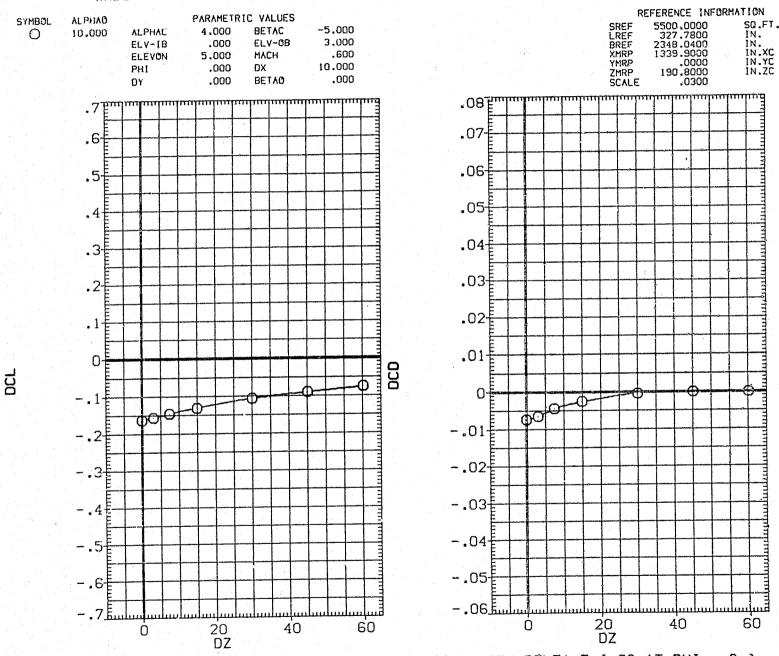


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1327

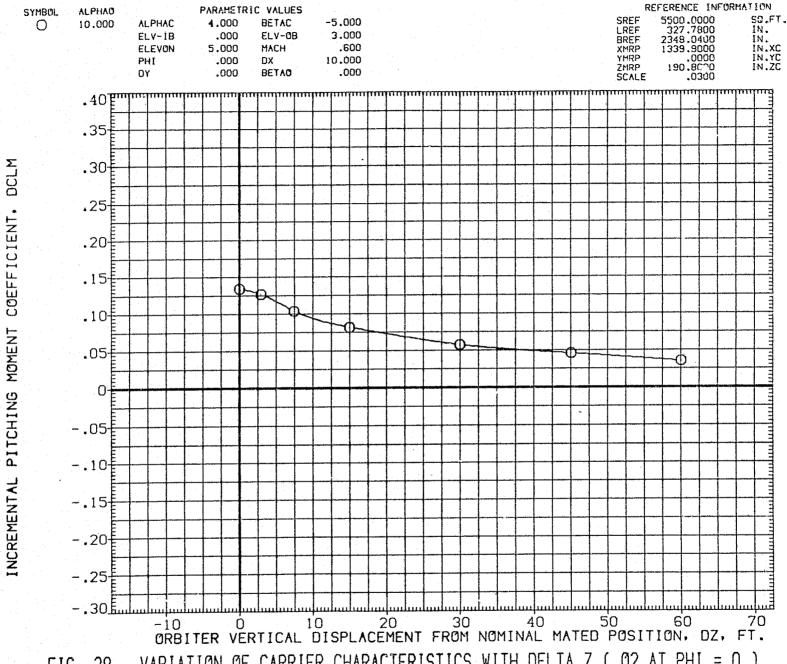
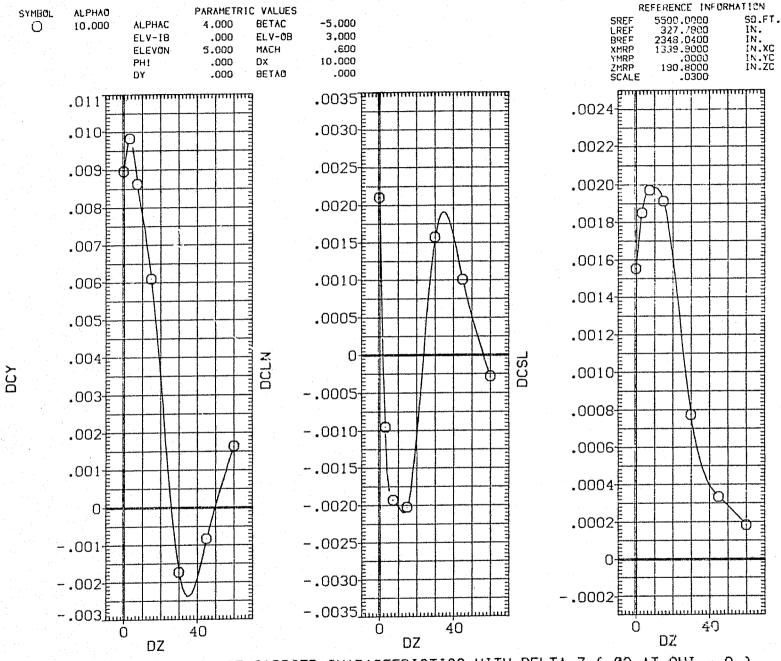


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1328

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) 28 FIG 1330 PAGE

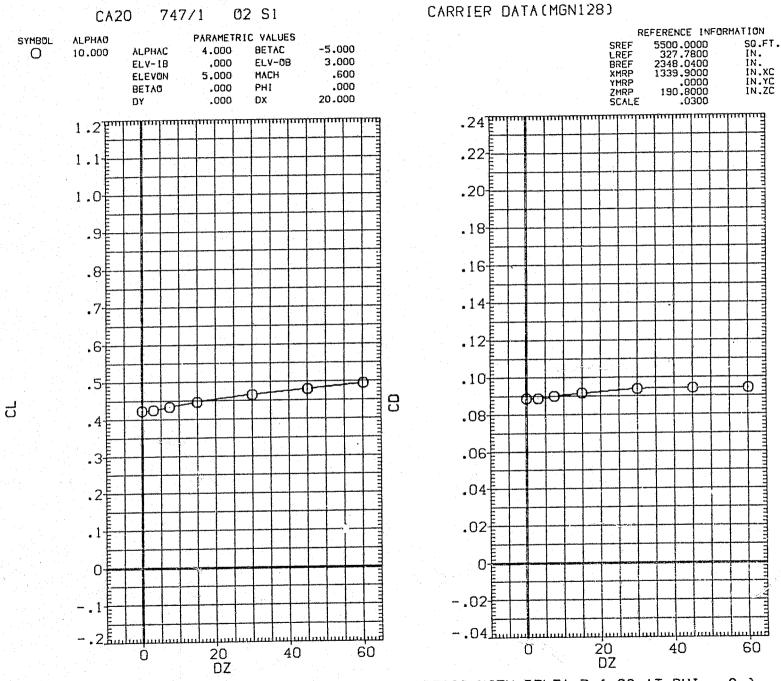


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1331

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE 1332

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
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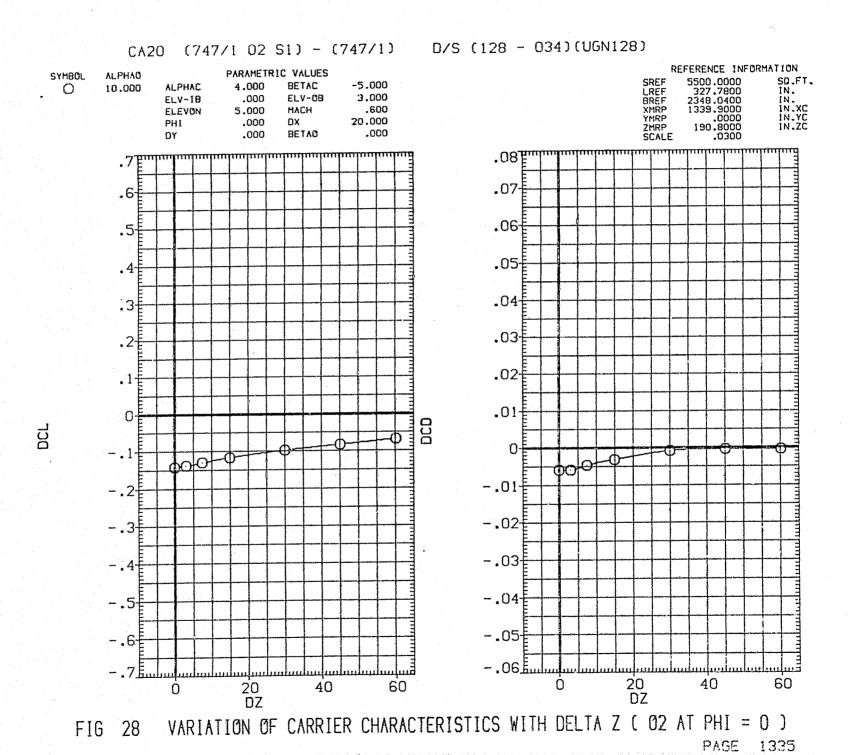
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FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1334



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) PAGE 1336

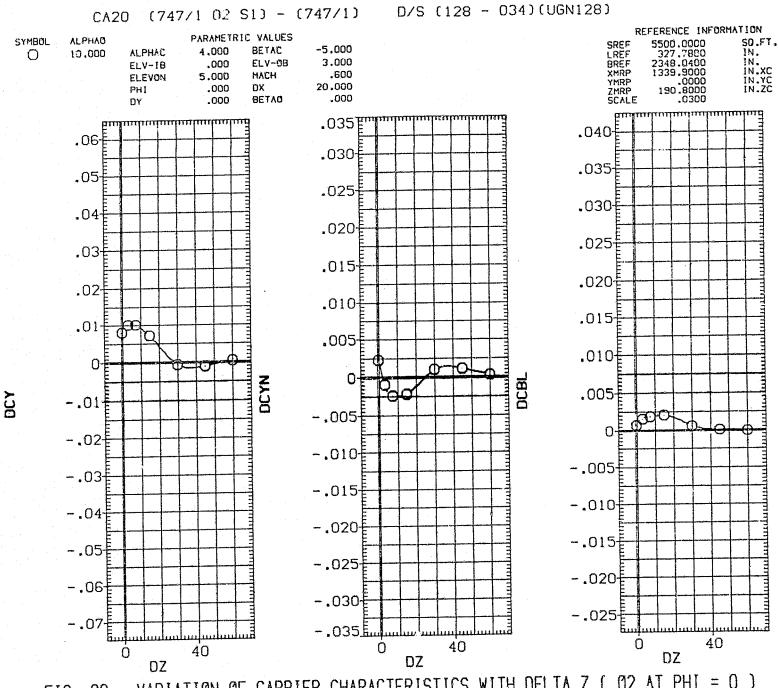


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1337

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )



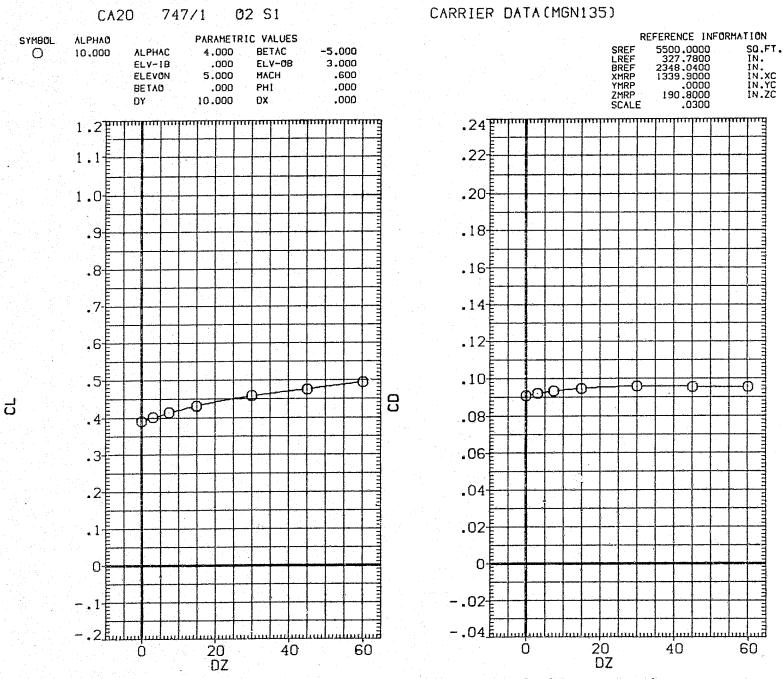
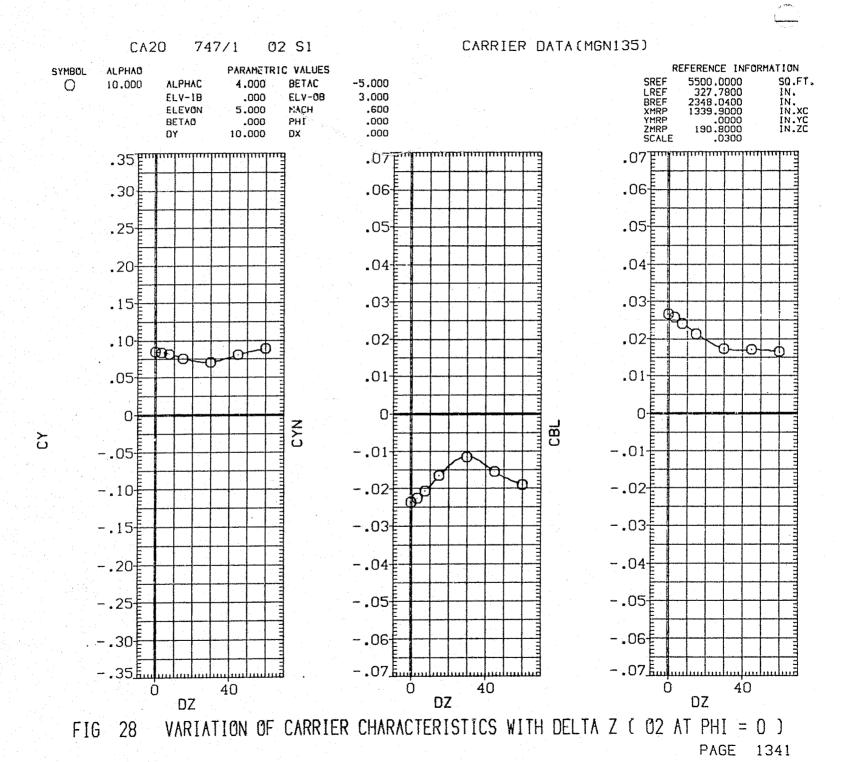
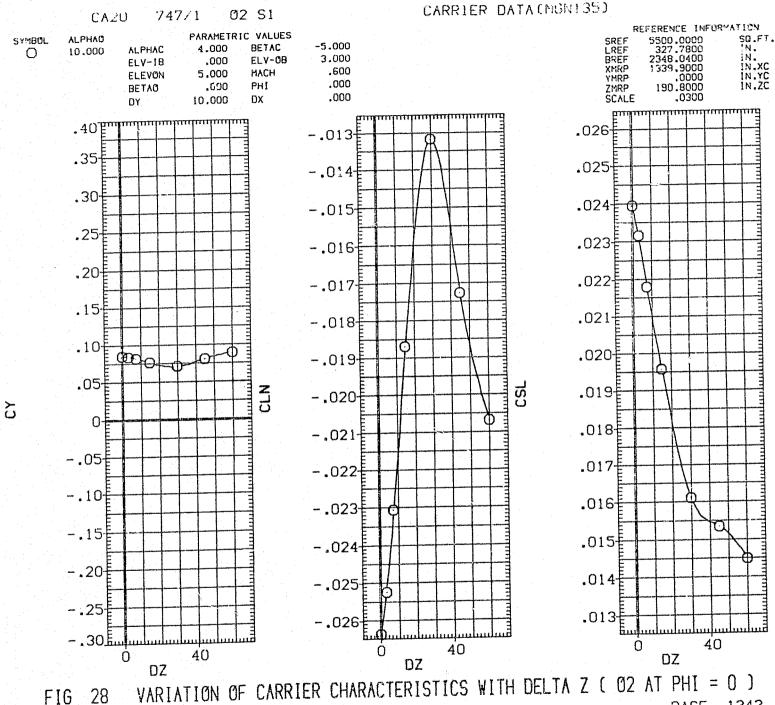


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
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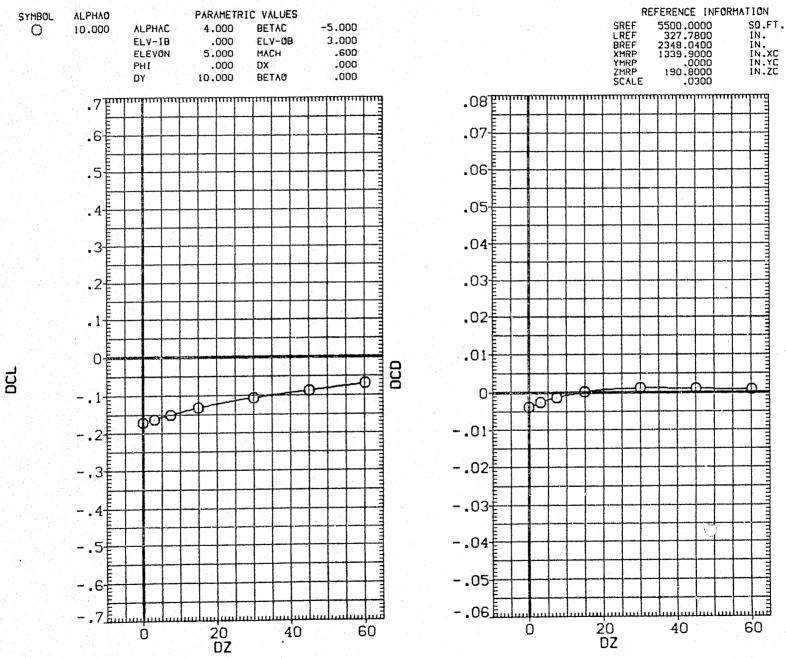


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1343

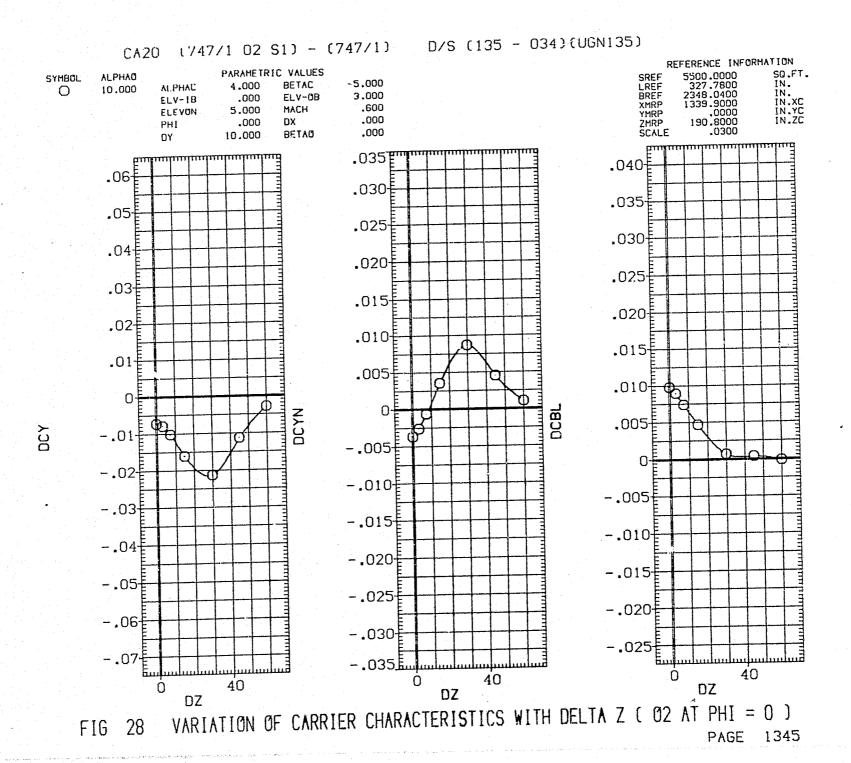
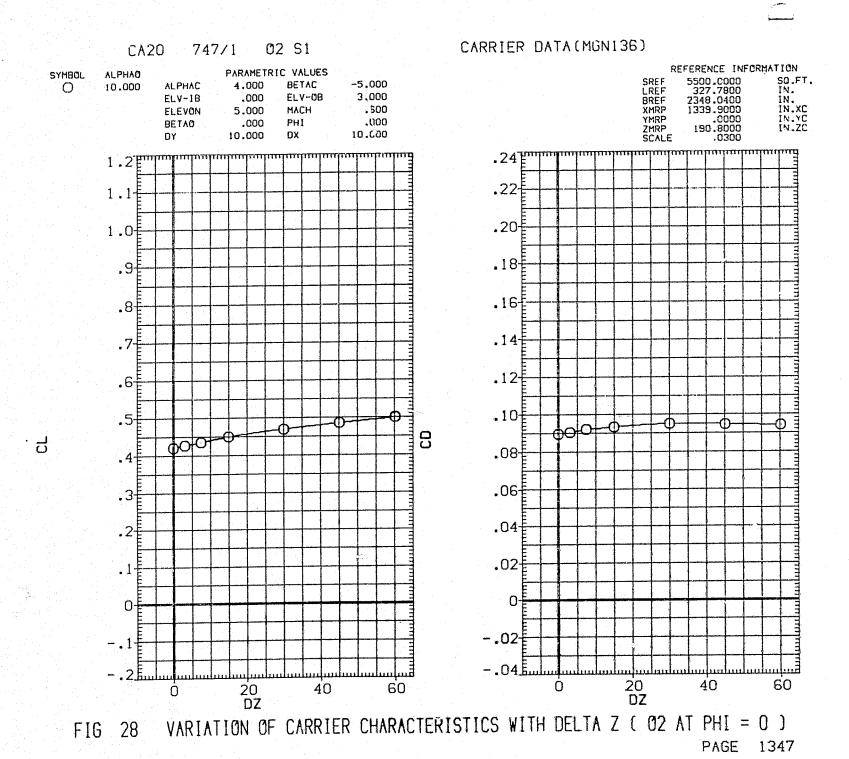


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1346



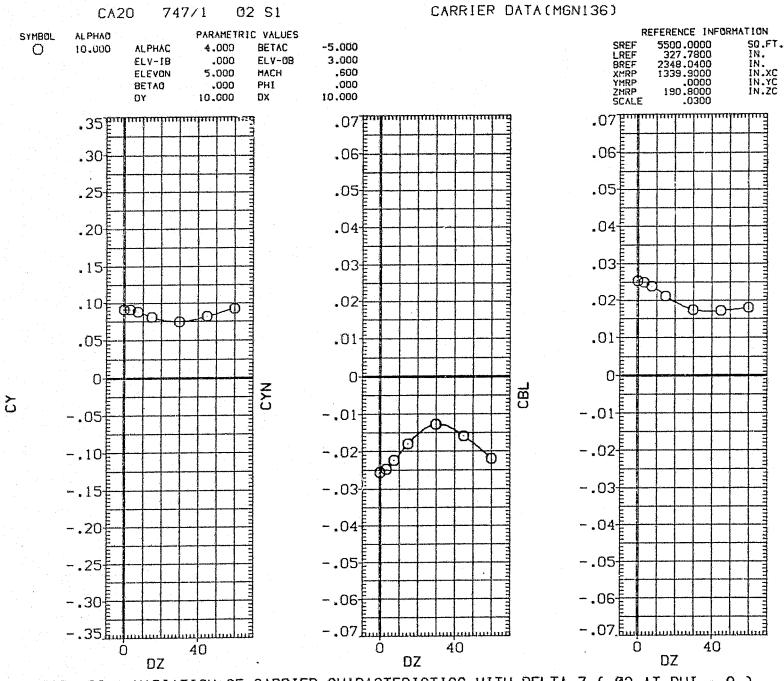


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1349

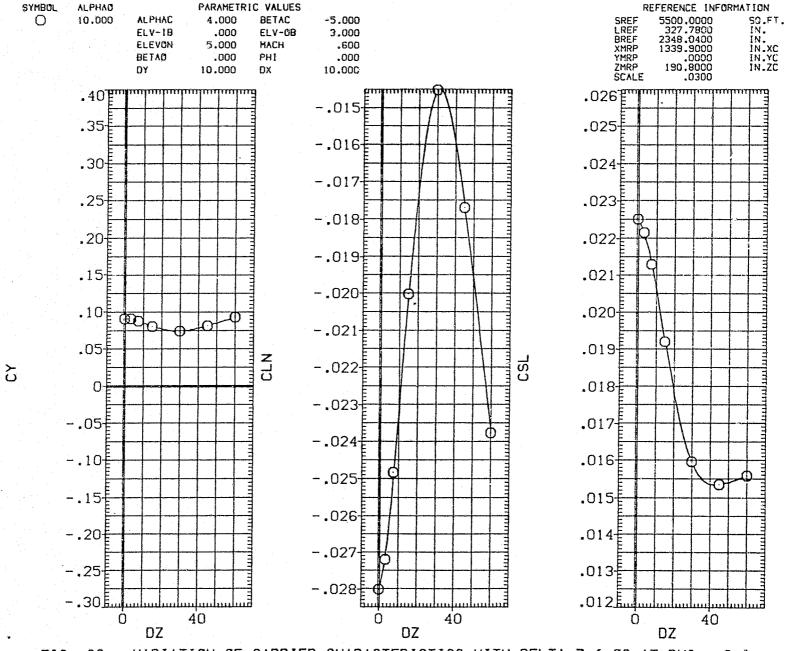


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1350

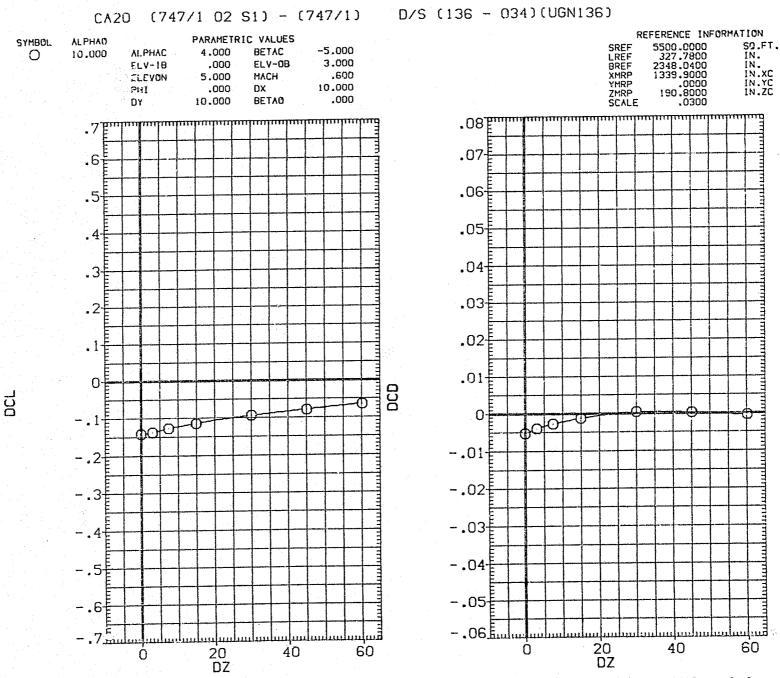


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1351

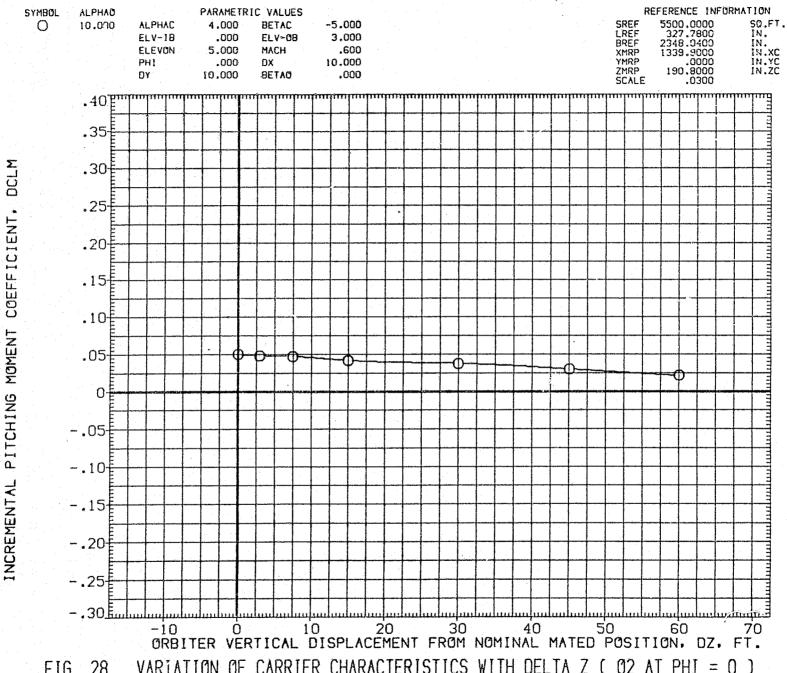


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 135

D/S (136 - 034)(UGN136)

 $(747/1 \ 02 \ S1) - (747/1)$ REFERENCE INFORMATION PARAMETRIC VALUES ALPHAO SYMBOL 5500.0000 327.7800 2348.0400 SO.FT. IN. IN. SREF 4.000 BETAC -5.000 0 10,000 **ALPHAC** LREF BREF XMRP YMRP ZMRP ELV-08 3,000 .000 ELV-IB 1339.9000 0000 190.8000 IN.XC IN.YC IN.ZC .600 MACH ELEVON 5.000 10.000 .000 DX PHI .000 BETAO 10.000 DY SCALE բույրուրուրորություն .035E .040 .06 .030 .035 .05 .025 .030 .04 .020 .025 .03± .015 .020 .02 .010 .015 .01= .005 .010E DCY .005 -.01<del></del> -.005<del></del> 0--.02 -.010<del>[</del> -.005 -.03<del>E</del> -.015<del>[</del> -.010 -.04 -.020  $-.015 \pm$ -.05 -.025<del>-</del> -.020 -.06E -.030 -.025 -.07 **- .**035<u>E</u>... 40 40 0 40 0 Ó DZ DZ DZ

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) 28 FIG PAGE 1353

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1354

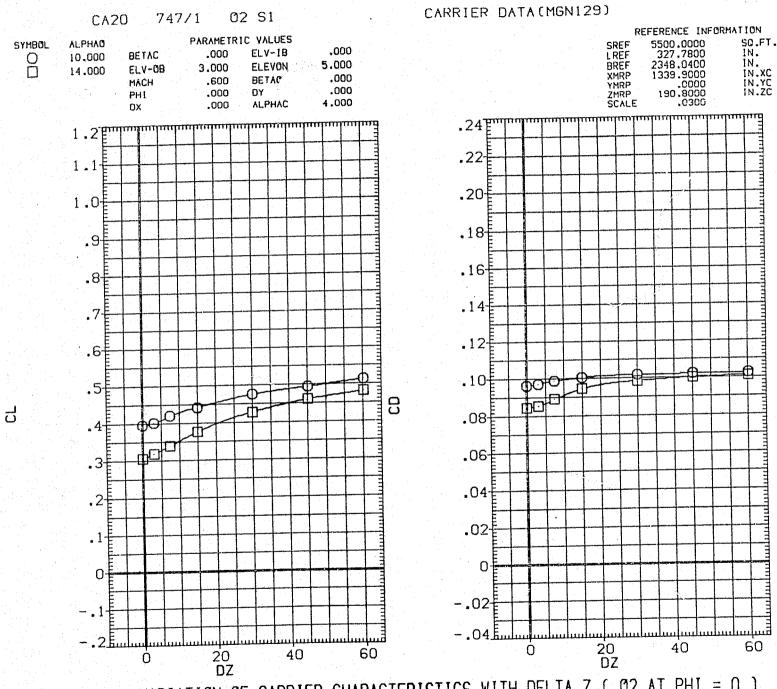
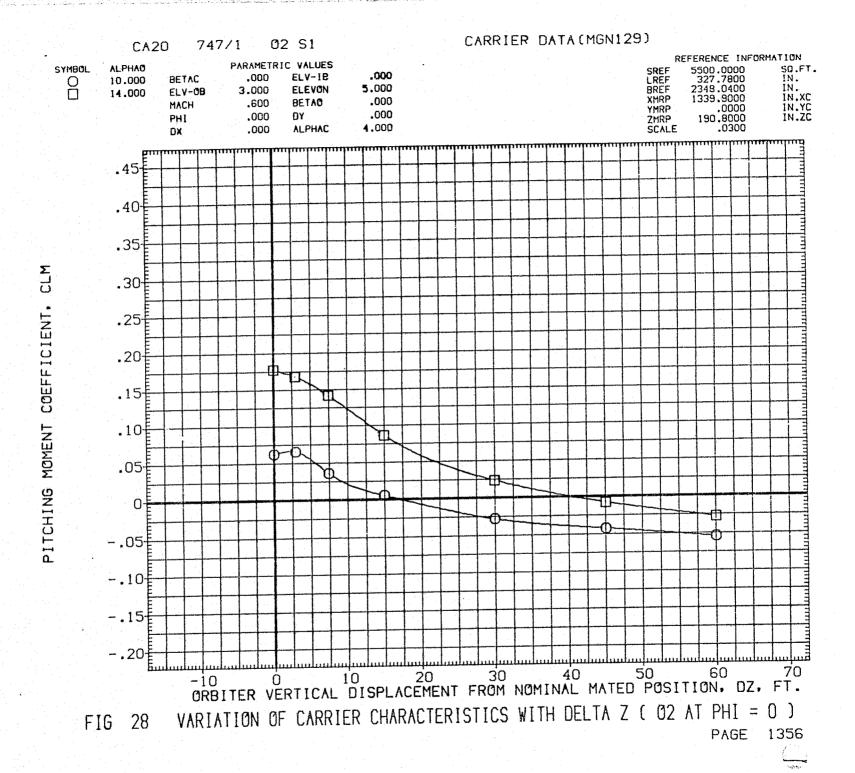
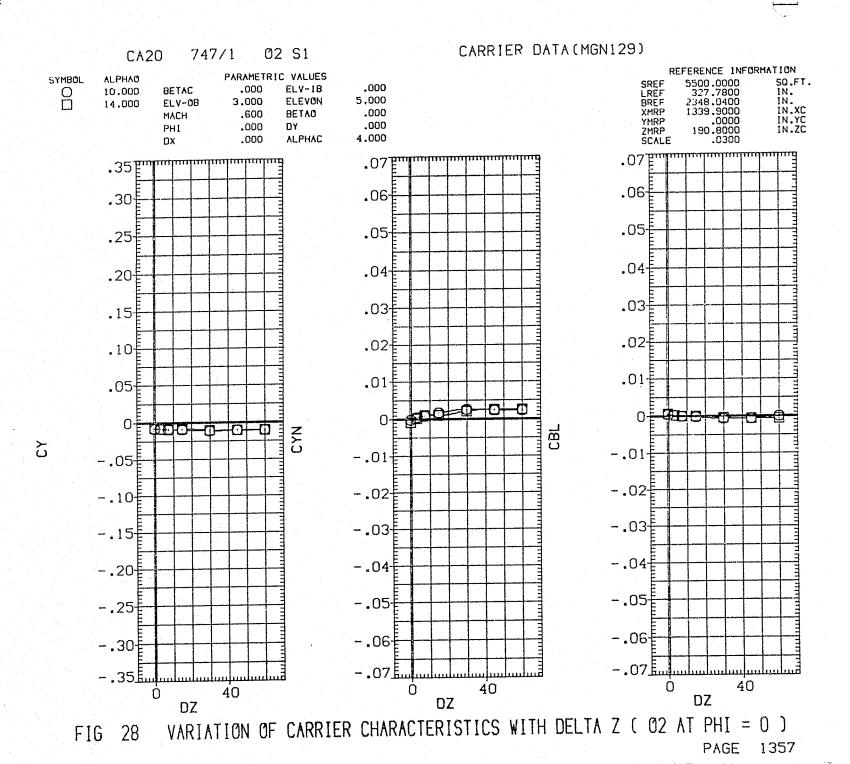
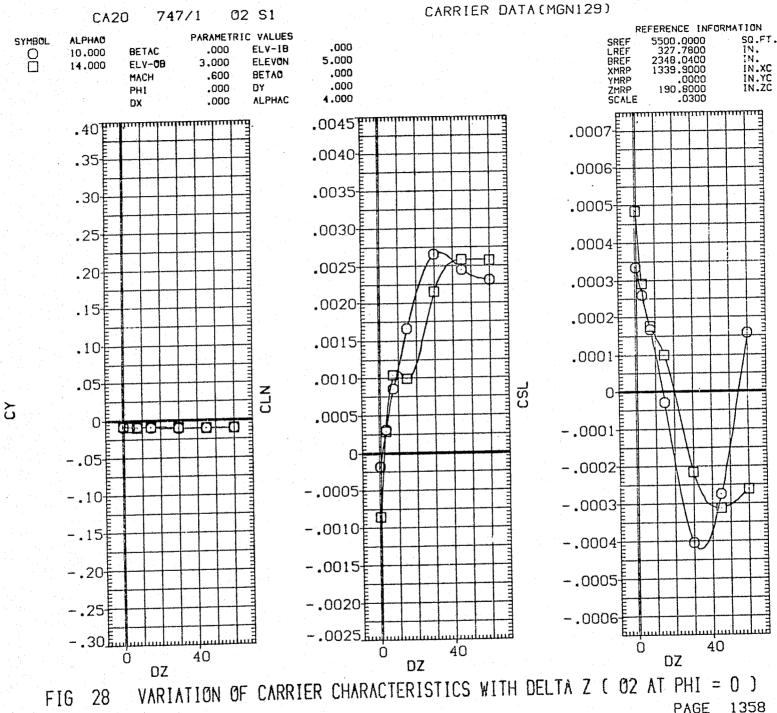


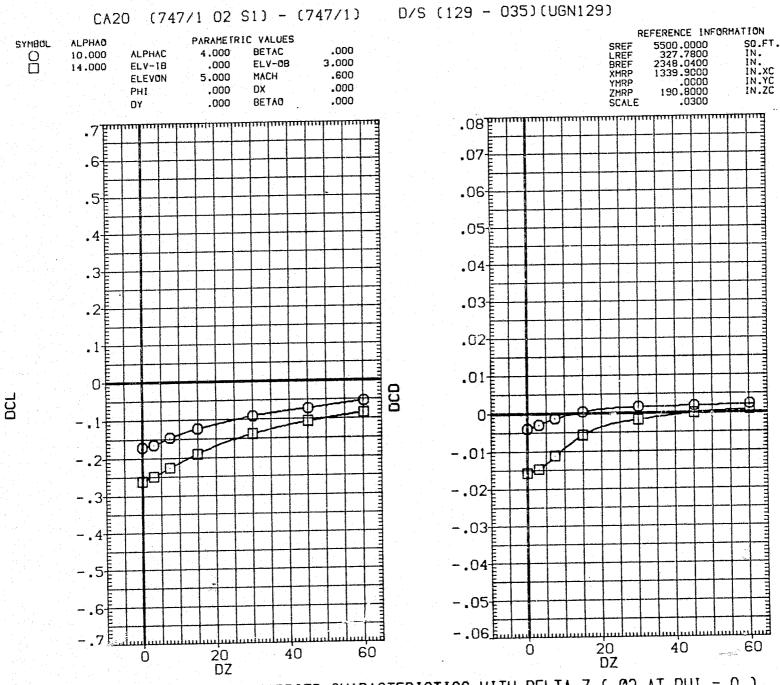
FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1355







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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1359

FIG

28

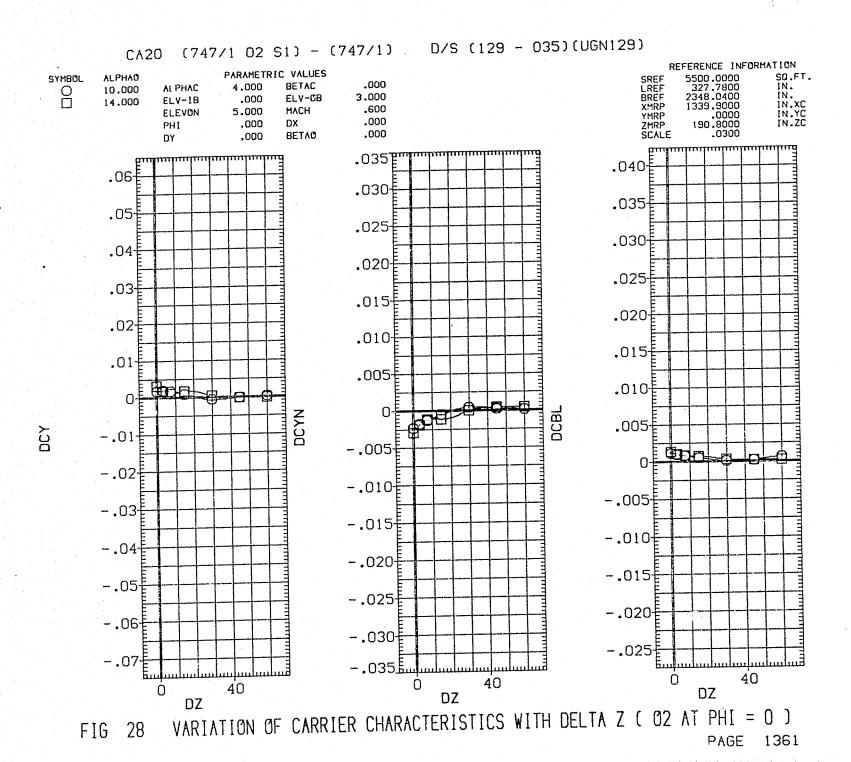


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1362

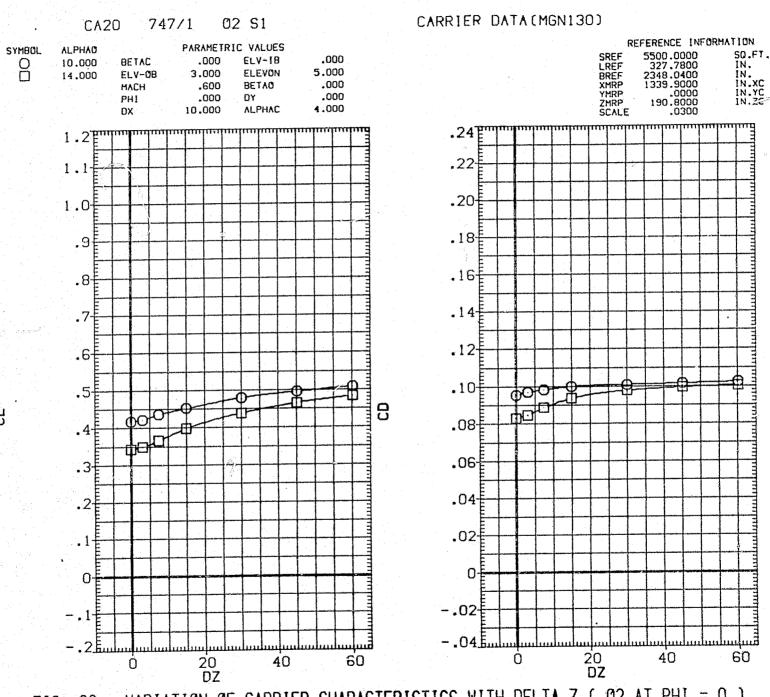
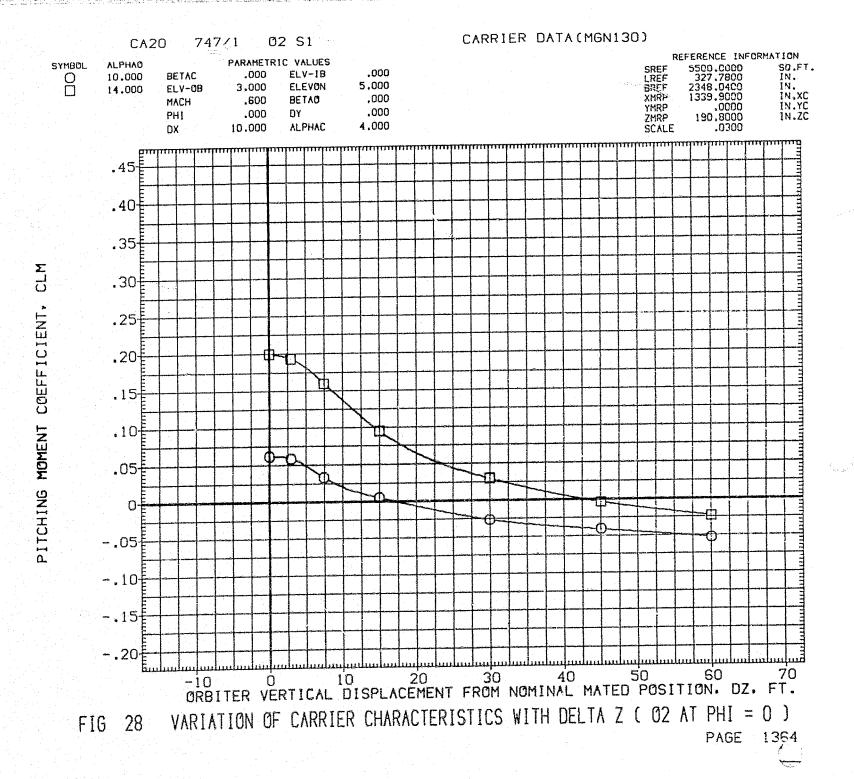


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1363





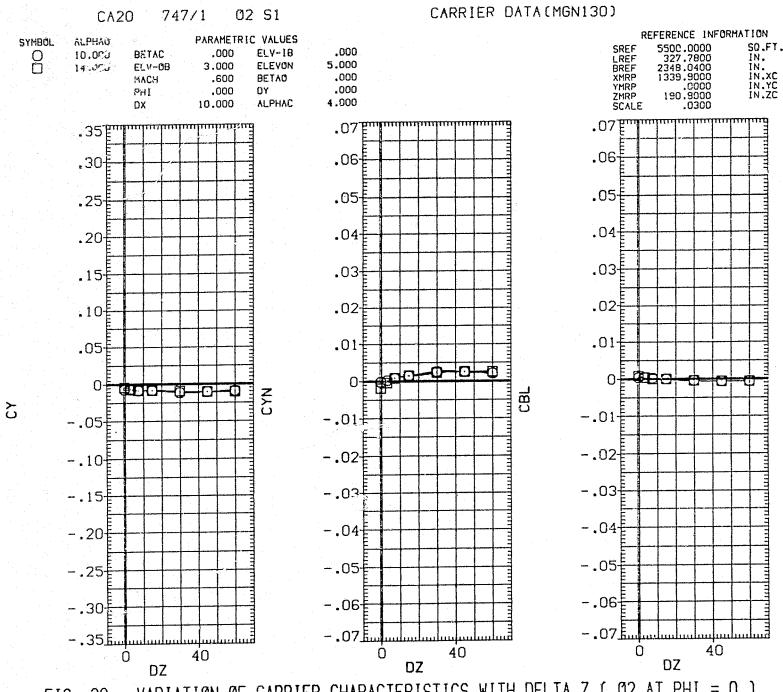


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1365

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE 136F

CA20 (747/1 02 S1) - (747/1) D/S (130 - 035)(UGN130) SYMBOL ALPHAO PARAMETRIC VALUES REFERENCE INFORMATION 0 10.000 **ALPHAC** 4.000 BETAC .000 5500.0000 SO.FT. SREF 14.000 ELV-18 .000 ELV-08 3.000 327,7800 IN. 2348.0400 ELEVON BREF 5,000 MACH .600 IN.XC IN.YC IN.ZC XMRP 1339.9000 PHI .000 DΧ 10.000 .0000 YMRP DY .000 BETAO .000 ZMRP .08Em .6-.07 .5 .06 .05 .4 .04 .03 .02<del>-</del> 0-.01 DCL 000 -.10--.01--.3+ -.02 -.03 -.04<del>-</del> -.6<del>E</del> -.05<del>[</del> -.06 E...l... 20 DZ 0 40 60 20 **DZ** 0 40 60

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
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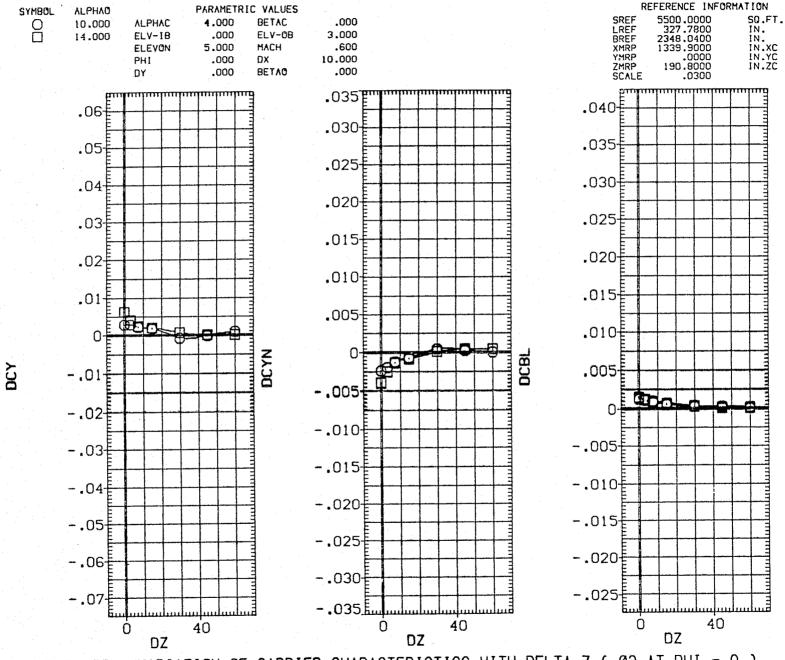


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1369

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
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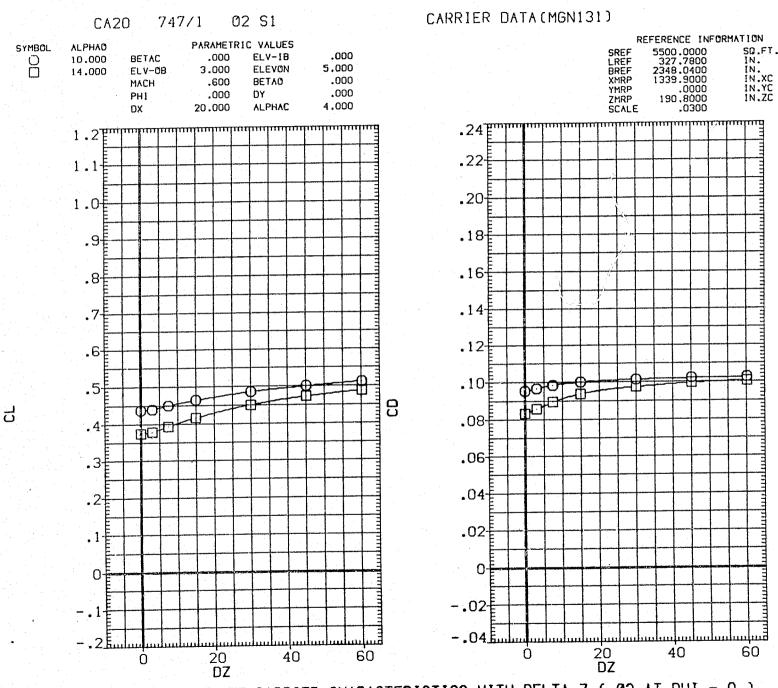
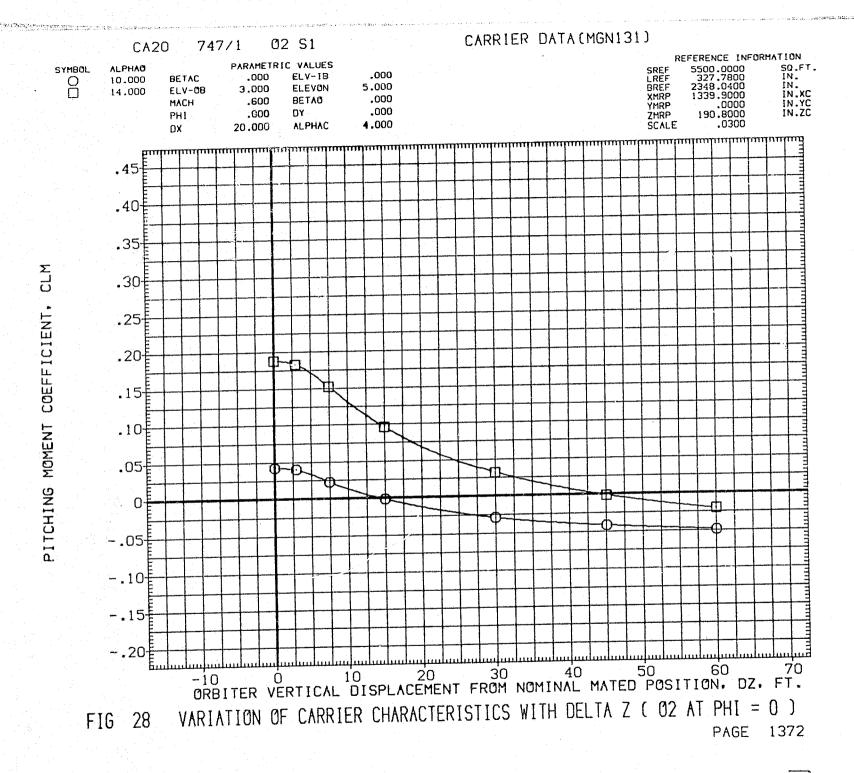


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1371



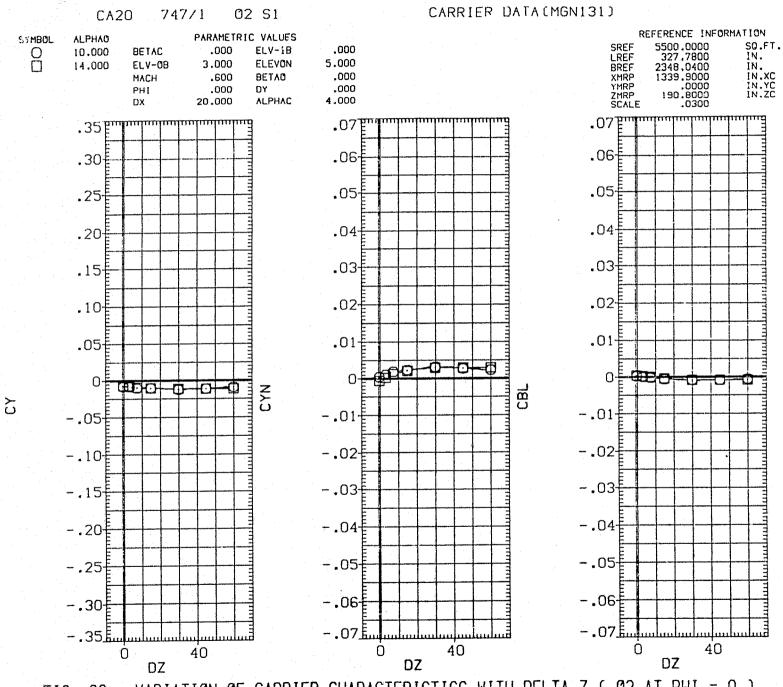


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1373

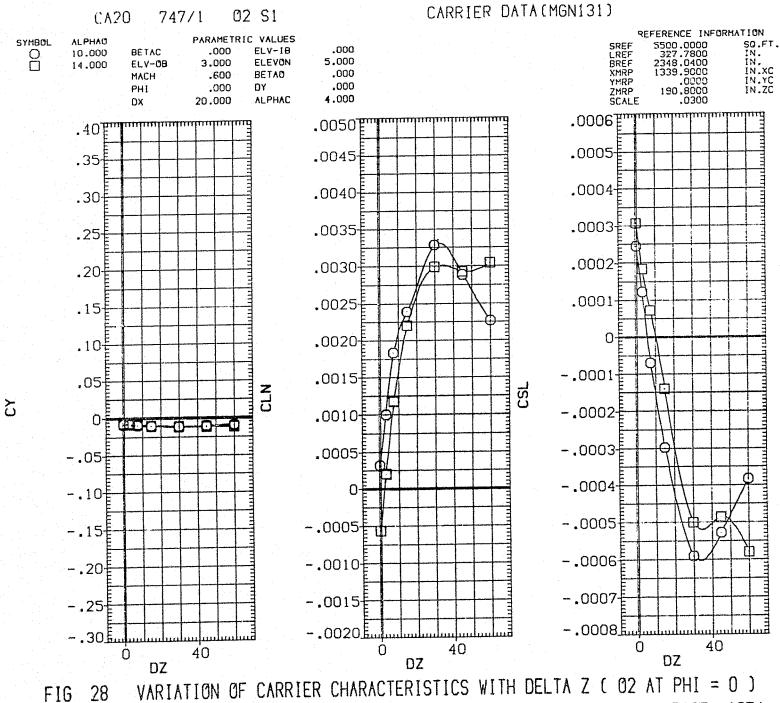


FIG 28 1374 PAGE

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1375

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT. FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 162 AT PHI = 0 )

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FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1378

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PAGE

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1380

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT.

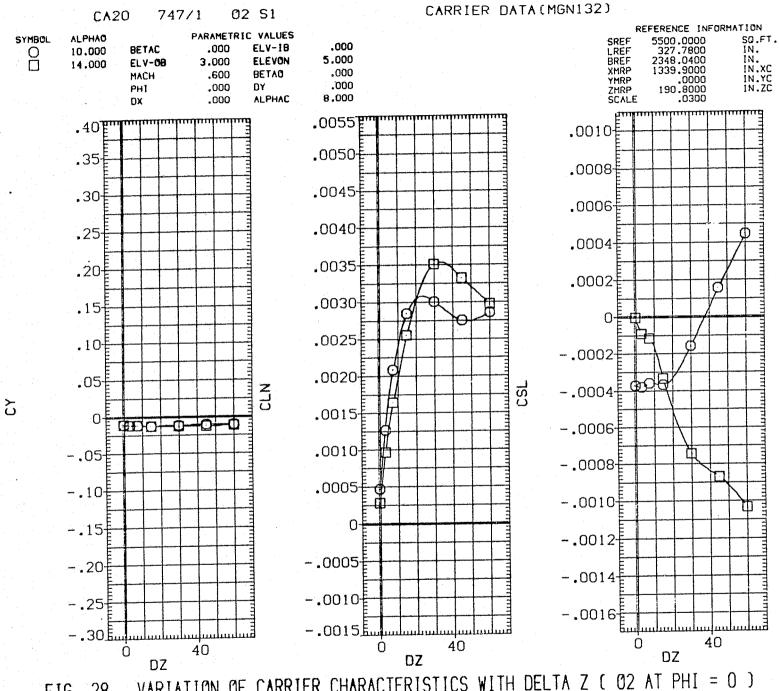
VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE 1381

DZ

DZ

0

DZ



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) 28 FIG PAGE 1382

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) 1383 PAGE

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1384

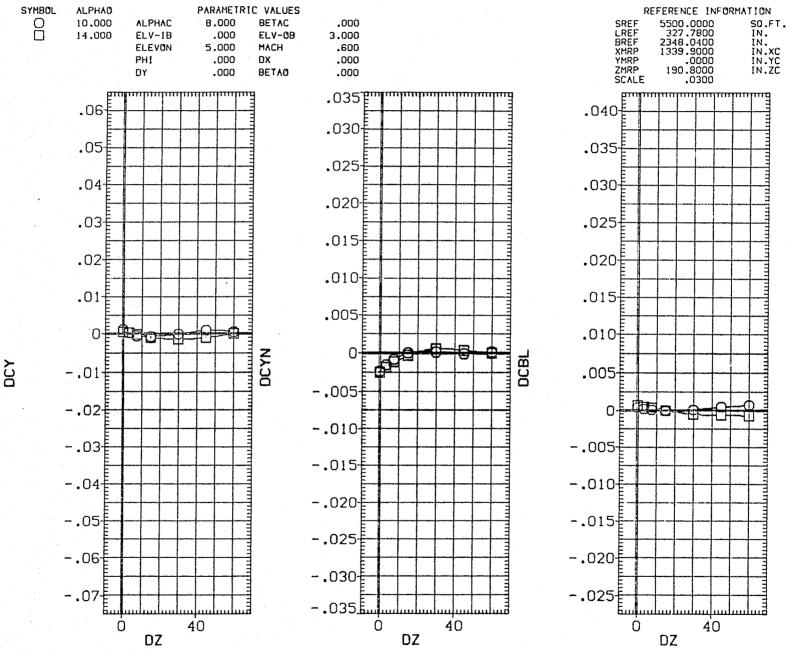


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1385

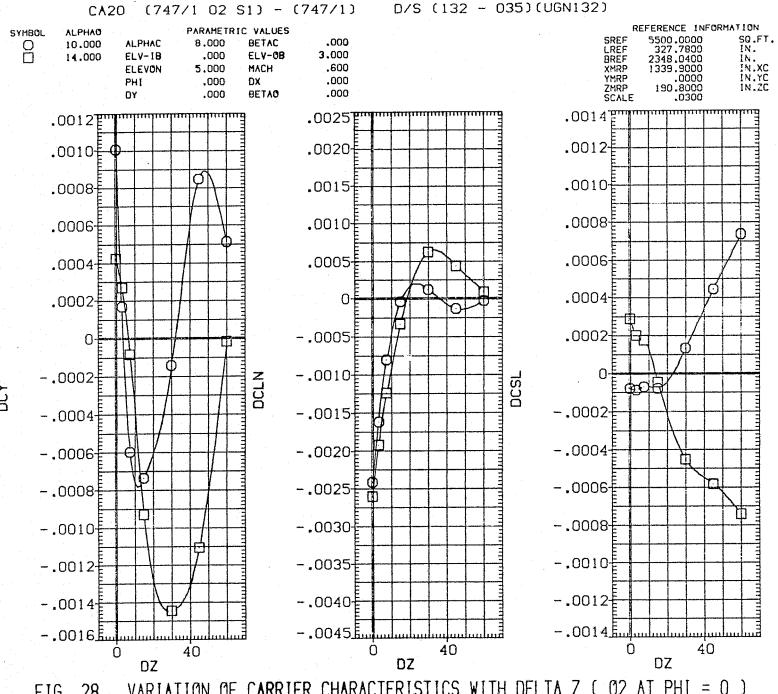


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1386

PAGE 1387

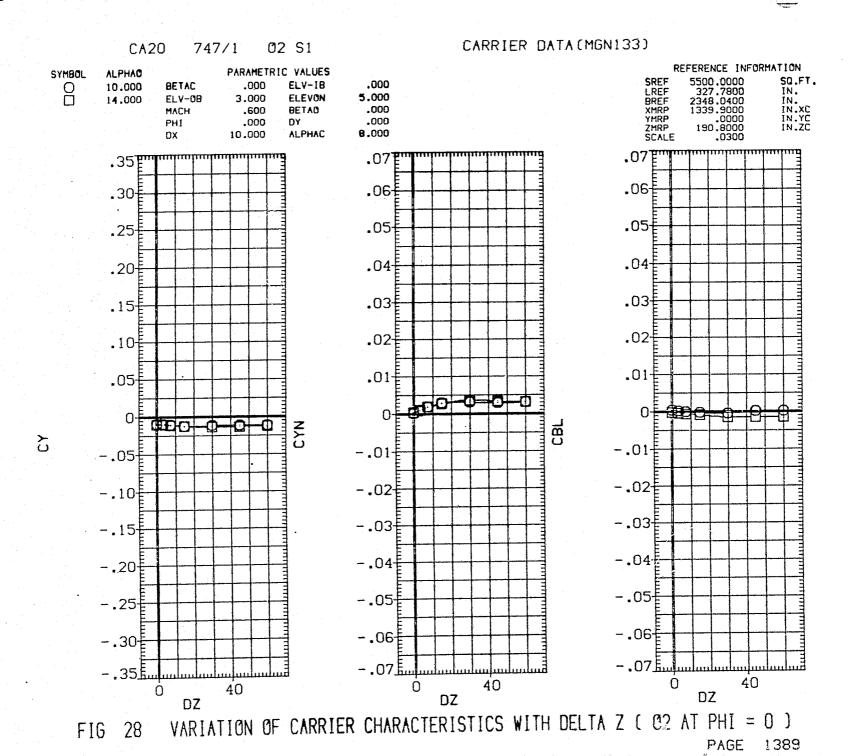


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1390

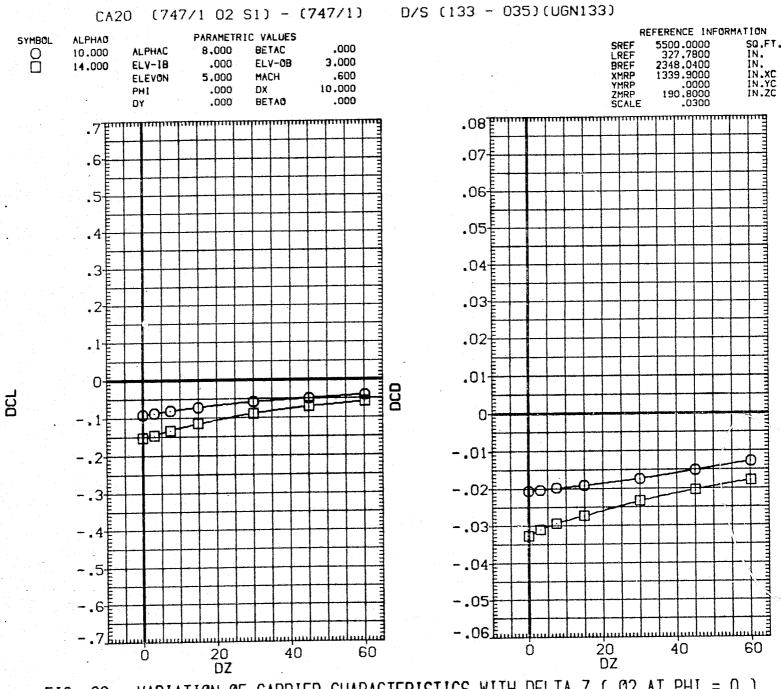


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1391

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT, FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

PAGE 1392

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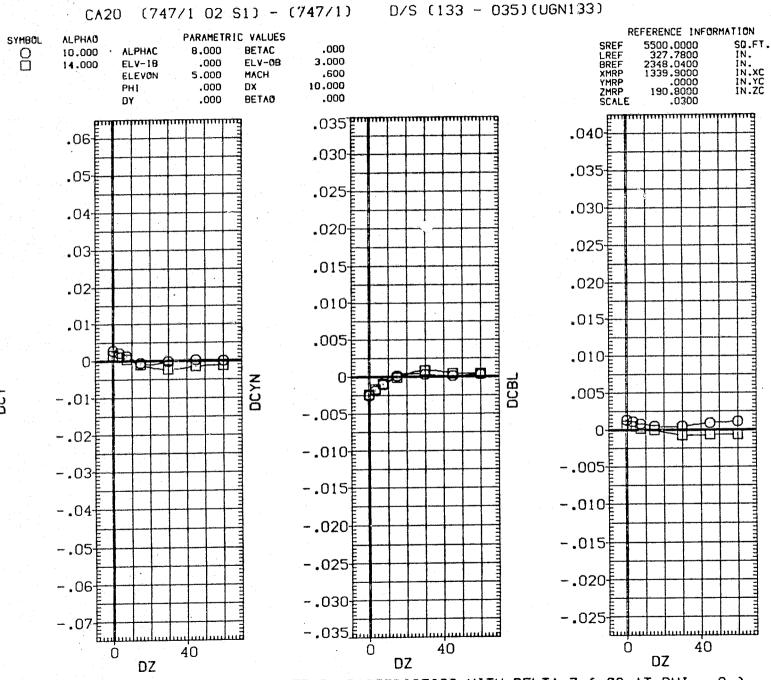


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1393

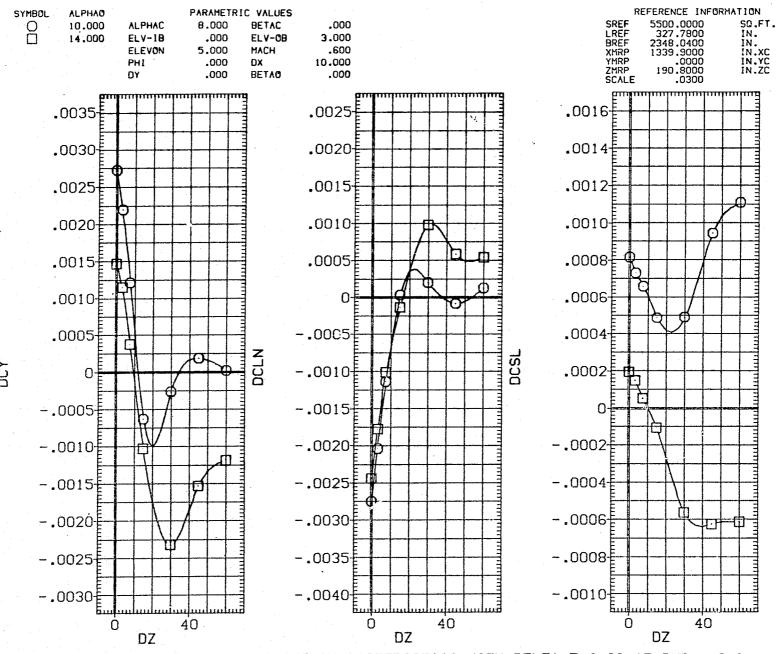


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1394

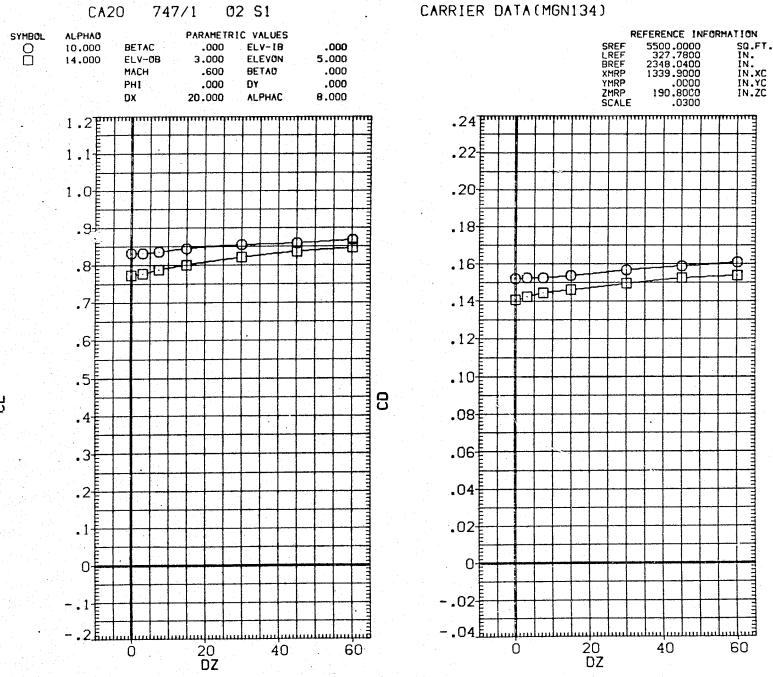


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 139

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1397

DZ

0

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DZ

40

-.07<u>L</u>L

-.35<u>k</u>...

0

DZ

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FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1398

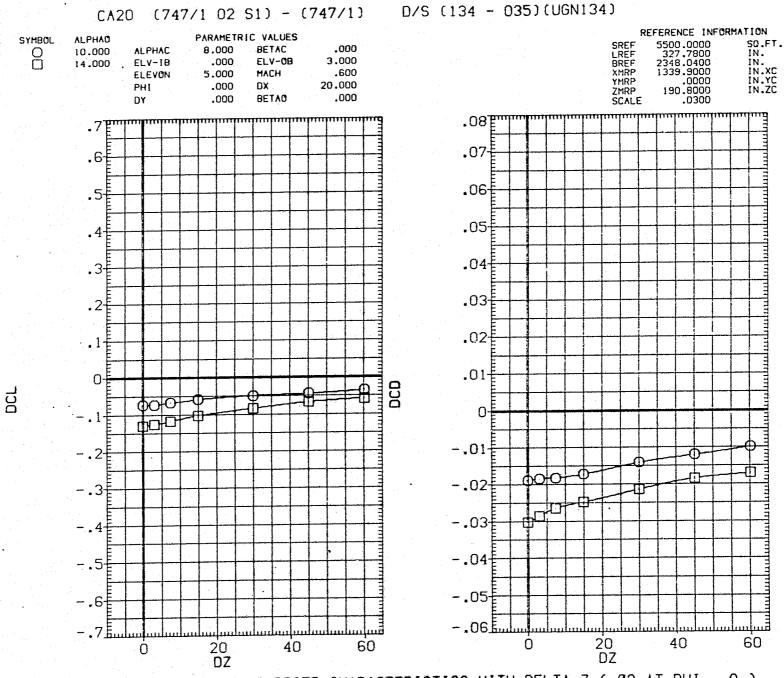


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1399

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT.

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

PAGE 1400

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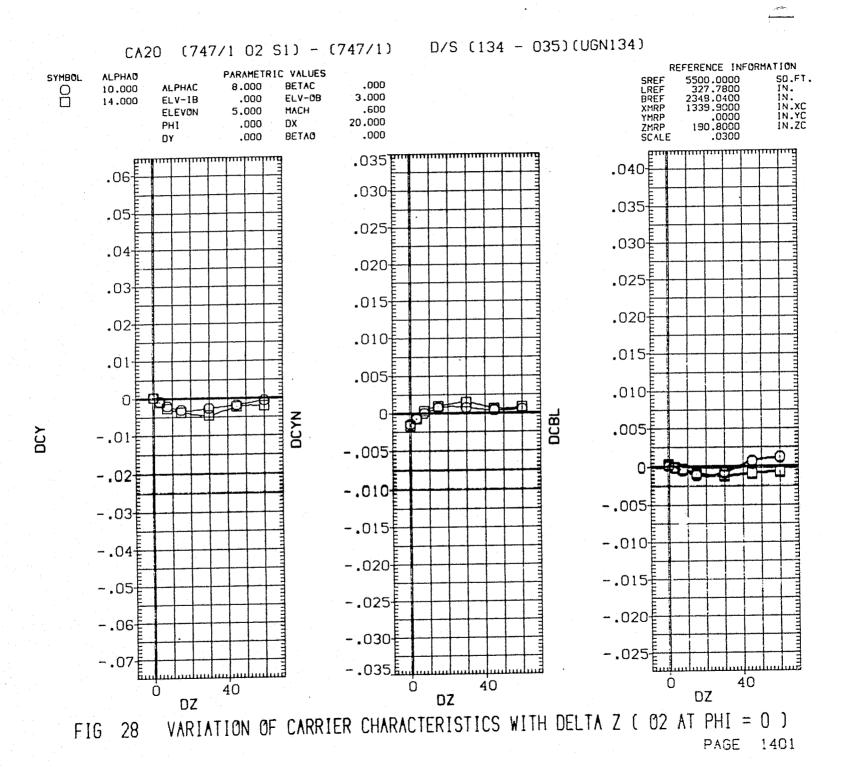
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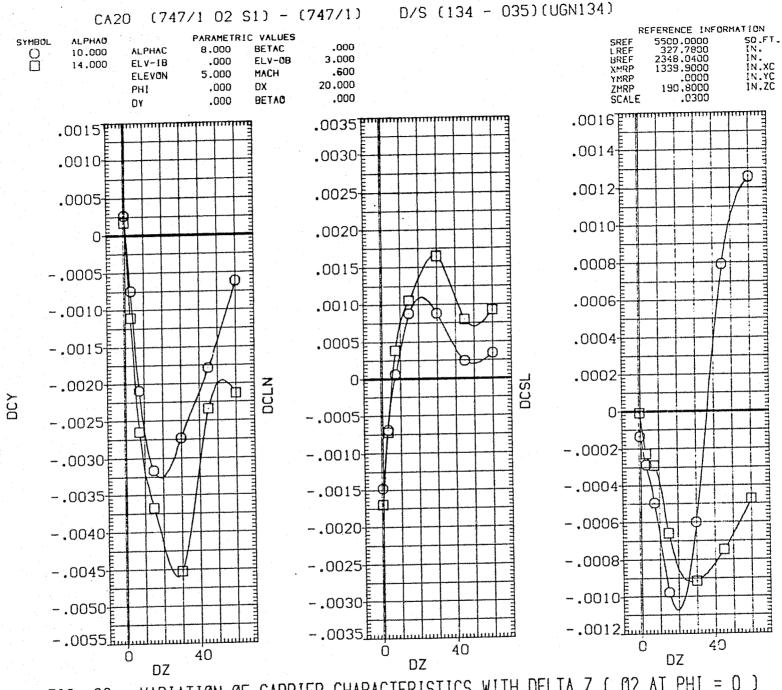


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1402



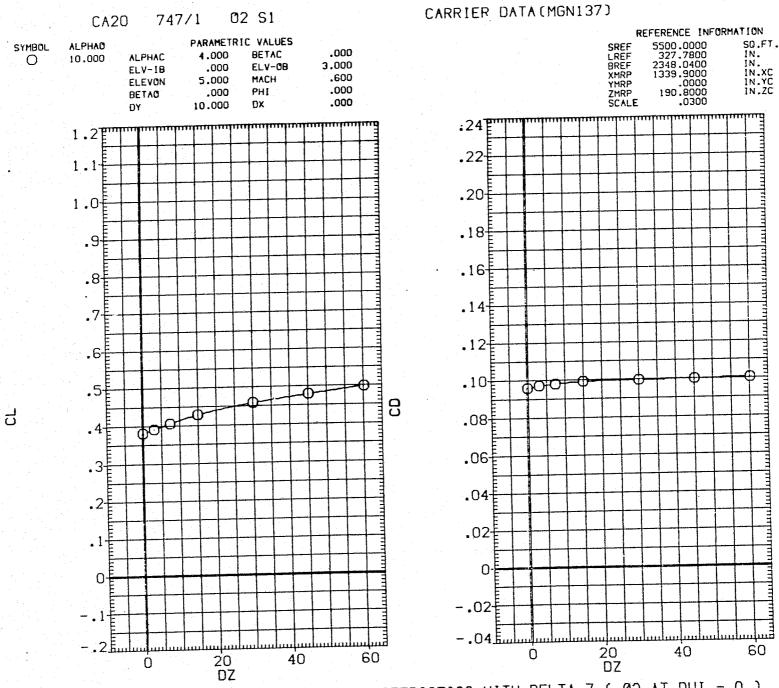


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1403

PAGE

FIG PAGE

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1407

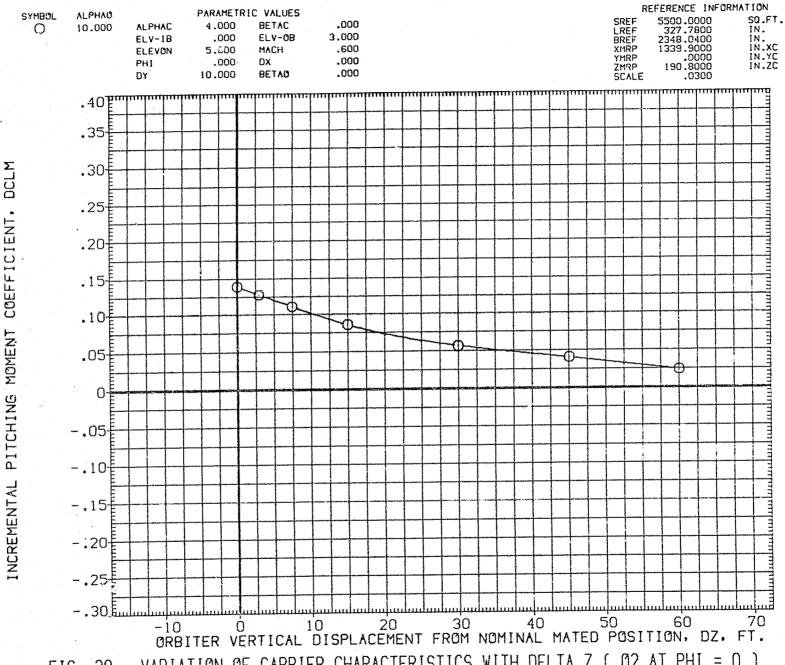


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1408

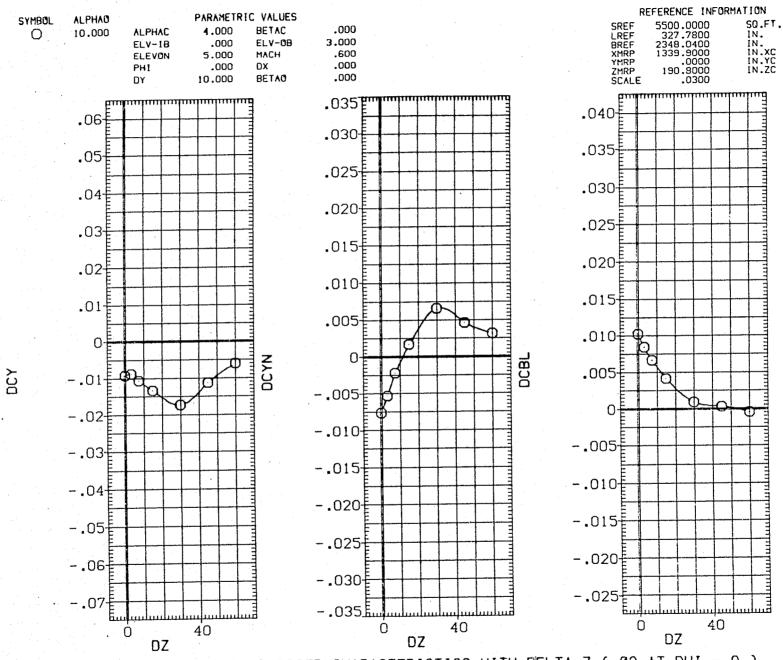


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

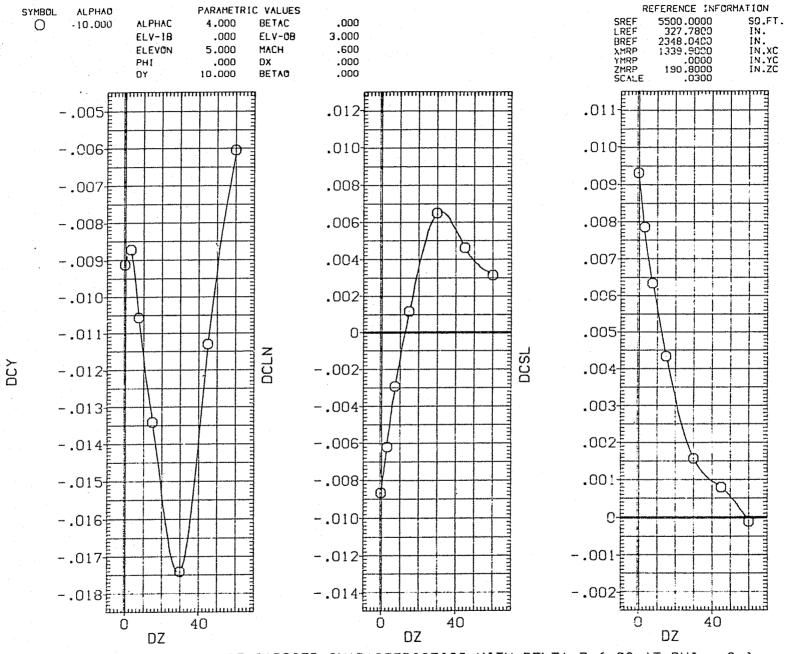


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1411

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 14:2

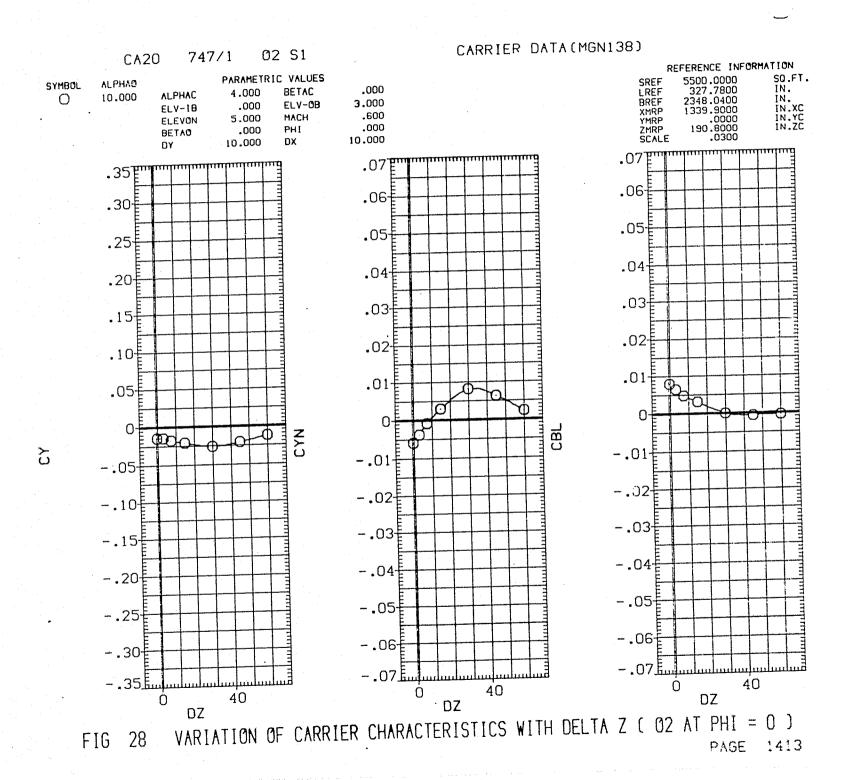


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

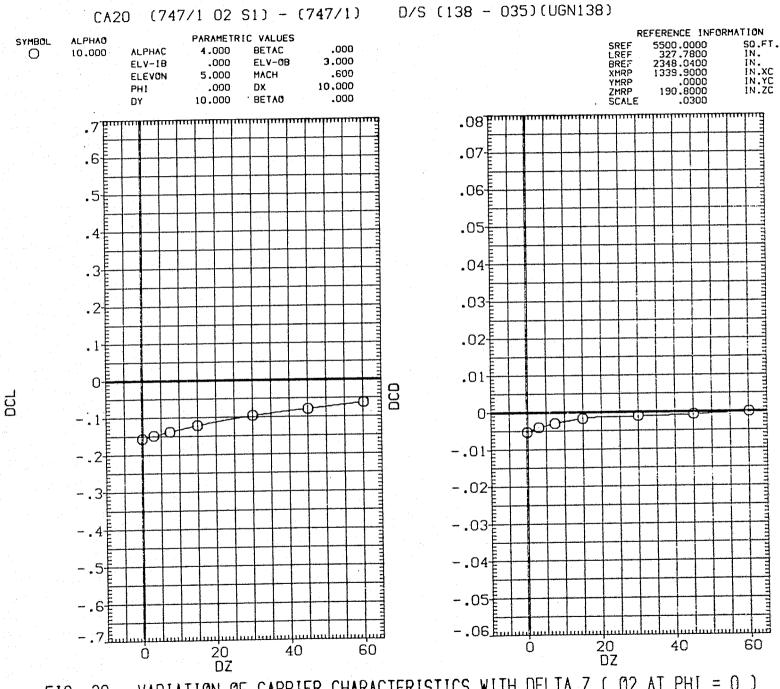
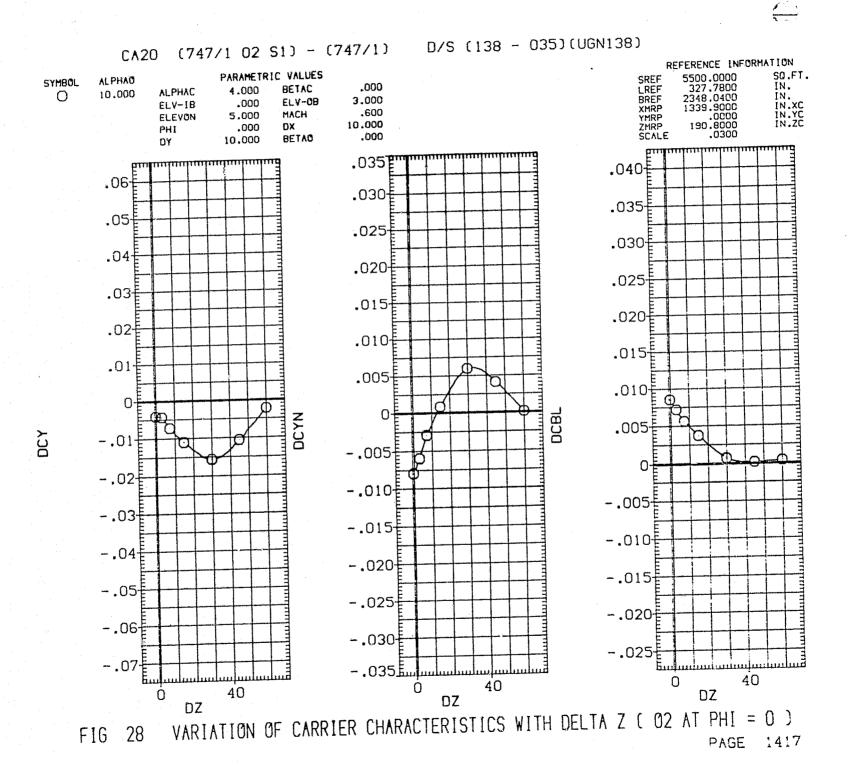


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1415

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG 28 PAGE



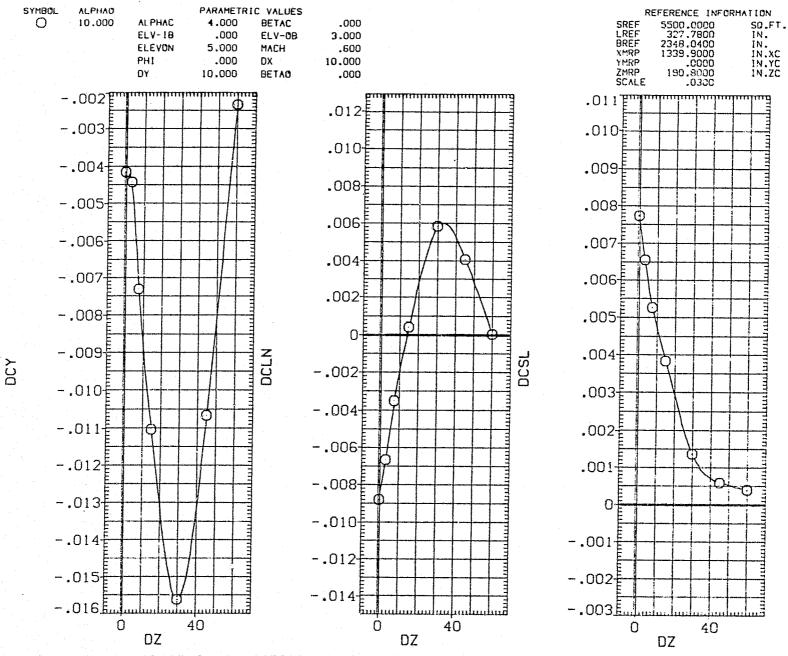
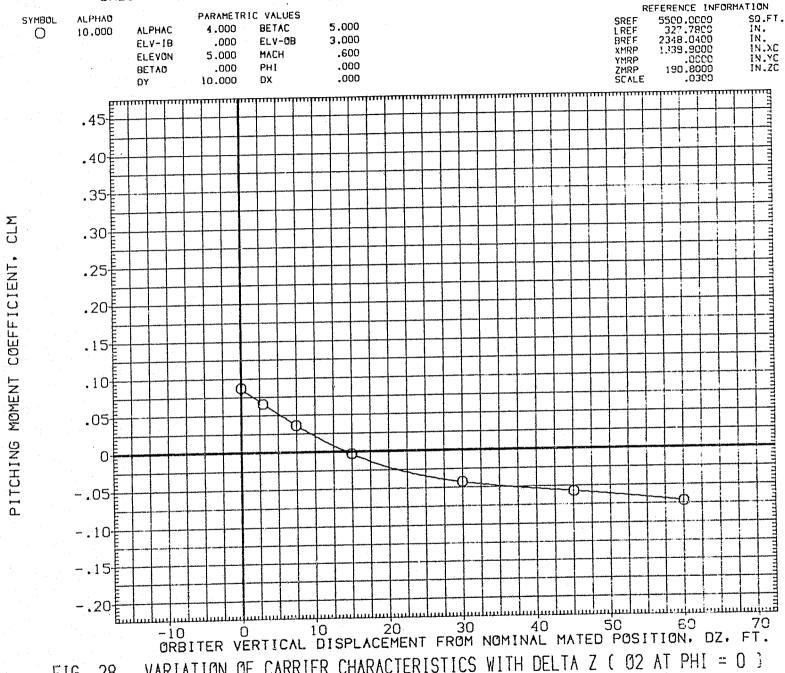


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

PAGE :418

PAGE

FIG



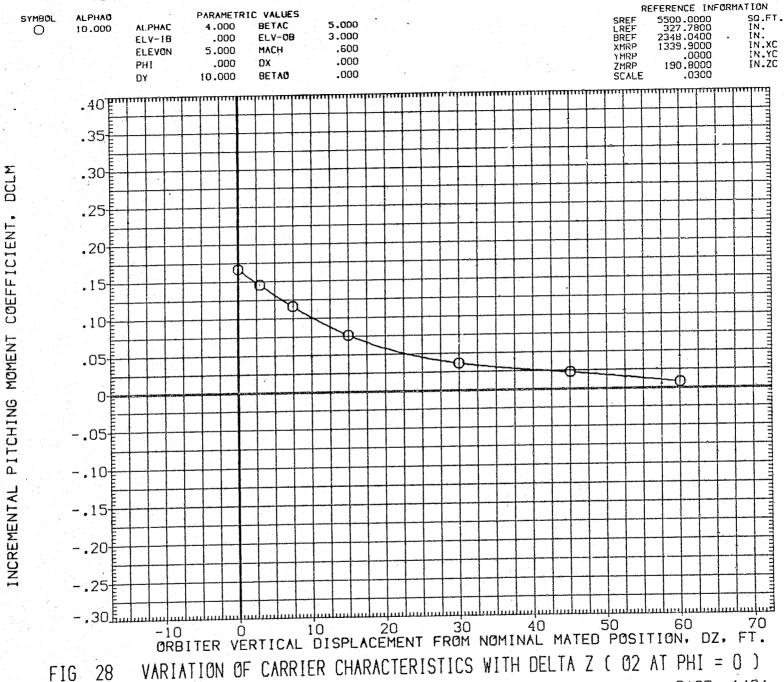
VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG 28 PAGE

PAGE

1421

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1422

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1423



PAGE 1424

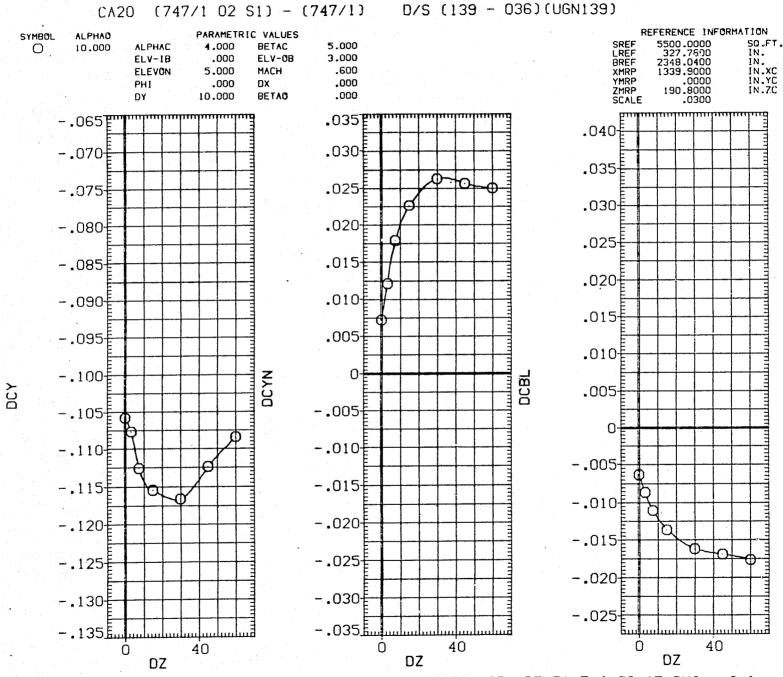


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1425

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 32 AT PHI = 0 ) FIG 28 PAGE 1426

DZ

747/1 02 S1 CA20 CARRIER DATA (MGN140) PARAMETRIC VALUES SYMBOL **ALPHAO** REFERENCE INFORMATION 5500.0000 327.7800 2348.0400 1339.9000 .0000 190.8000 .0306 0 10.000 **ALPHAC** 4.000 BETAC SREF LREF SQ.FT. 5.000 **ELV-18** .000 ELV-0B 3.000 BREF IN. ELEVON 5.000 .600 XMRP YMRP ZMRP SCALE IN XC IN YC IN ZC BETAO .000 PHI .000 DY 10.000 DX 10.000 1.2 Emphalmanamana .22 1.1 1.0 .20 .18<del>[</del> .8<del>[</del> .16 .14 •6<del>-</del> .12 .10<del>[</del> 김 900 .08 .06 .04 .02 -.02<del></del> -.04点 20 DZ 20 DZ 60 40 40 60

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) PAGE 1428

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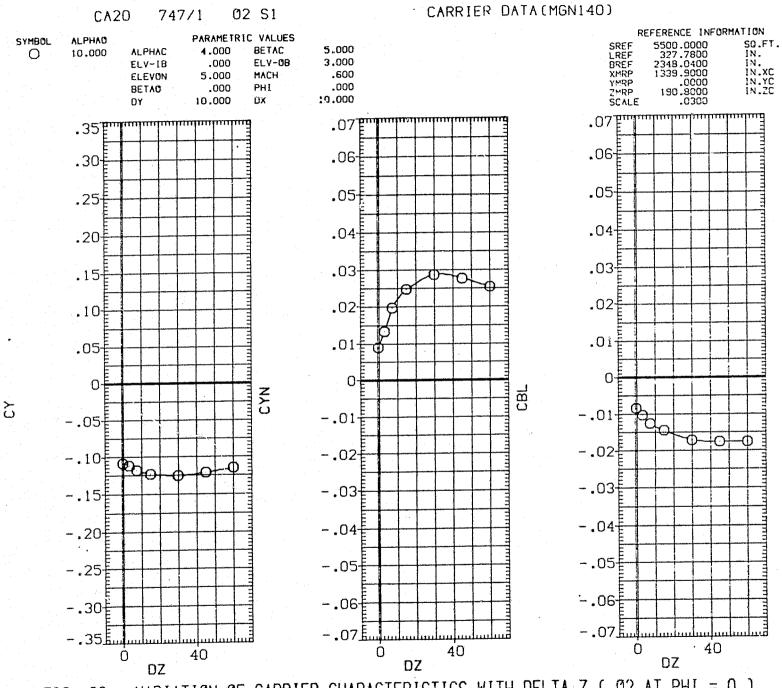


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1430

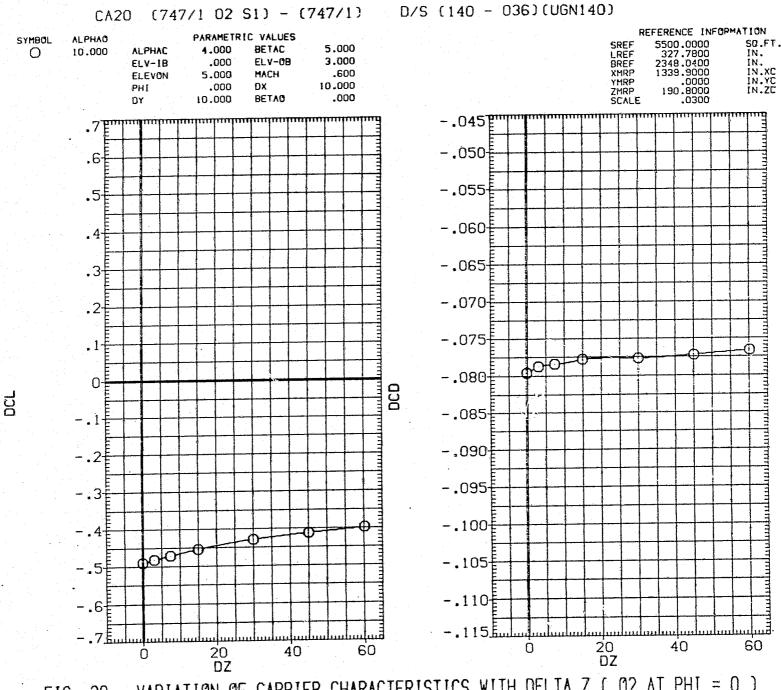
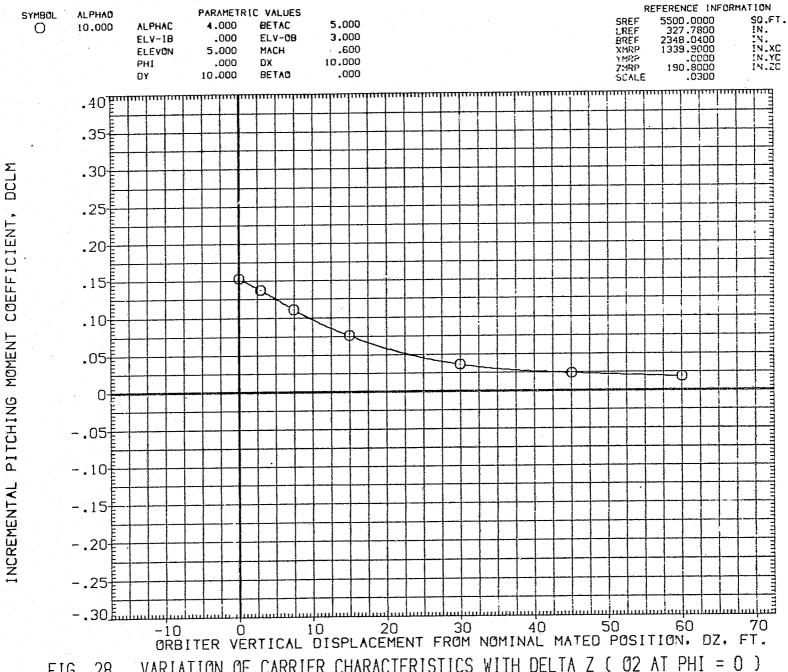


FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1431



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

DZ VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE 1433

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-.025

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DZ

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FIG 28 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1434

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DZ

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DZ

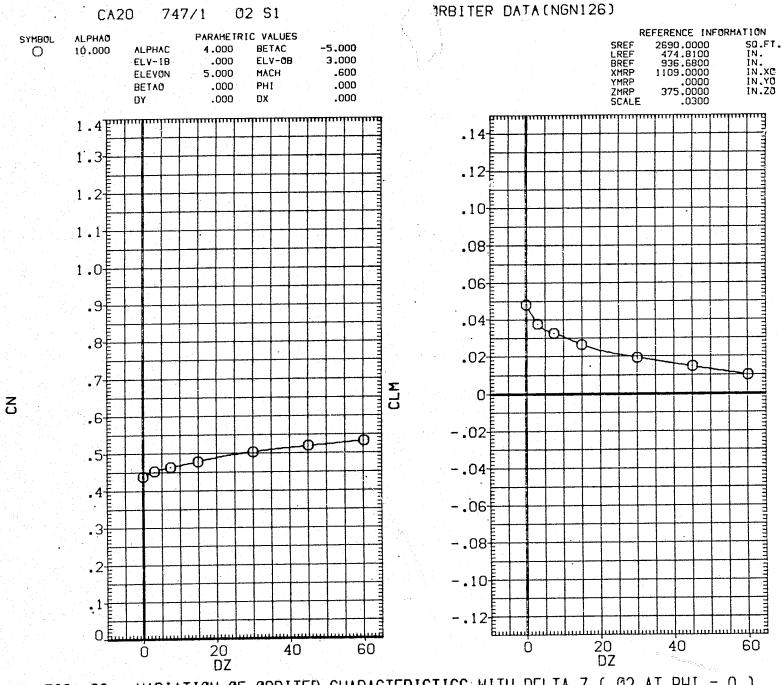
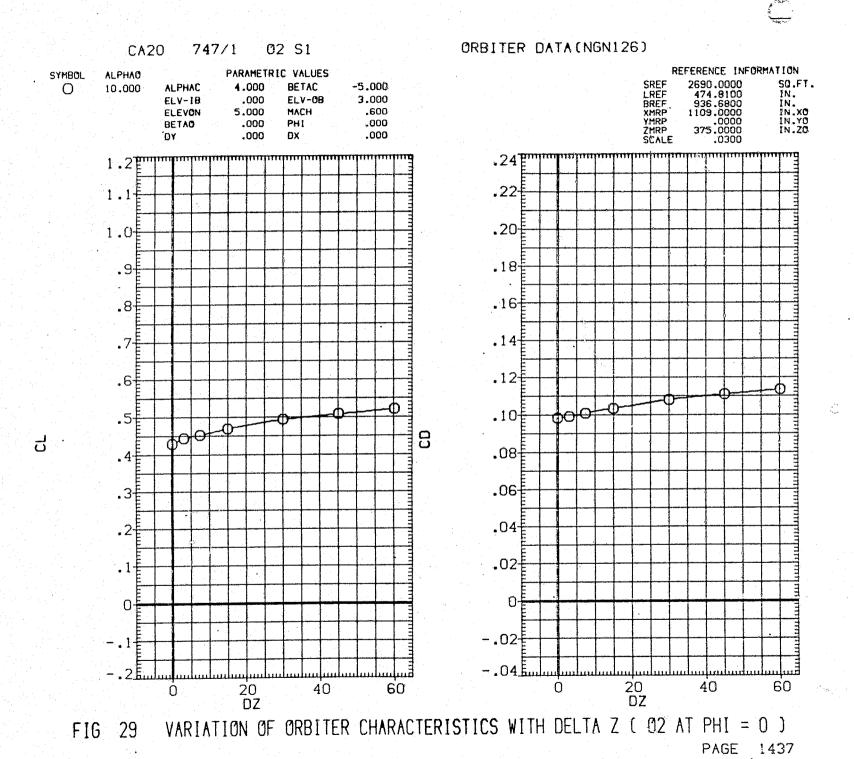


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1435

rangangula gayra ku gang su agga dikang ing na ung palkot dikang bibang na kikala su di na dikang bibang pilaba

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1436



<mark>taniga pamanga panggang</mark>a pangganga pangga panggan penggangan panggangan penggangangan ang penggangan panggangan

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1438

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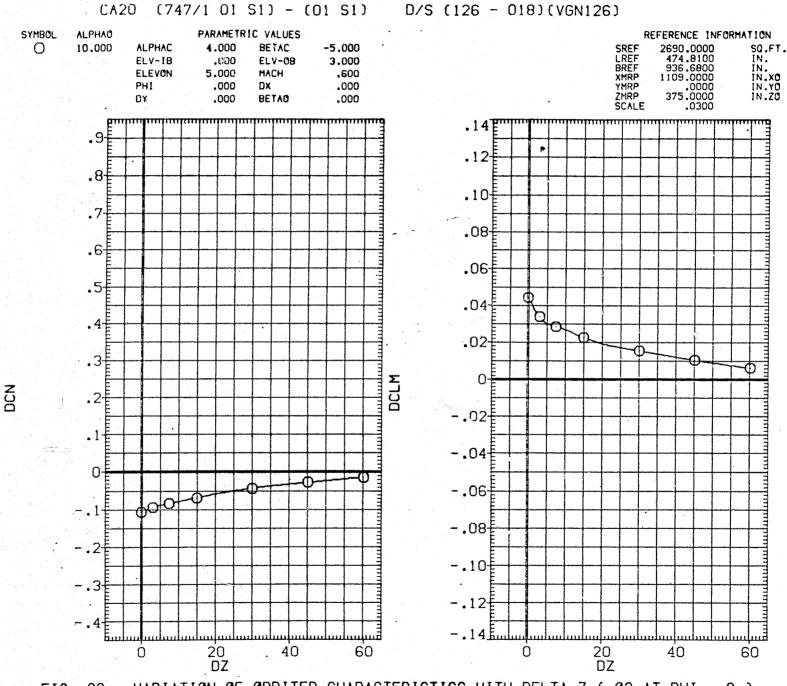


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1440

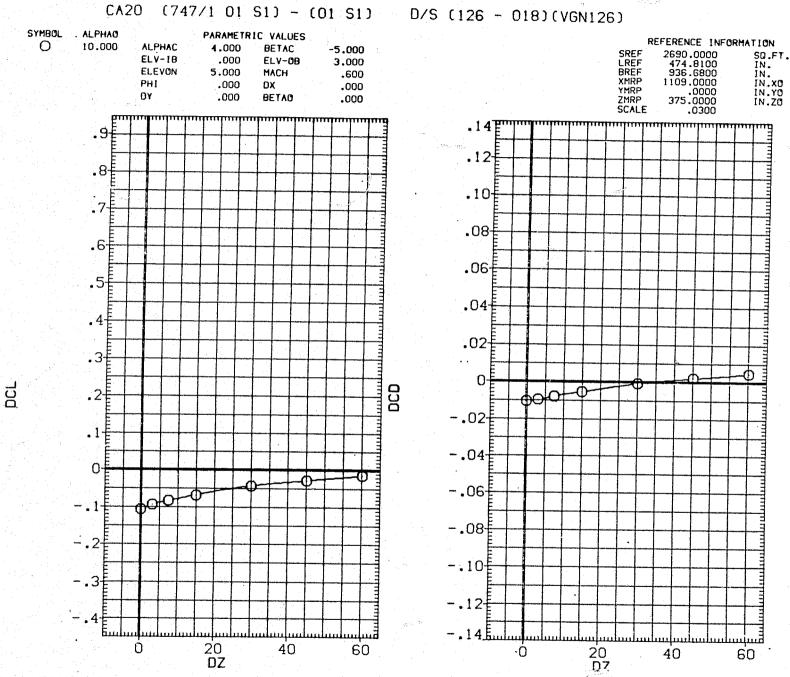
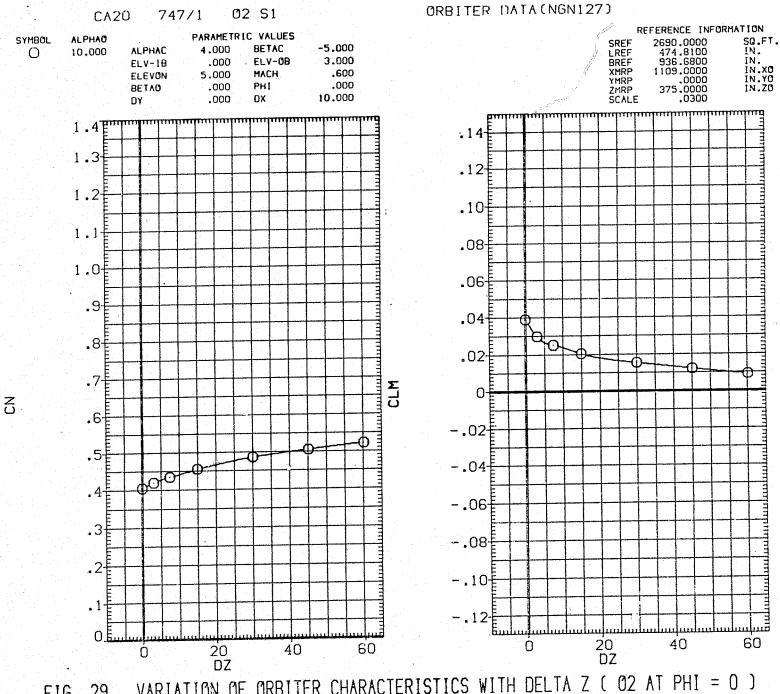


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1441



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) 29 FIG PAGE 1442

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1444

SYMBOL

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DCN

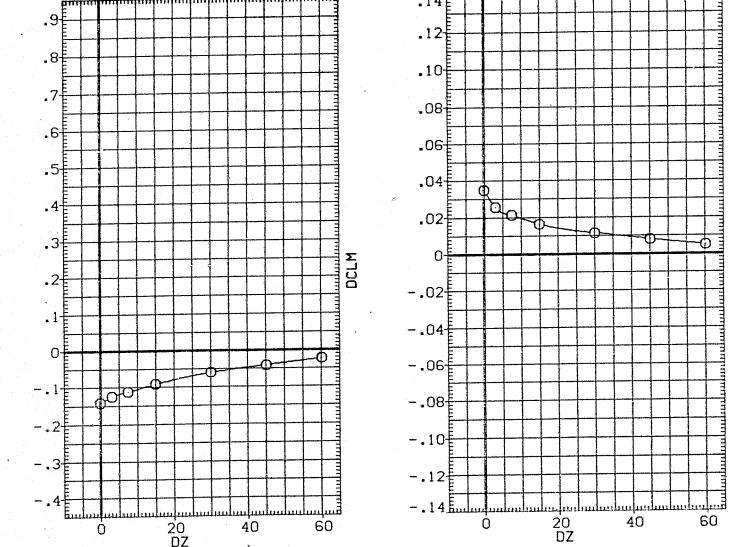


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1446

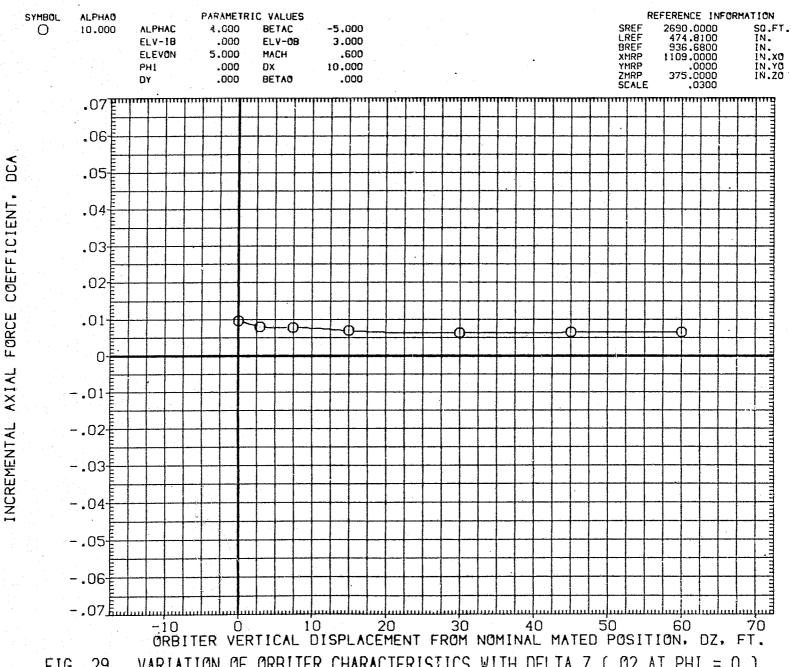


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1447

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1448

60

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20 DZ 20 DZ 60

FIG 29 VARÍATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1449

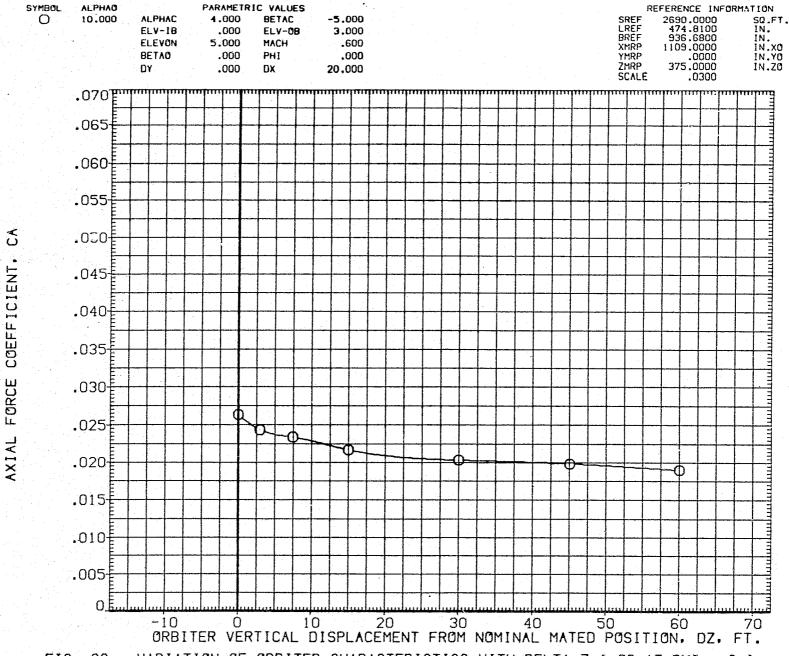
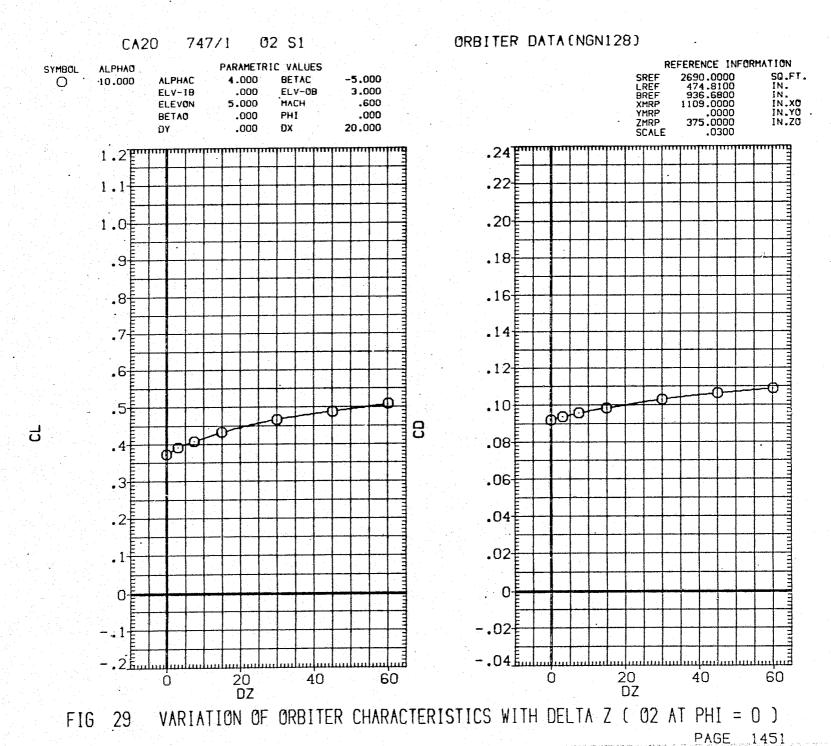


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1450





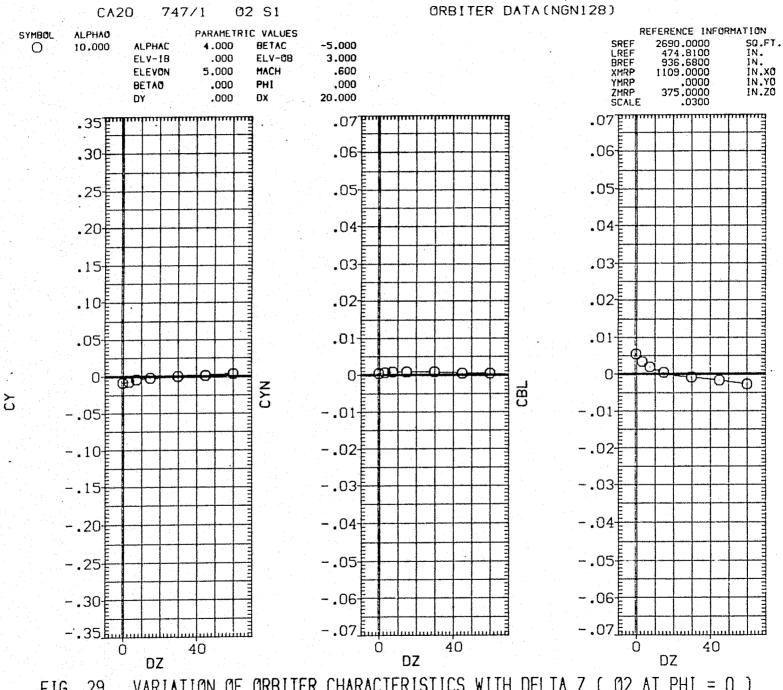
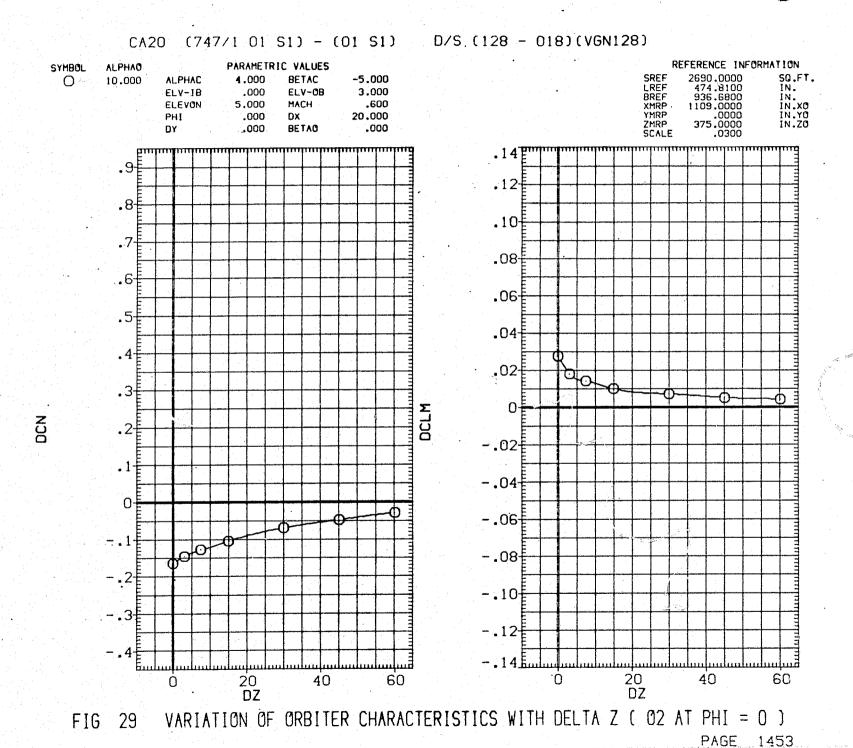


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 145





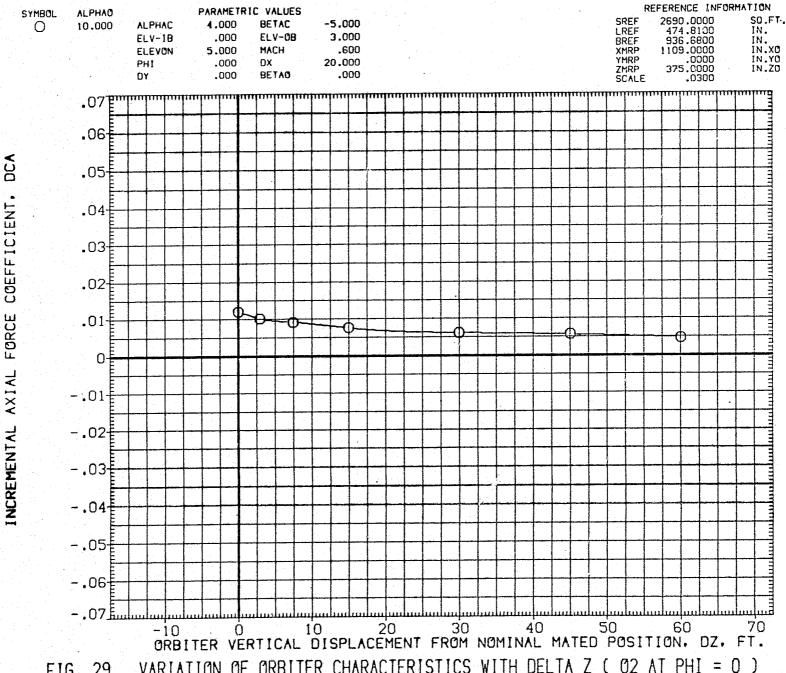


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1454

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1455

CA20 747/1 02 S1

## ORBITER DATA (NGN135)

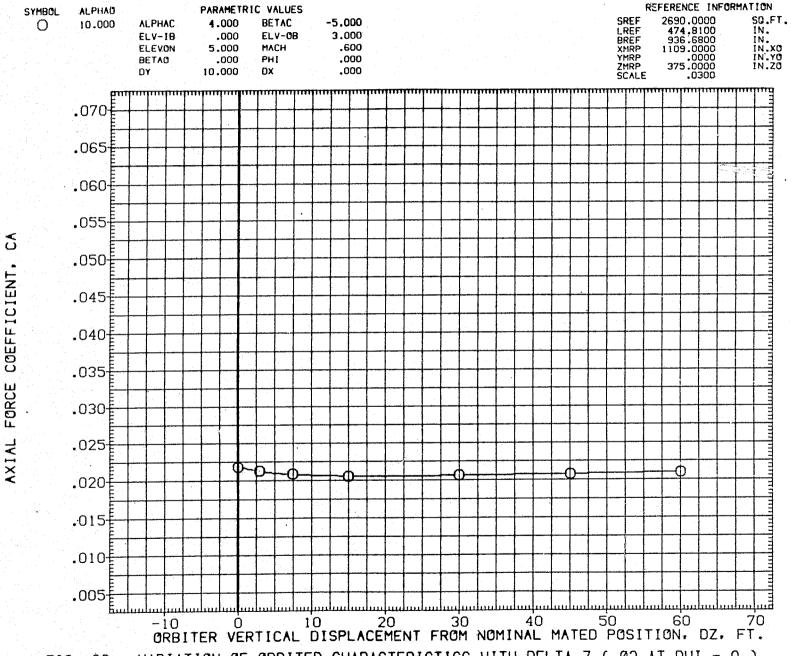
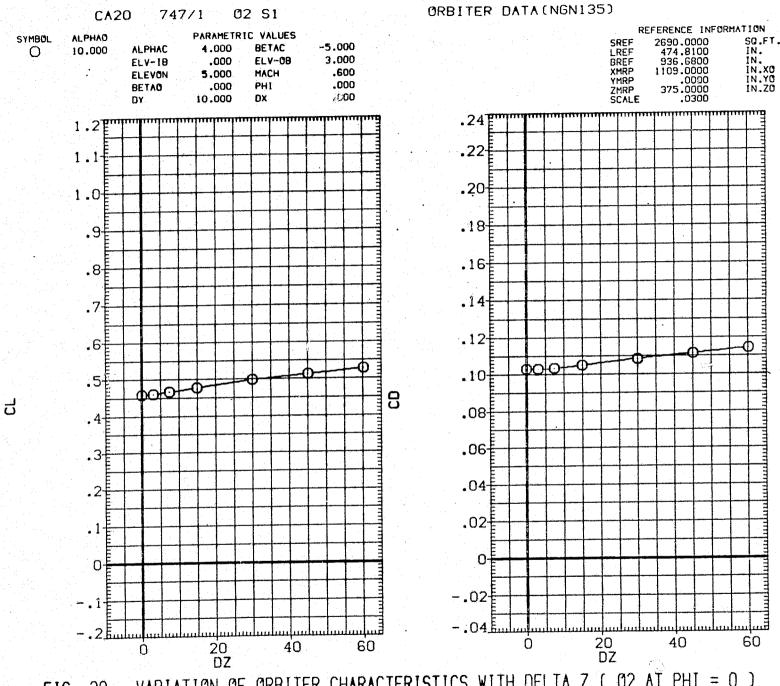


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1457



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE

747/1 02 S1 ORBITER DATA (NGN135) **CA20** REFERENCE INFORMATION PARAMETRIC VALUES ALPHAO SYMBOL 2690.0000 474.8100 SQ.FT. 4.000 BETAC -5.000 Ó 10.000 **ALPHAC** IN. LREF 3.000 ELV-IB ELV-0B .000 936.6800 1109.0000 .0000 IN. IN.XO IN.YO IN.ZO BREF ELEVON 5.000 MACH .600 XMRP YMRP .000 .000 PHI BETAO ZMRP SCALE 375.0000 DY 10.000 ÐΧ .000 .0300 .07 բարարարարա .07F .35En .06 .06<del>-</del> .30 .05 .05 .25 .04 .20 .04 .03<del>[</del> .03 .15 .02 .02 .10 .01 .01 .05<del>‡</del> (B) CBL S -.01 -.01<del>[</del> -.05 -.02 -.02<del>[</del> -.10<del>[</del> -.03 -.15<del>[</del> -.03 -.04 -.04 -.20<del>[</del> -.05<del>[</del> -.05<del>{</del> -.25<del>[</del> -.06 -.06 -.30<del>[</del> - .07 <u>L</u>... - .07 盂 -.35<u>E</u>. 40 40 Ó 40 0 Ó DΖ DZ DZ VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG

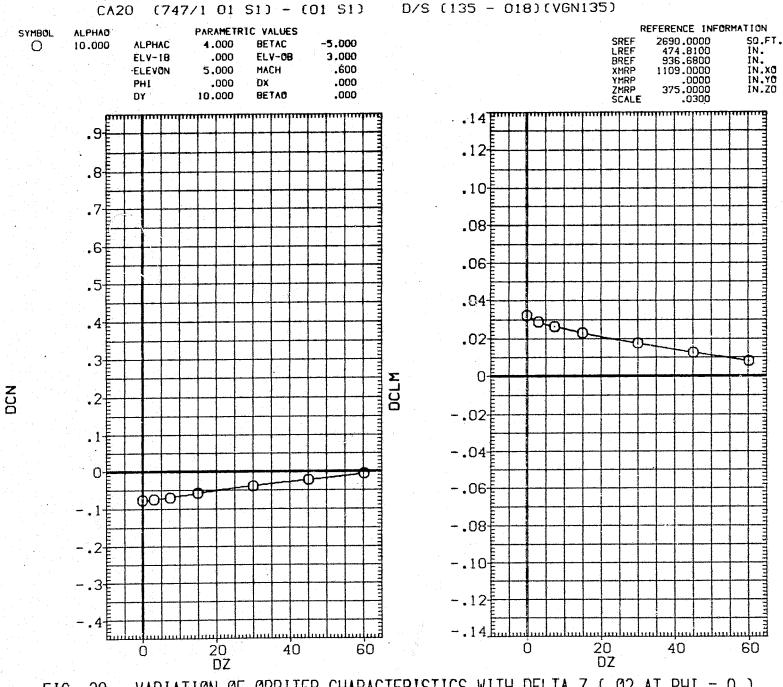
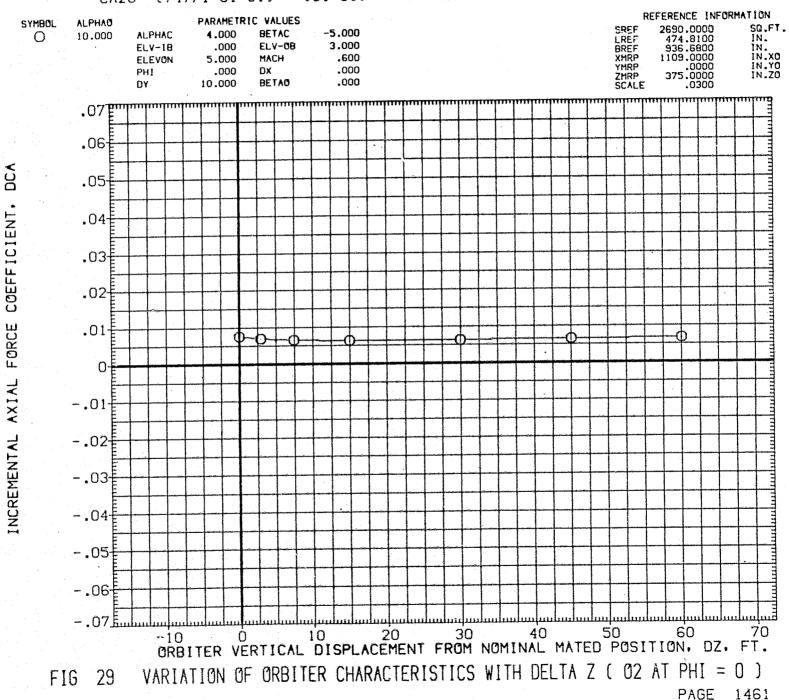


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1460



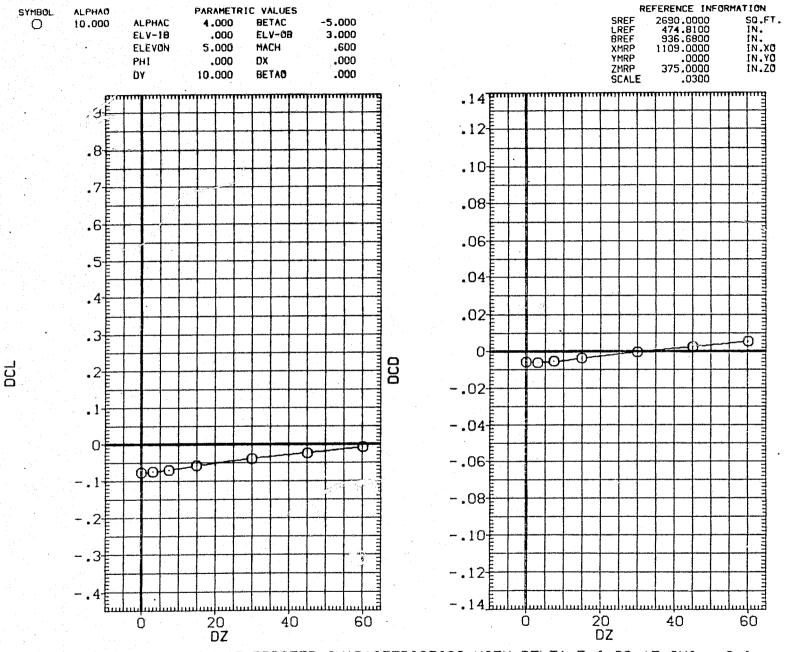
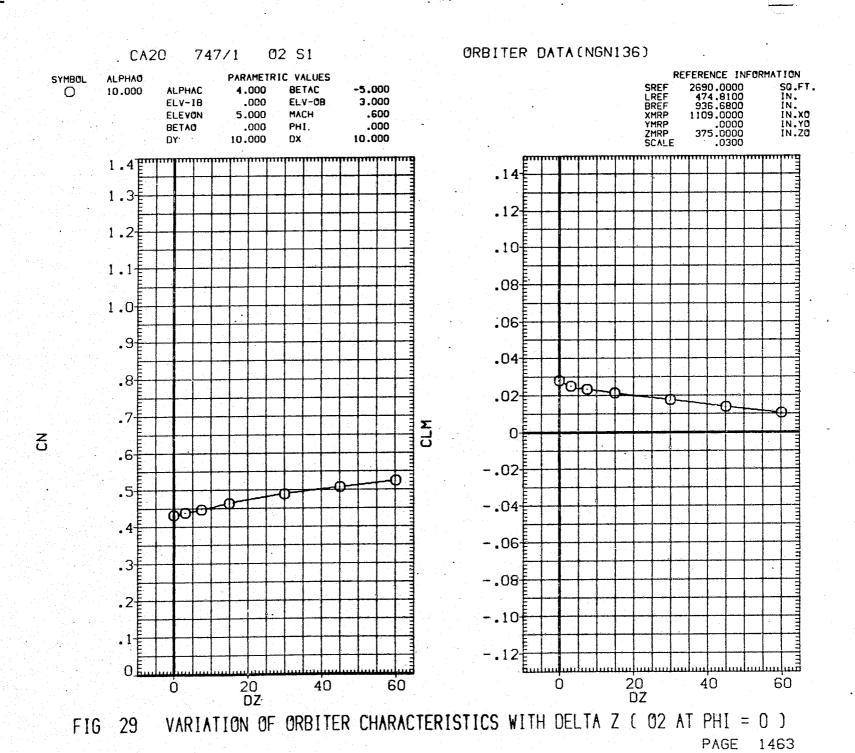
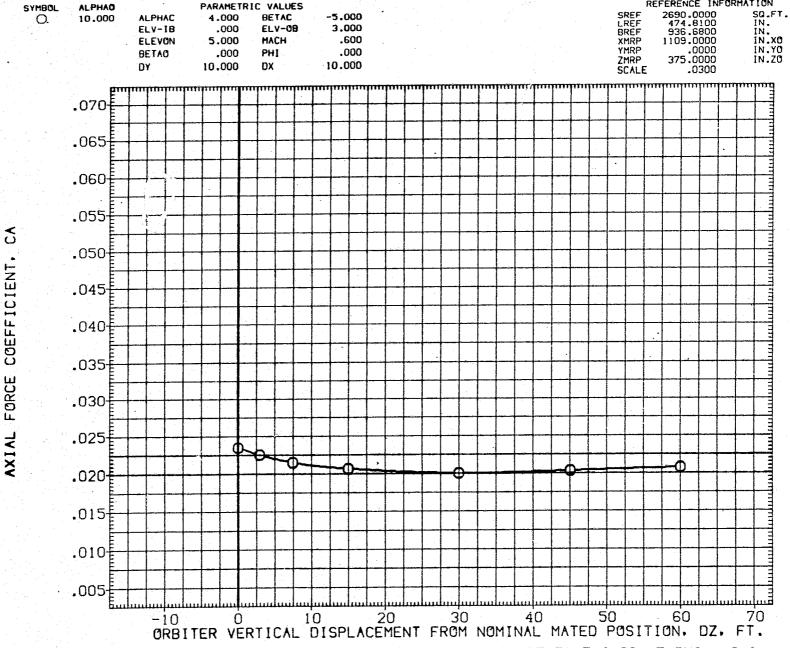


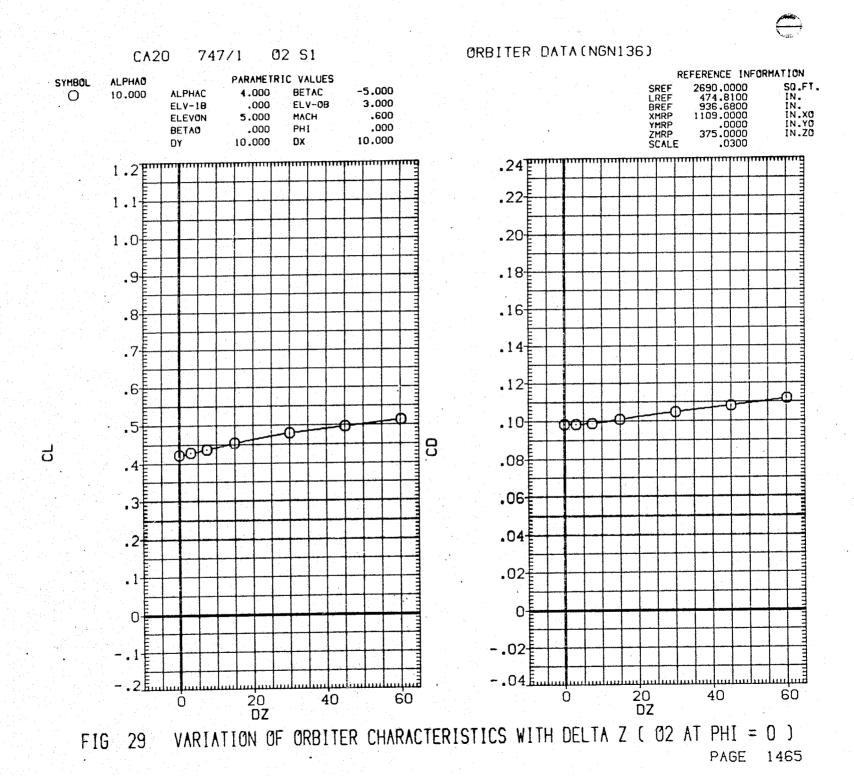
FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1462



REFERENCE INFORMATION



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG 29 PAGE 1464



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG

PAGE 1466 CA20 (747/1 01 S1) - (01 S1) D/S (136 - 018)(VGN136)

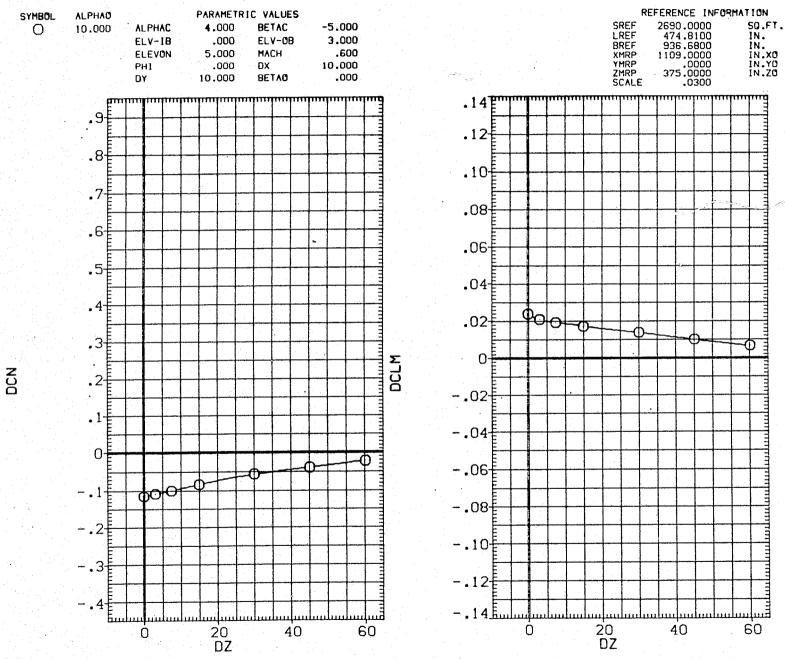


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1467

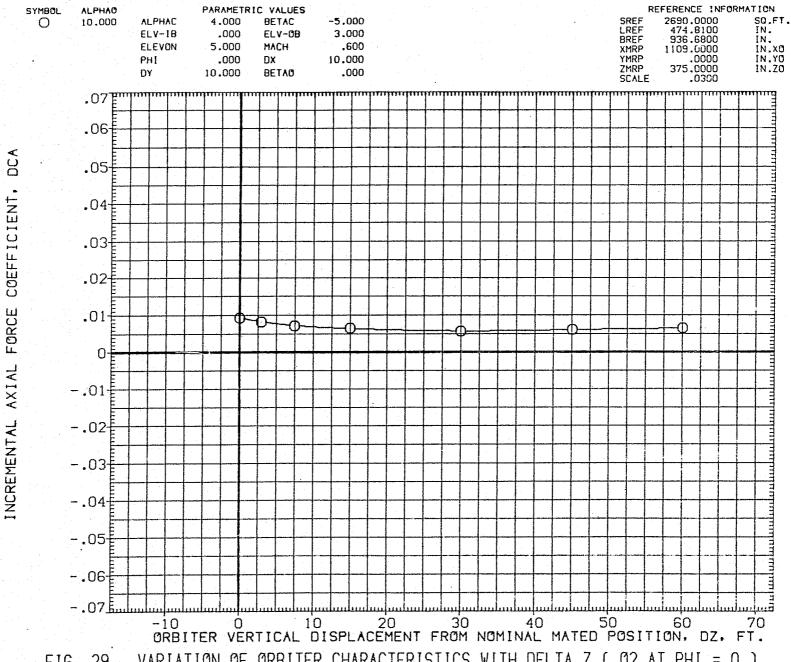


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1468

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) PAGE 1469

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1473

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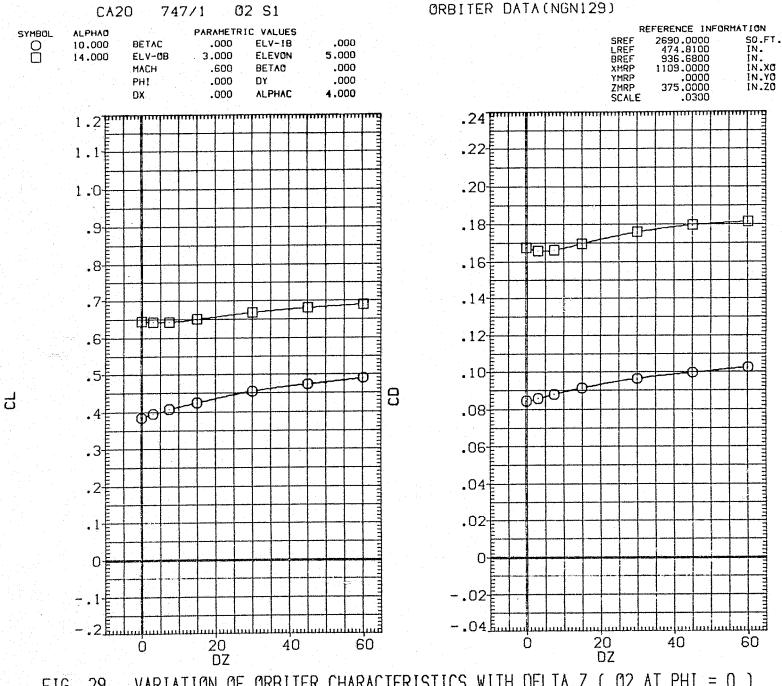
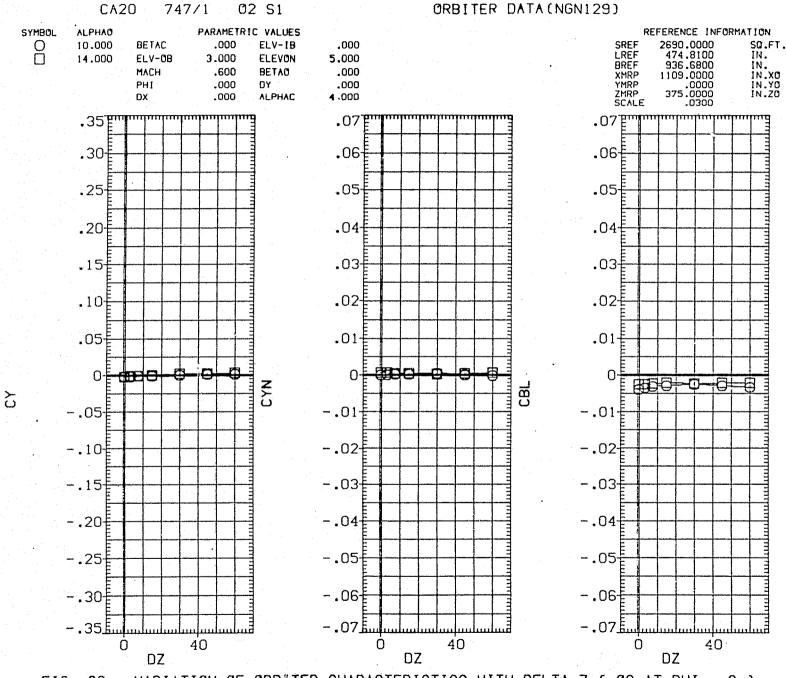


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1472



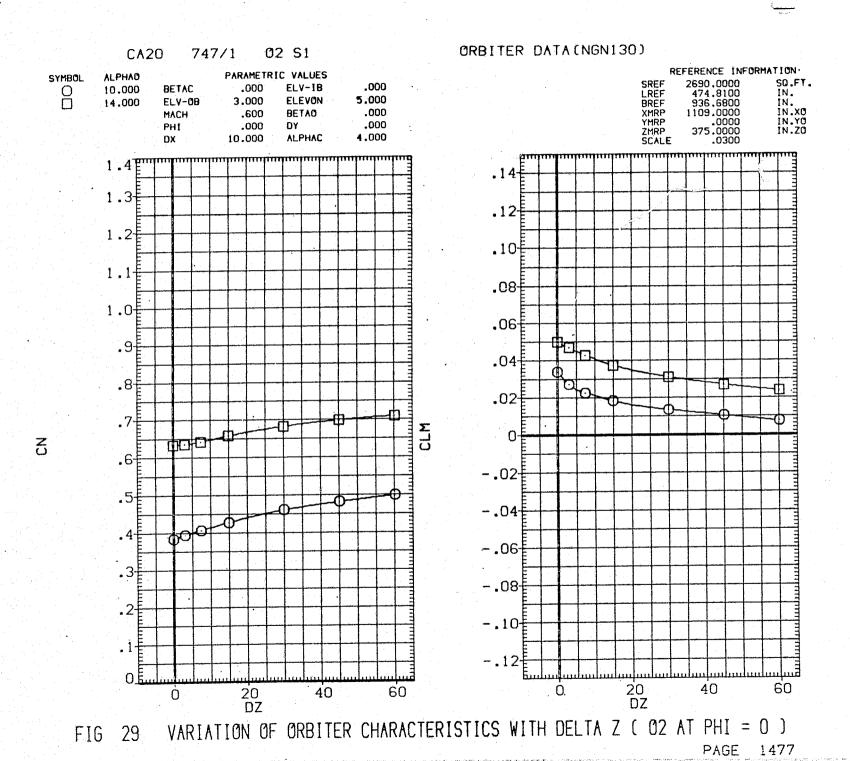


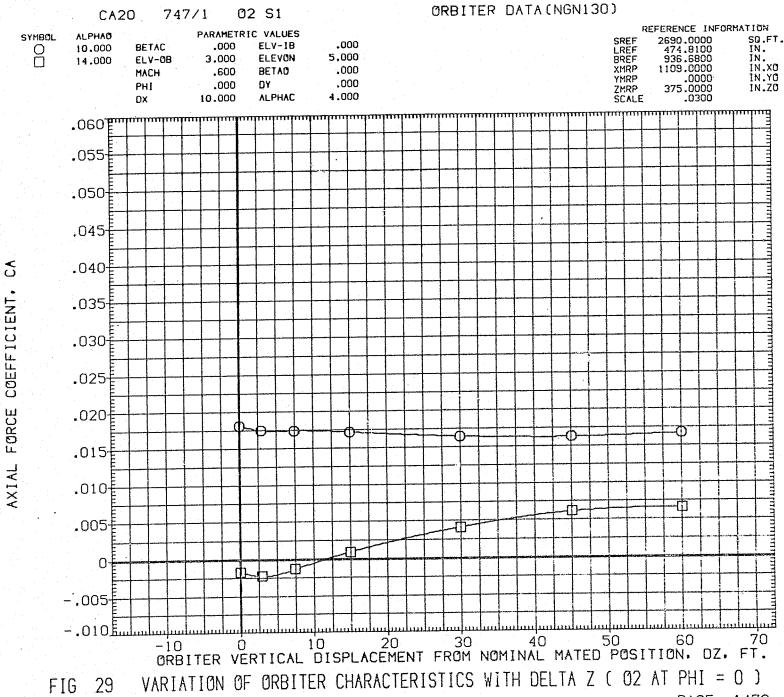
VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG

PAGE

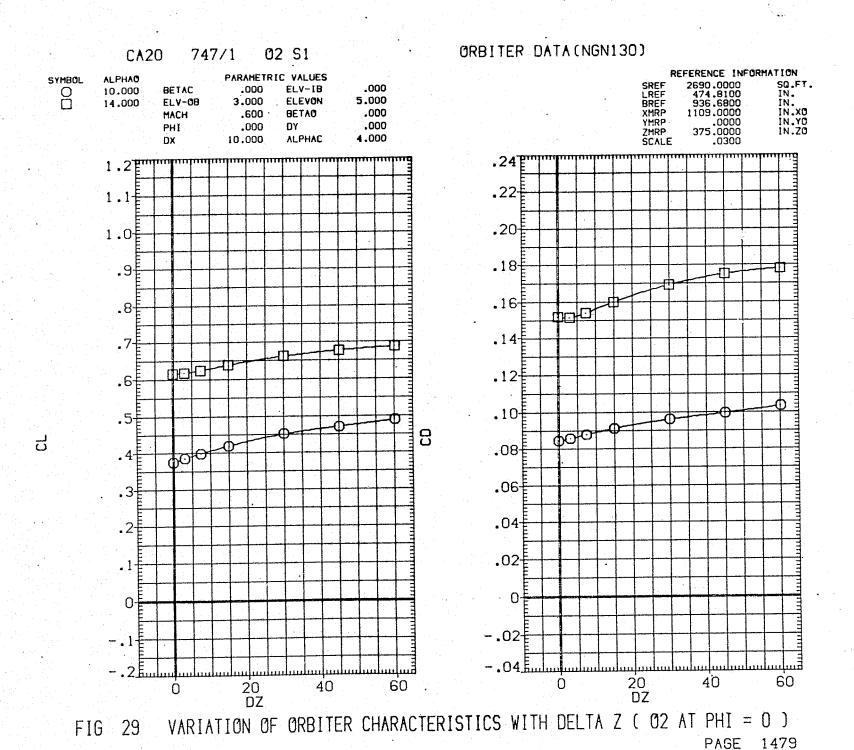
FIG PAGE

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1476





PAGE 1478



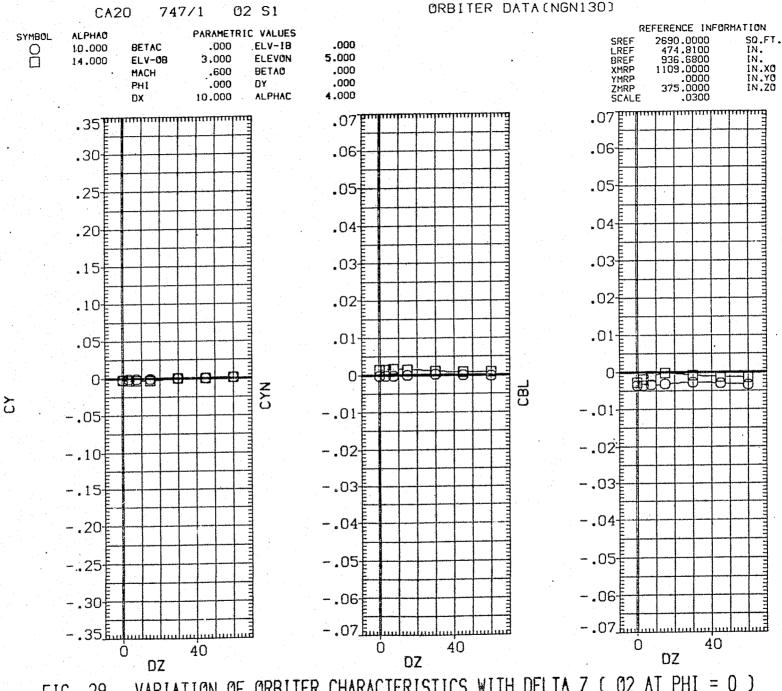


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

 $(747/1 \ 01 \ S1) - (01 \ S1)$  D/S (130 - 018)(VGN130)CA20 REFERENCE INFORMATION SYMBOL **ALPHAO** PARAMETRIC VALUES 2690.0000 474.8100 936.6800 1109.0000 SO.FT. SREF 4.000 0 BETAC .000 10.000 ALPHAC LREF IN. 3.000 .000 ELV-0B 14.000 ELV-IB IN.XO BREF MACH .600 ELEVON 5.000 XMRP YMRP .0000 IN.YO 10.000 .000 ĐΧ PHI 375.0000 ZMRP IN.ZO BETAO .000 .000 DY. SCALE .14 Emmin .9-.12 .8€ .10 .08 .6 .06<del>-</del> .04 .02<del>[</del> ╢ .3= DCLM 0 DCN .2= -.02<del></del> -.04<del></del> -.06 - .08 -.10 -.12<del>[</del> 20 DZ 40 60 60 Ò 20 DZ 40

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1481

SYMBOL

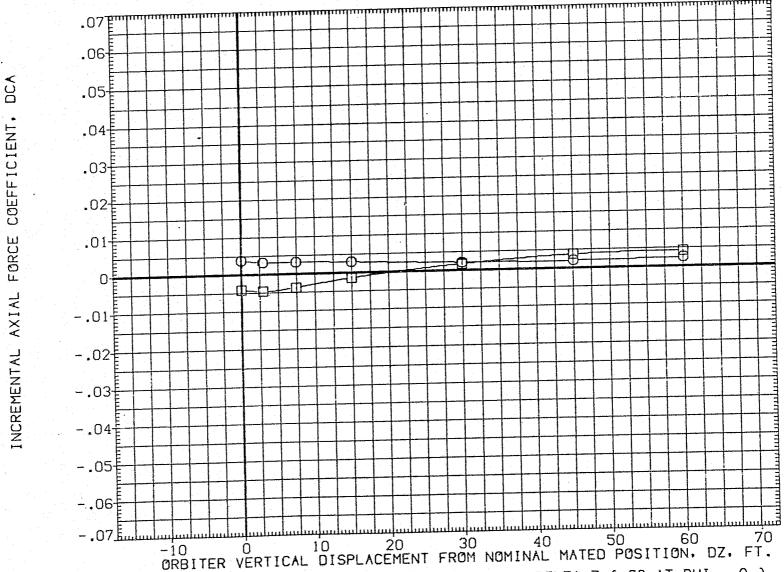


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1482

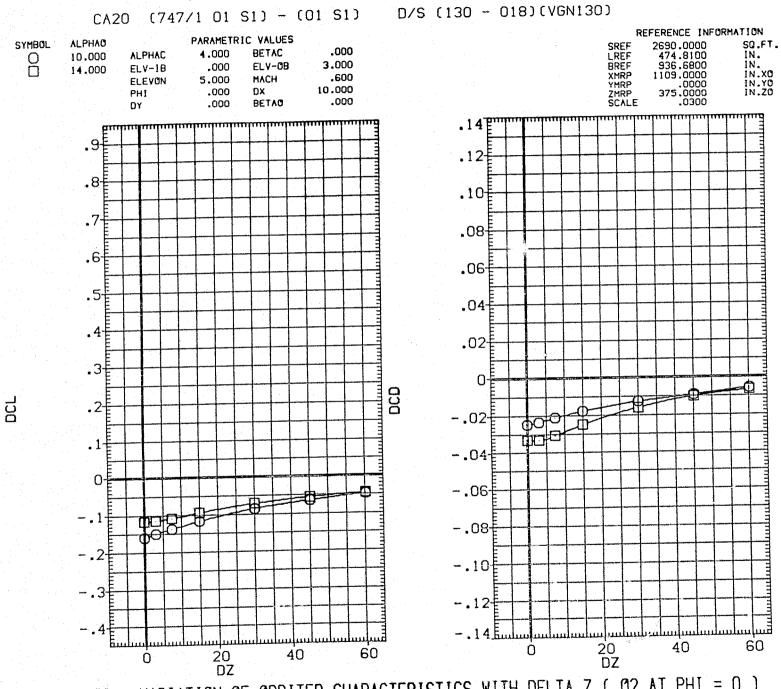
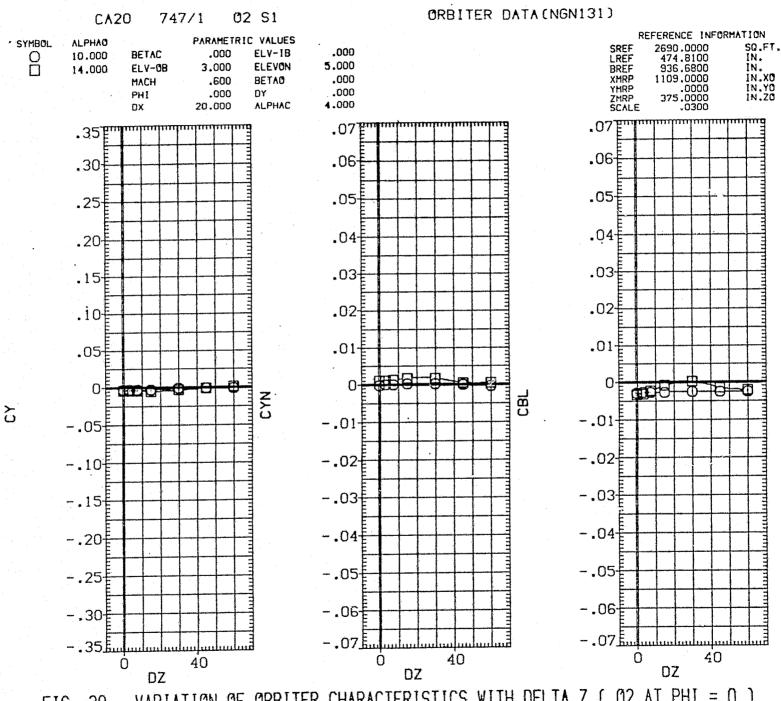


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1483

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) FIG PAGE 1484

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1486



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FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1487

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1488

CA20 (747/1 01 S1) - (01 S1) D/S (131 - 018)(VGN131)

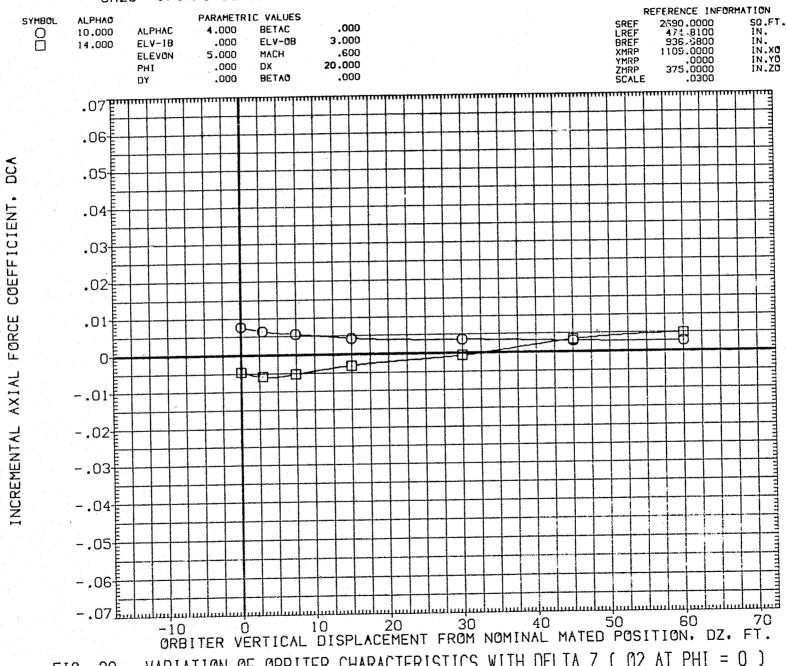


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1489

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1490

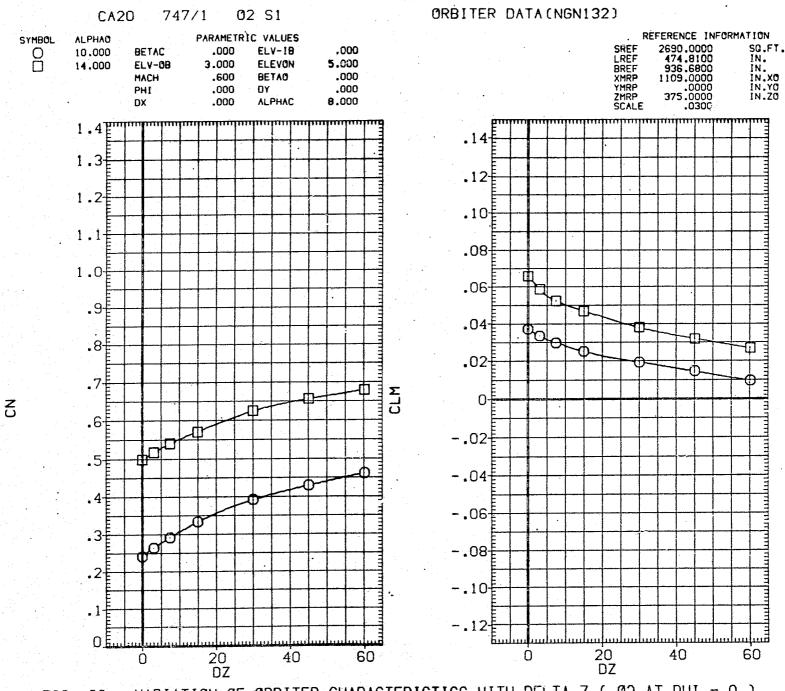


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1491

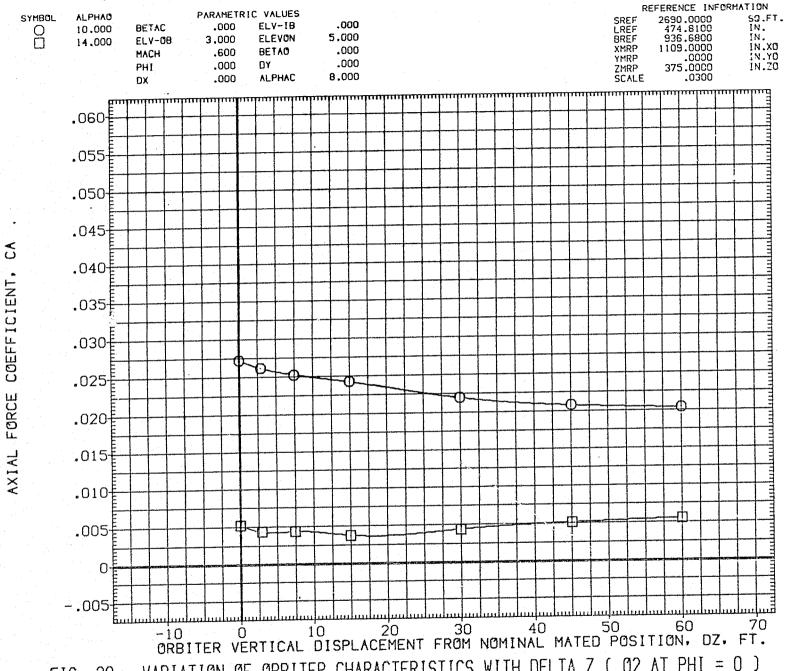
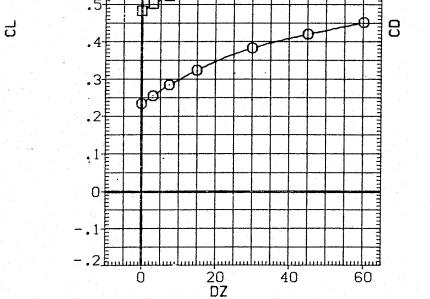


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )



SYMBOL

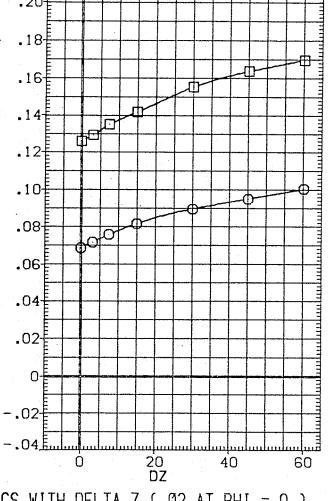


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1493

CA20 (747/1 01 S1) - (01 S1)

D/S (132 - 018)(VGN132)

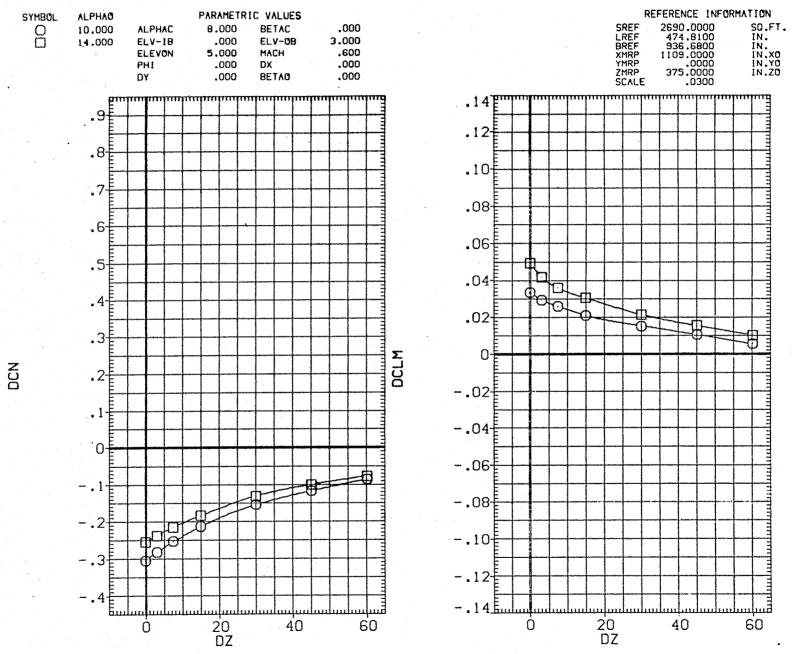


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1495

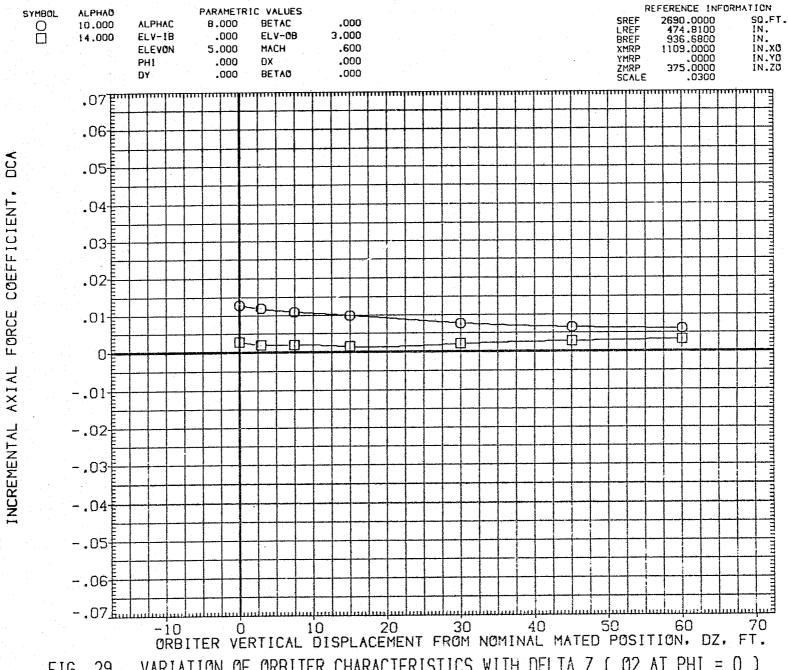


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1496

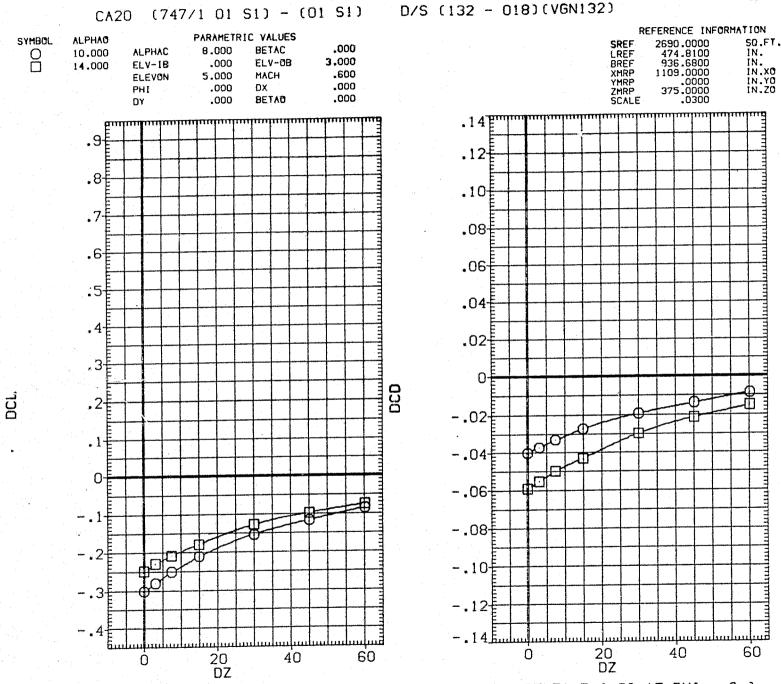


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1497

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1498

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )

PAGE

1499

FIG

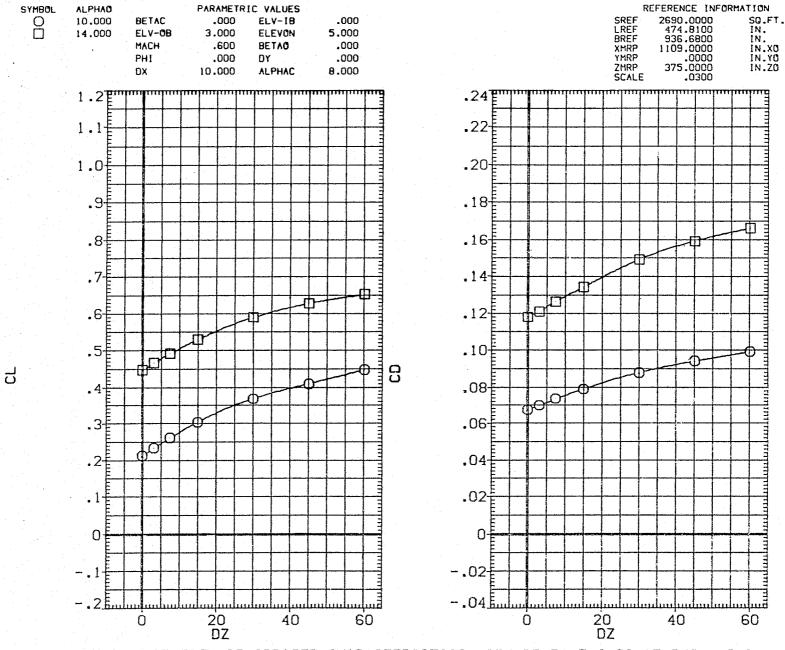


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1500

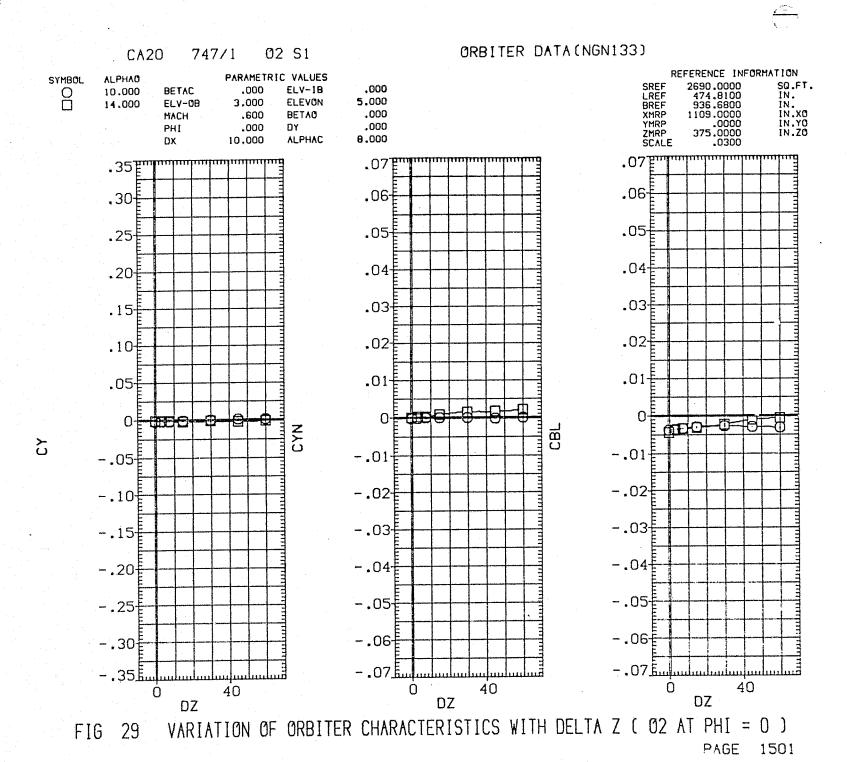
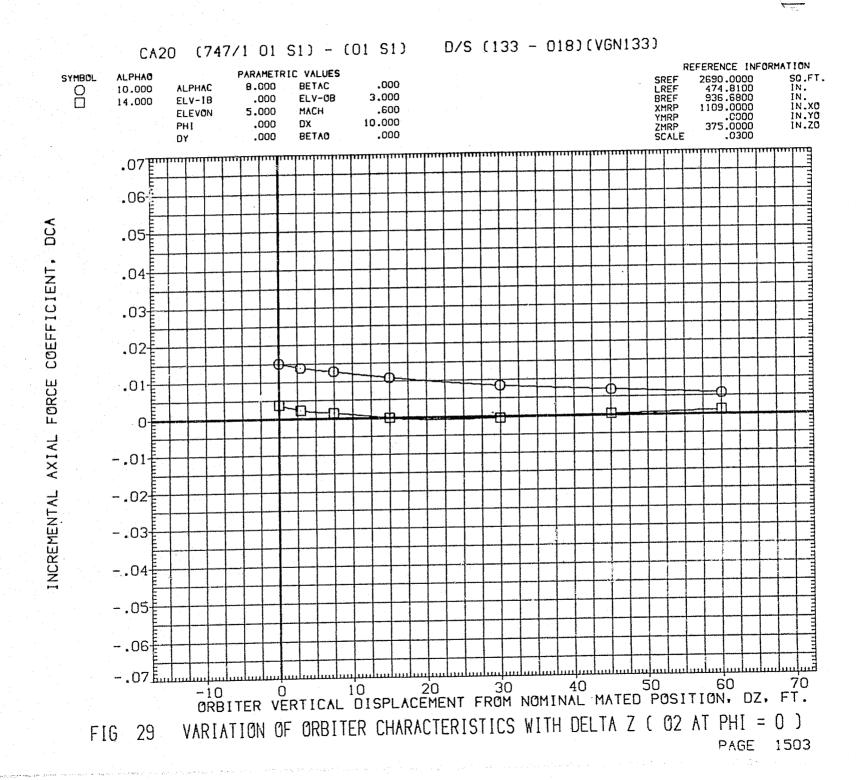


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1502



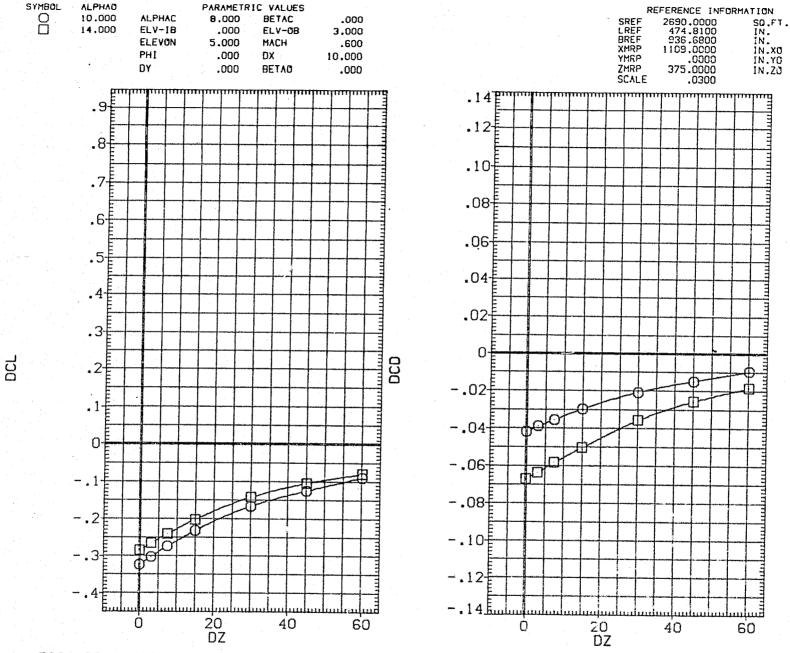


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1504

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1505

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1506

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ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT.

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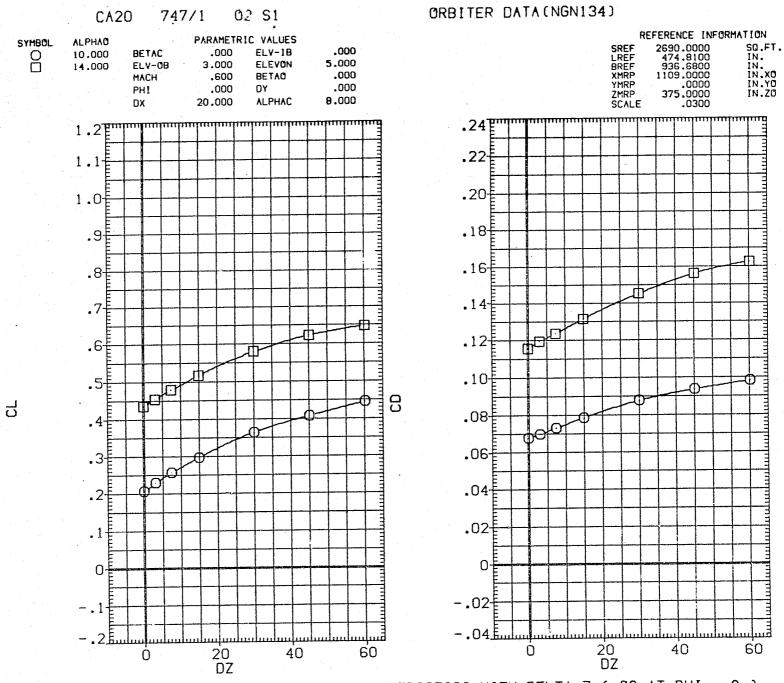


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1507

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1508

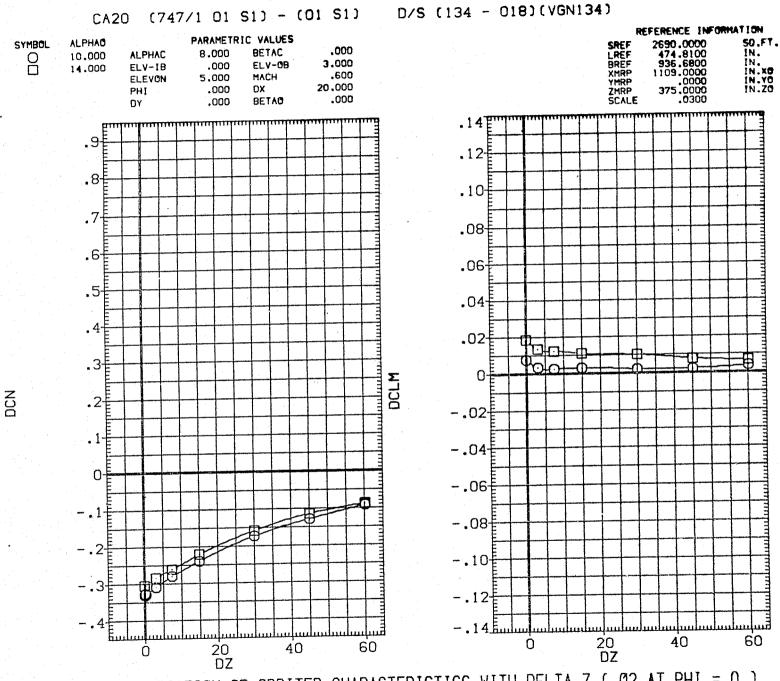
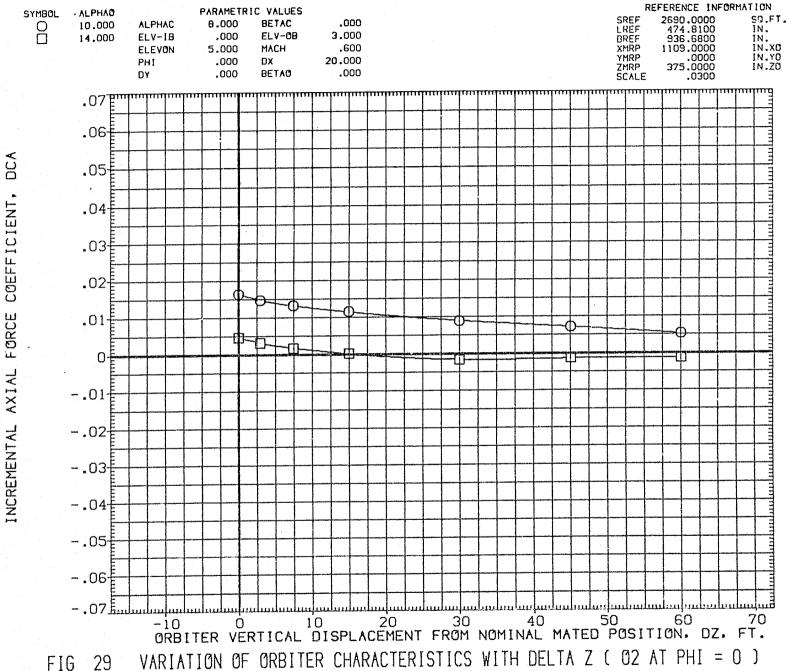


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 ) PAGE 1510

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1511

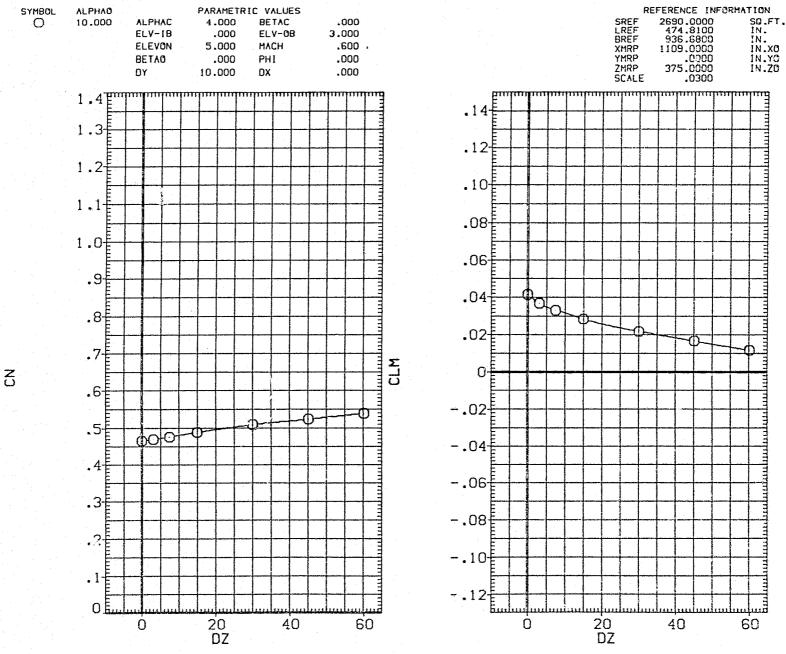
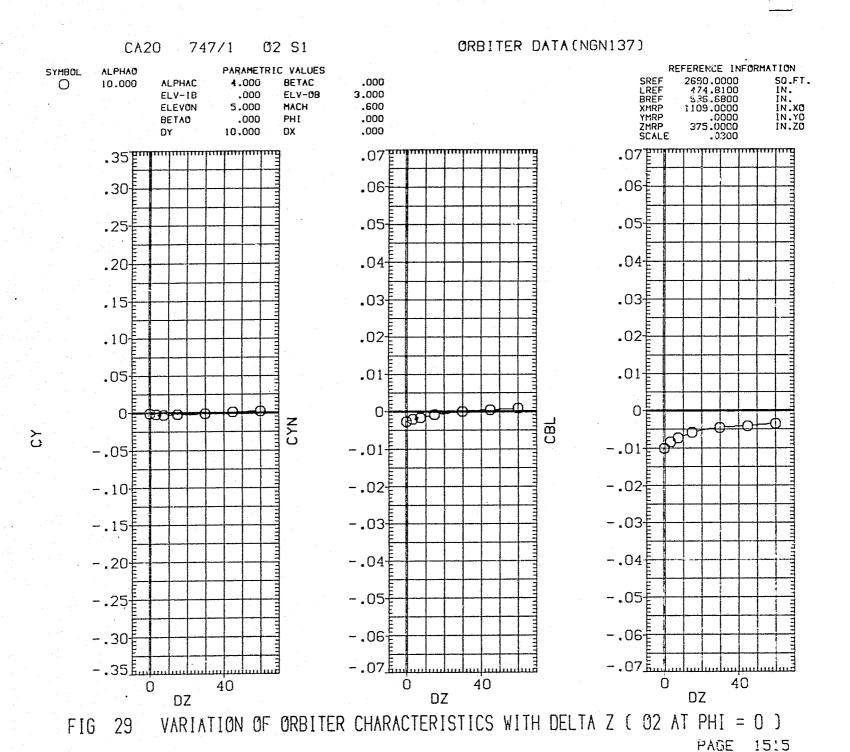


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1512

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1513

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FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 15:4



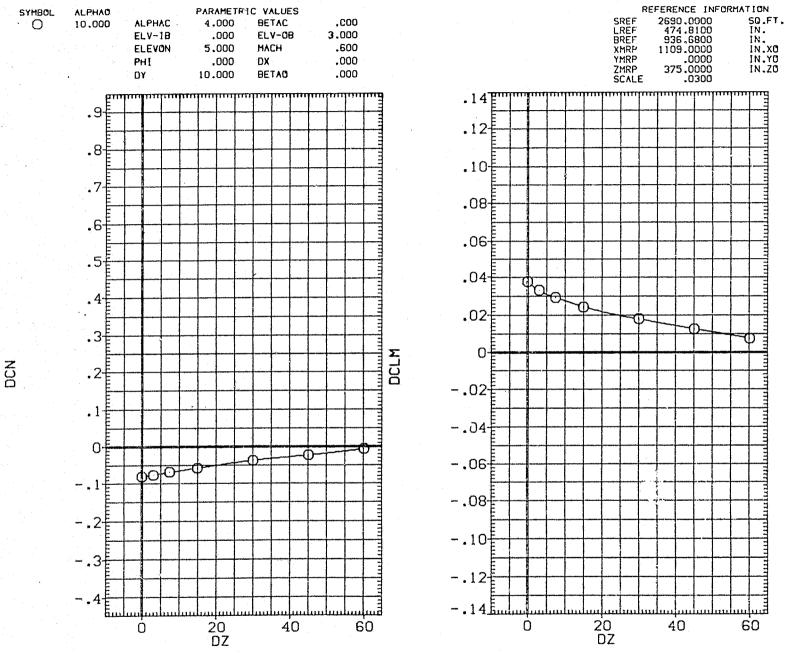
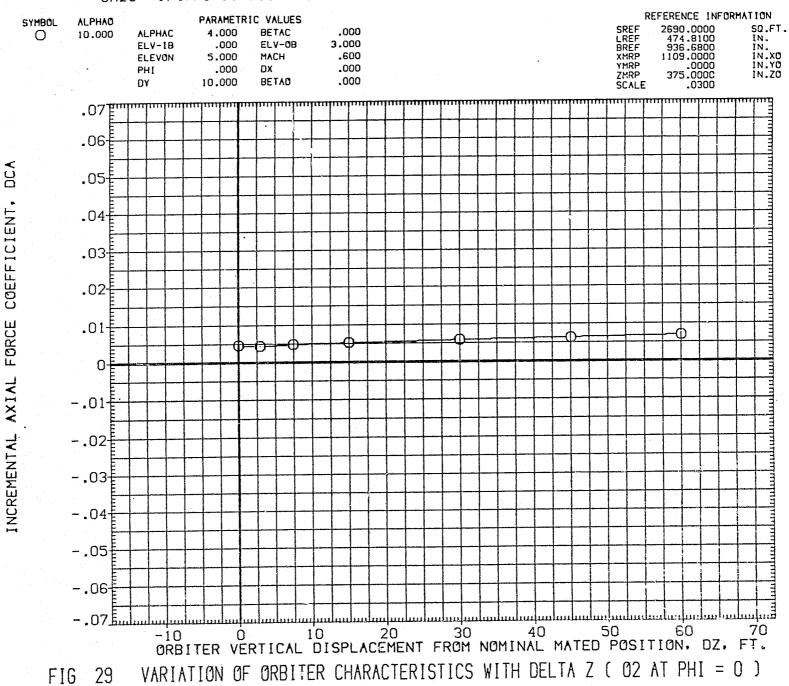


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1516



PAGE 1517

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1518

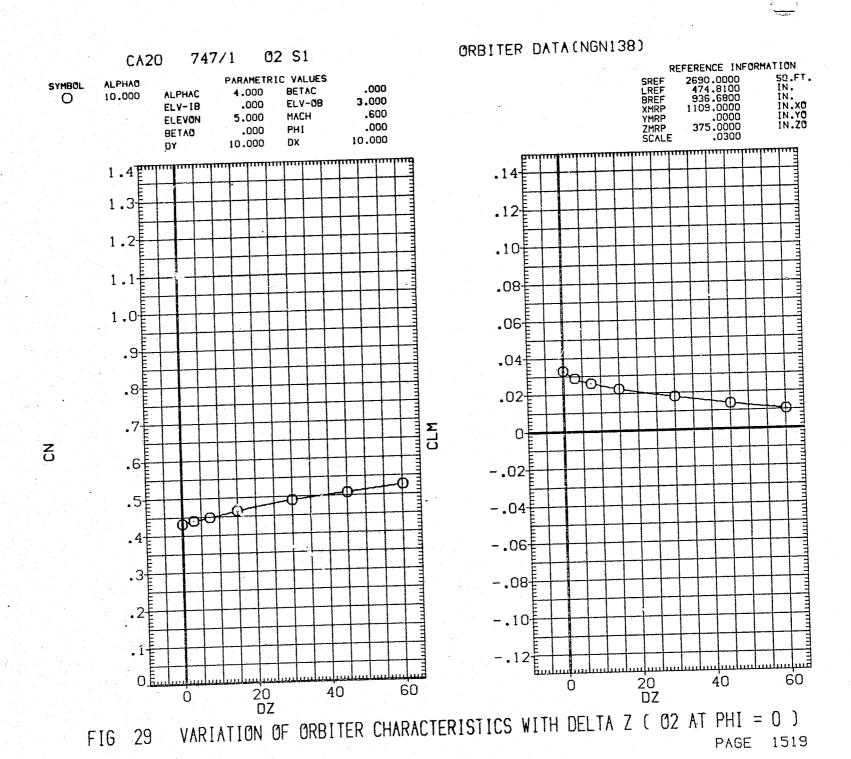
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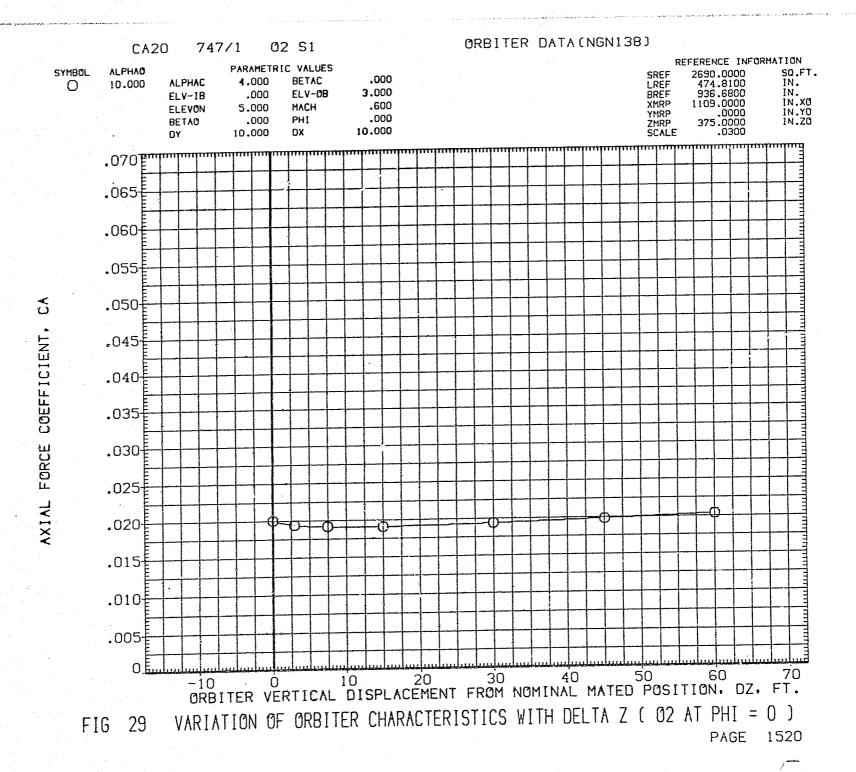


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1521

FIG PAGE 1522

CA20 (747/1 01 S1) - (01 S1) D/S (138 - 018)(VGN138) REFERENCE INFORMATION ALPHAO PARAMETRIC VALUES SYMBOL 2690.0000 474.8100 936.6800 1109.0000 SQ.FT. 10.000 ALPHAC 4.000 BETAC .000 0 IN. ELV-IB .000 ELV-08 3.000 IN. IN.XO IN.YO IN.ZO ELEVON 5.000 MACH .600 PHI .000 DX 10.000 YMRP .0000 ZMRP SCALE 375.0000 10.000 BETAO .000 DY .0300 .9 .12 .8 F .10 .08 .06 .04 Polo .02 .3+ DCLM 0-DCN -.02 .1ŧ -.04 0--.06 -.08 -.10 -.12<del>[</del> 20 DZ 20 DZ 40 60 40 60

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1523

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1524



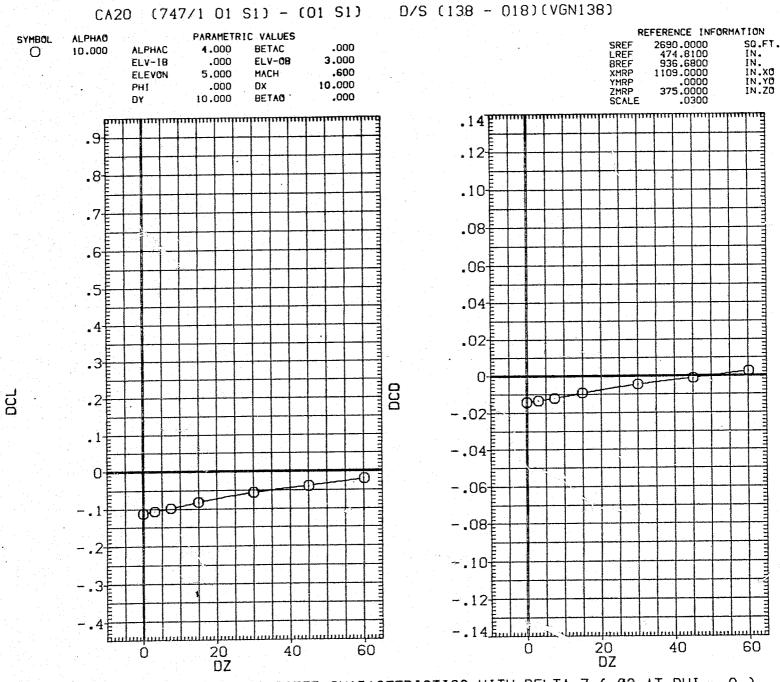


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1525

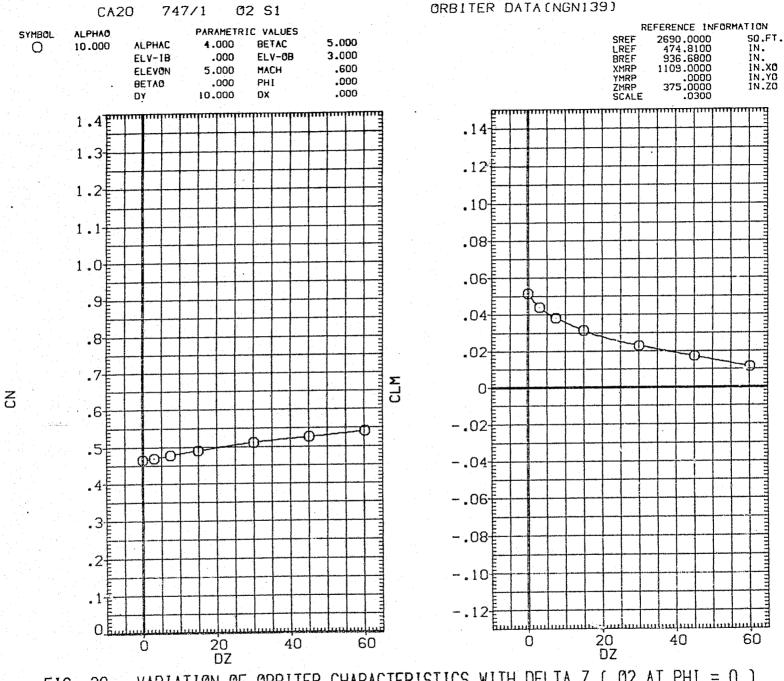
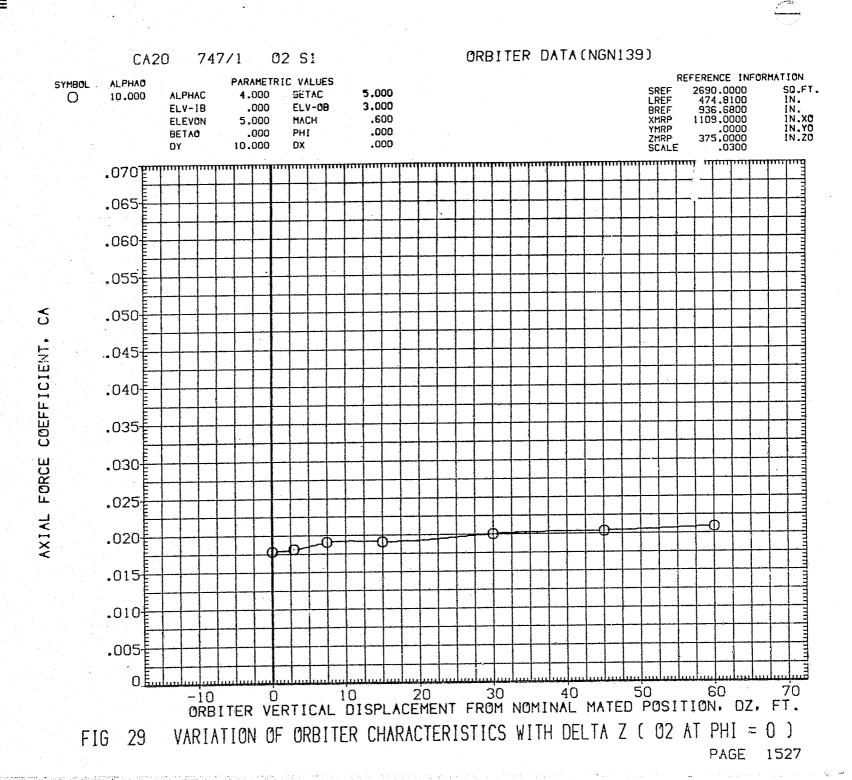


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1526



PAGE 1529

FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1530

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FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1532

PAGE

1533

FIG



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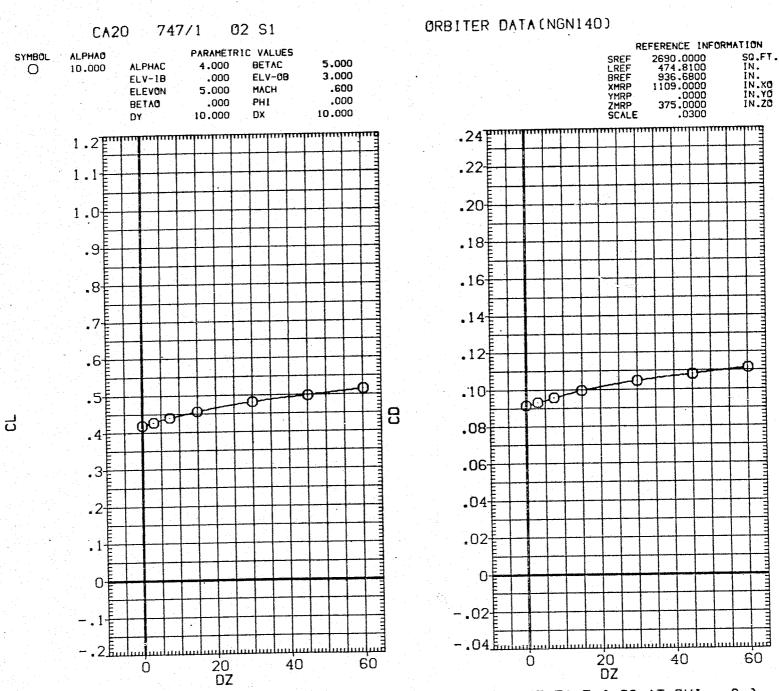


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1535

ORBITER DATA (NGN140)

PAGE 1536

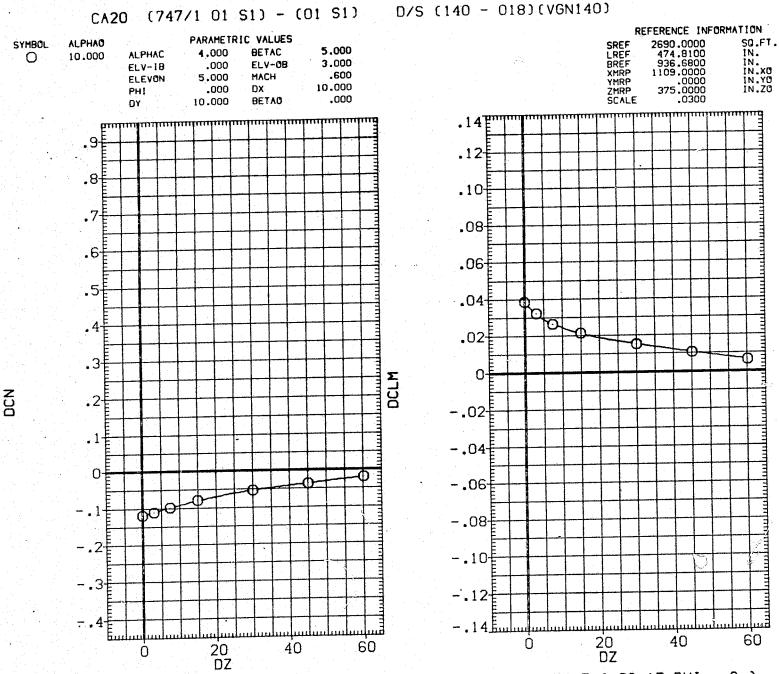


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1537

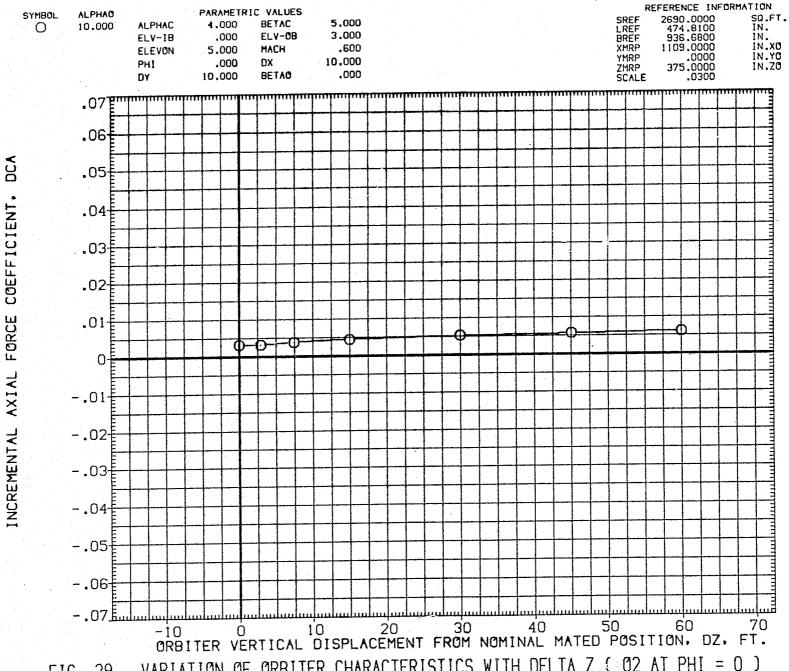


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1538

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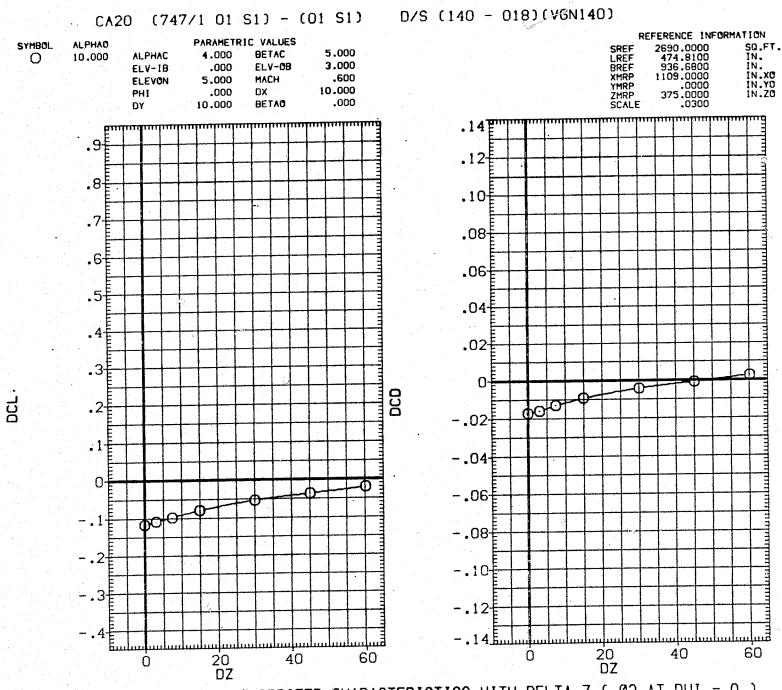


FIG 29 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( 02 AT PHI = 0 )
PAGE 1539

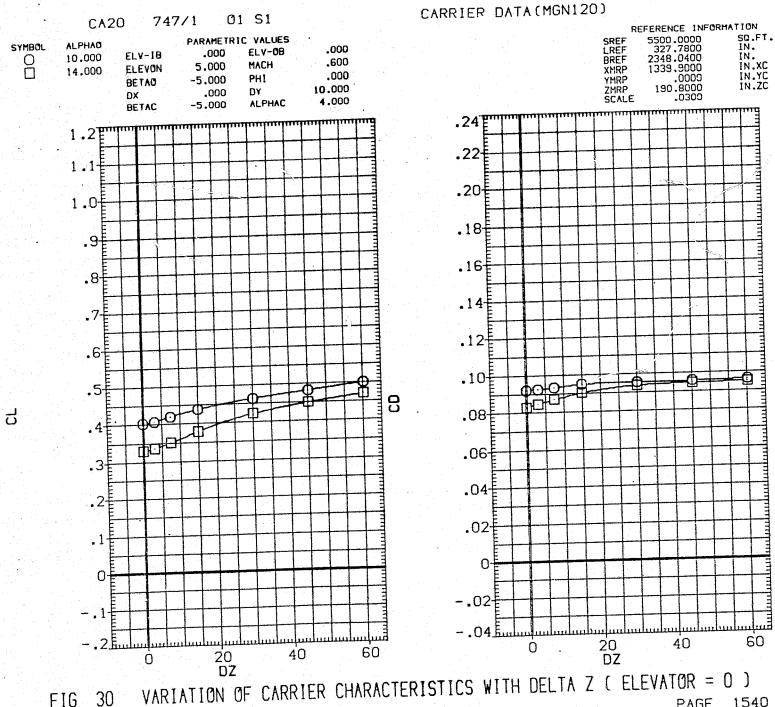
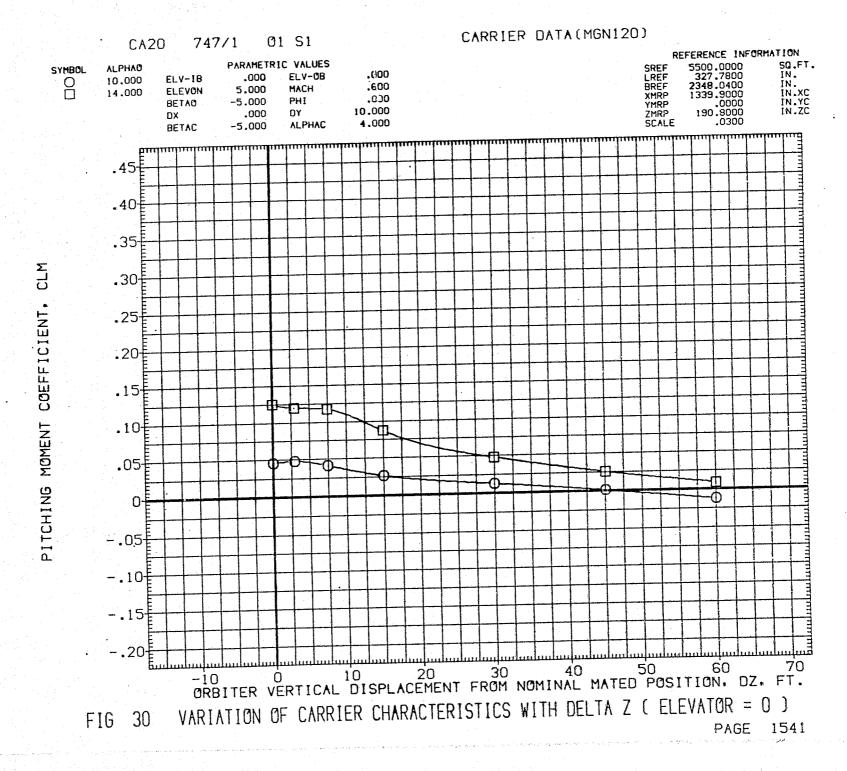
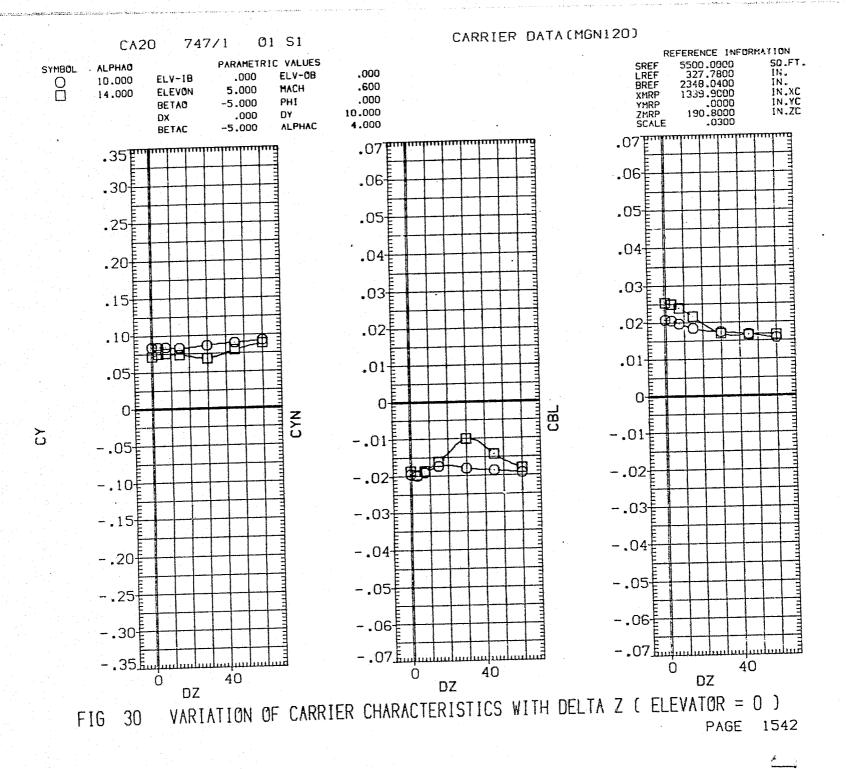


FIG 1540 PAGE







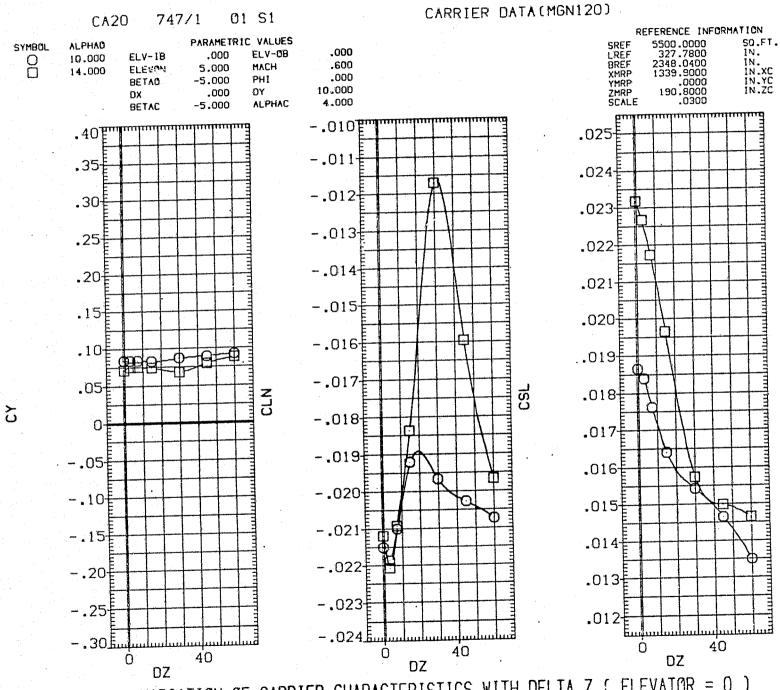
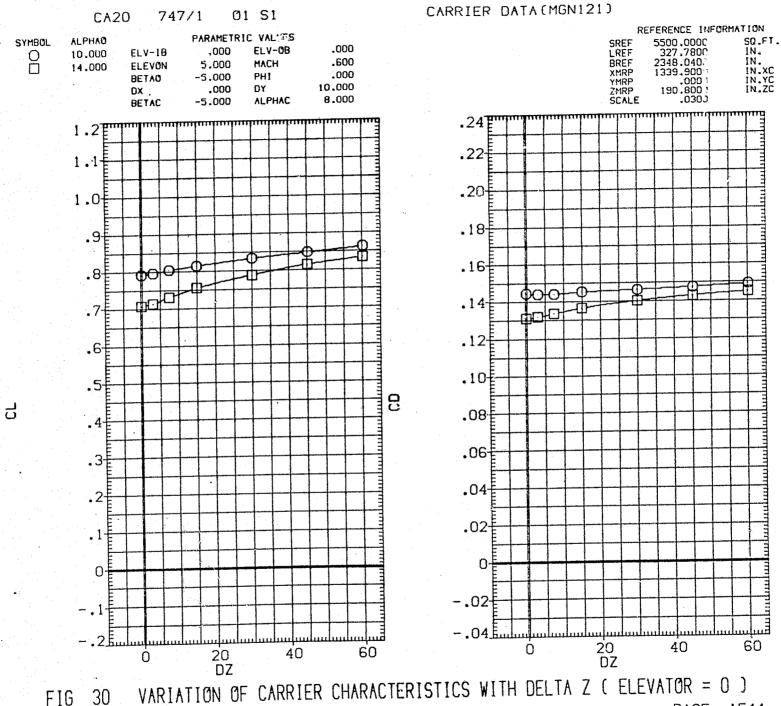
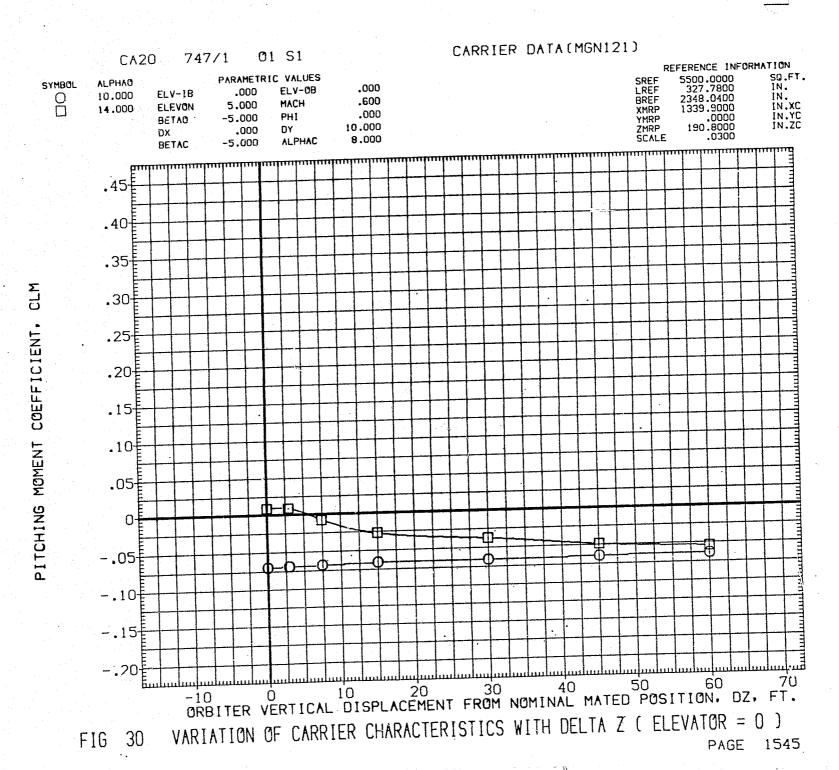


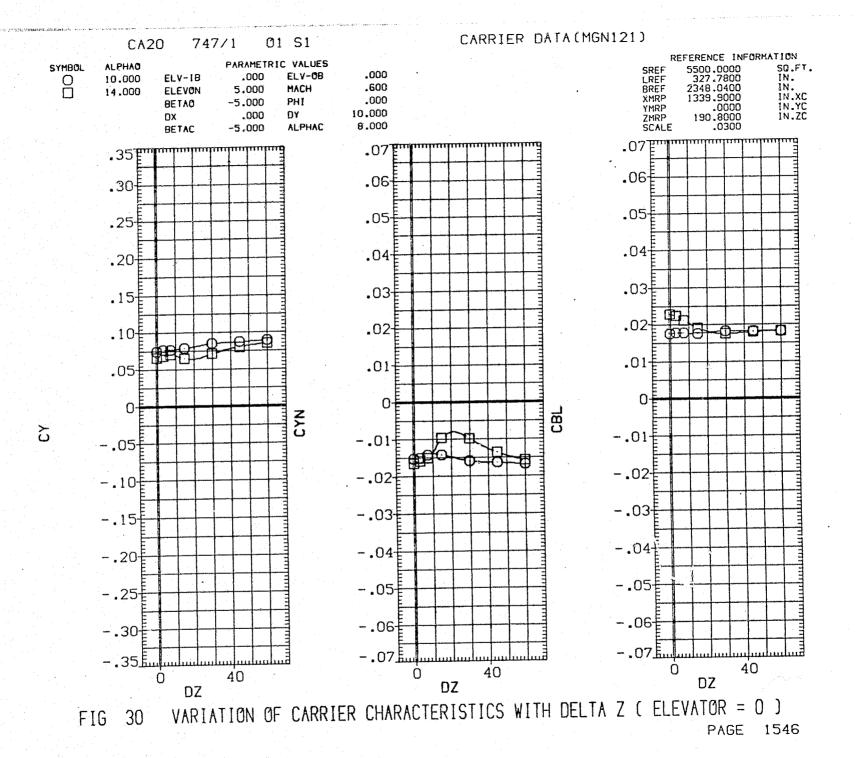
FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )



30 FIG 1544 PAGE







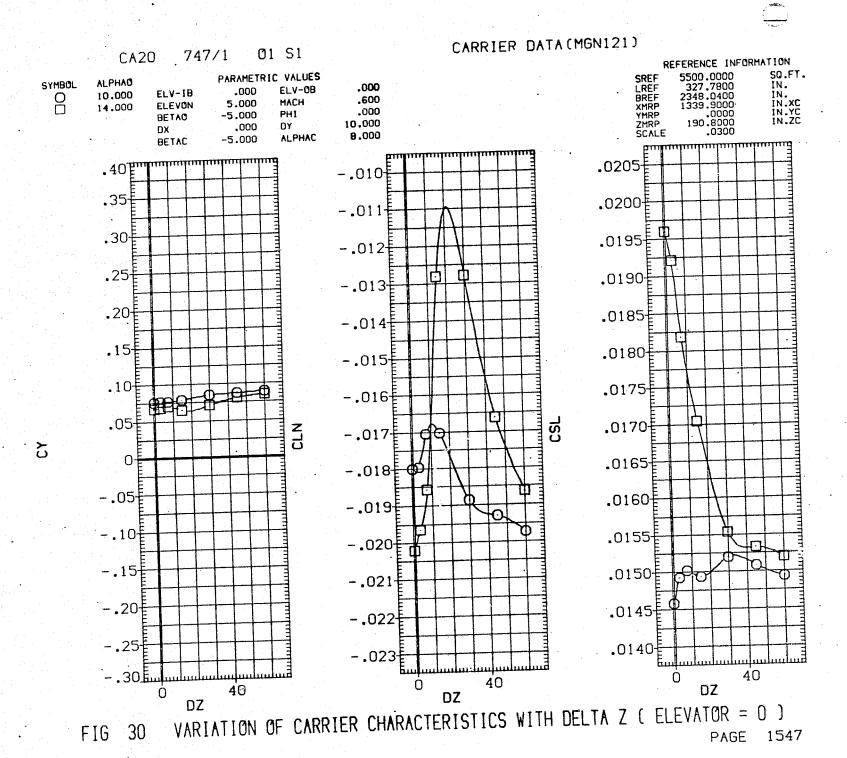


FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1549

ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT.

FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1550





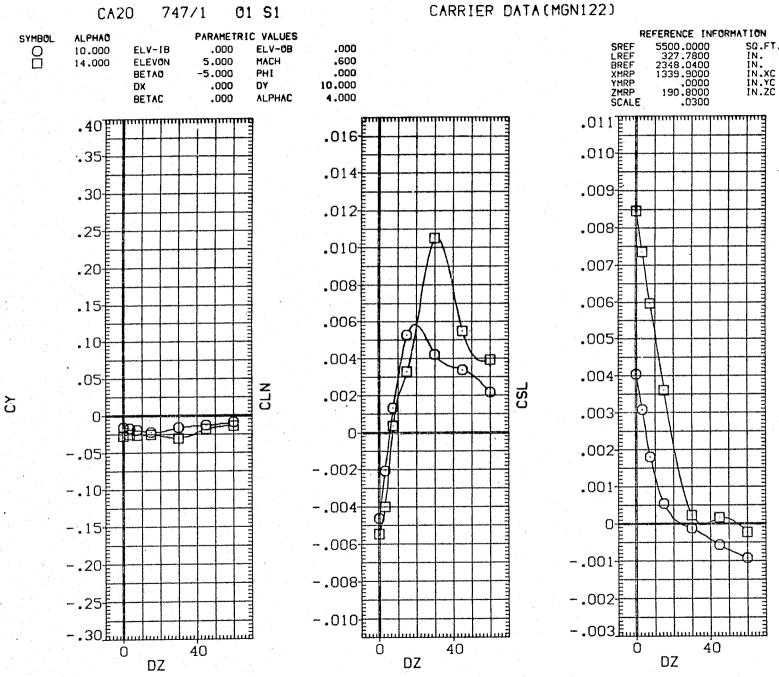
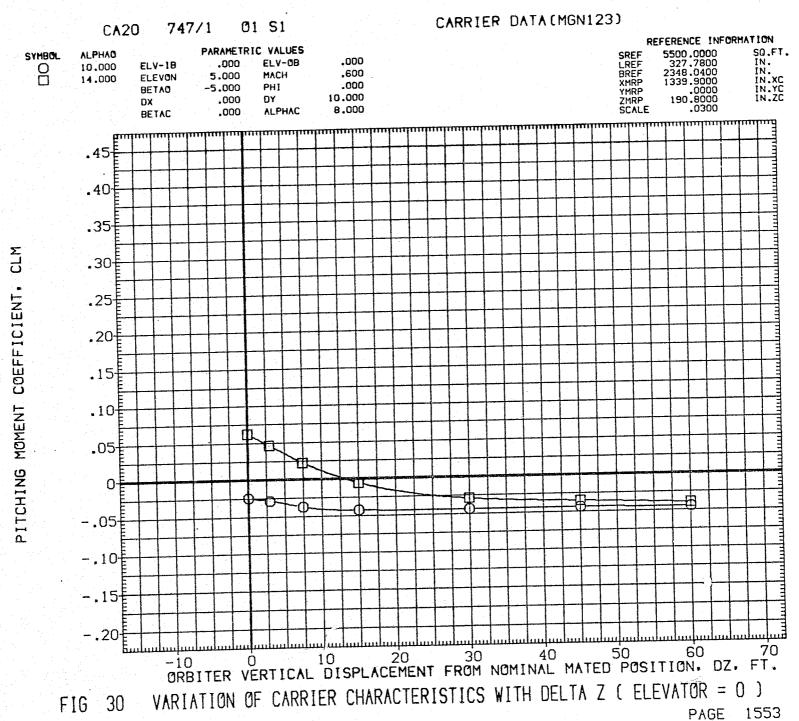


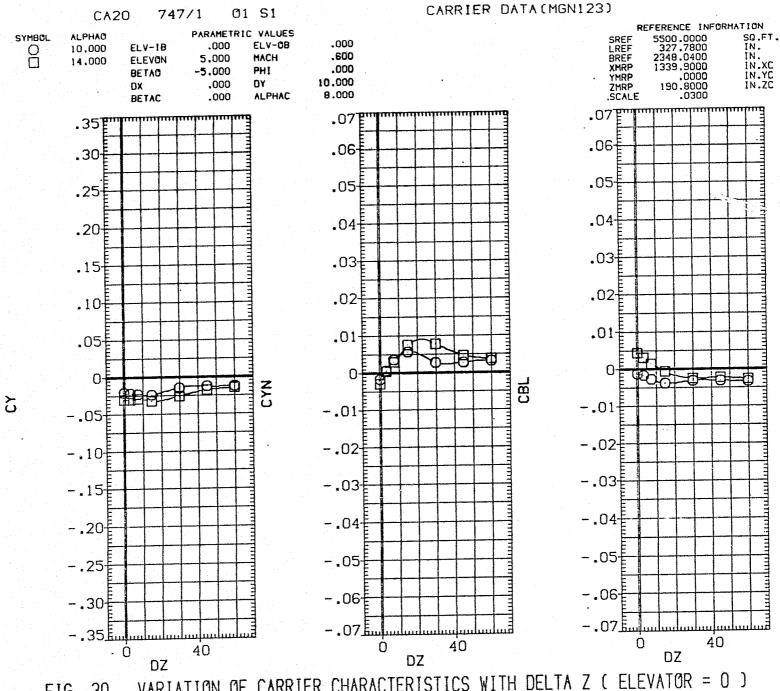
FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1551

FIG PAGE 1552









VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 ) FIG PAGE

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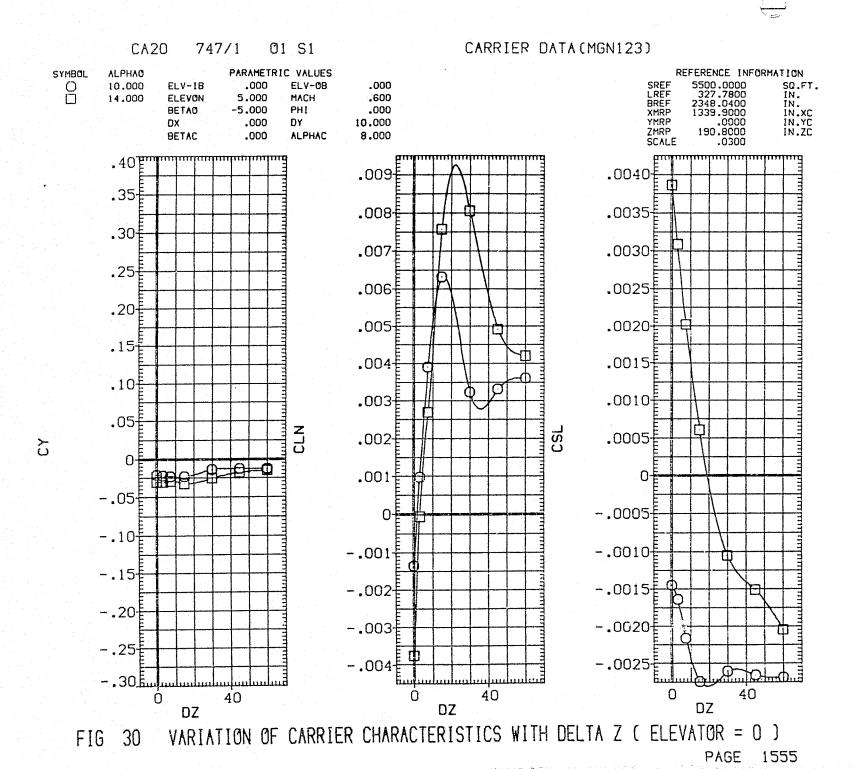
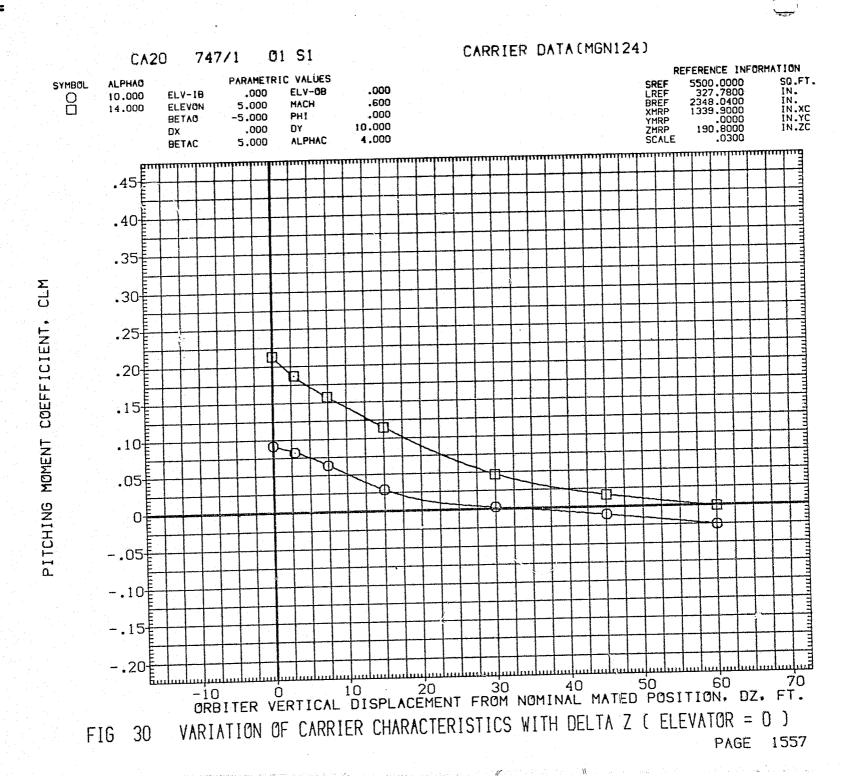
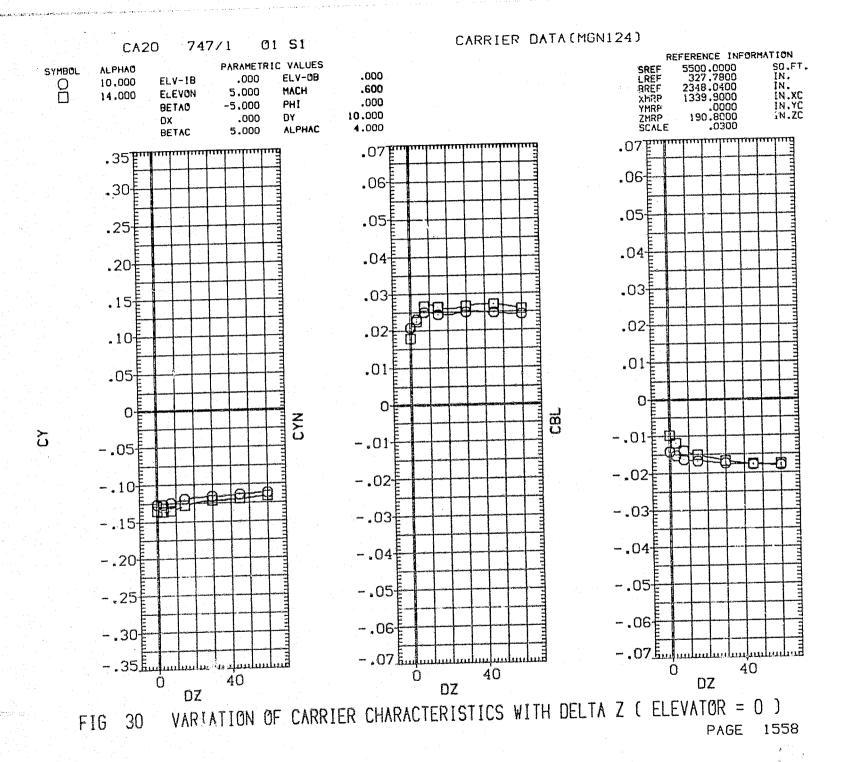
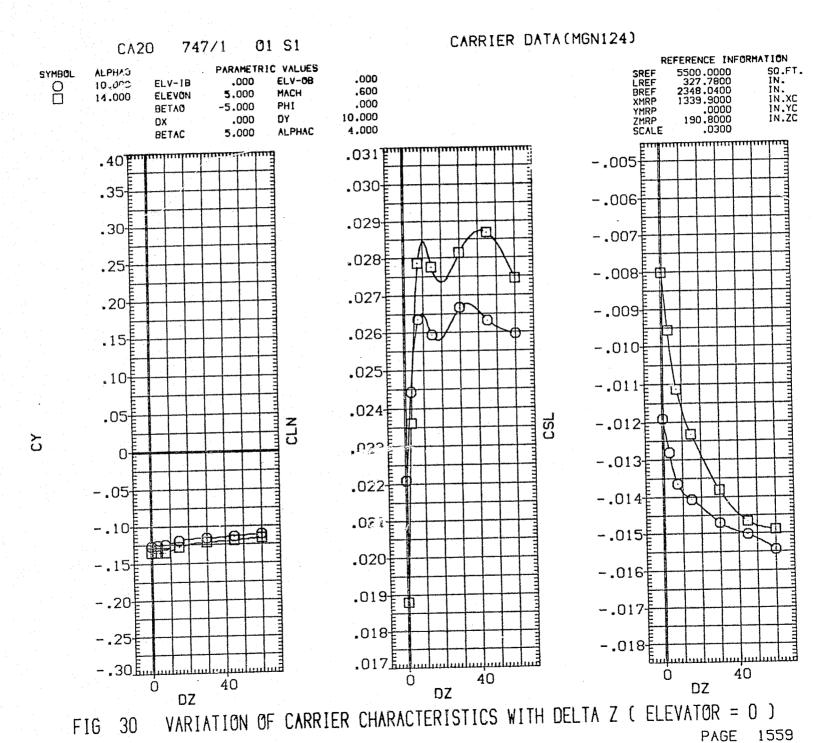


FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 15









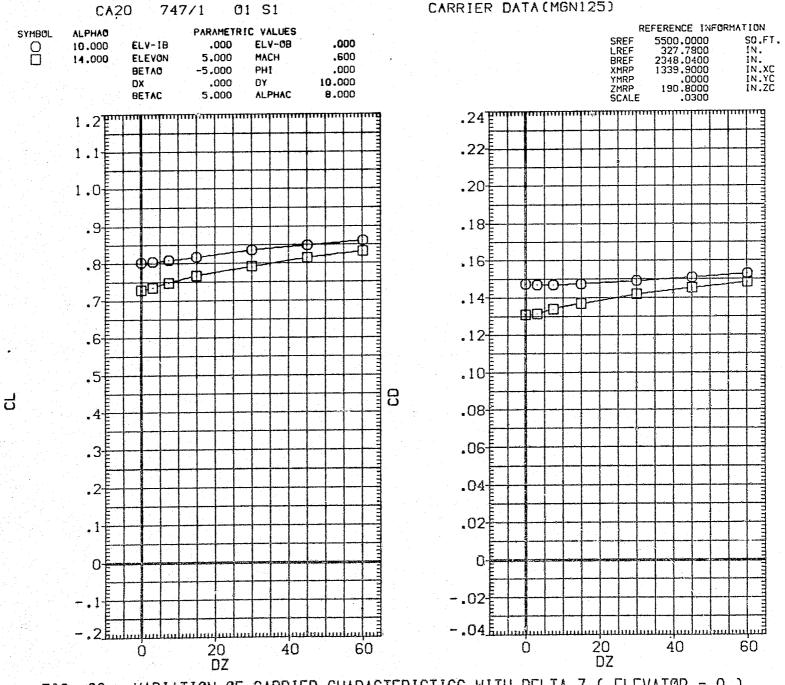
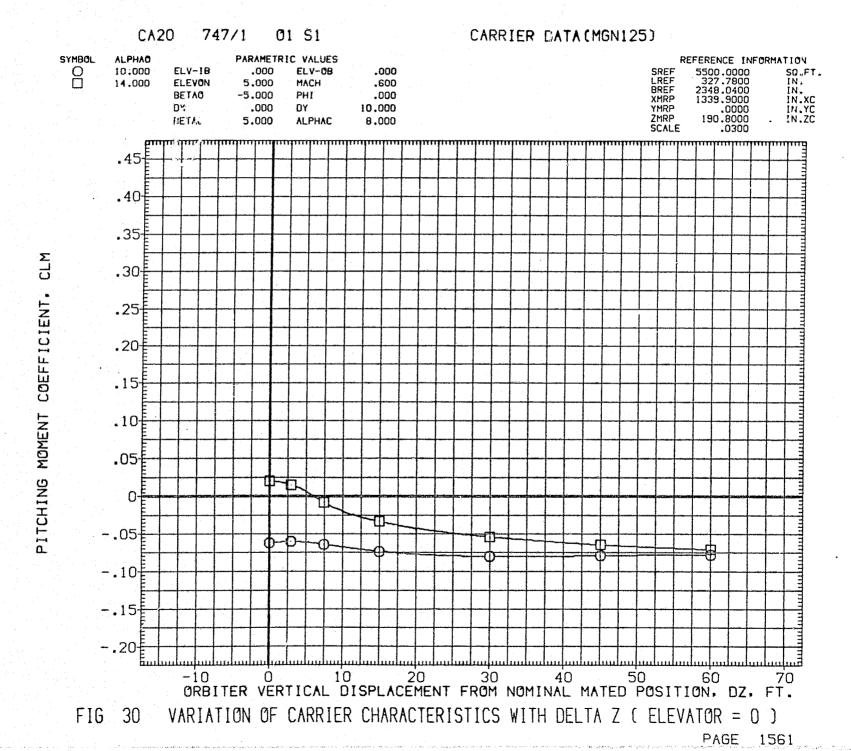


FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1560





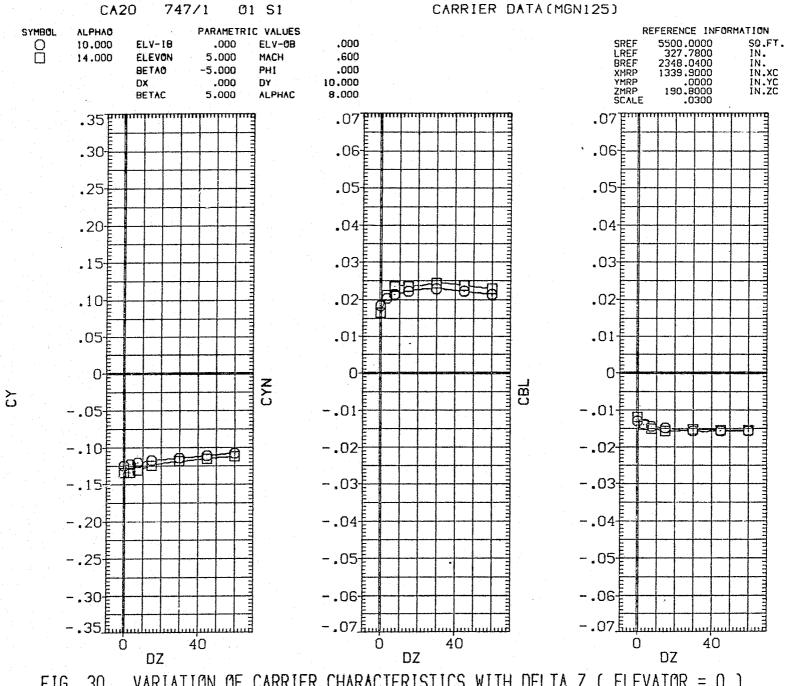


FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1562



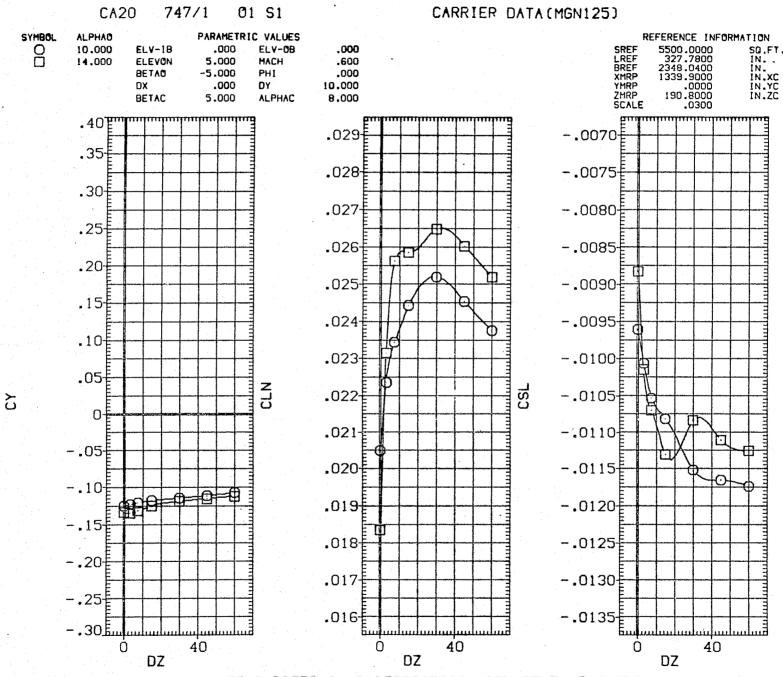


FIG 30 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1563

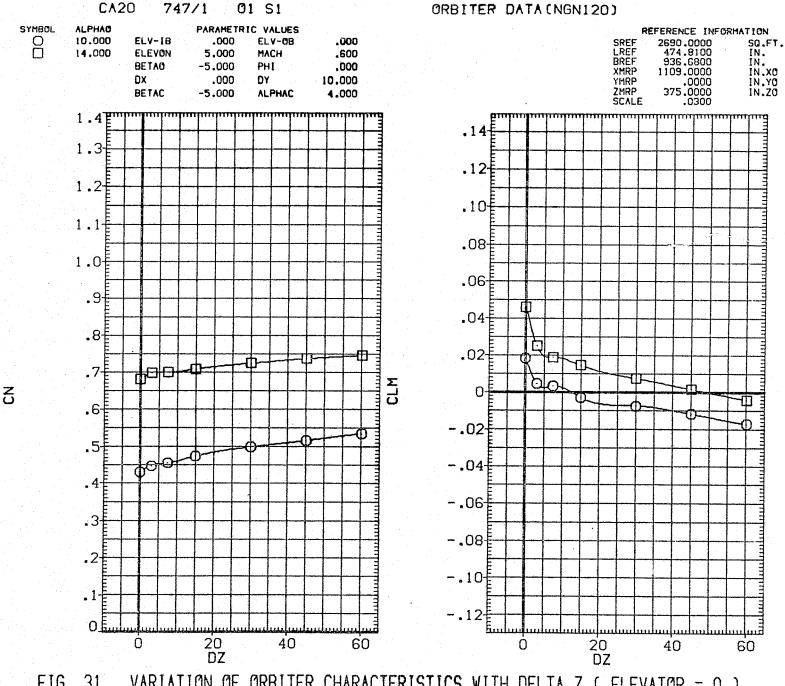


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1564

PAGE 1565

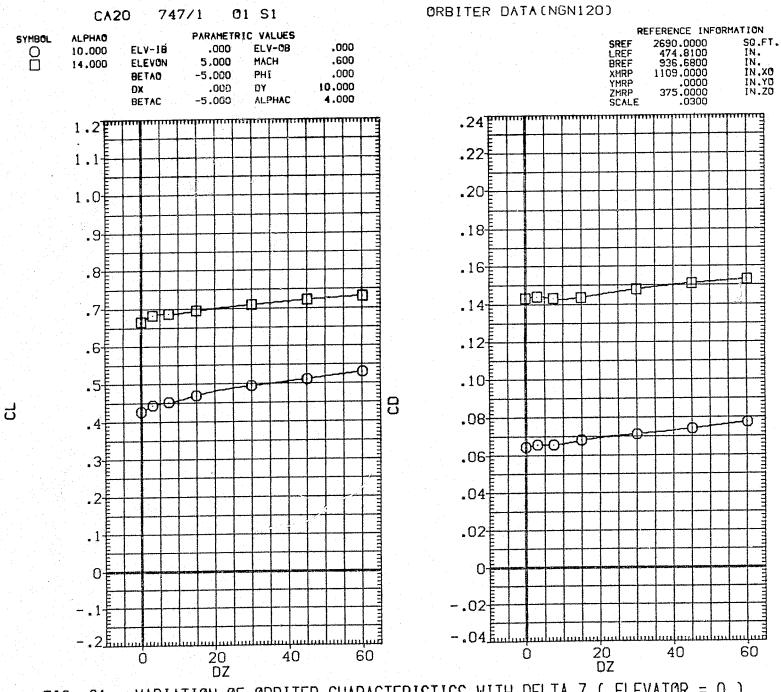


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1566

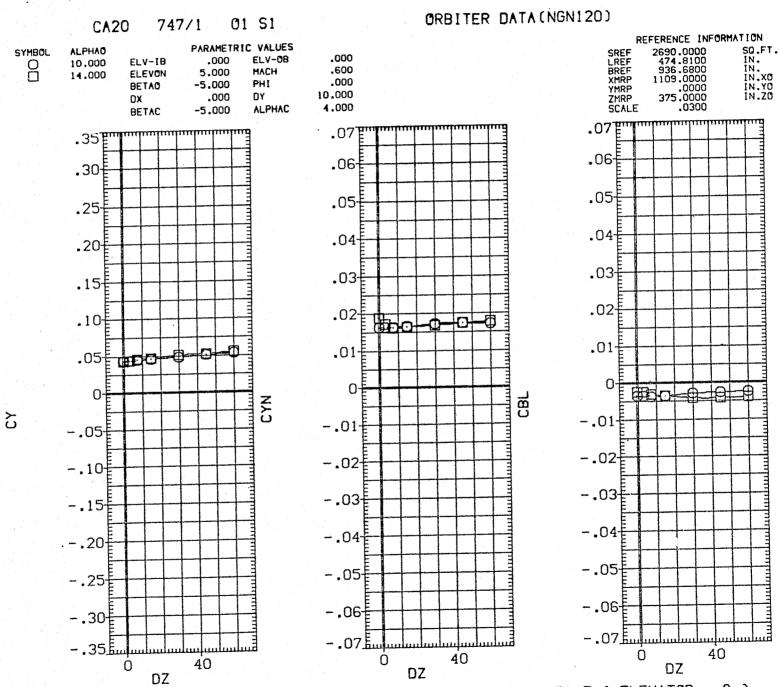


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )

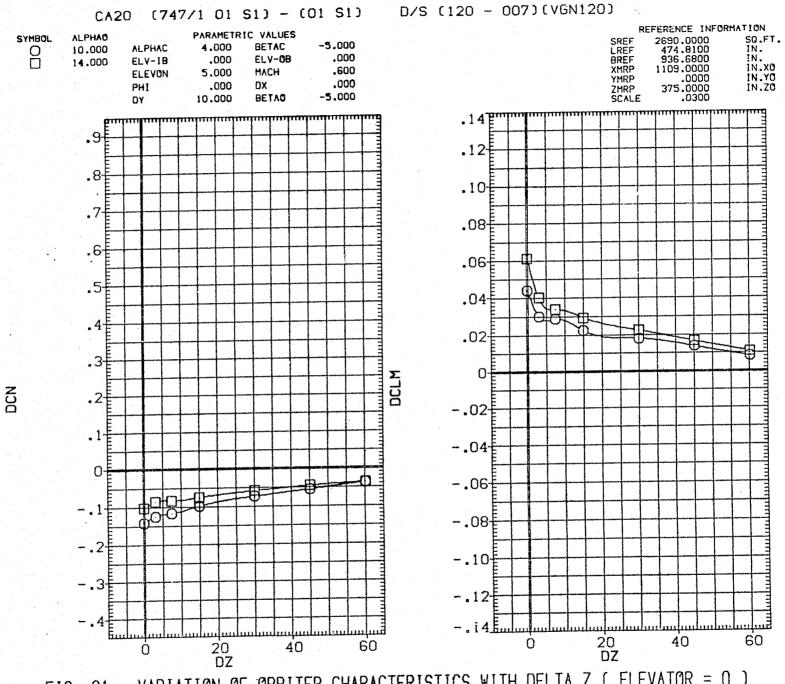


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1568

D/S (120 - 007)(VGN120)  $(747/1 \ 01 \ S1) - (01 \ S1)$ REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL ALPHAO 2690.0000 474.8100 SQ.FT. 4.000 BETAC -5.000 10.000 ALPHAC IN. IN.XO .000 .000 ELV-0B ELV-1B 936.6800 14.000 1109.0000 .0000 375.0000 .0300 .600 XMRP ELEVON 5.000 MACH YMRP ZMRP .000 .000 DX PHI IN.ZO -5,000 10.000 BETAO .07 Էարարարարարարա .06<del>[</del> .05 COEFFICIENT .04 .03<del>[</del> .02<del>-</del> FORCE .01 0-AXIAL -.01<del></del> INCREMENTAL -.02<del>[</del> -.03<del>[</del> -.04 -.05<del>[</del> -.06<del>[</del> -.07 Eulunlin -10 0 10 20 30 40 50 60 70 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ. FT. VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )

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PAGE

31

FIG

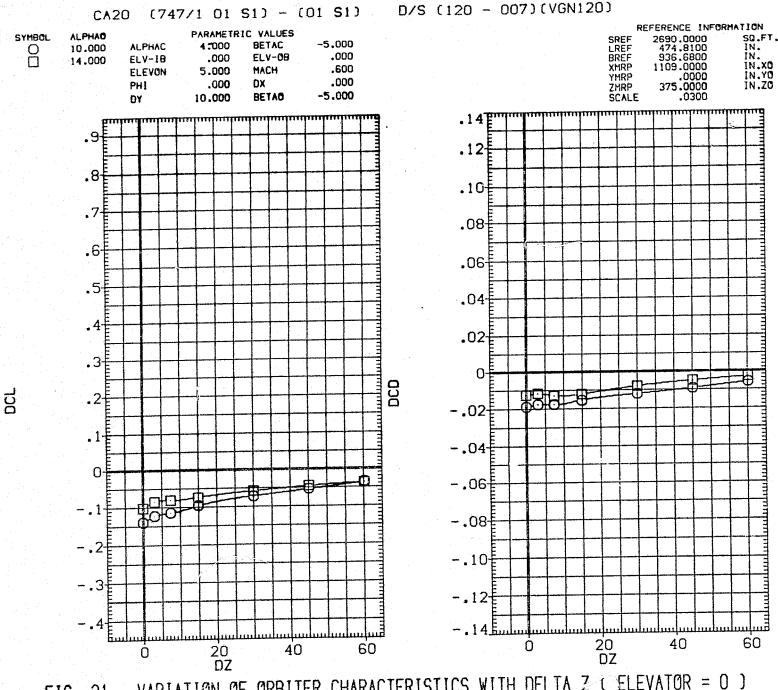
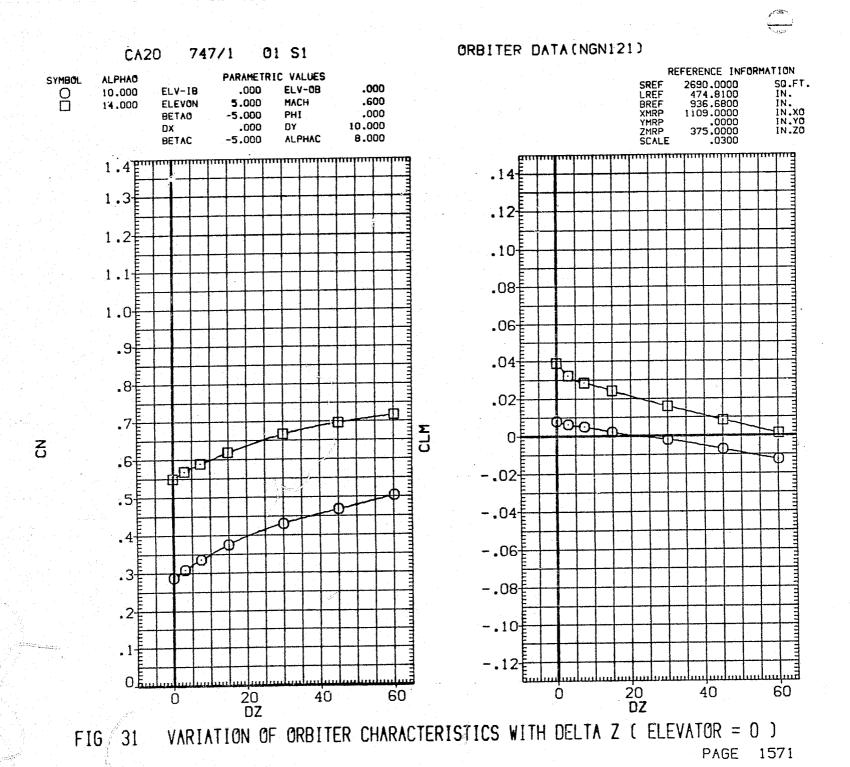


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1570



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VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 ) FIG 31 PAGE 1572

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ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT.

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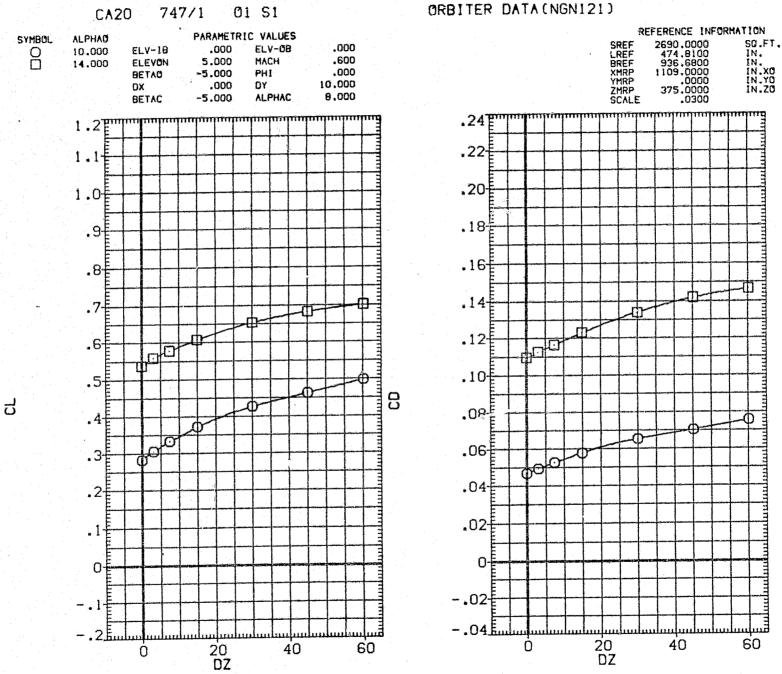


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1573

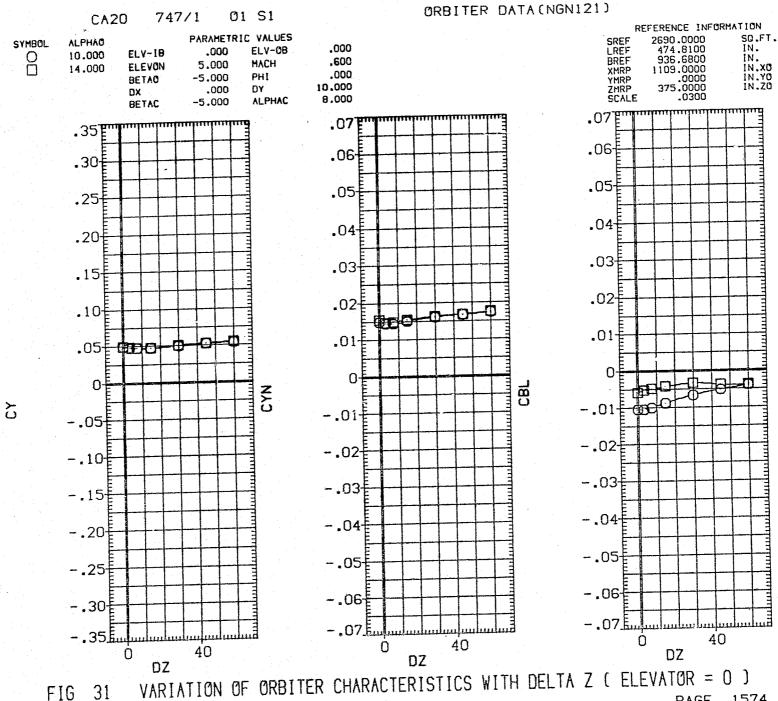


FIG 1574 PAGE

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1575

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1576

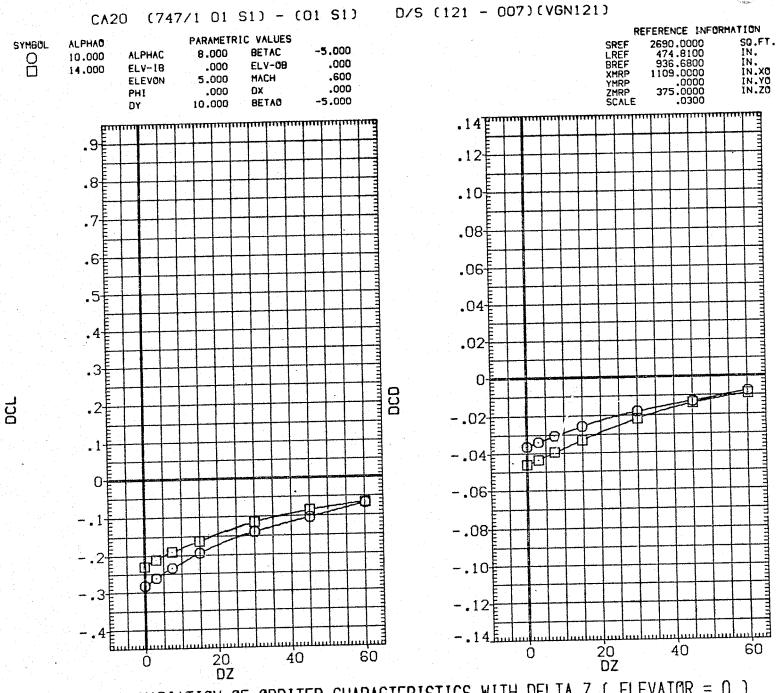
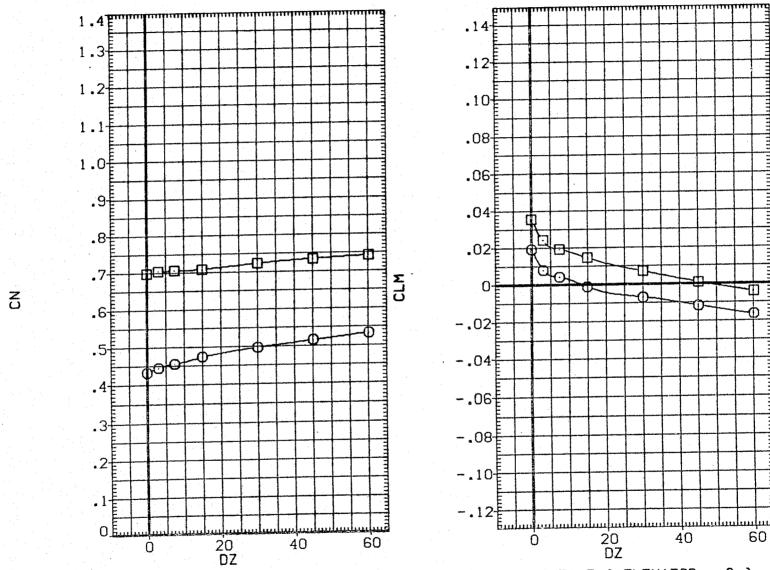


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1577



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10.000

4.000

PARAMETRIC VALUES

.000

.000

.000

5.000

-5.000

747/1

ELV-18

ELEVON

BETAO

BETAC

DX

CA20

ALPHAO

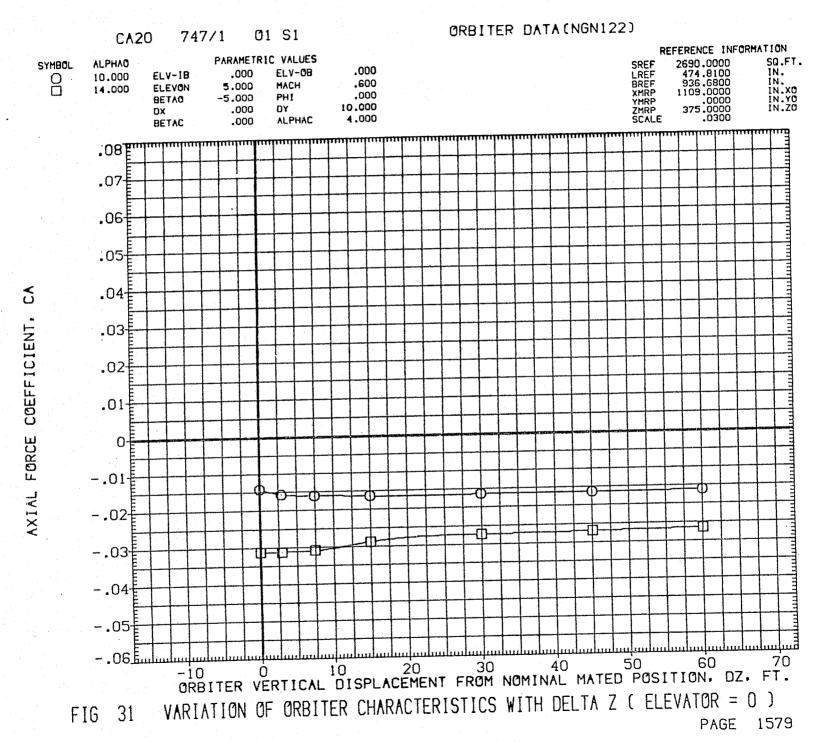
10.000

14.000

SYMBOL

VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 ) 31 FIG PAGE 1578





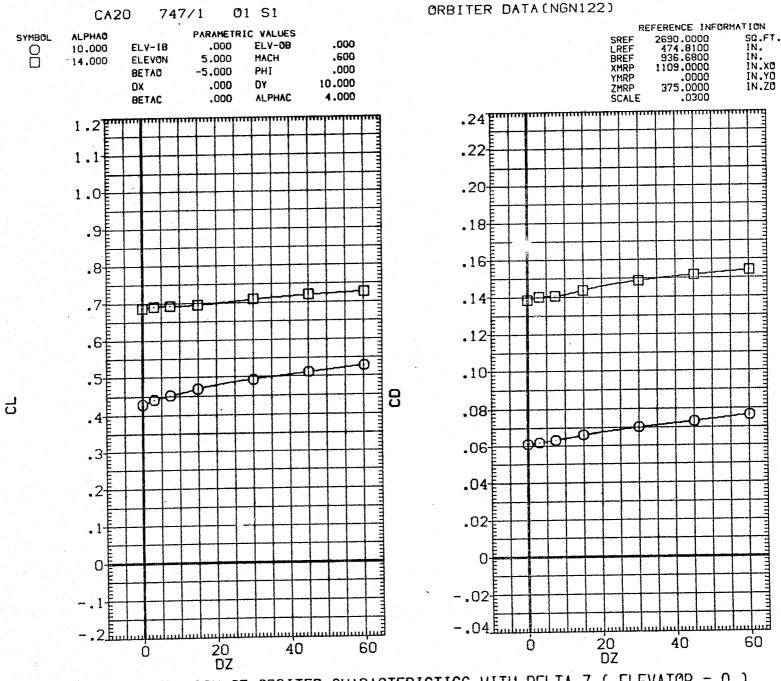
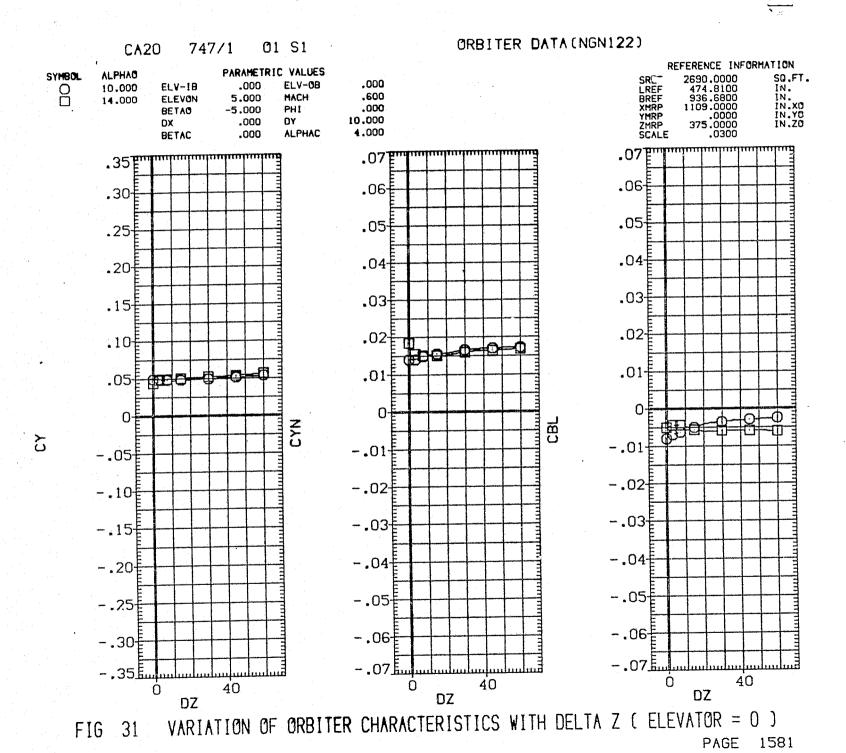


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1580





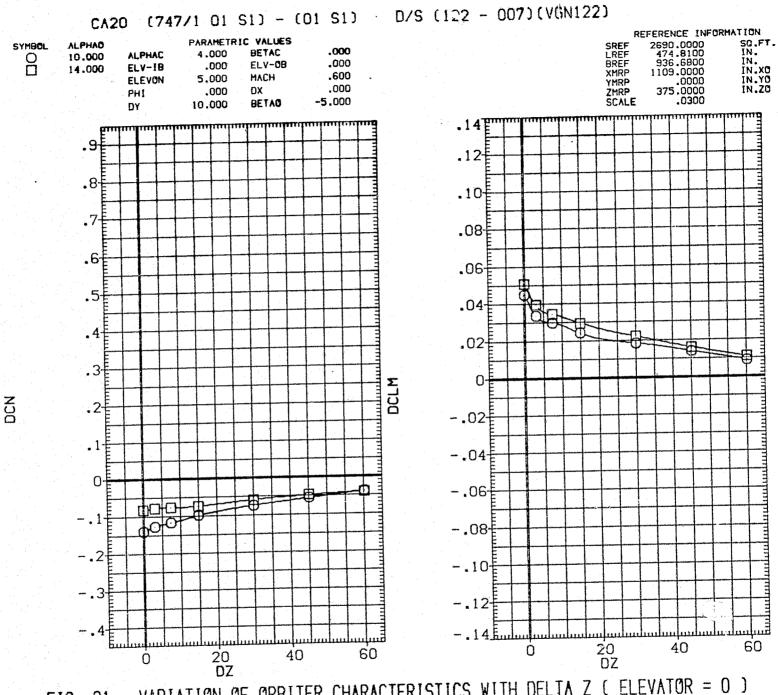


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1582

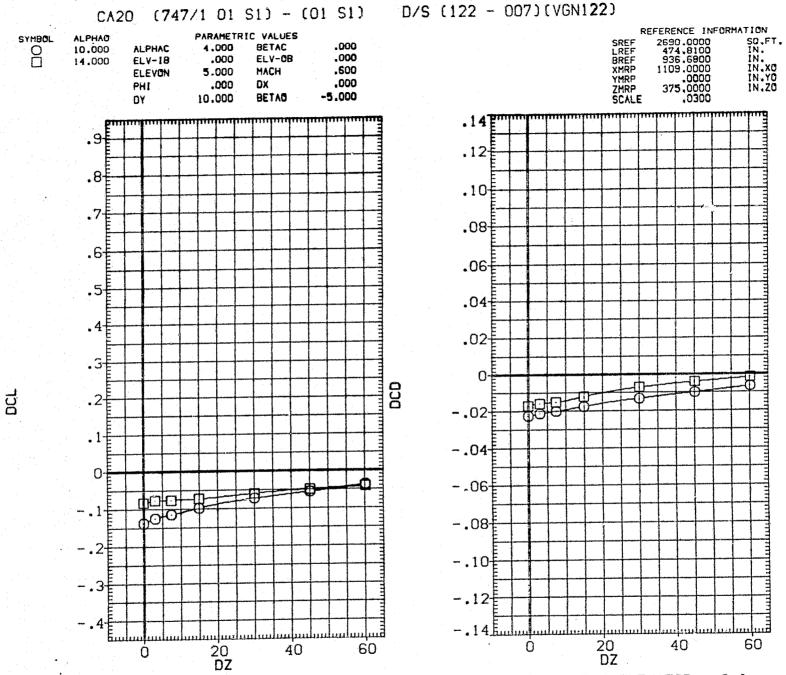


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1584

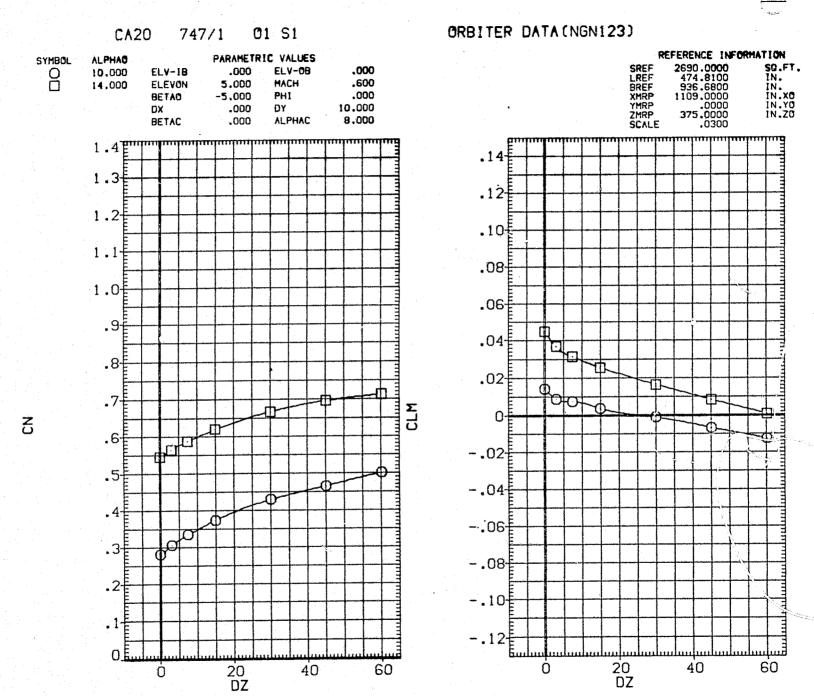


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1585

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1586

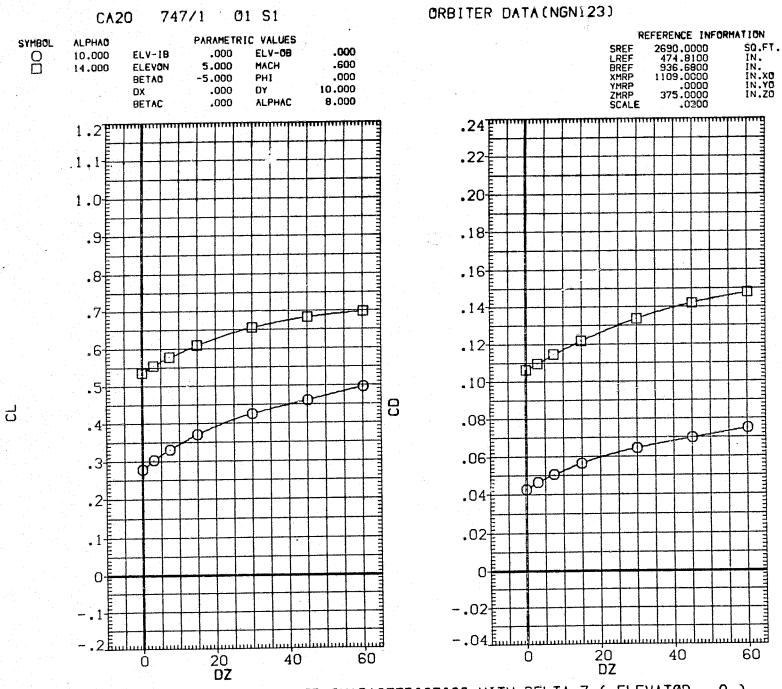


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1587

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE · 1588

DZ

0

0

DZ

40

40

0

DZ

CA20 (747/1 01 S1) - (01 S1) D/S (123 - 007)(VGN123)

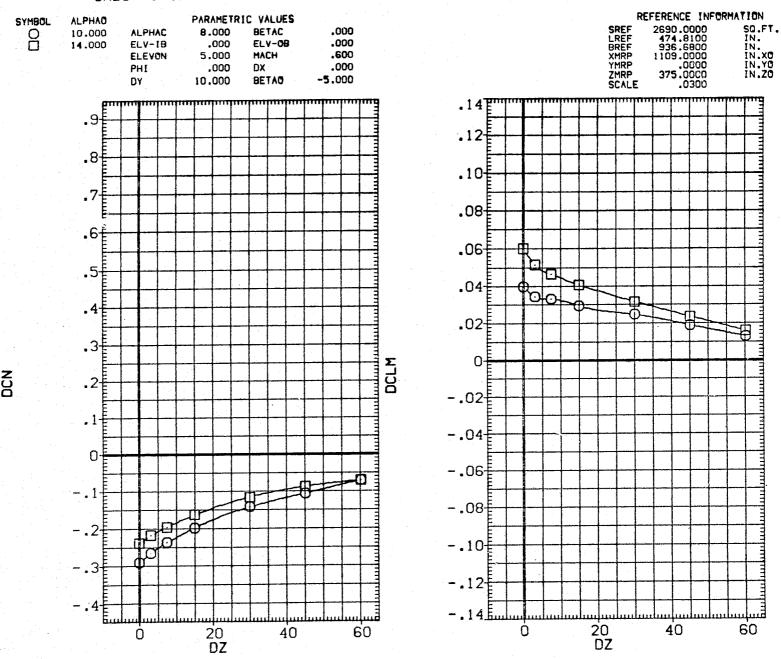


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )

PAGE 1589

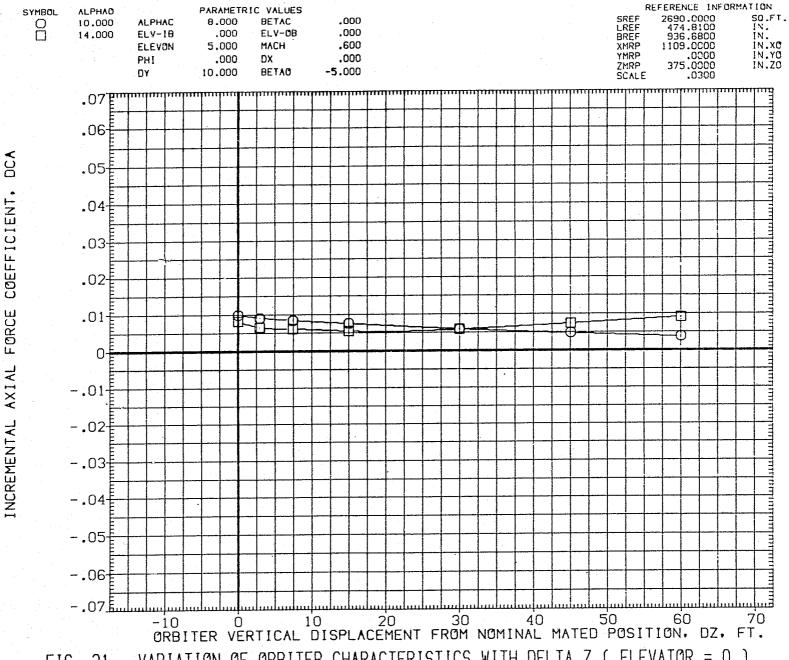


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1590

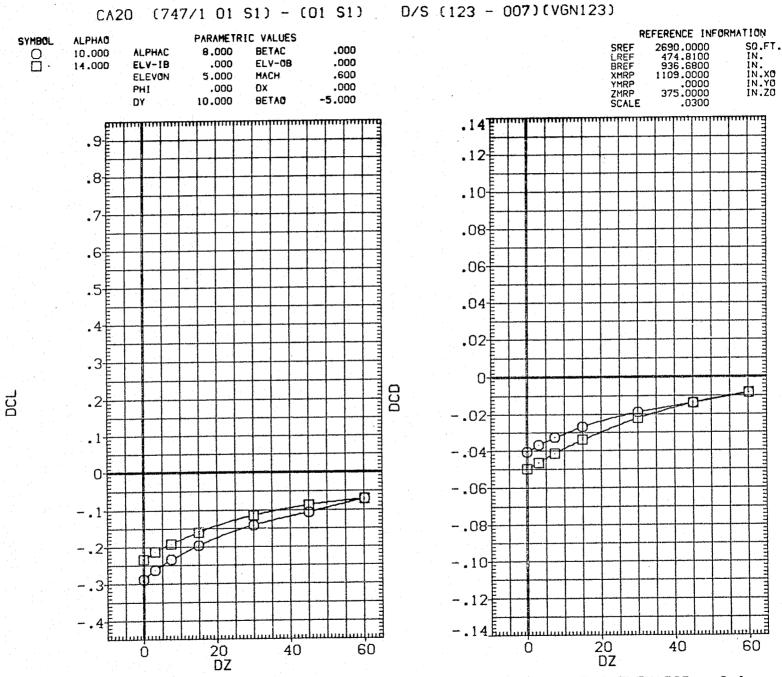


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1591

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1592

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1593

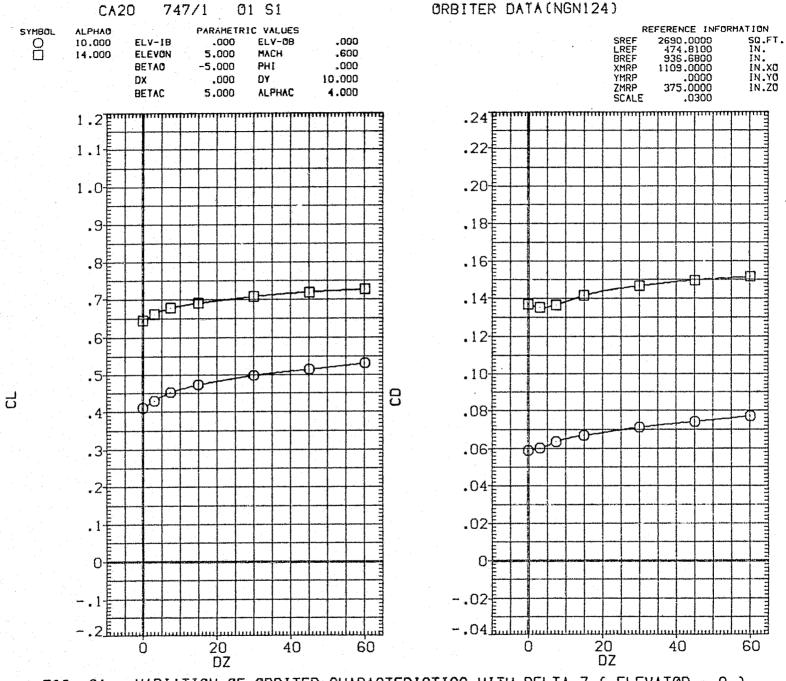
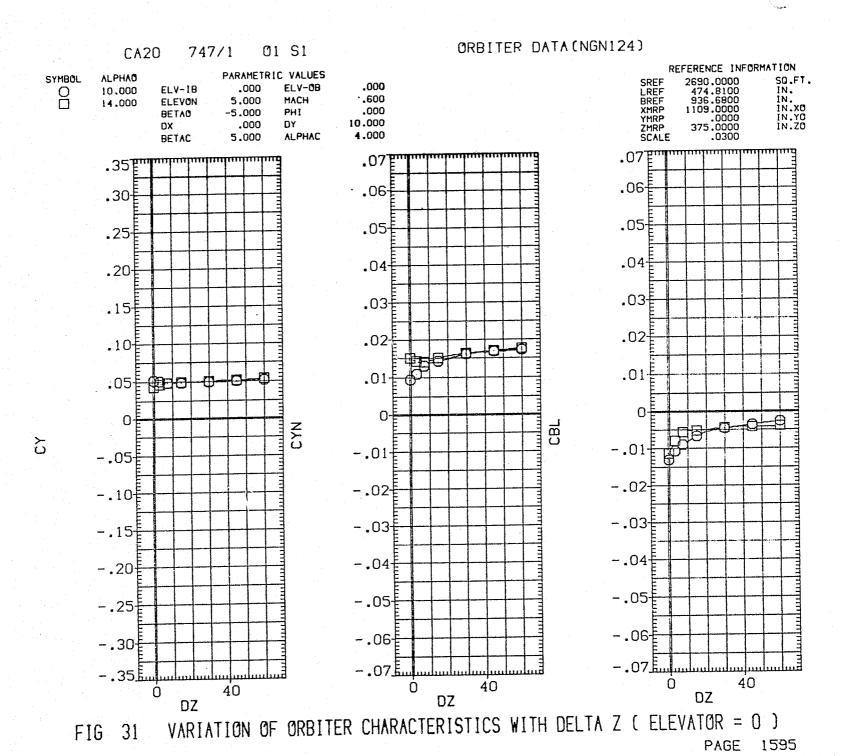
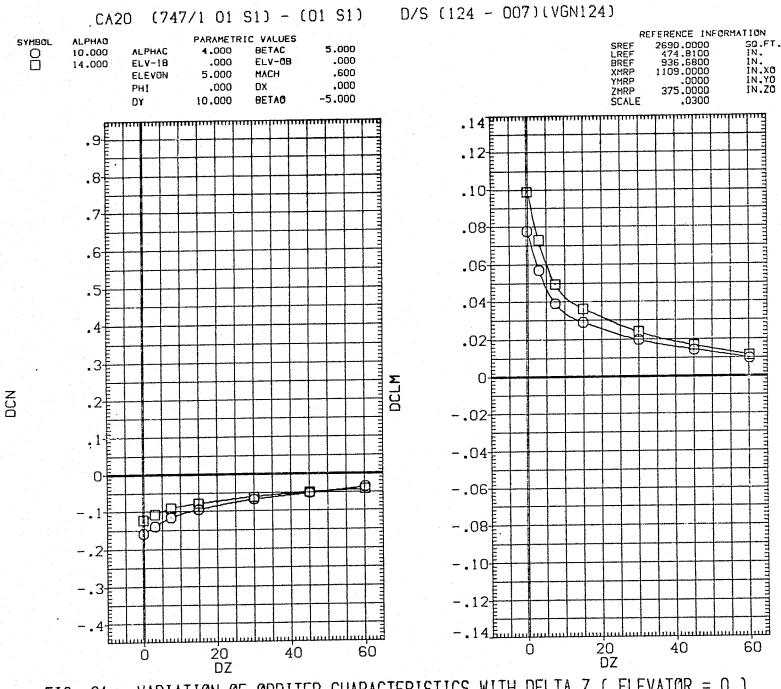


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1594







Carried of Manager States

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1596



CA20 (747/1 01 S1) - (01 S1) D/S (124 - 007)(VGN124) SYMBOL ALPHAO PARAMETRIC VALUES REFERENCE INFORMATION 0 10,000 ALPHAC 4,000 BETAC 5.000 2690,0000 SREF 474.8100 936.6800 1109.0000 .0000 375.0000 IN. LREF 14,000 ELV-18 .000 .000 ELV-0B IN. IN.XO IN.YO IN.ZO BREF ELEVON 5.000 .600 MACH .000 PHI .000 YMRP ZMRP SCALE 10.000 BETAO -5.000 .06£ DCA .05 COEFFICIENT, .04 .03<del>-</del> .02€ FORCE .01 AXIAL -.01 INCREMENTAL - .02 -.03 -.04 -.05<del>‡</del> -.06<del>-</del> 20 50 -10 10 30 40 60 ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT.

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1597

FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1598



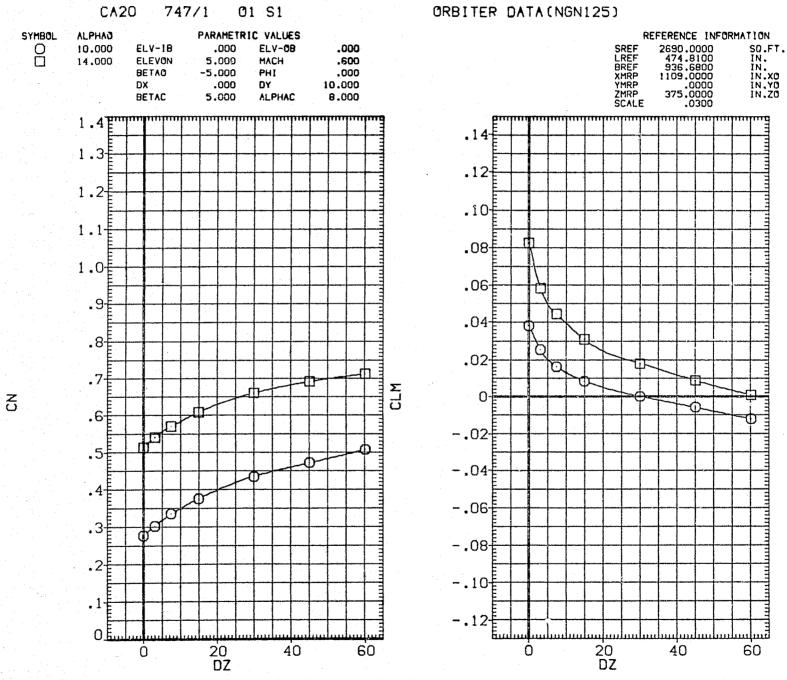


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1599

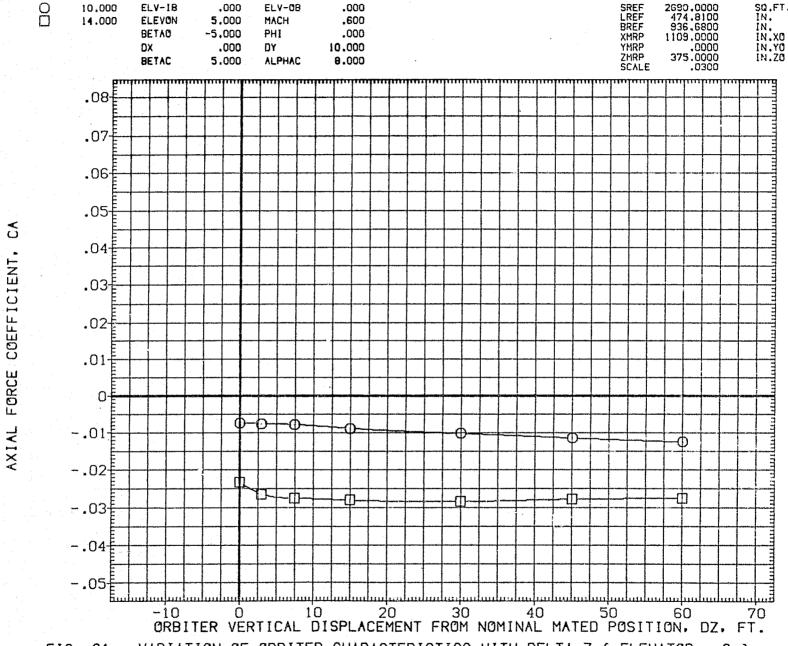
SYMBOL

REFERENCE INFORMATION

SQ.FT.

2690,0000

SREF



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 ) FIG 31 PAGE 1600

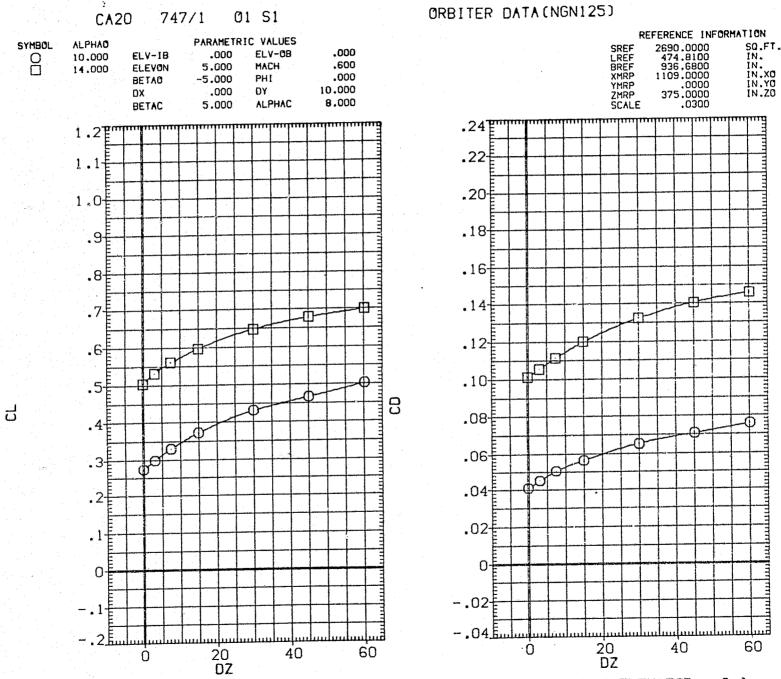
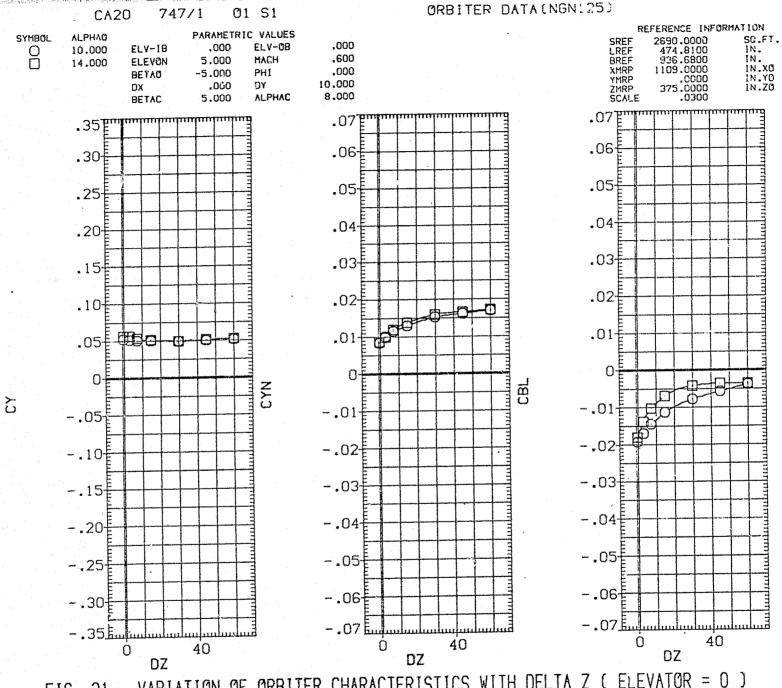


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1601



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 ) FIG 31 PAGE 1602



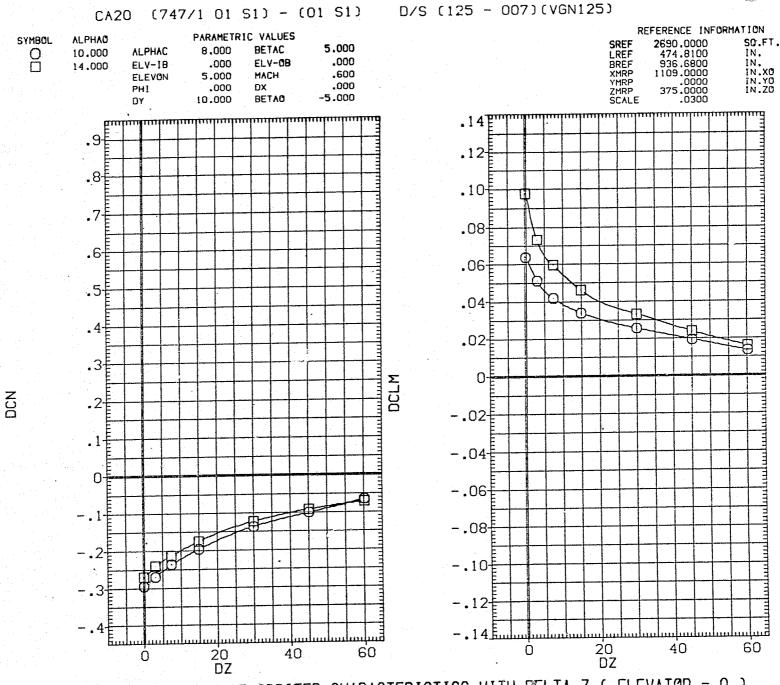


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1603

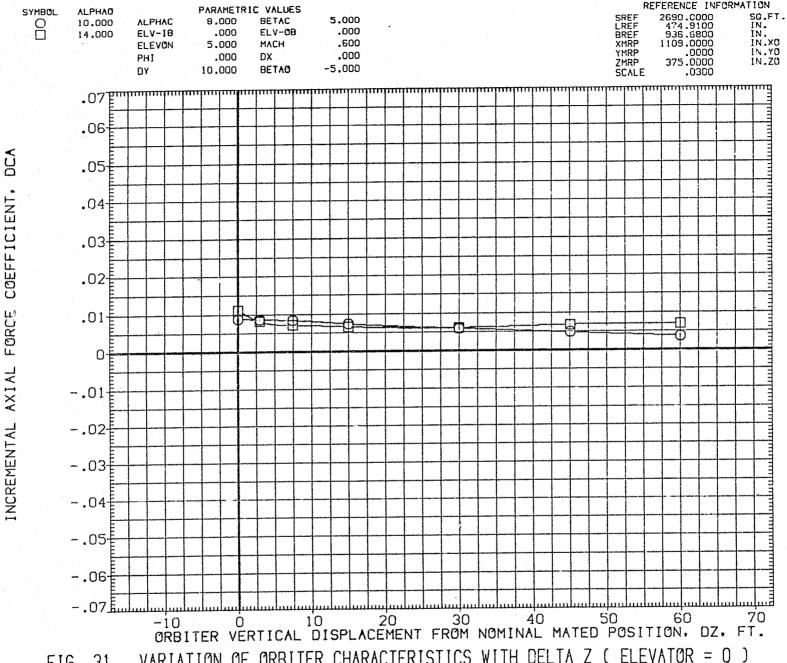
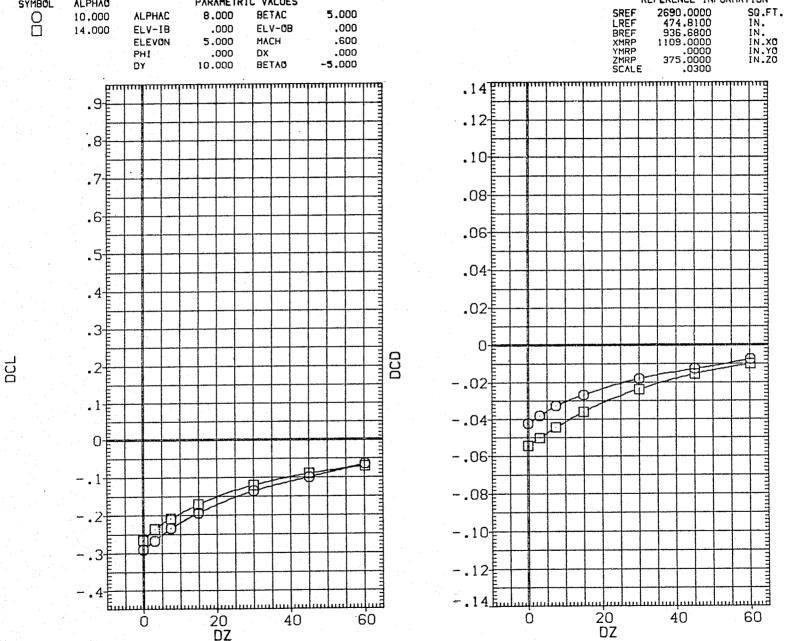


FIG 31 VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 )
PAGE 1604

SYMBOL



VARIATION OF ORBITER CHARACTERISTICS WITH DELTA Z ( ELEVATOR = 0 ) 31 FIG PAGE 1605

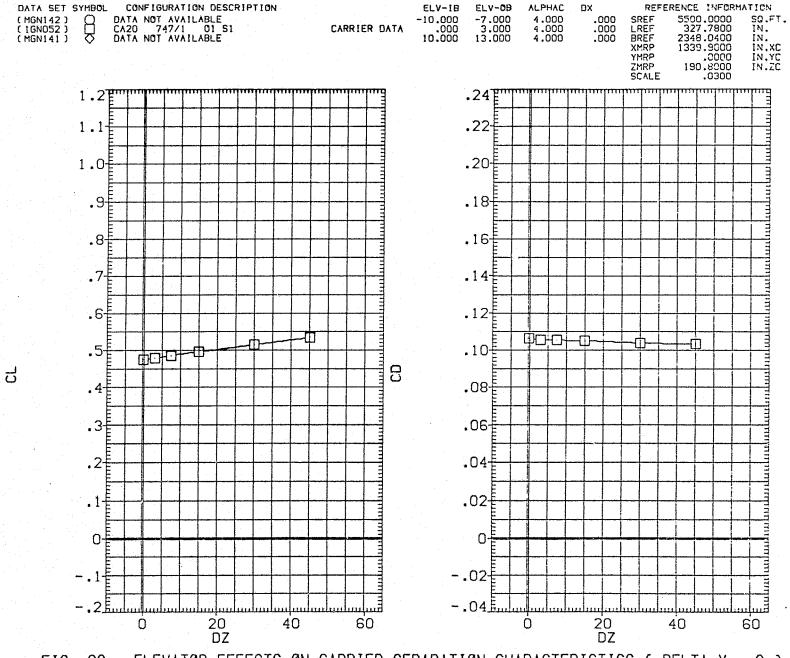


FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(A)ALPHAO= 6.00

PAGE 1606

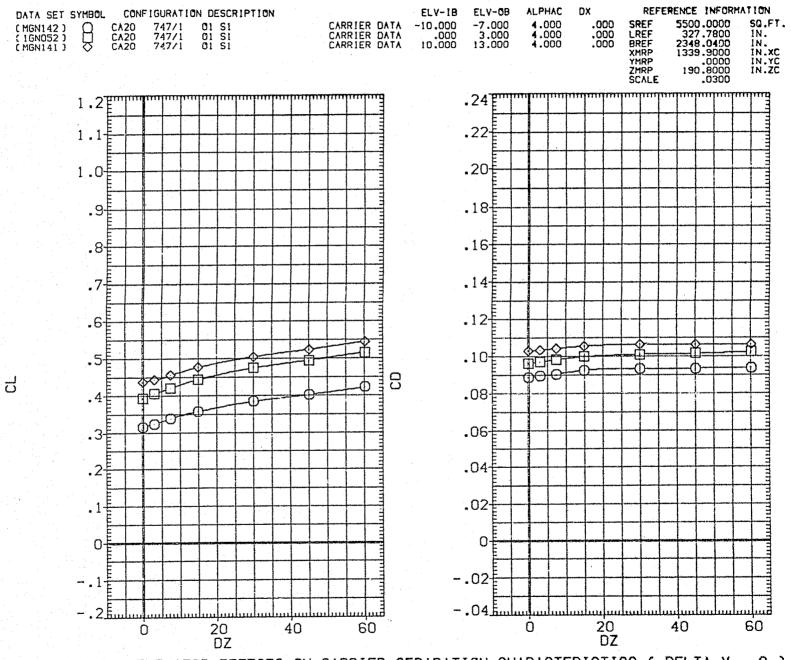


FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B) ALPHAO = 10.00

PAGE 1607

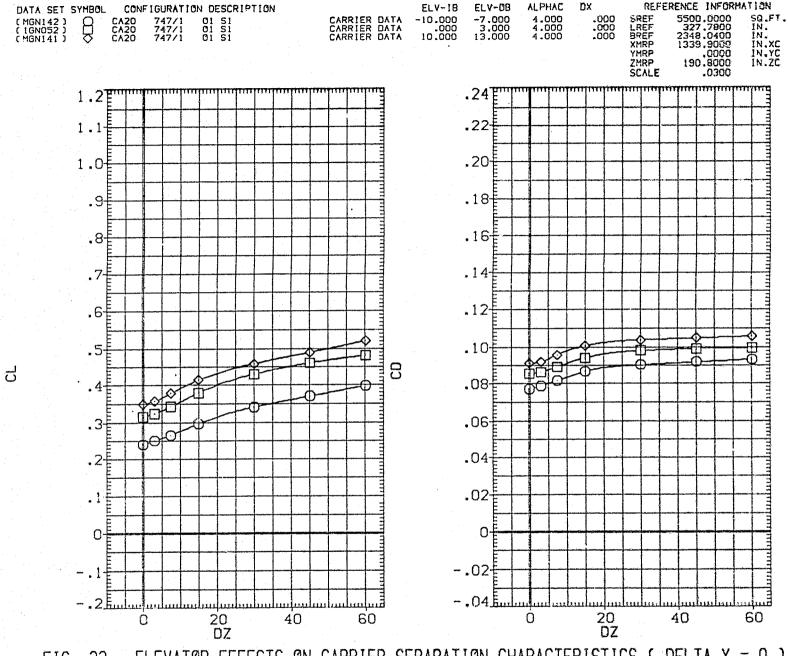
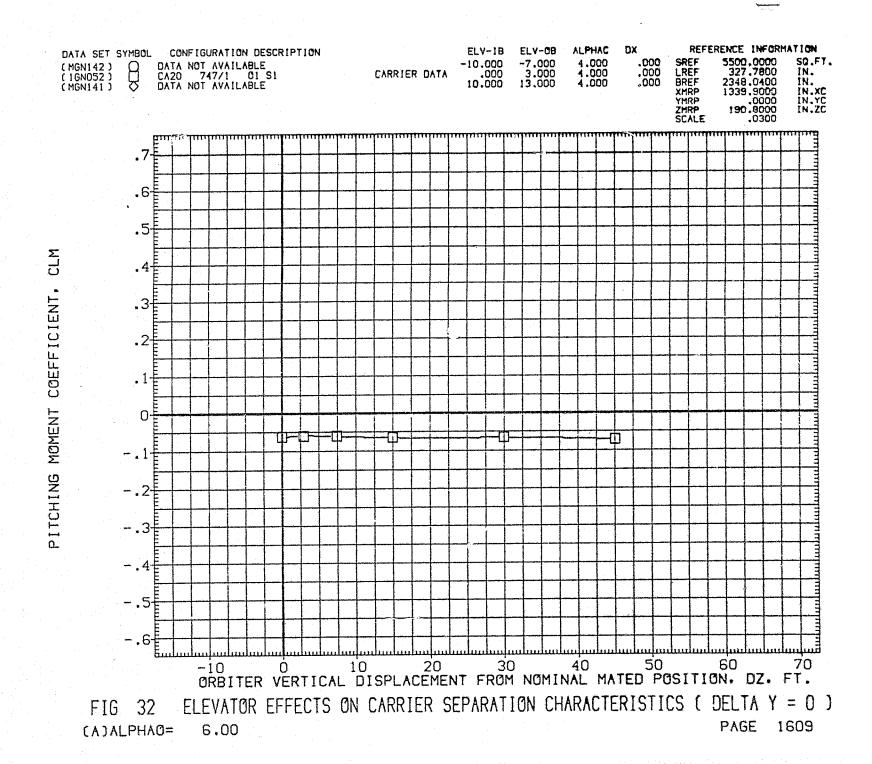
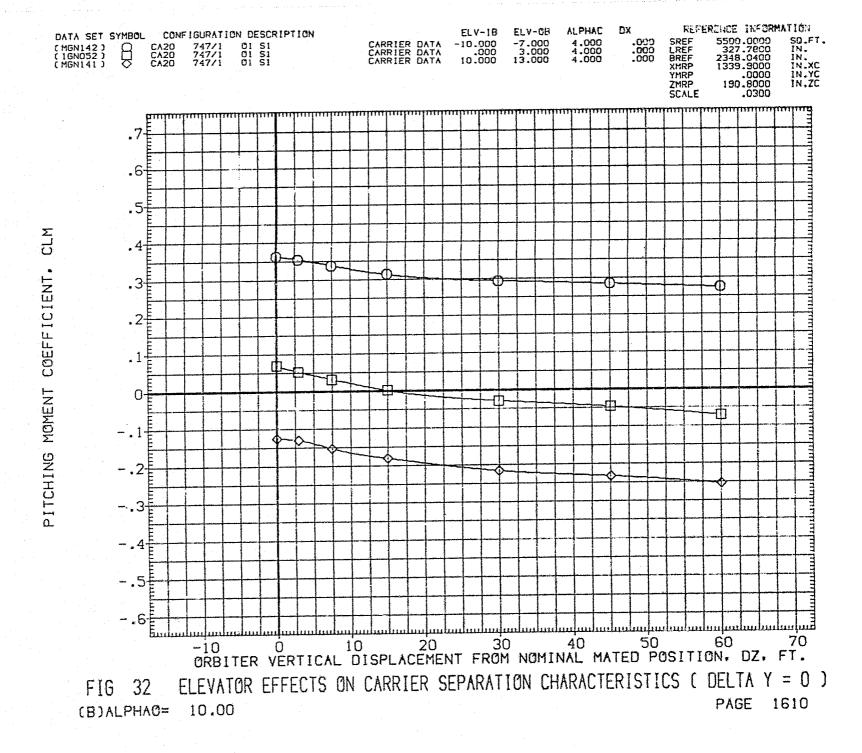
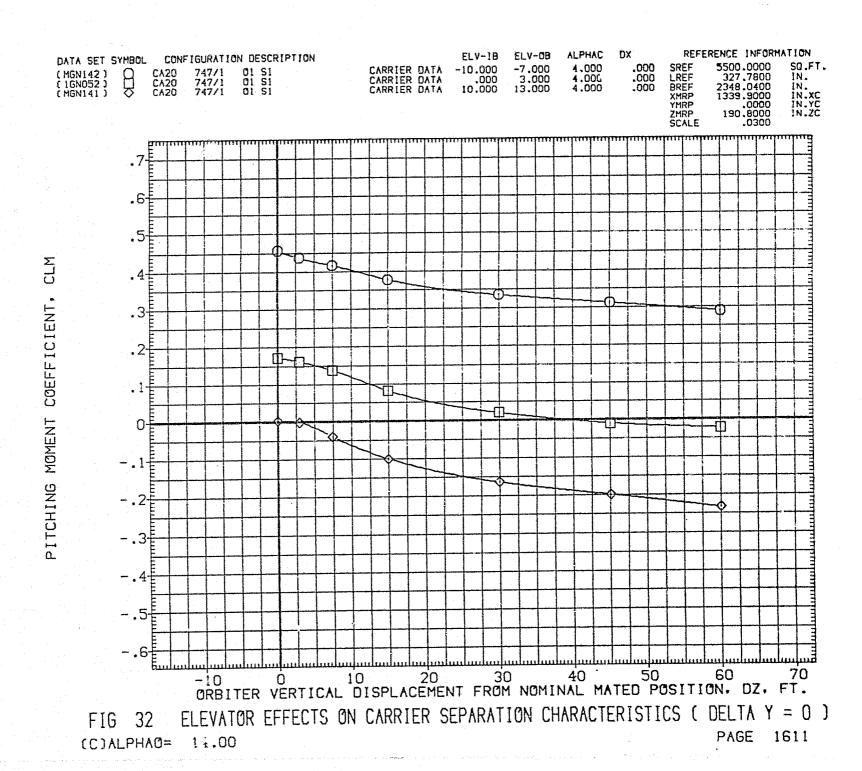


FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(C)ALPHAO= 14.00 PAGE 1608







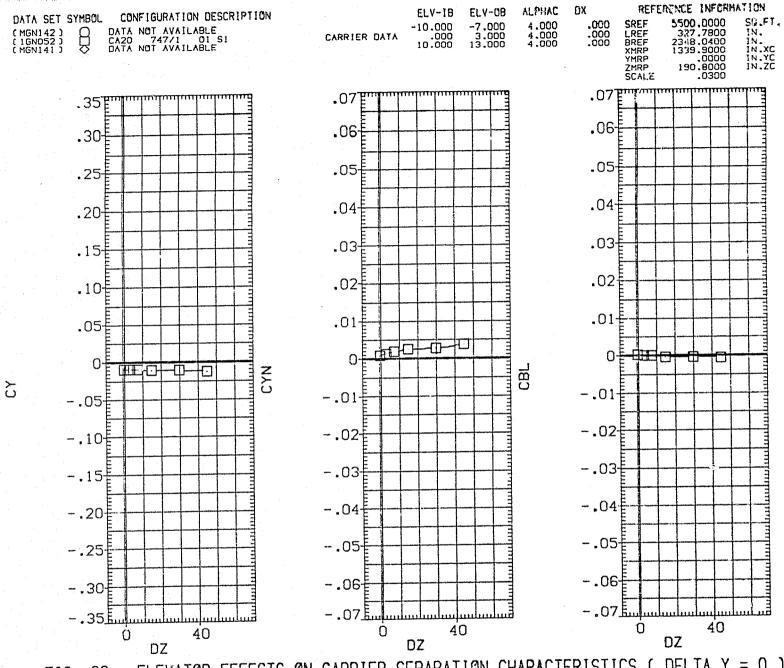


FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

PAGE 1612

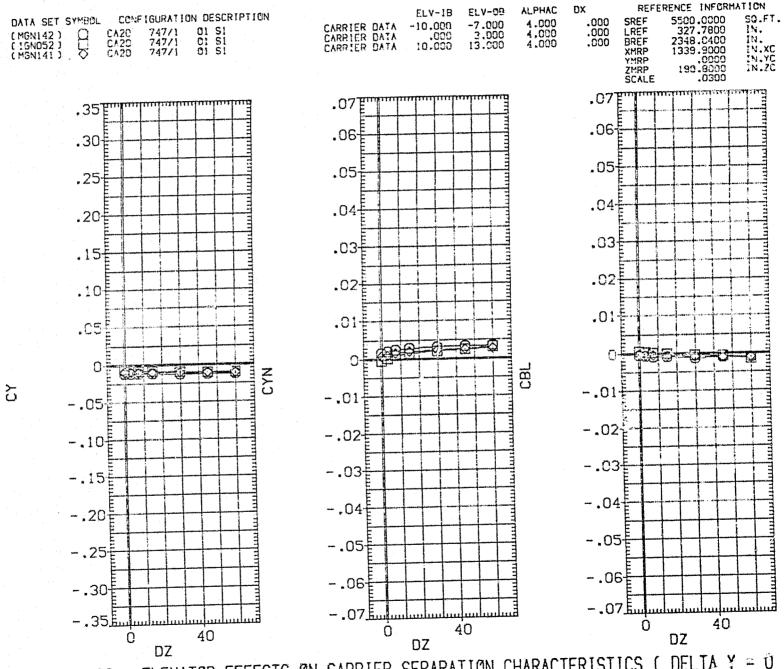
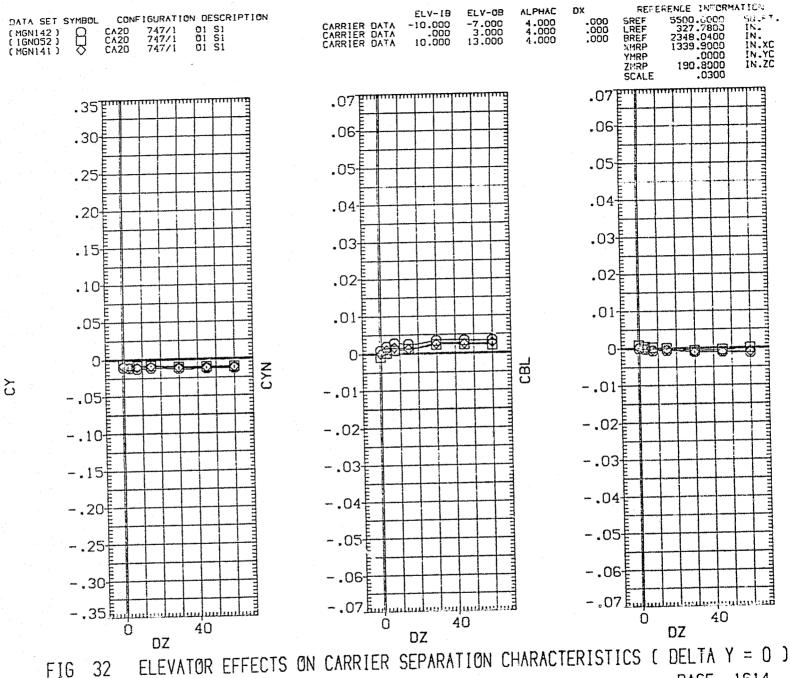


FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B) ALPHAG= 10.00



PAGE 1614 (C)ALPHAO= 14.00

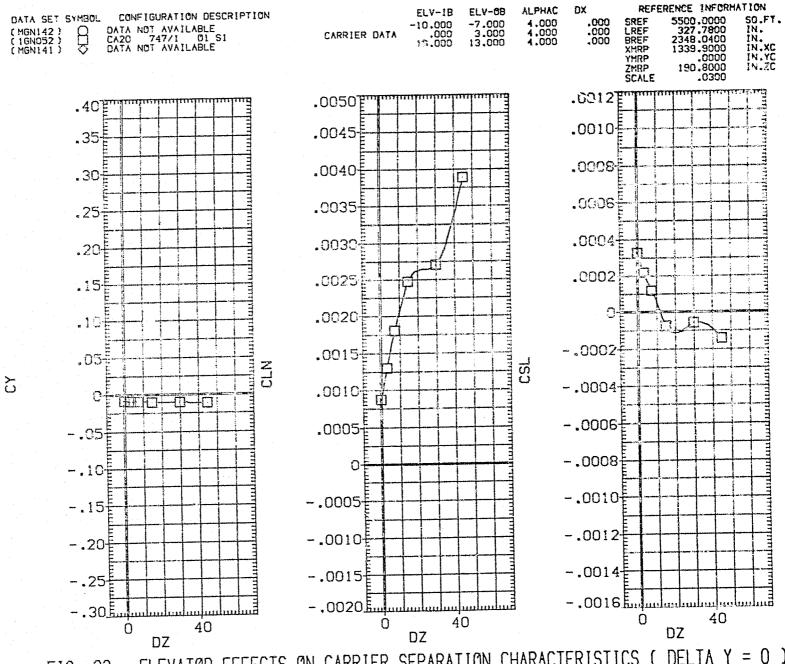
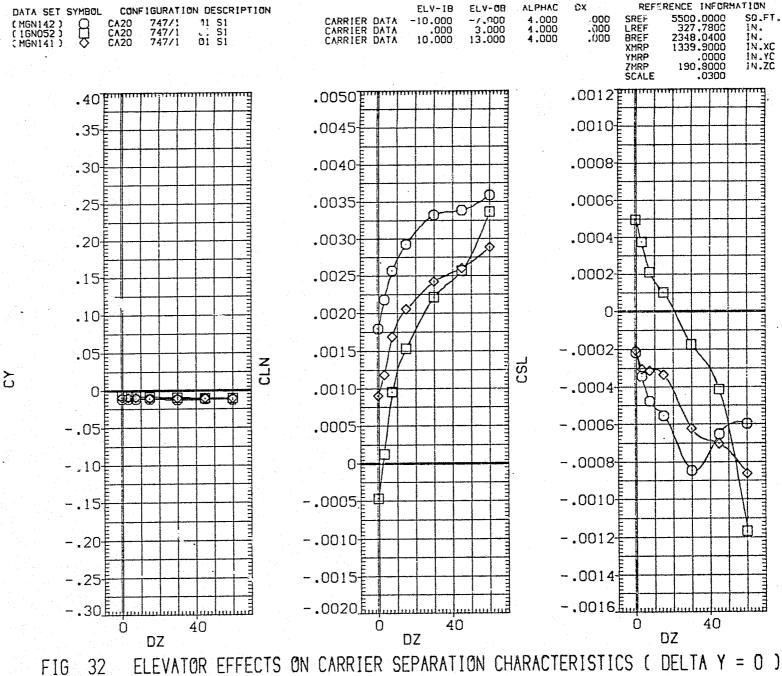


FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 :



PAGE 1616 (B)ALPHAO= 10.00

FIG 32 ELEVATOR EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )
PAGE 1617

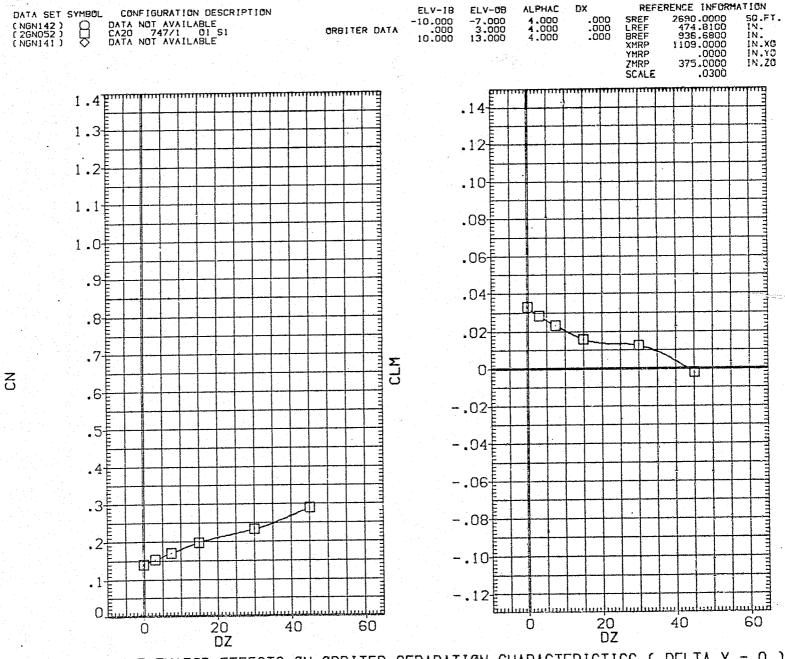


FIG 33 ELEVATOR EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(A)ALPHAO = 6.00

PAGE 1618

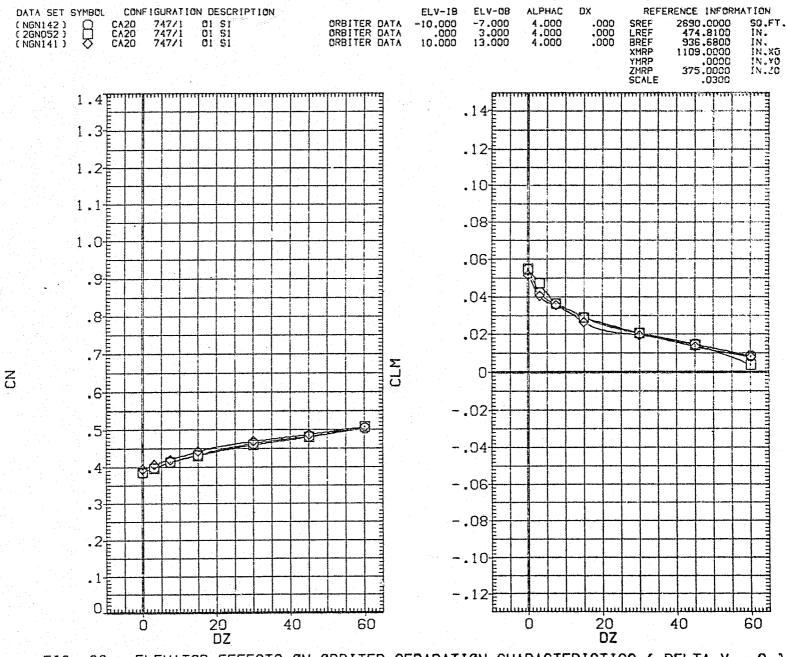


FIG 33 ELEVATOR EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B) ALPHAO = 10.00 PAGE 1619

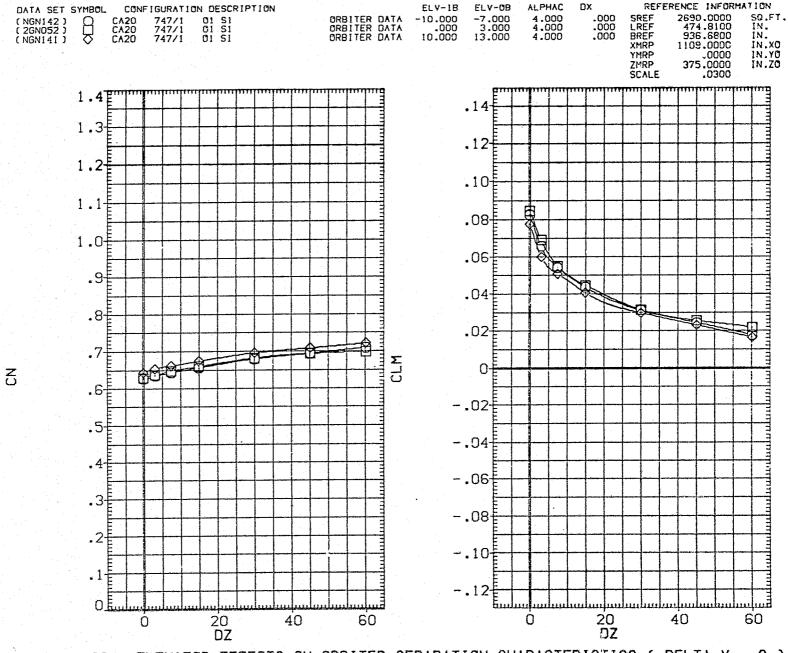
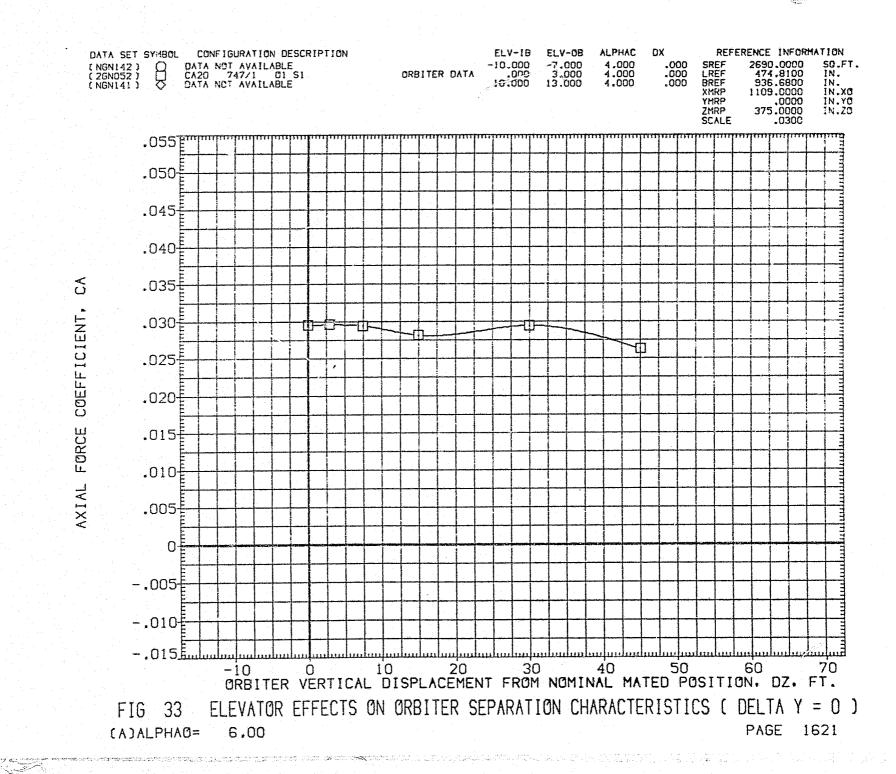
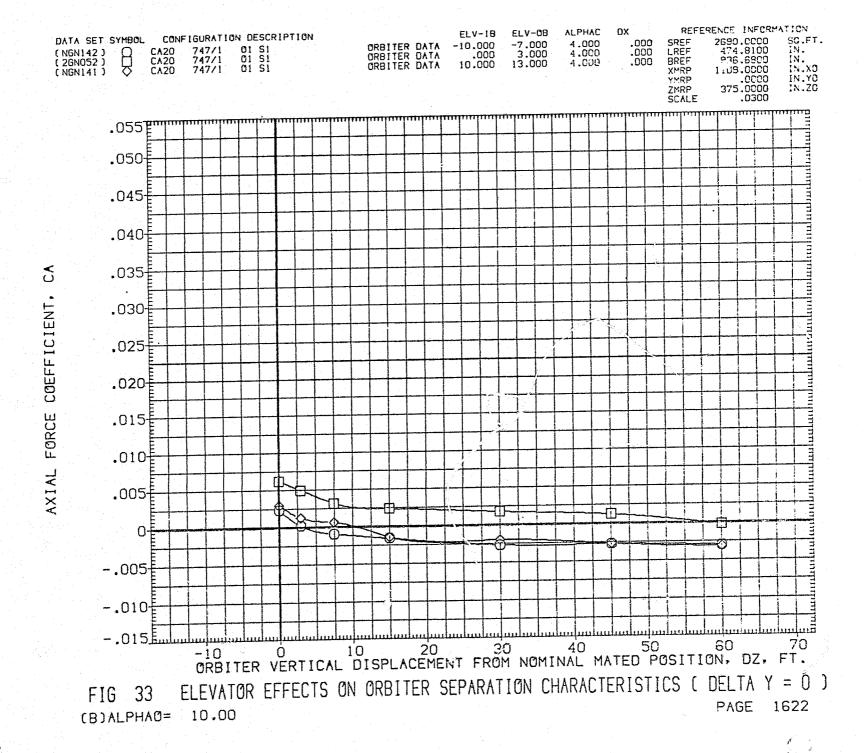
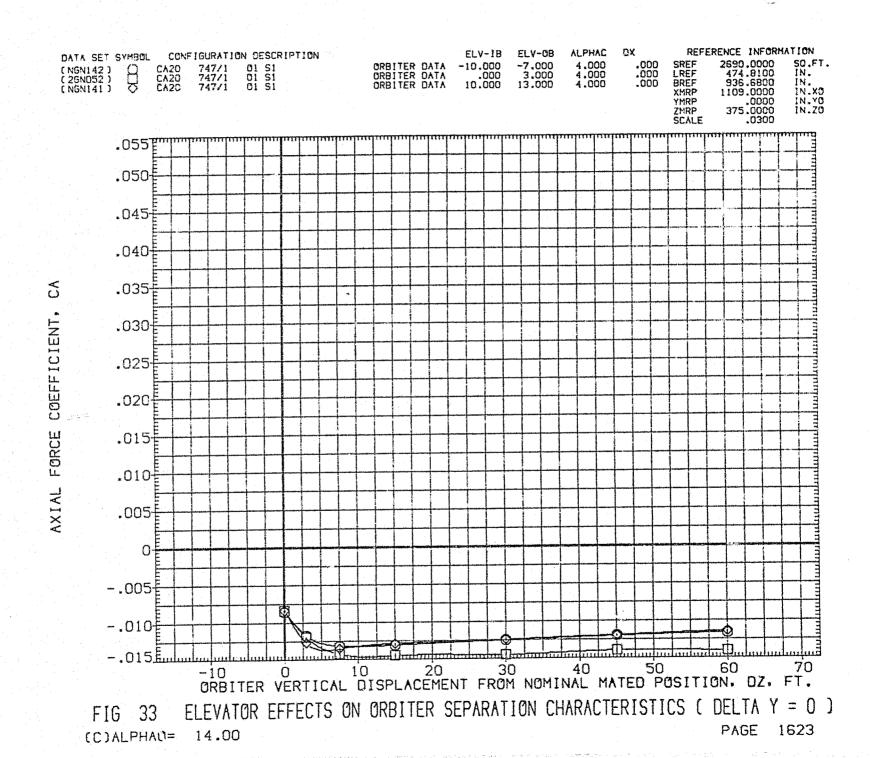


FIG 33 ELEVATOR EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

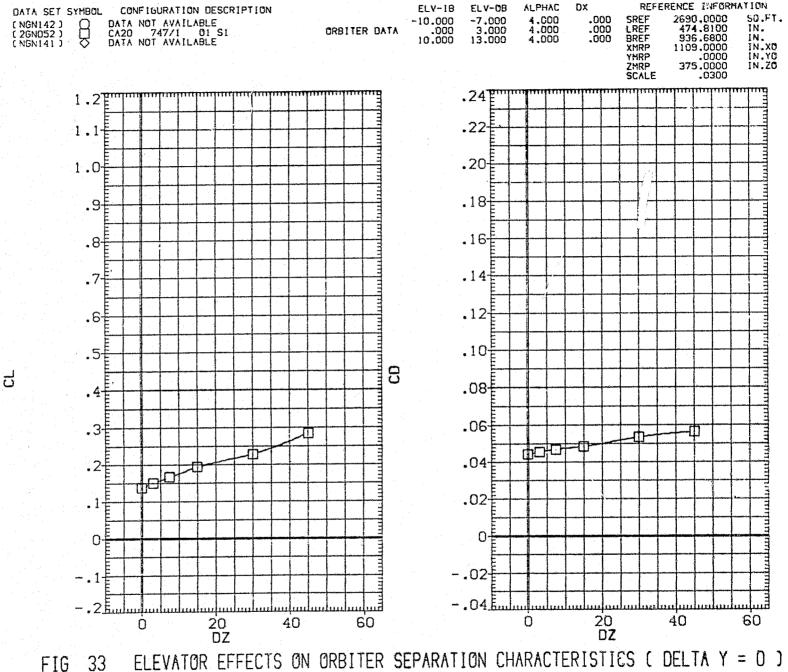
(C) ALPHAO = 14.00 PAGE 1620





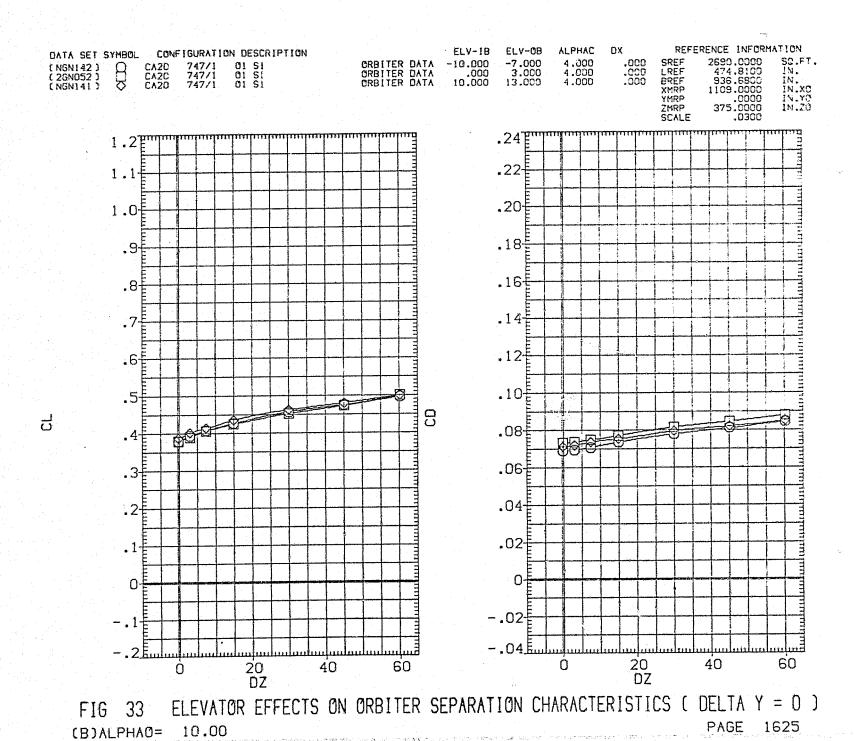


ANTER PROPERTY AND ASSESSED.



PAGE 1624 6.00 (A)ALPHAO=





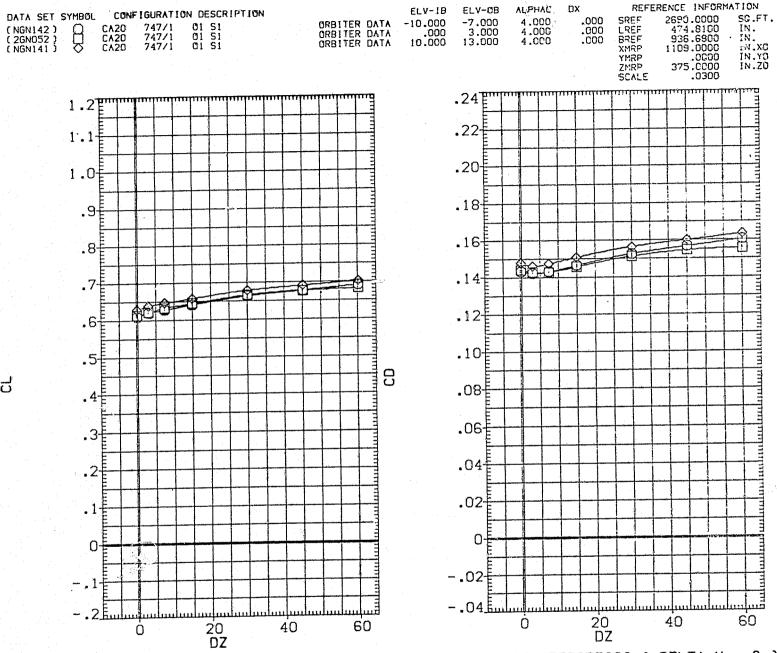


FIG 33 ELEVATOR EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(C) ALPHAG= 14.00



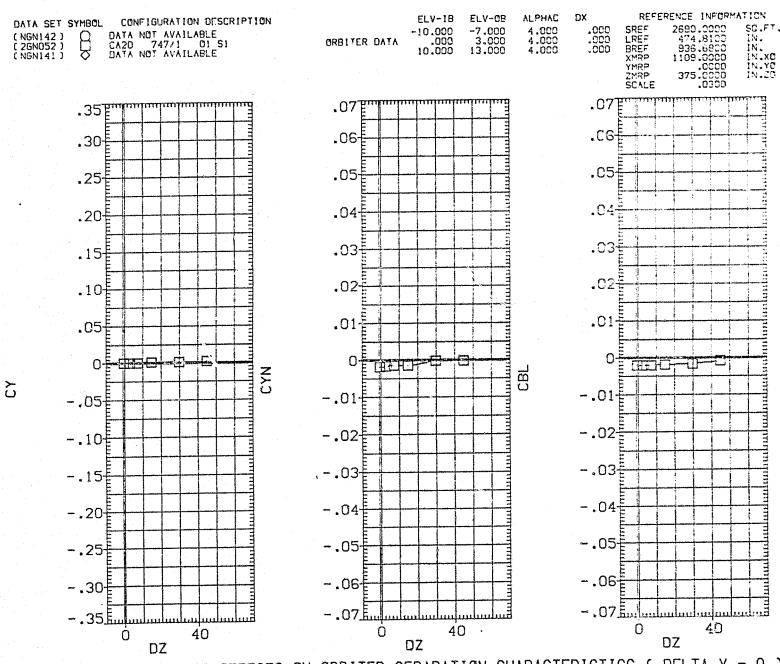
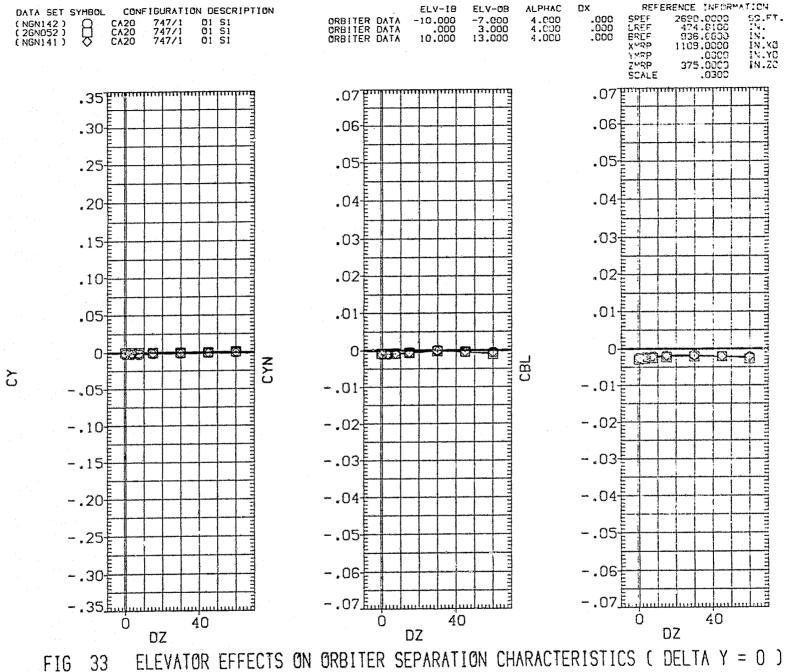


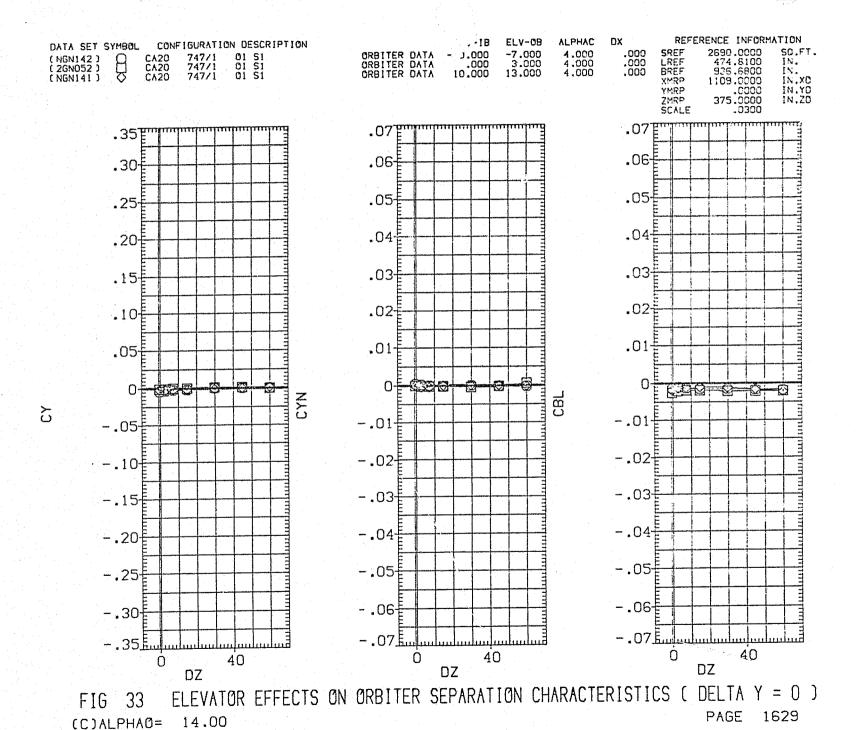
FIG 33 ELEVATOR EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

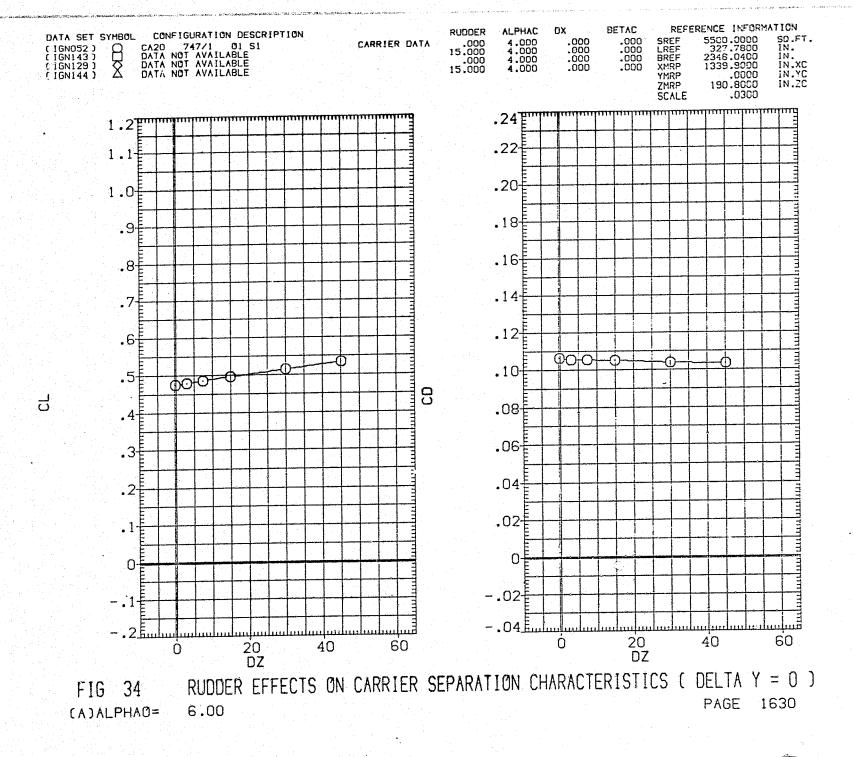
(A)ALPHAO= 6.00

PAGE 1627



PAGE 1628 10.00 (B)ALPHAO=





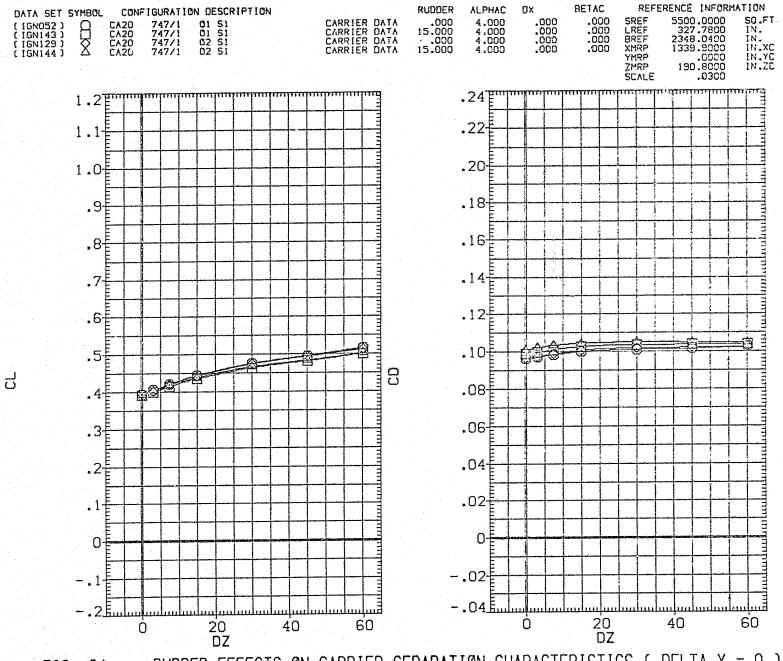
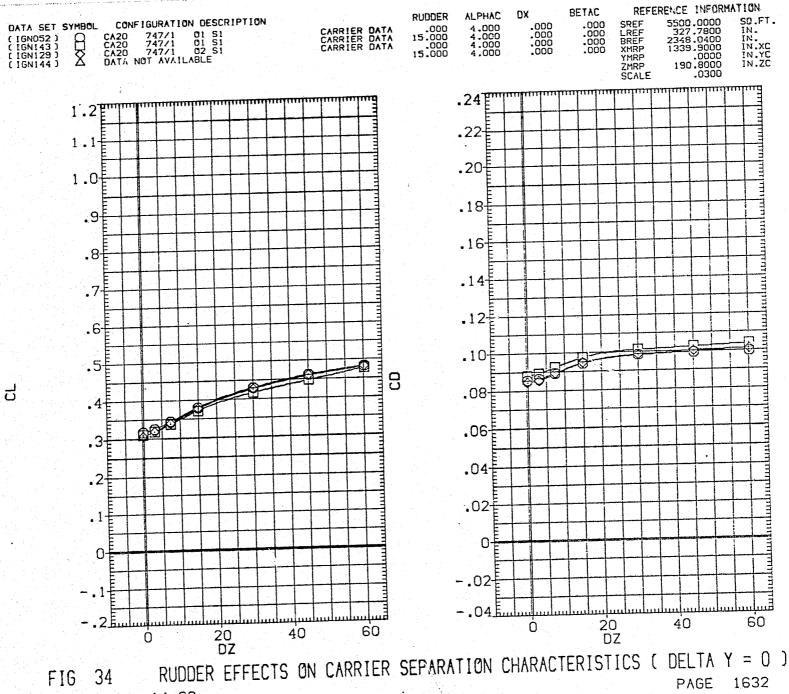


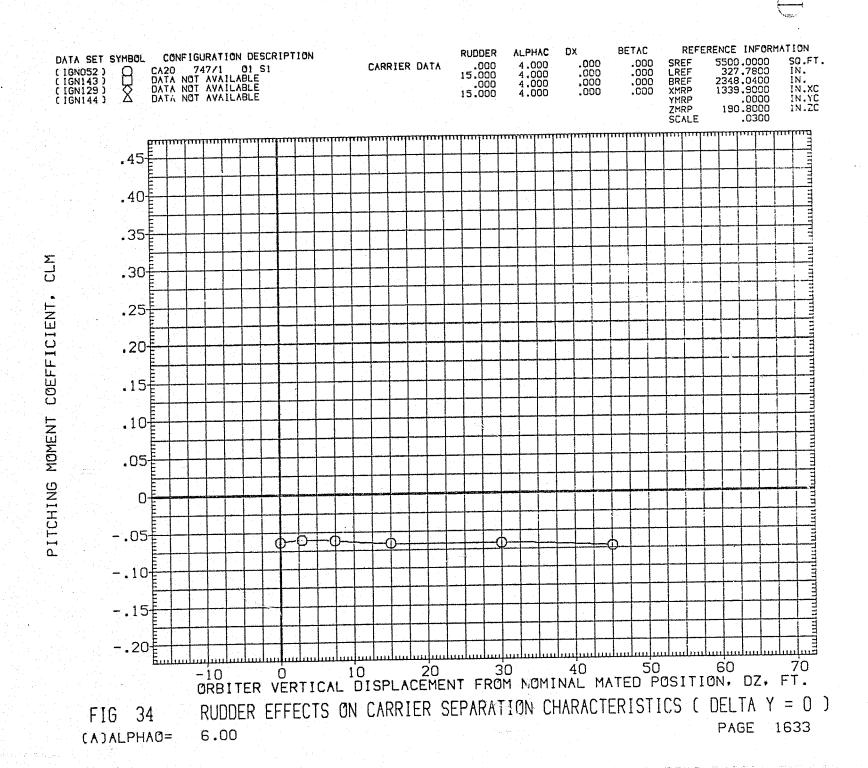
FIG 34 RUDDER EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

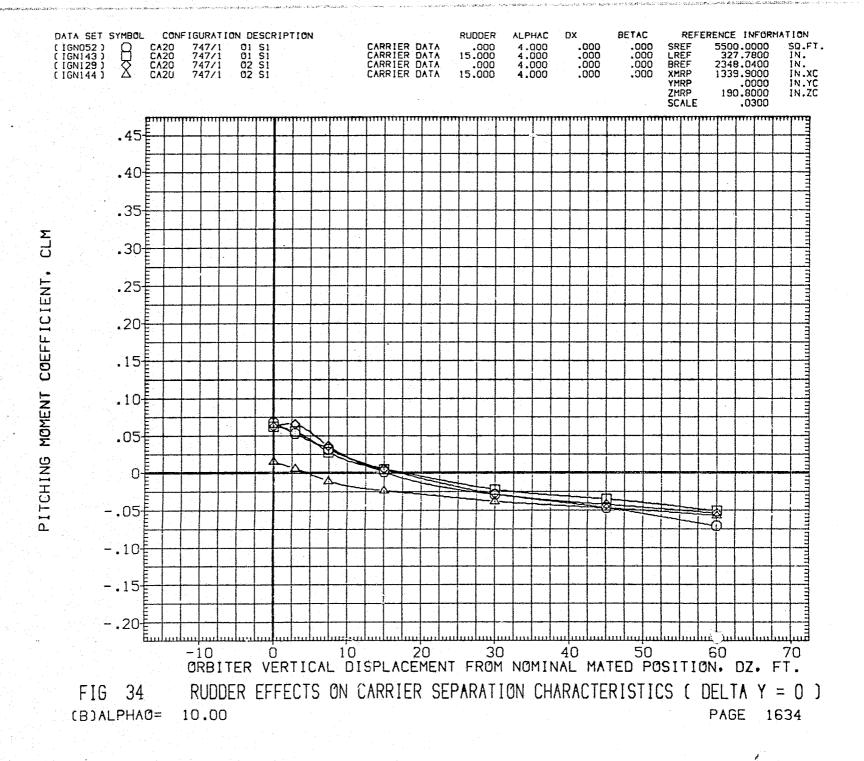
(B) ALPHAO = 10.00

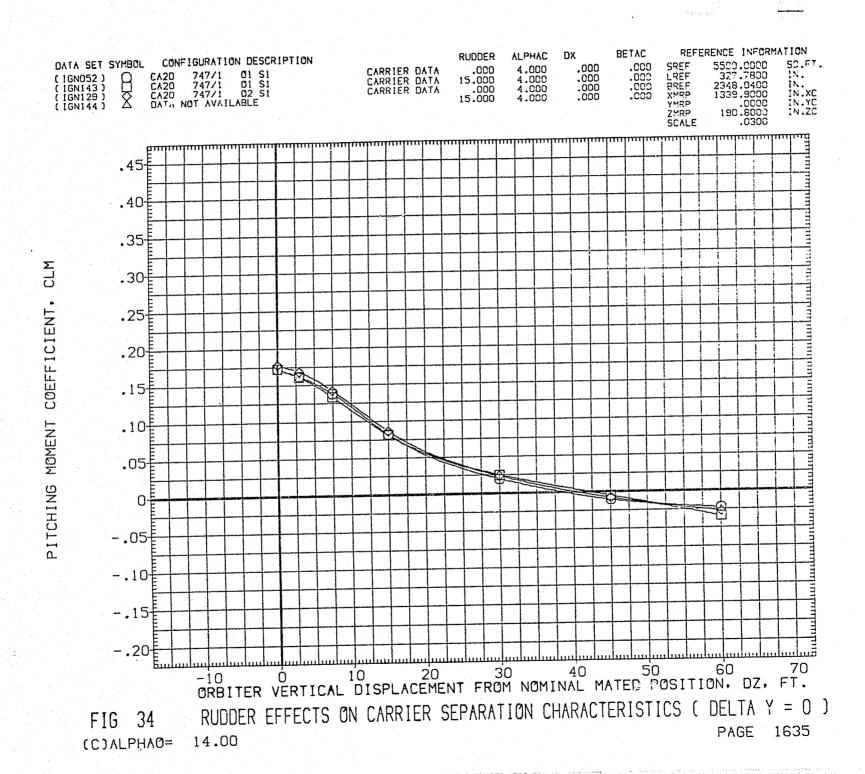
PAGE 1631

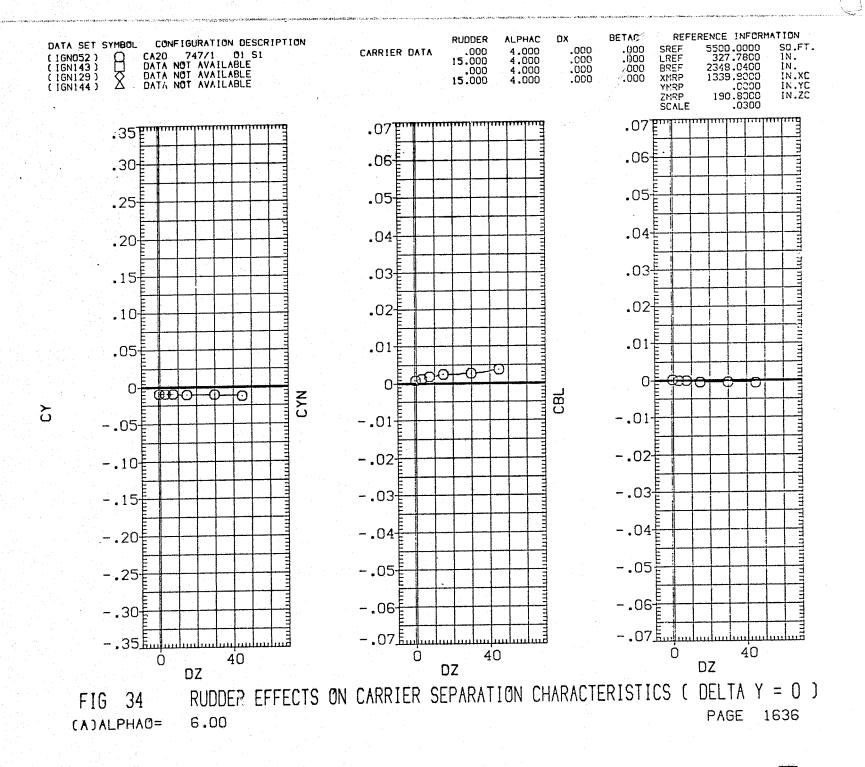


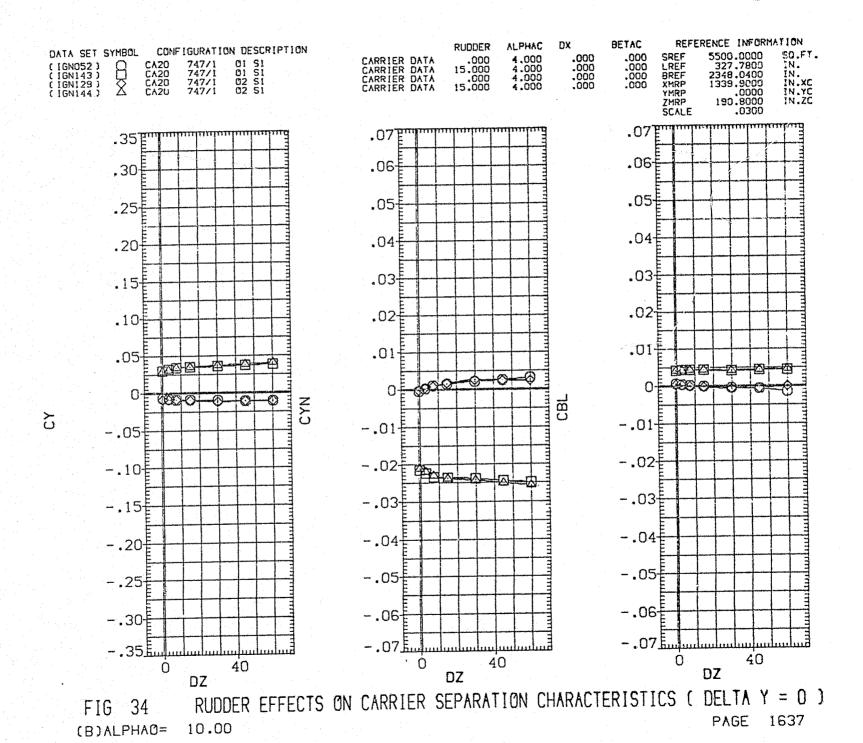
14.00 (C)ALPHAO=

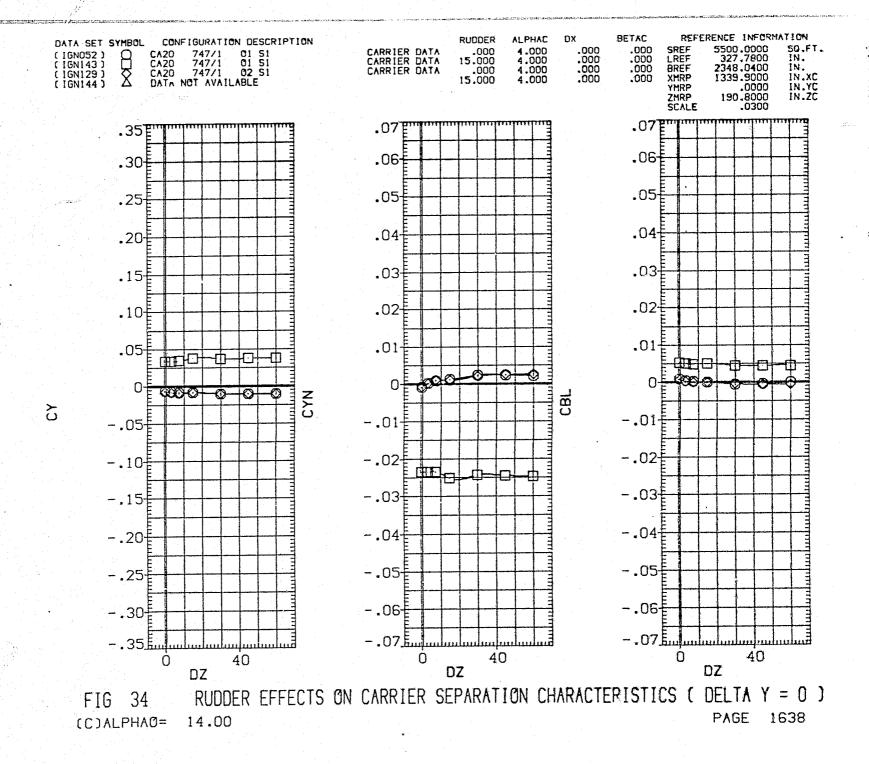


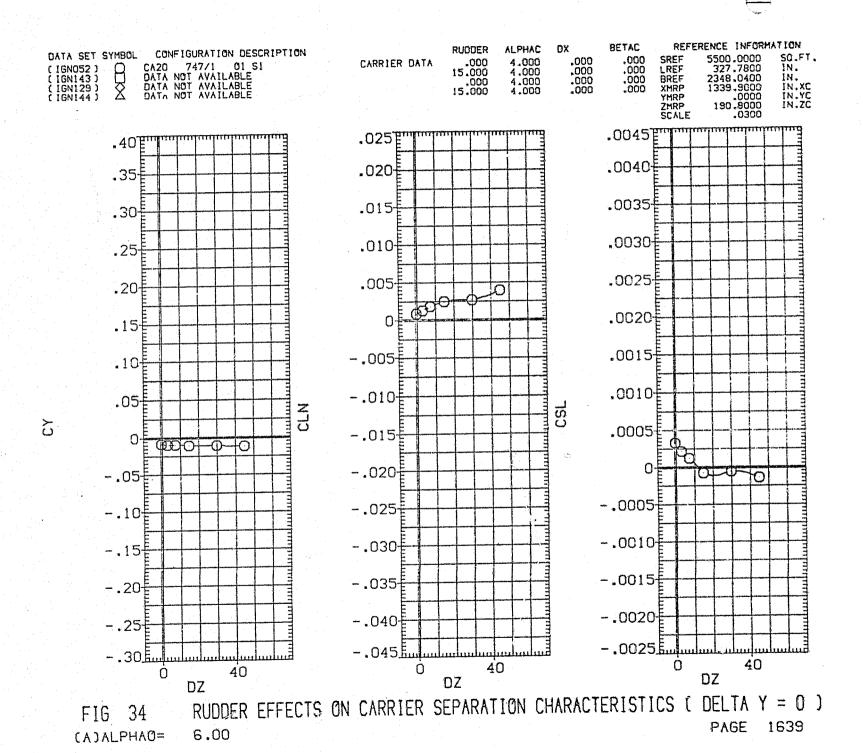


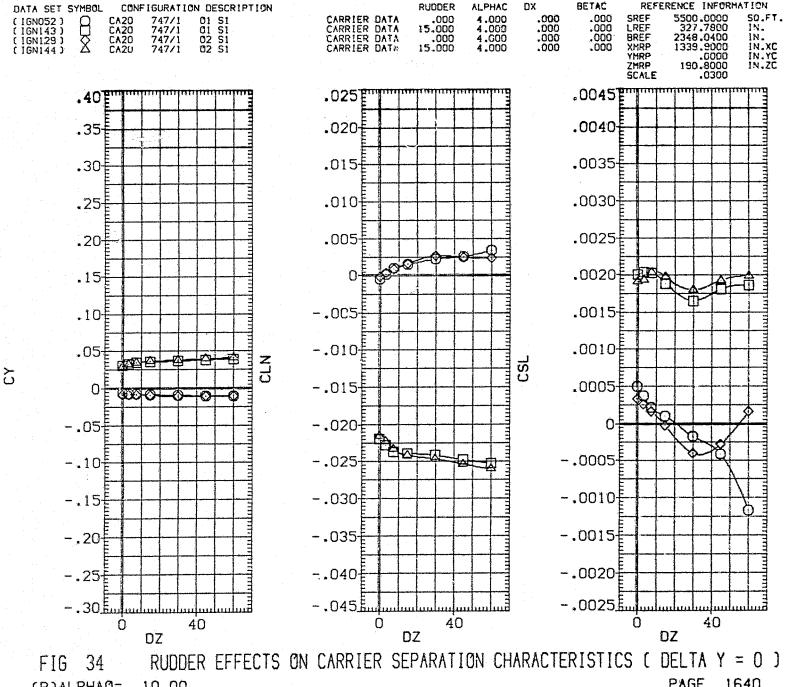




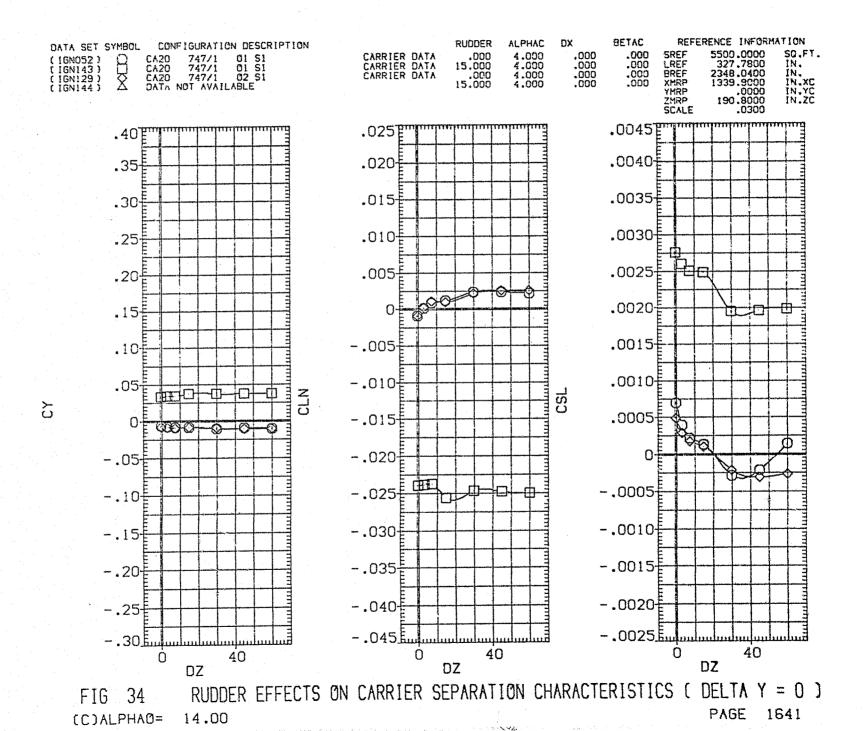


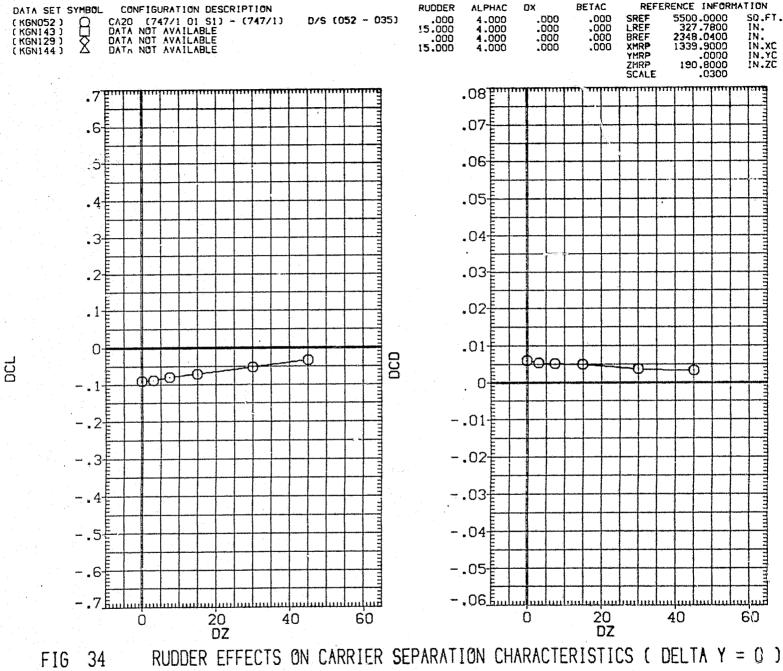




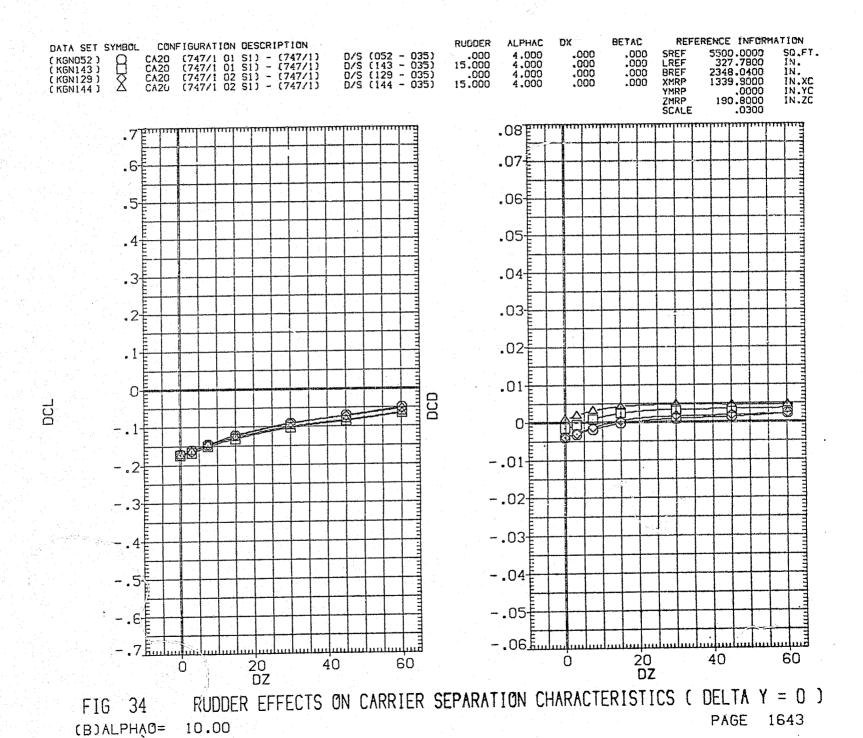


PAGE 1640 (B)ALPHAO= 10.00





PAGE 1642 6.00 (A)ALPHAØ=



a security if 119 propagation are

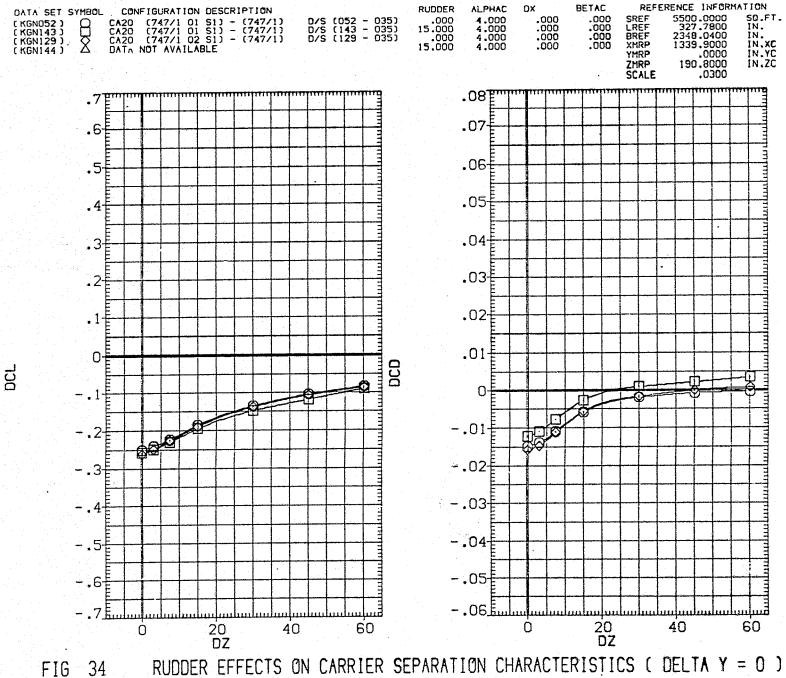
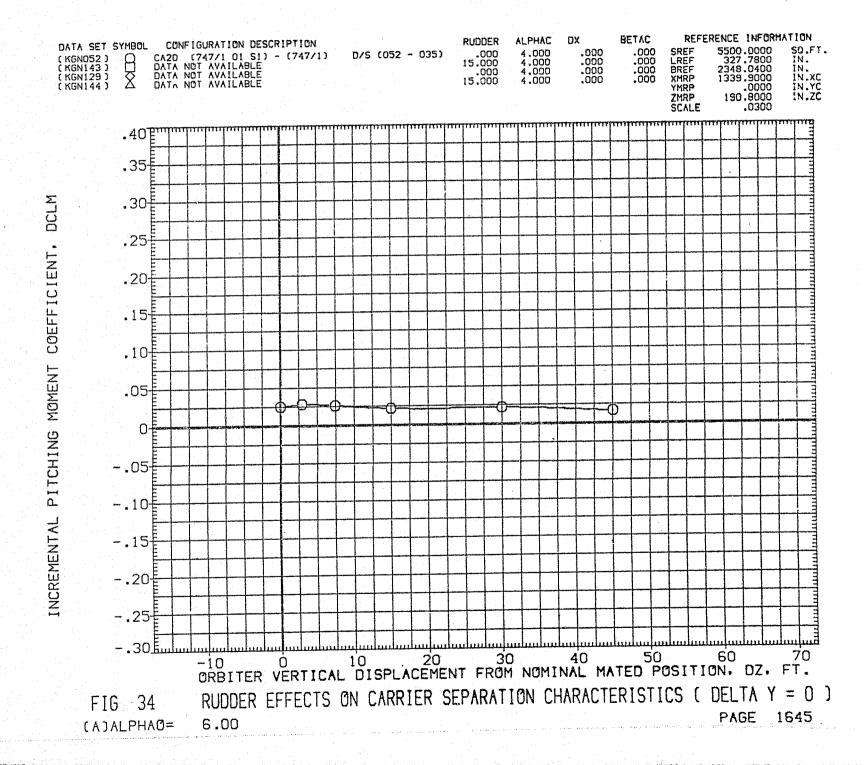
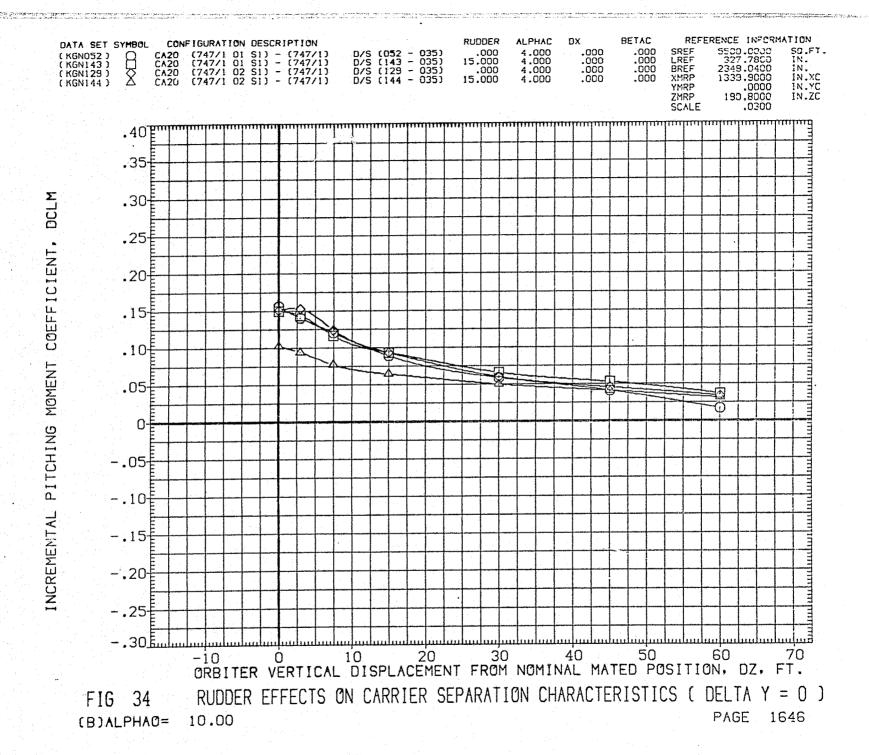
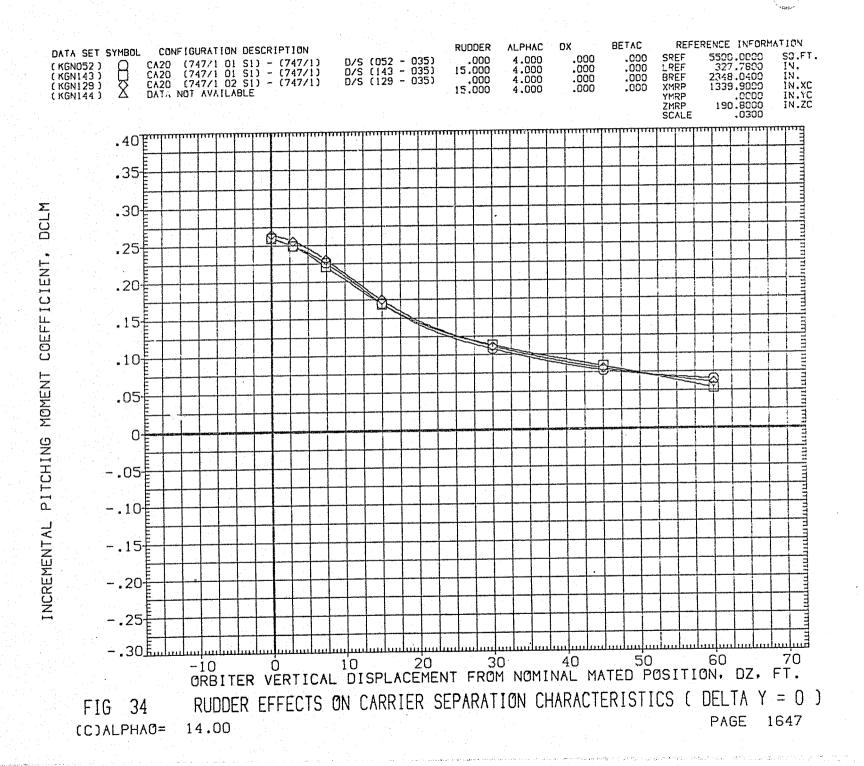


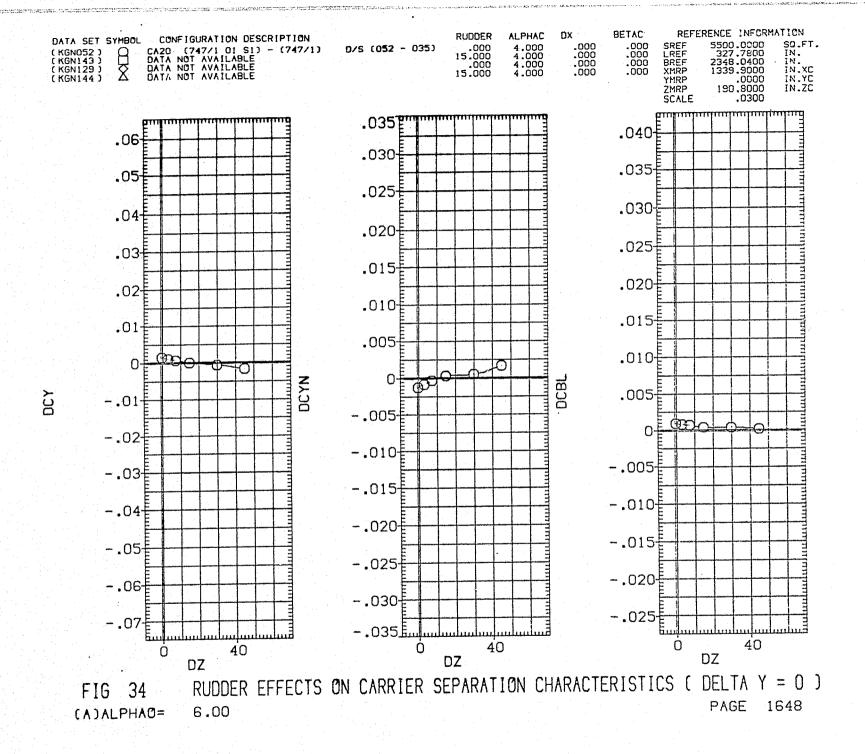
FIG 34 RUDDER EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 (C)ALPHAO= 14.00 PAGE 1644

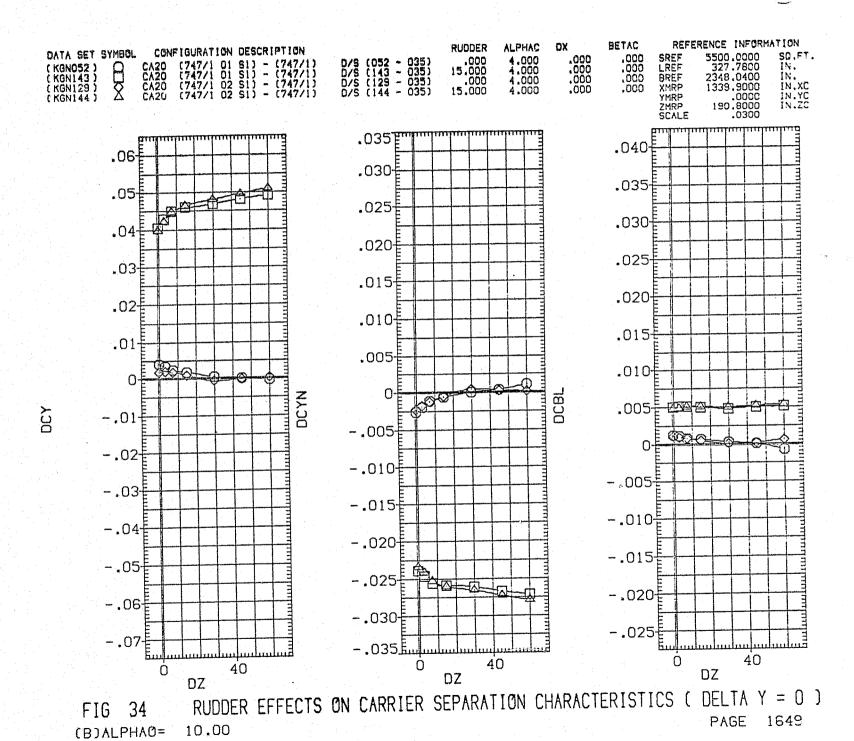


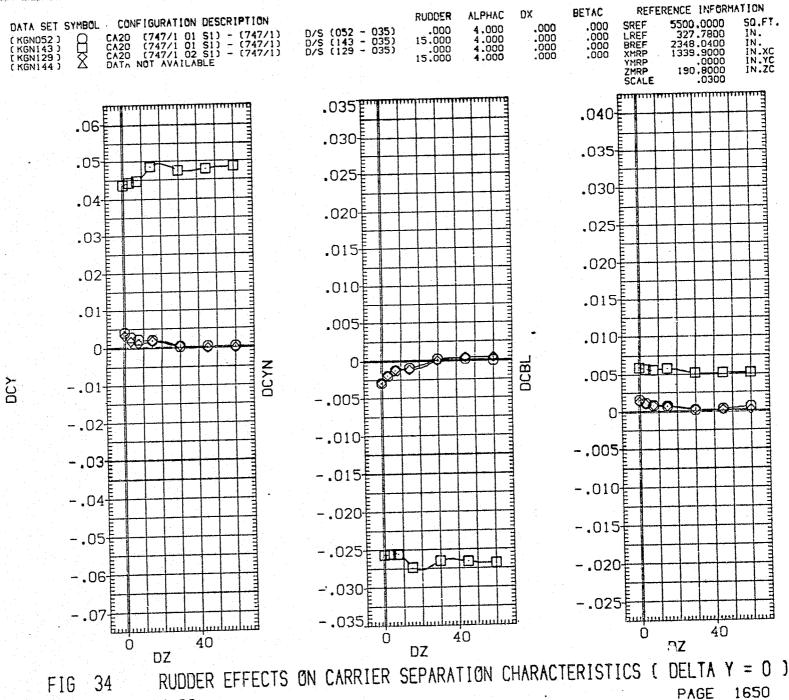




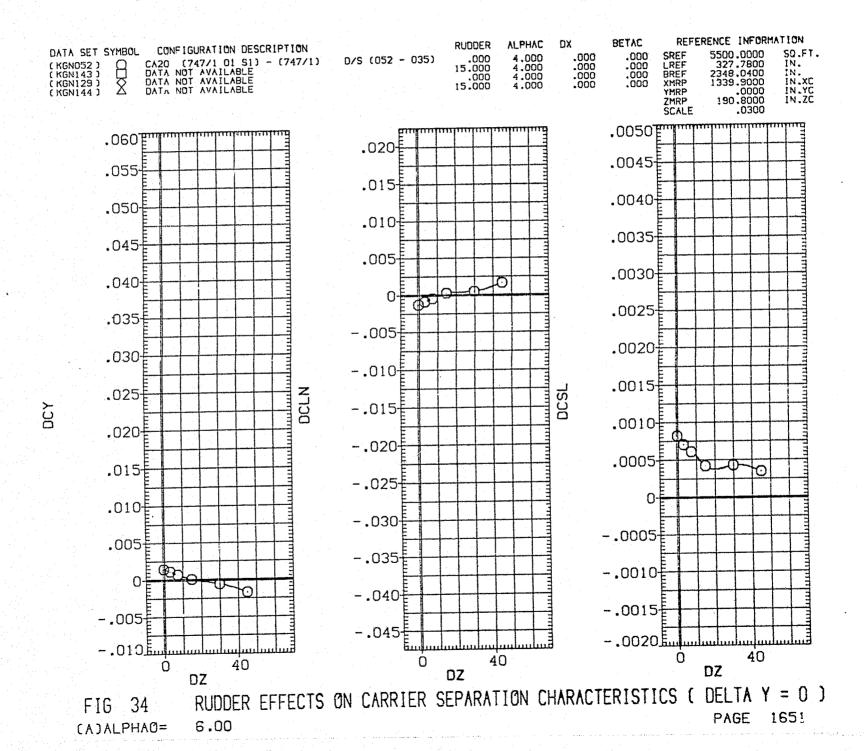


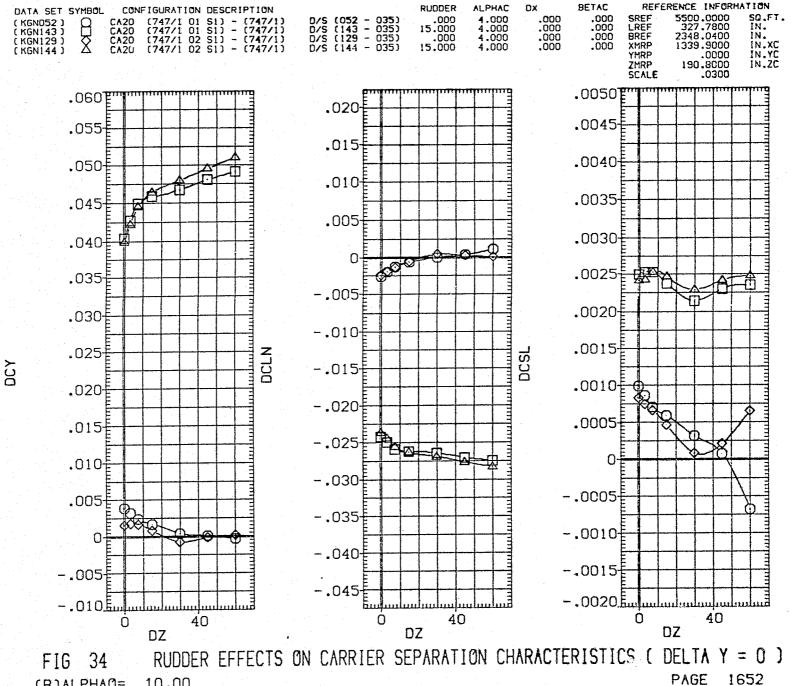




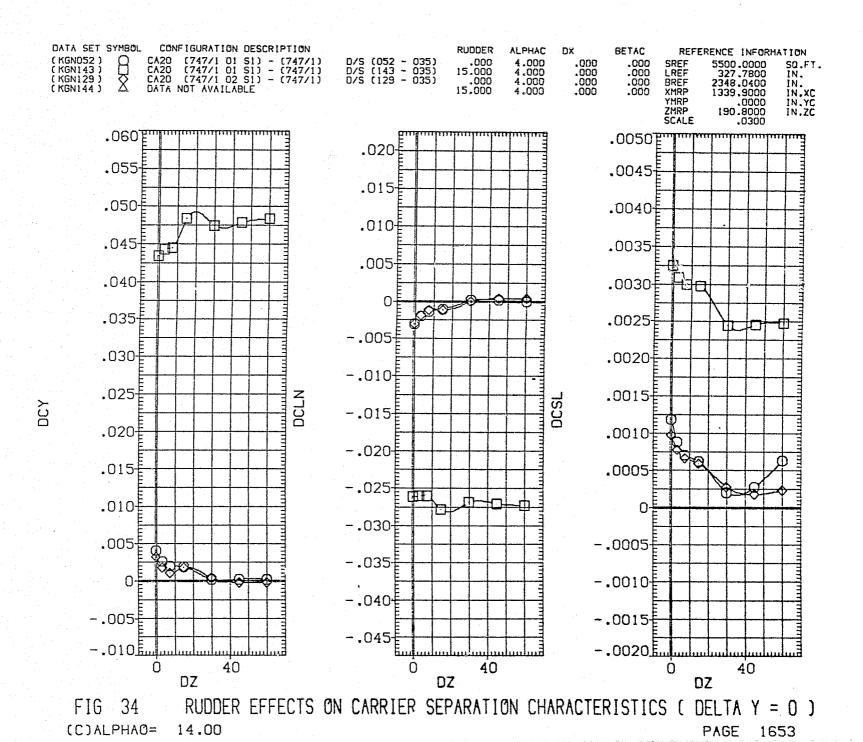


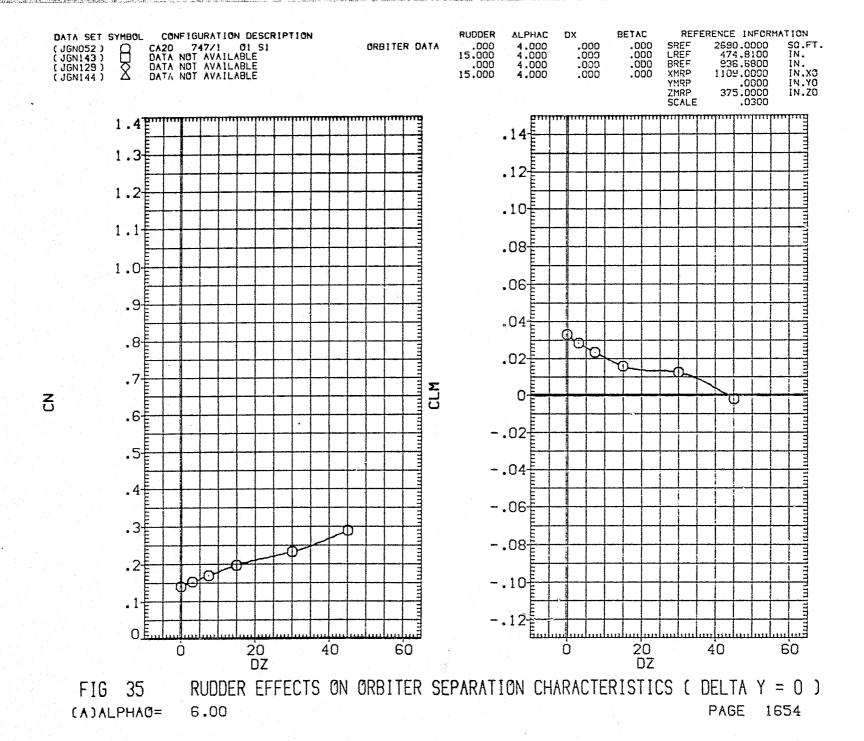
14.00 (C)ALPHAO=

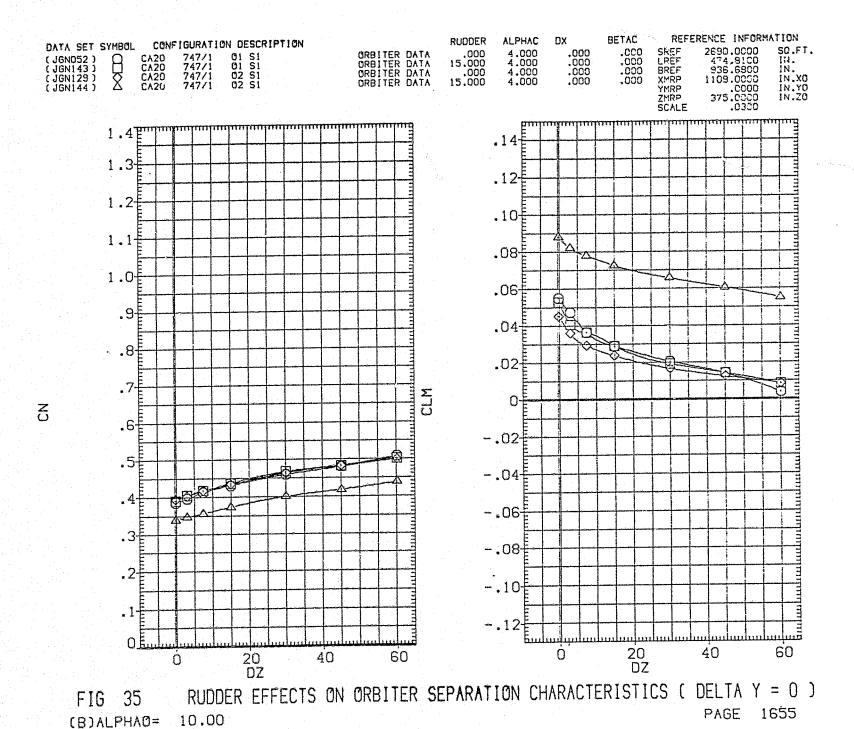


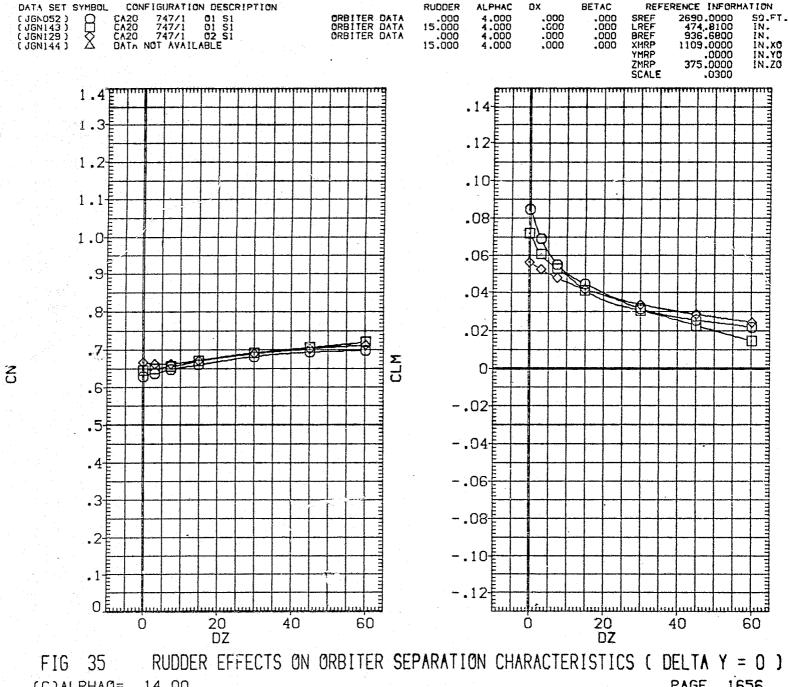


1652 10.00 (B)ALPHAO=

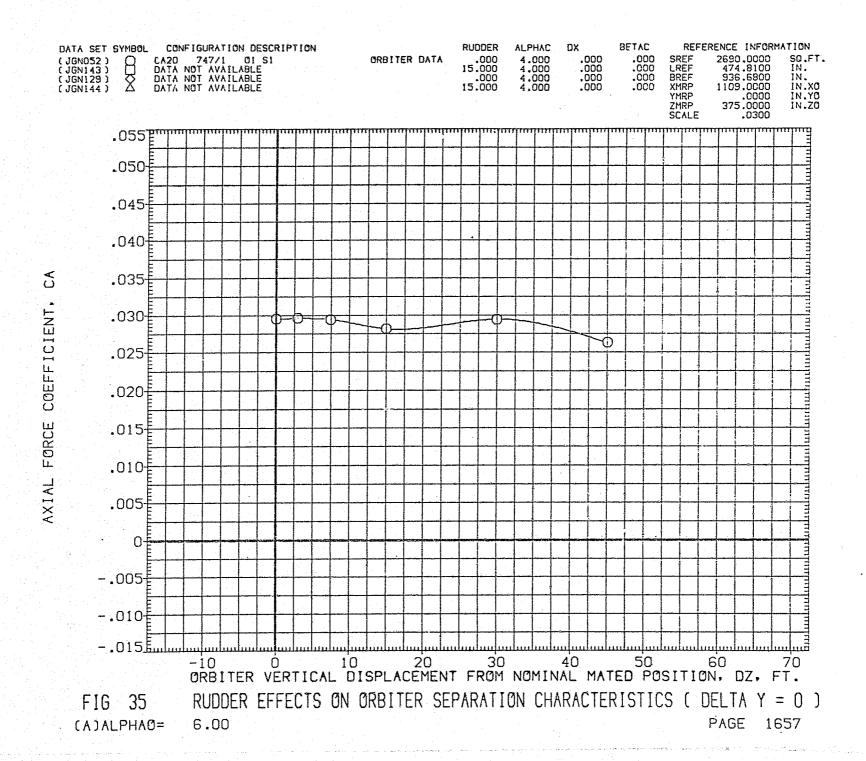


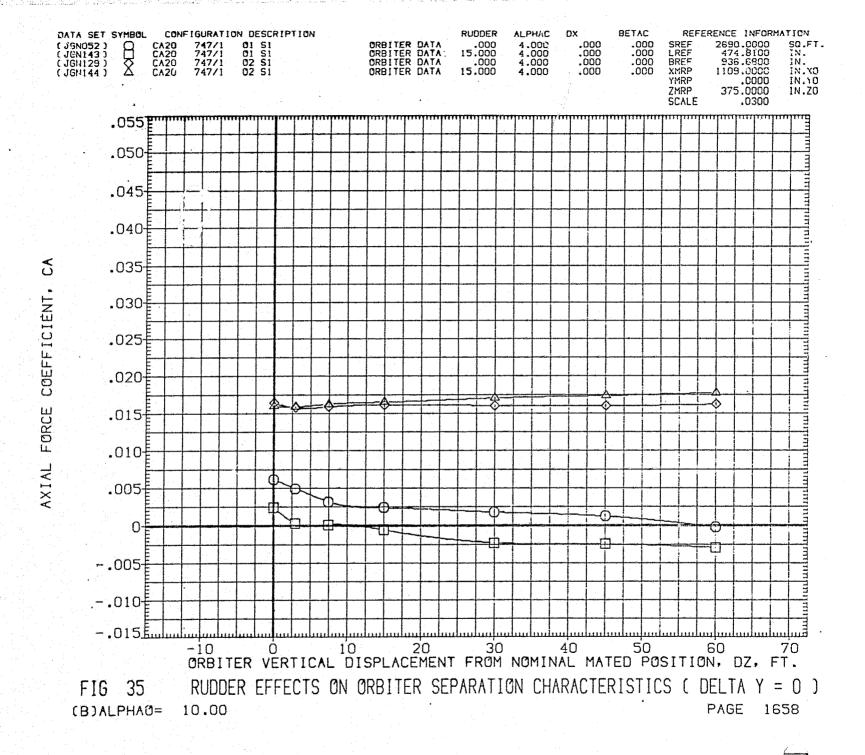




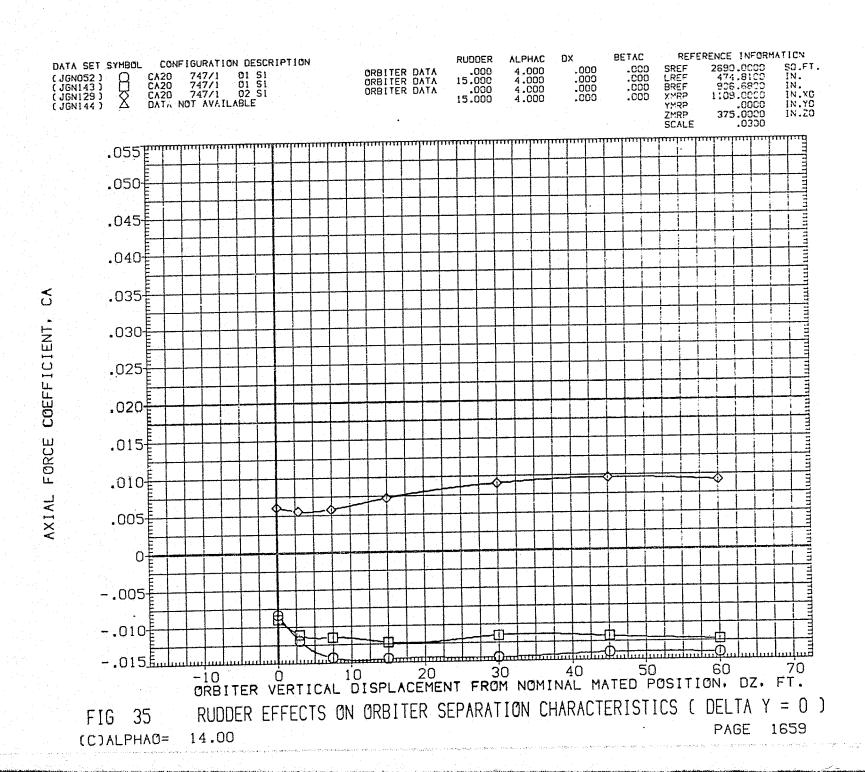


(C)ALPHAO= 14.00 PAGE 1656

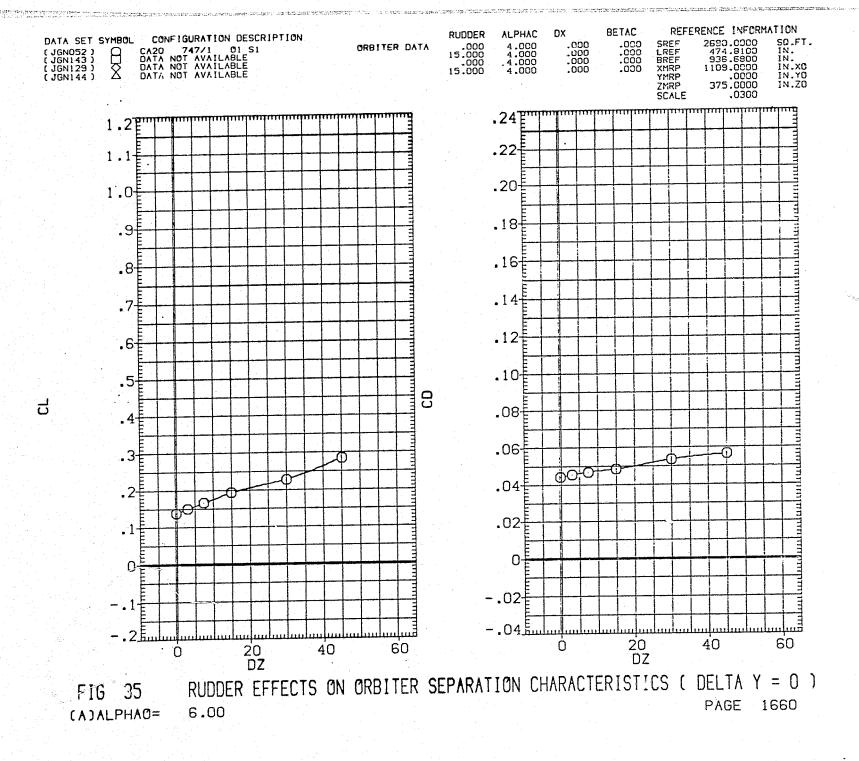








and the property of the second



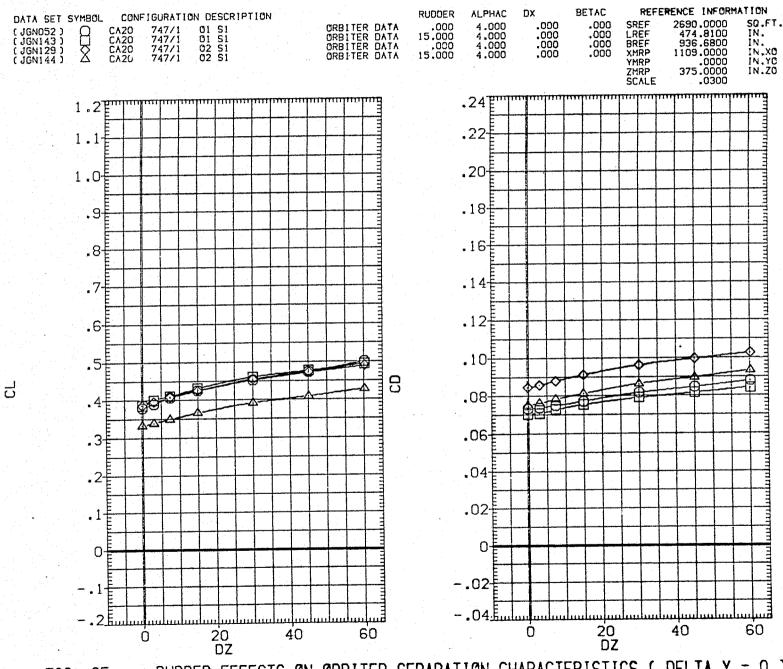


FIG 35 RUDDER EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B) ALPHAO = 10.00

PAGE 1661

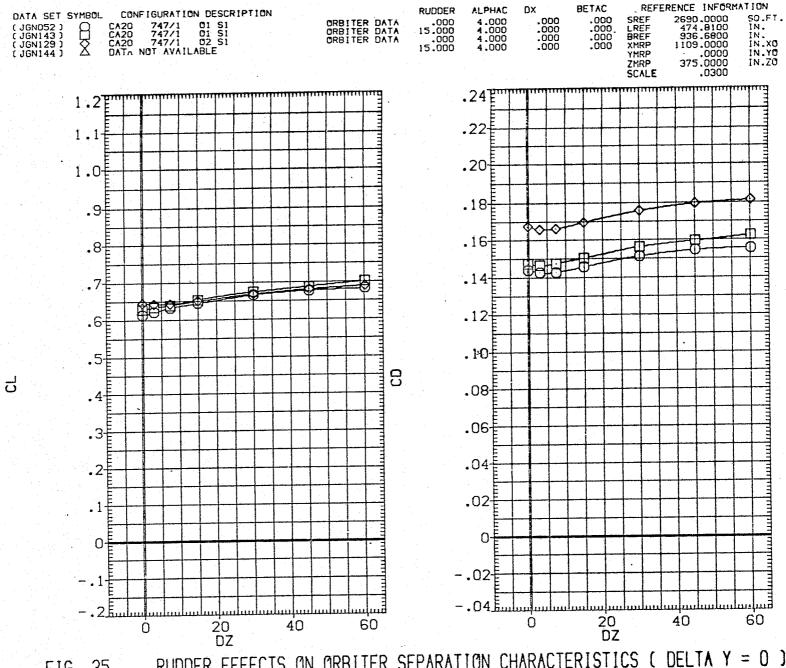
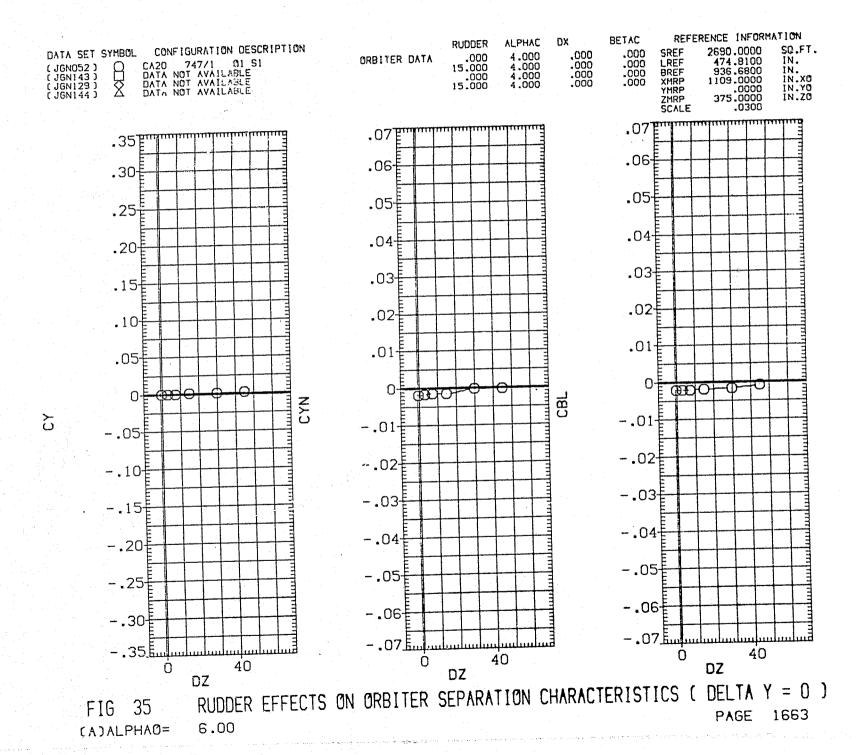
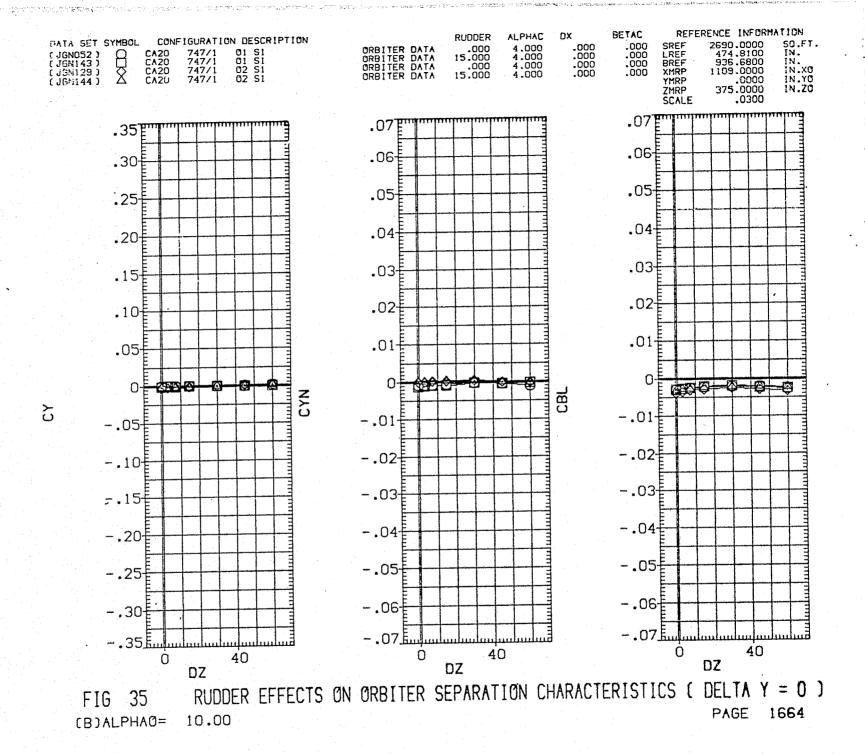


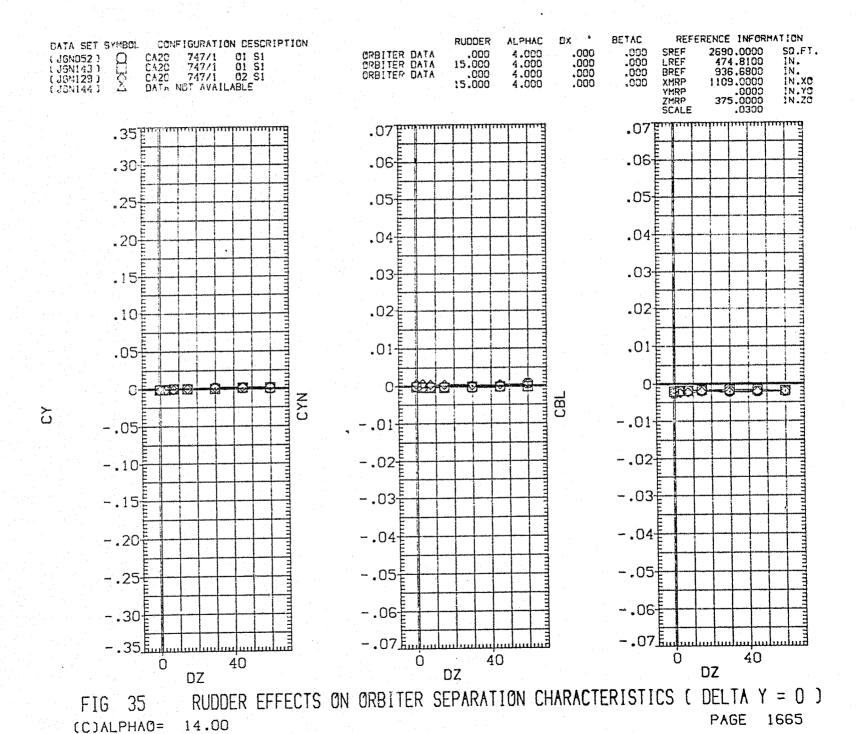
FIG 35 RUDDER EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

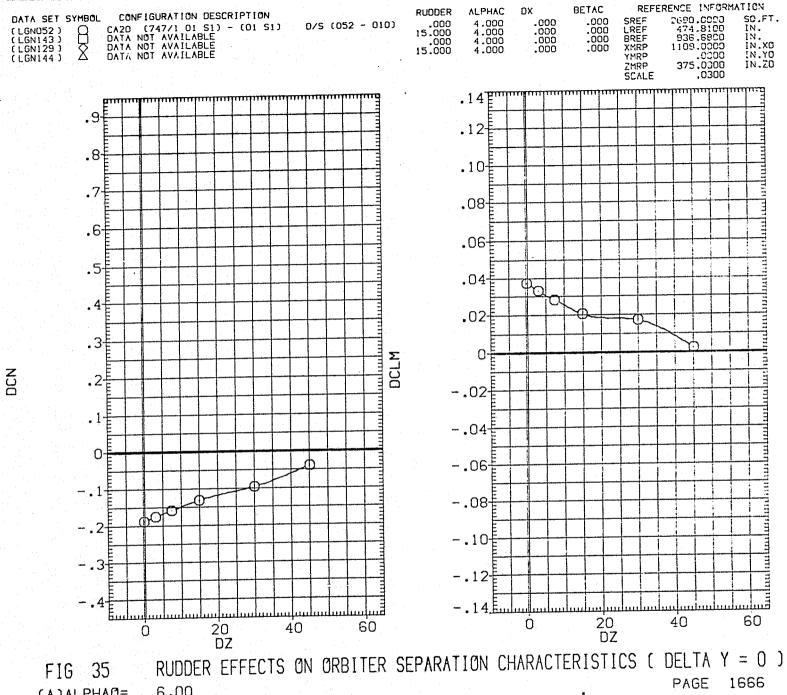
COALPHAGE 14.00

PAGE 1662









(A)ALPHAO= 6.00

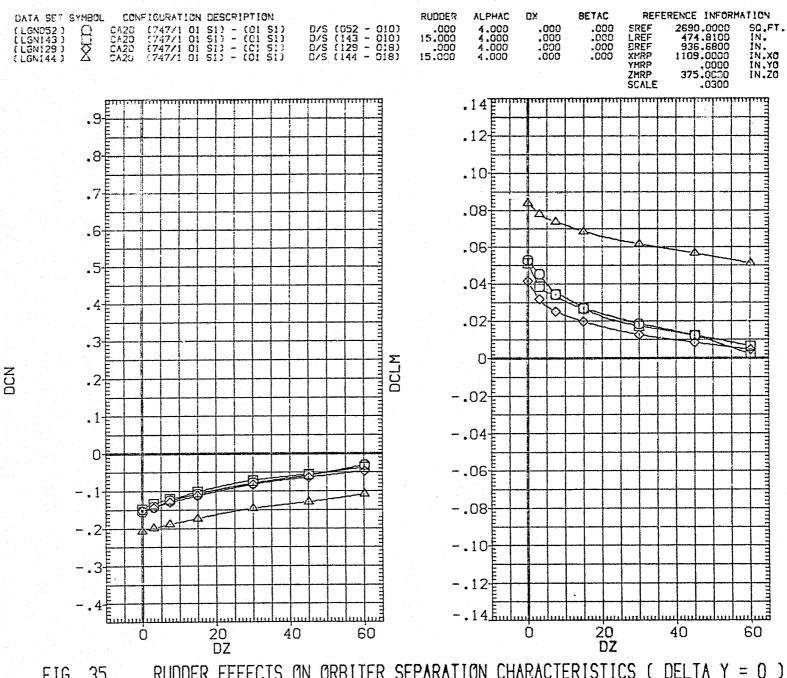


FIG 35 RUDDER EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B)ALPHAO= 10.00 PAGE 1667

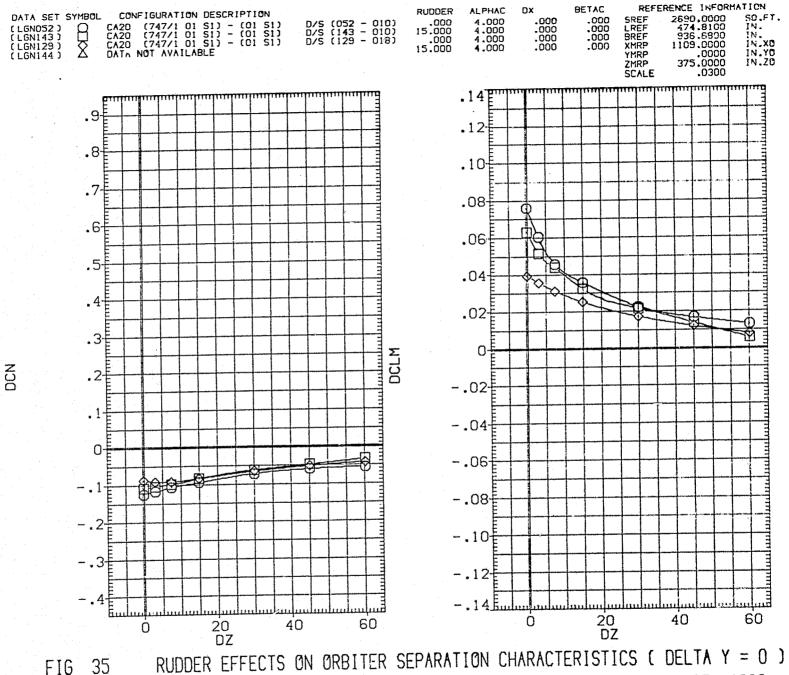
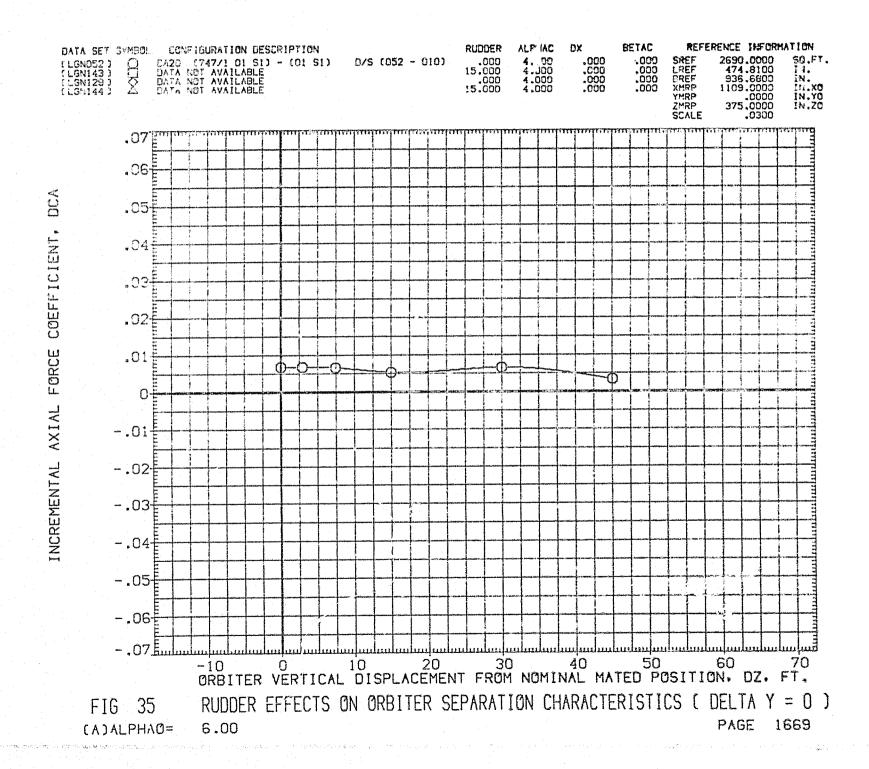
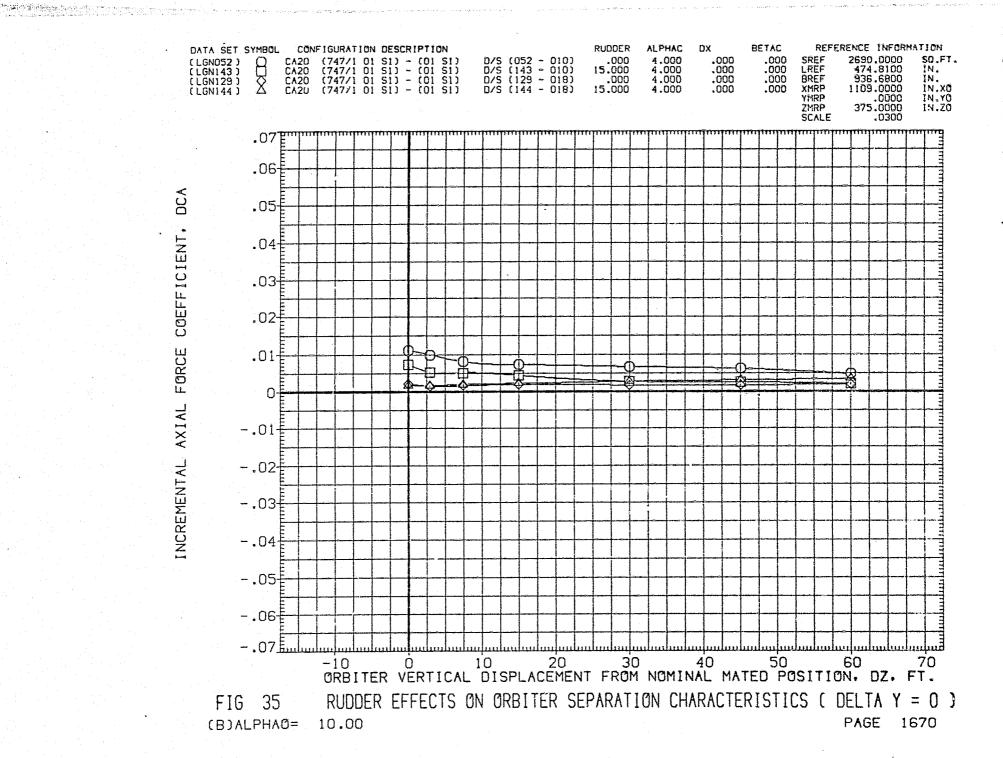
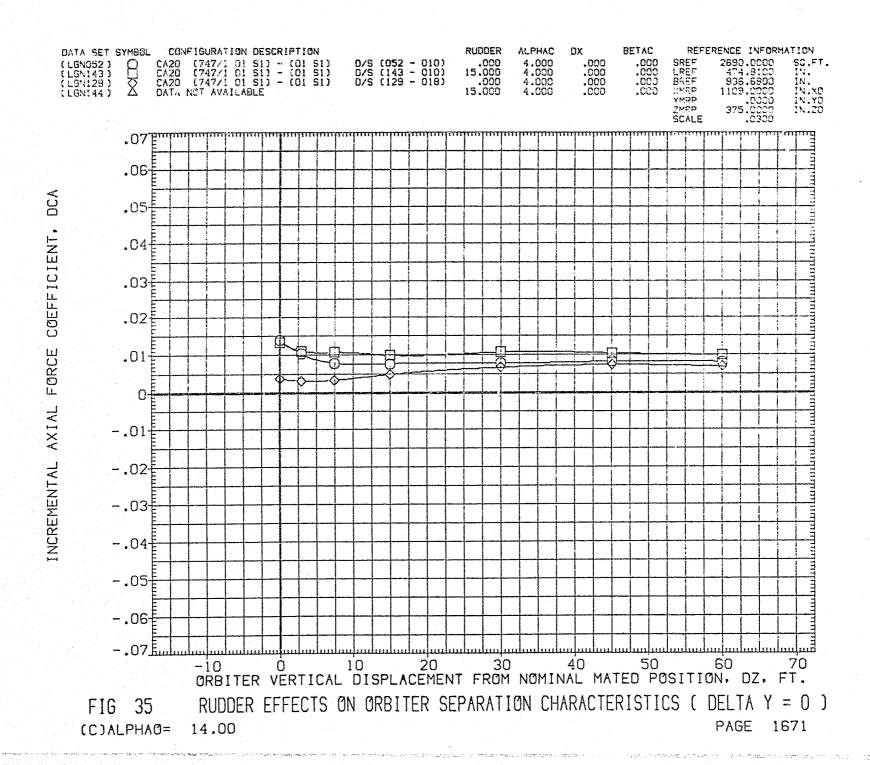


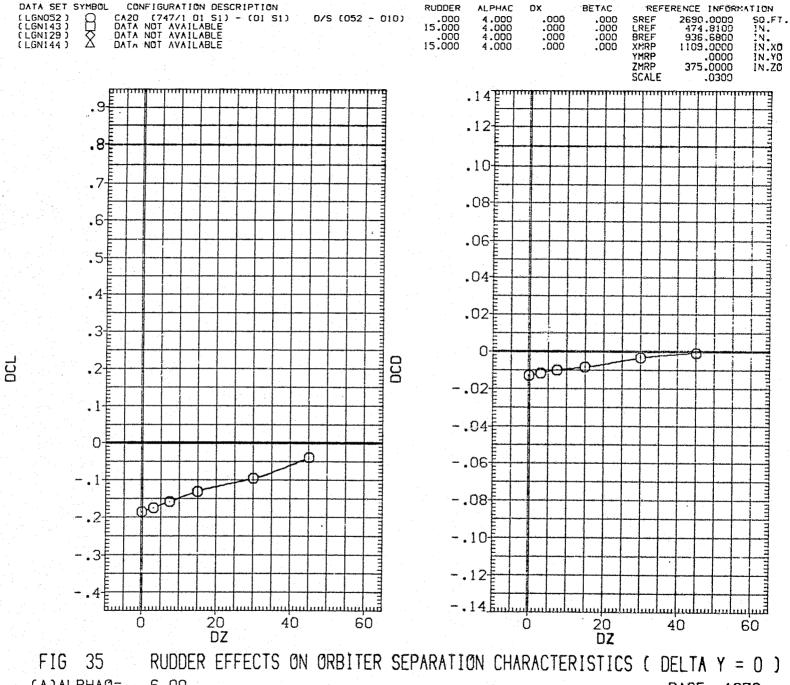
FIG 35 RUDDER EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELIA Y = U )

PAGE 1668









(A)ALPHAO= 6.00 PAGE 1672

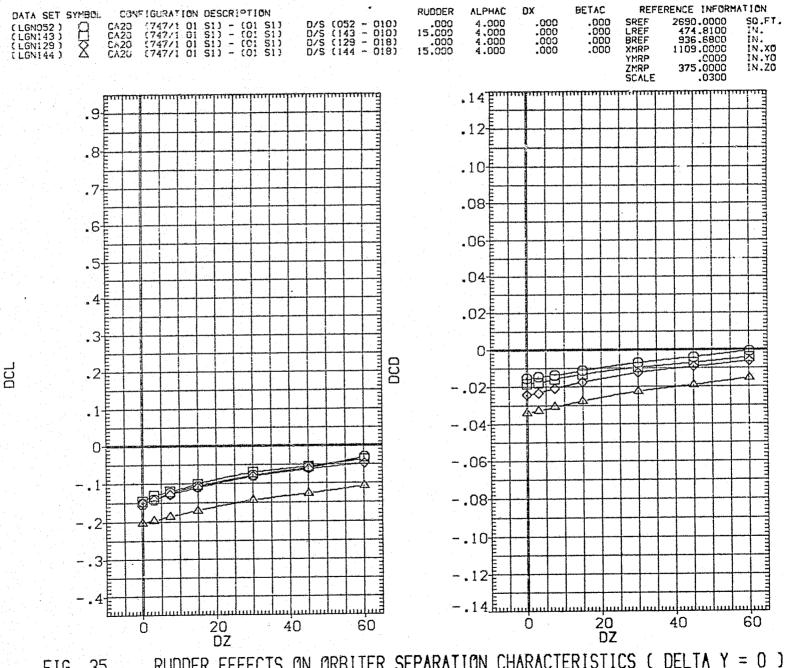


FIG 35 RUDDER EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B) ALPHAO = 10.00 PAGE 1673

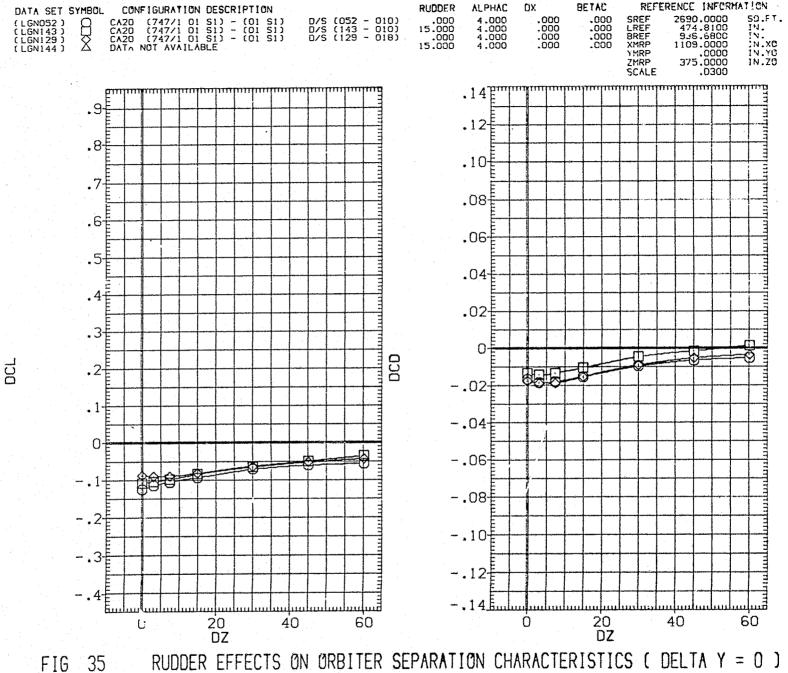


FIG 35 RUDDER EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(C)ALPHAO = 14.00 PAGE 1674



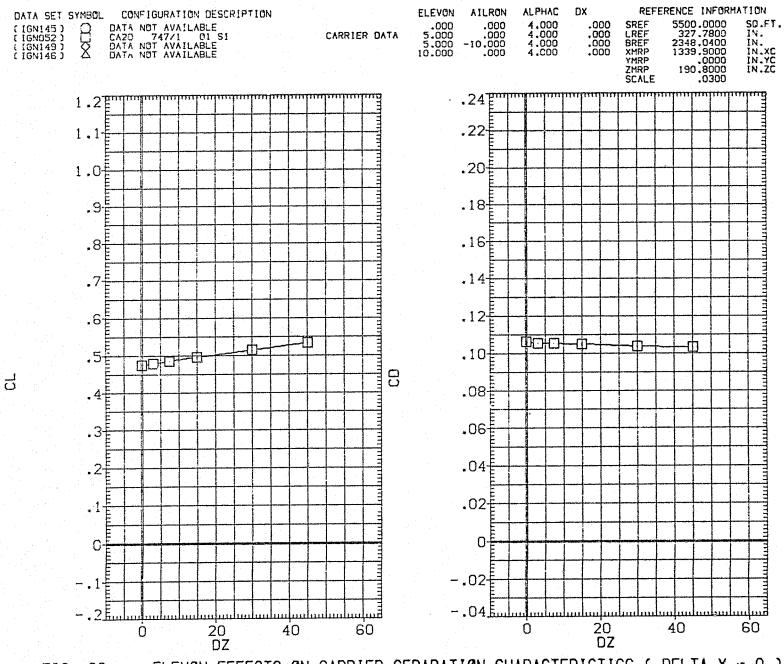
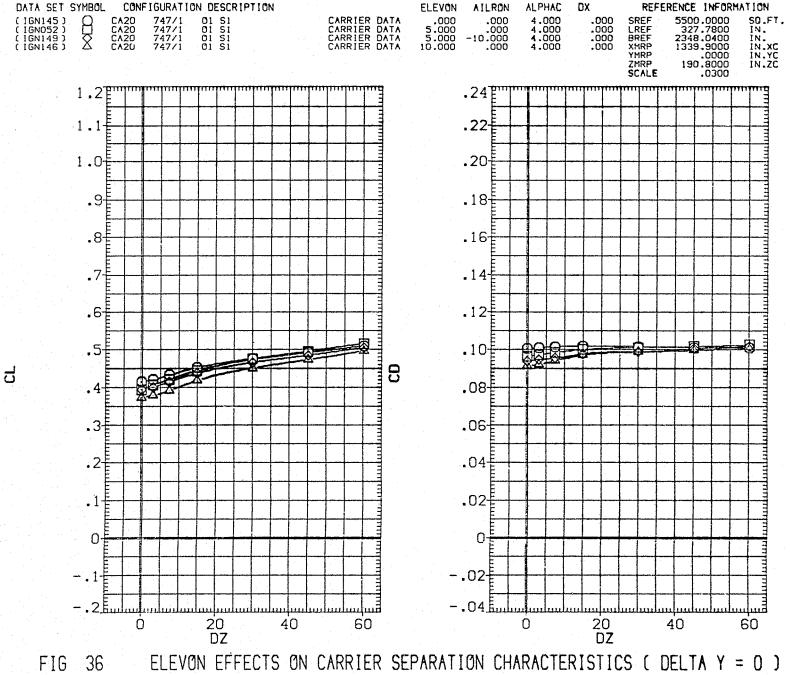


FIG 36 ELEVON EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(A)ALPHAO= 6.00 PAGE 1675

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(B)ALPHAO= 10.00 PAGE 1676

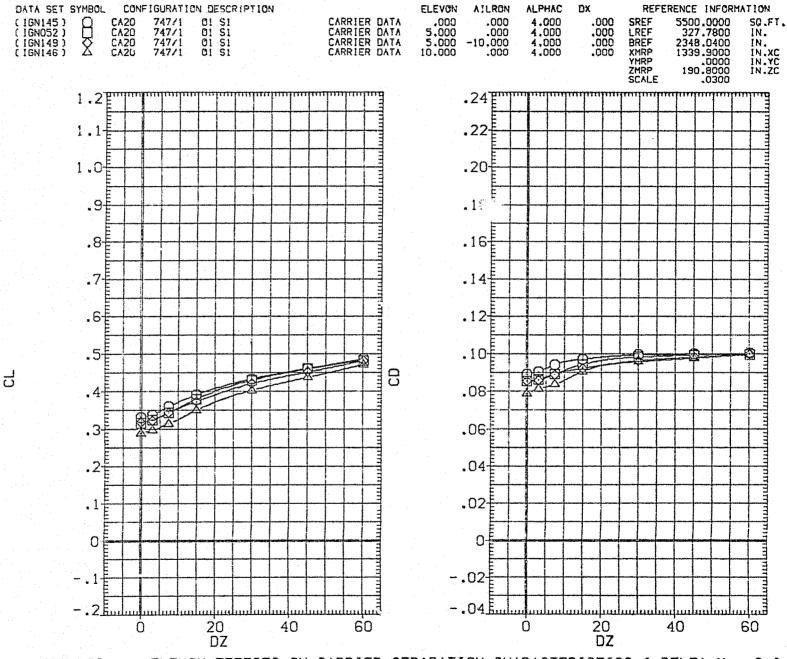
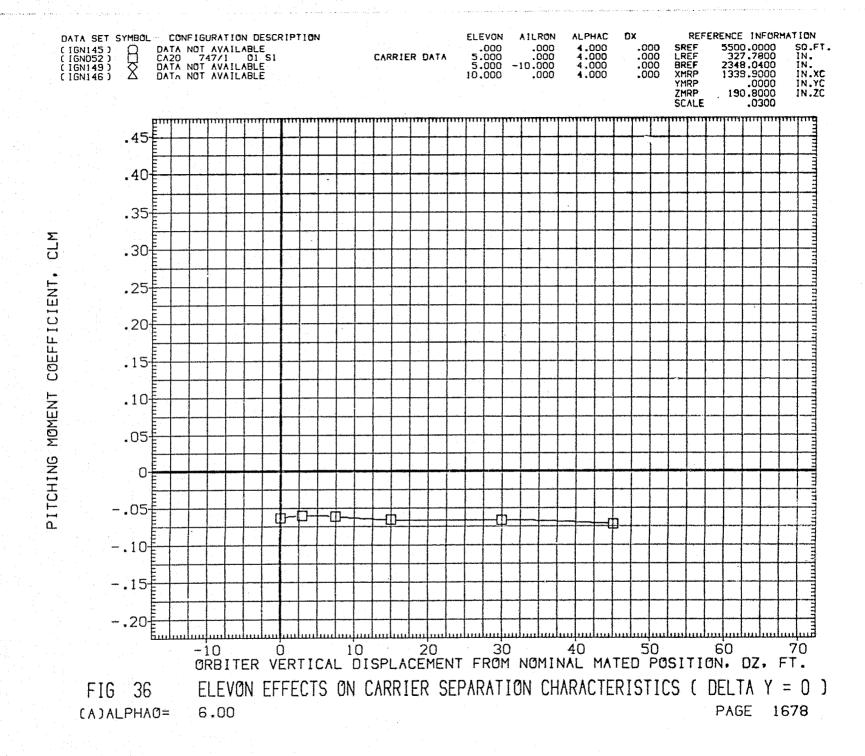
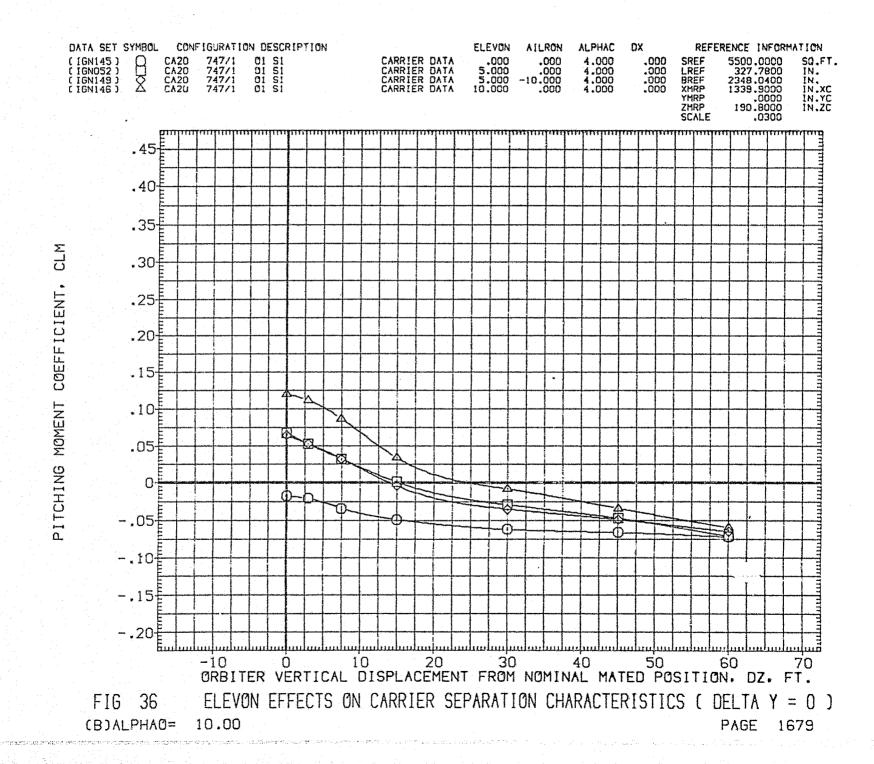
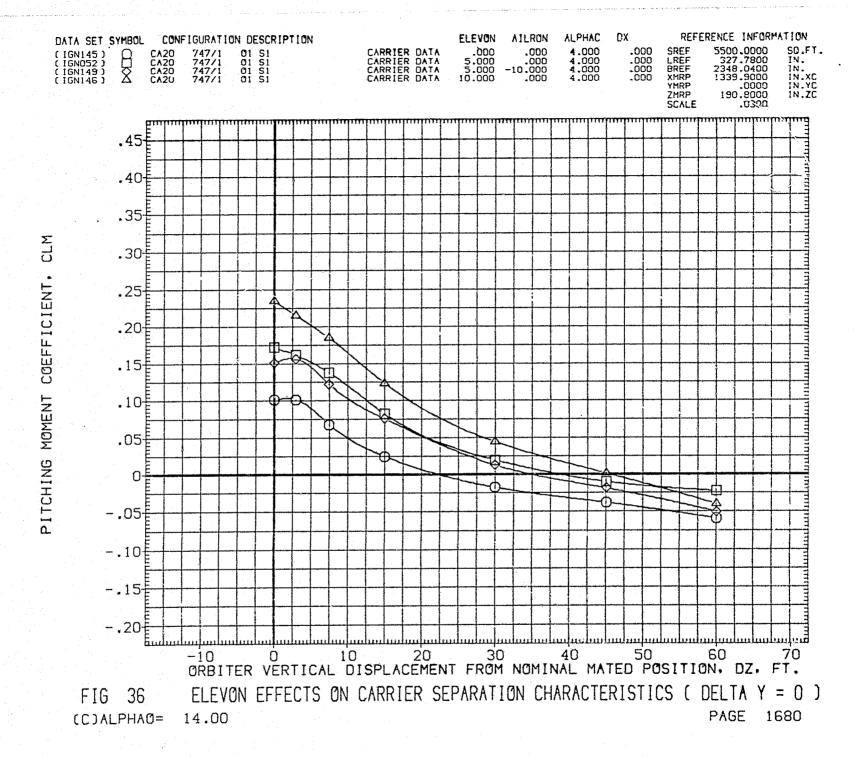


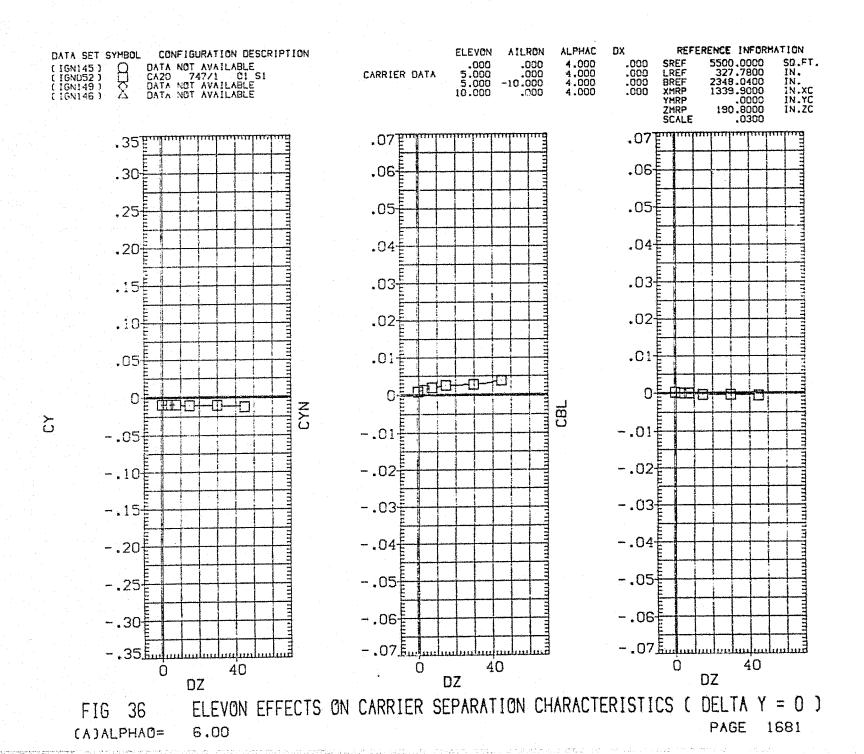
FIG 36 ELEVON EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

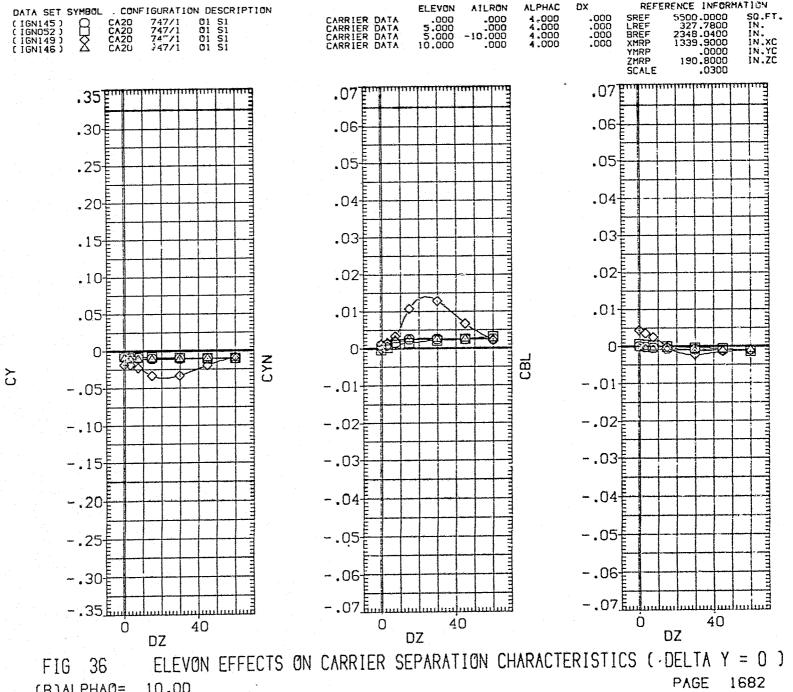
(C)ALPHAO= 14.00 PAGE 1677



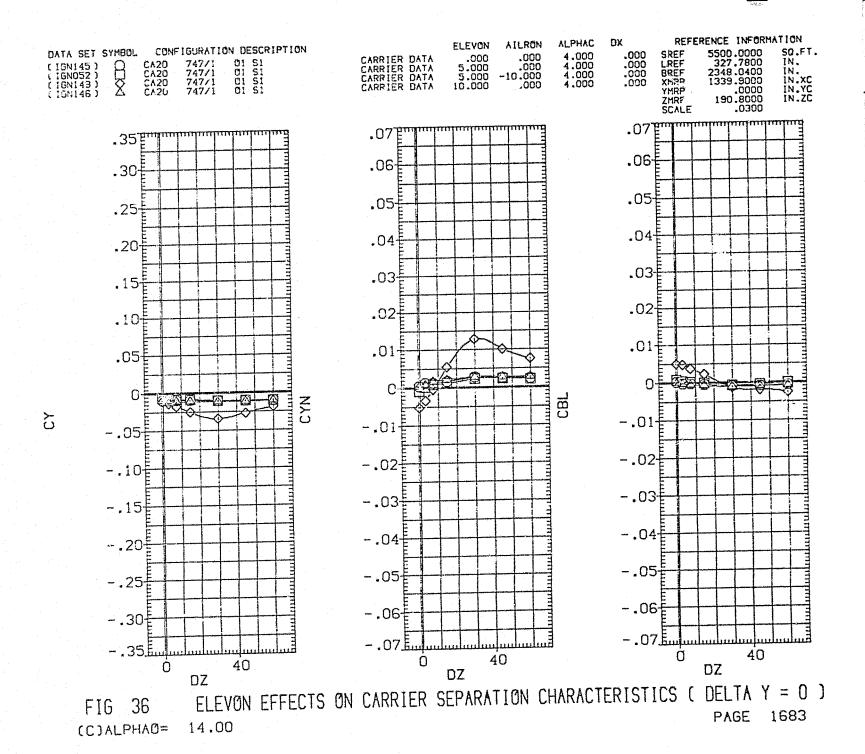


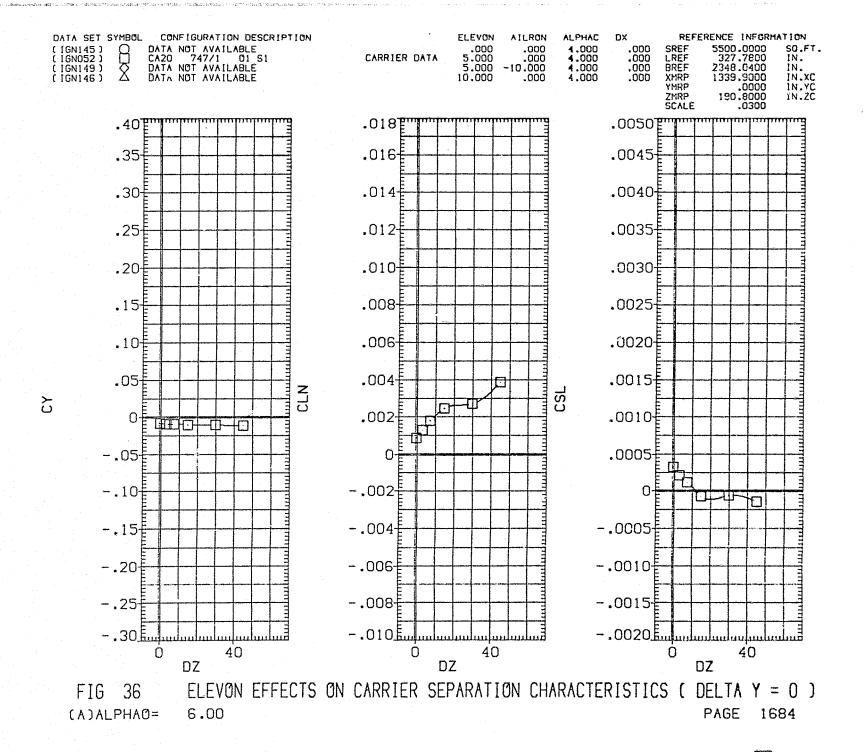


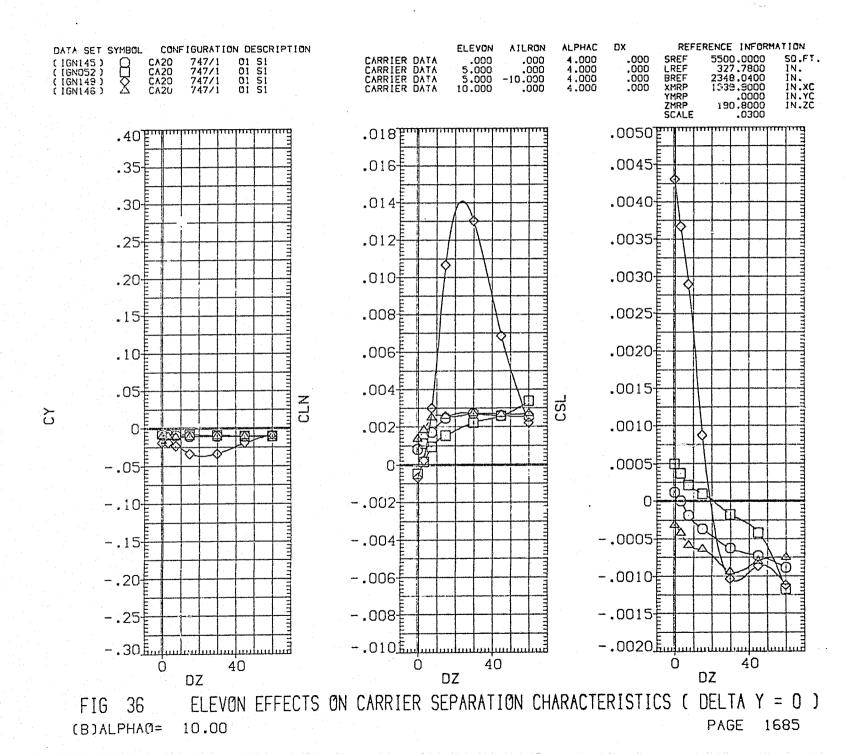


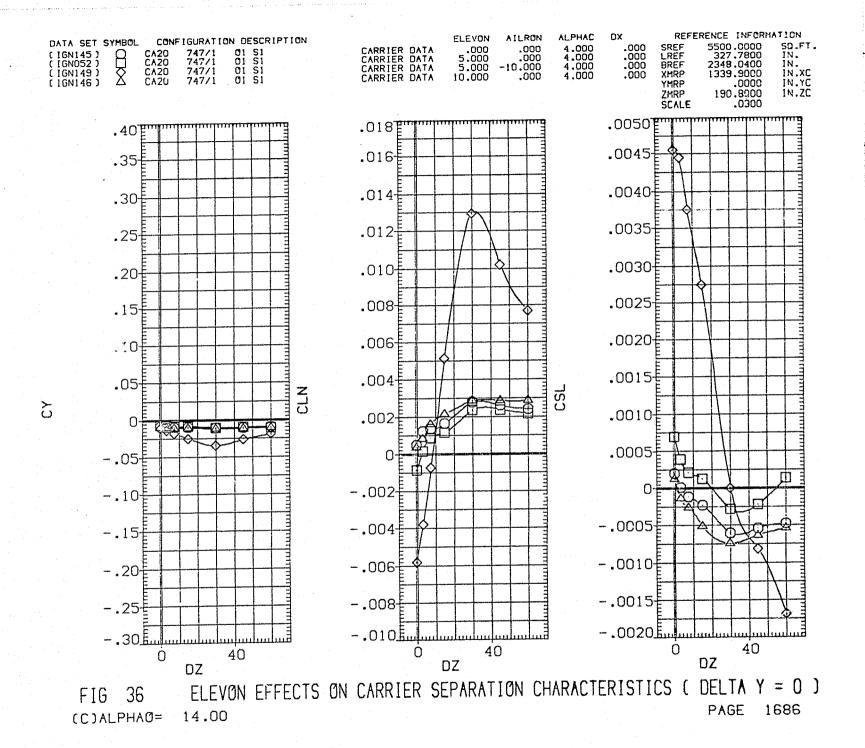


(B)ALPHAO= 10.00

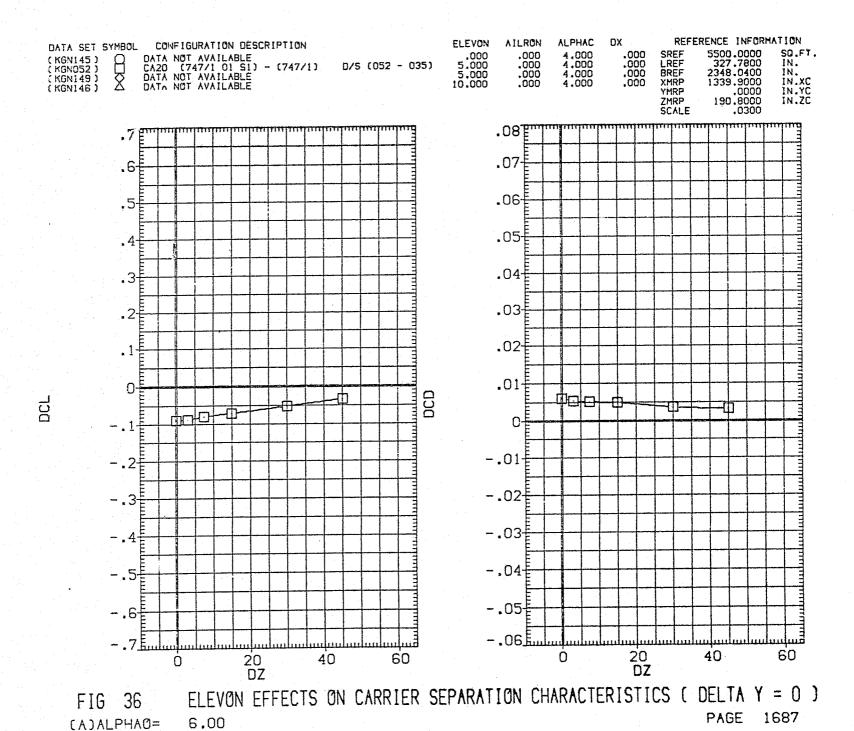












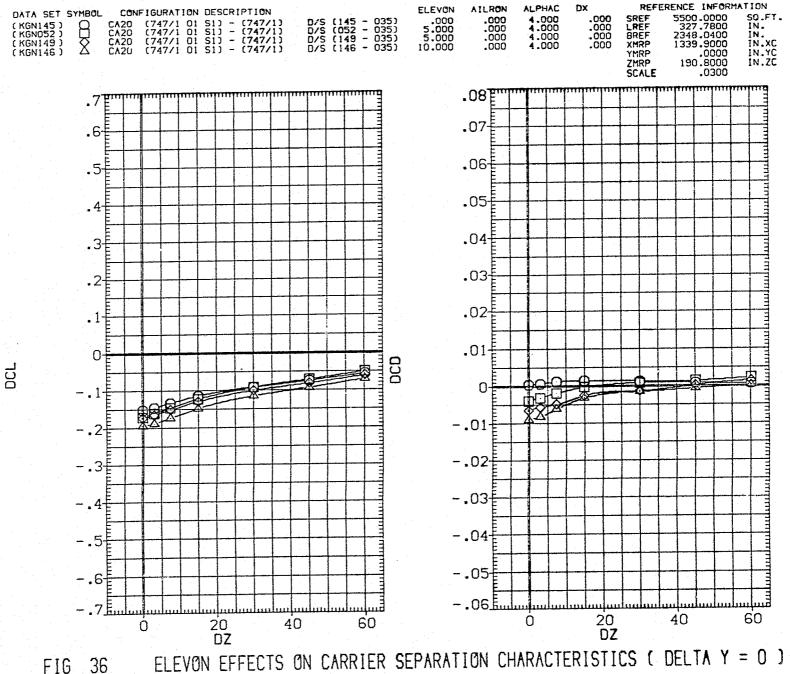


FIG 36 ELEVON EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B)ALPHAO = 10.00

PAGE 1688

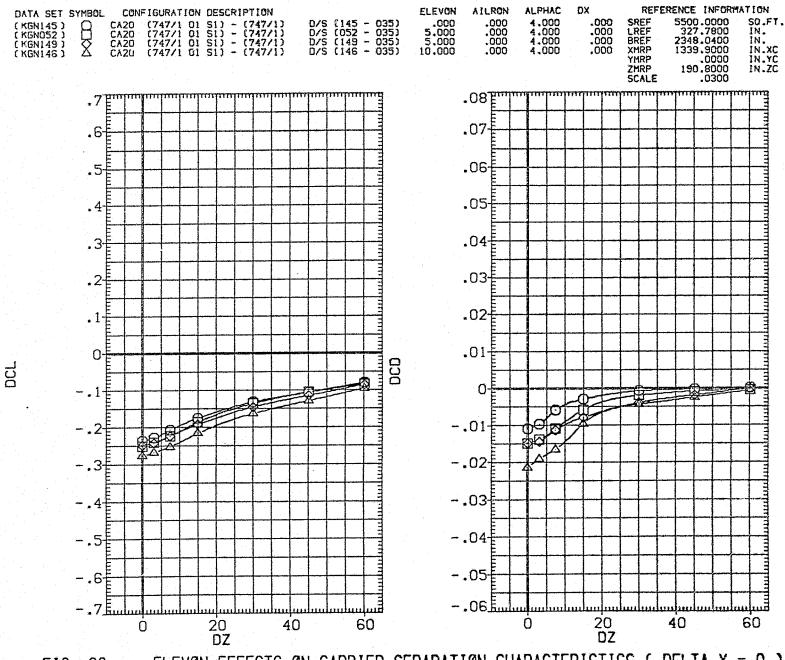
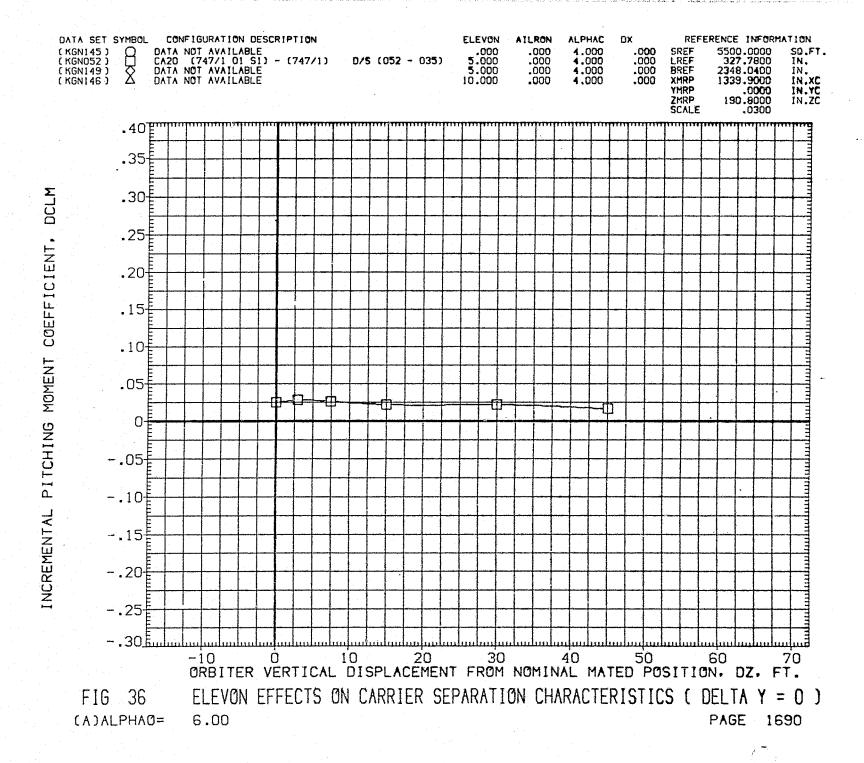
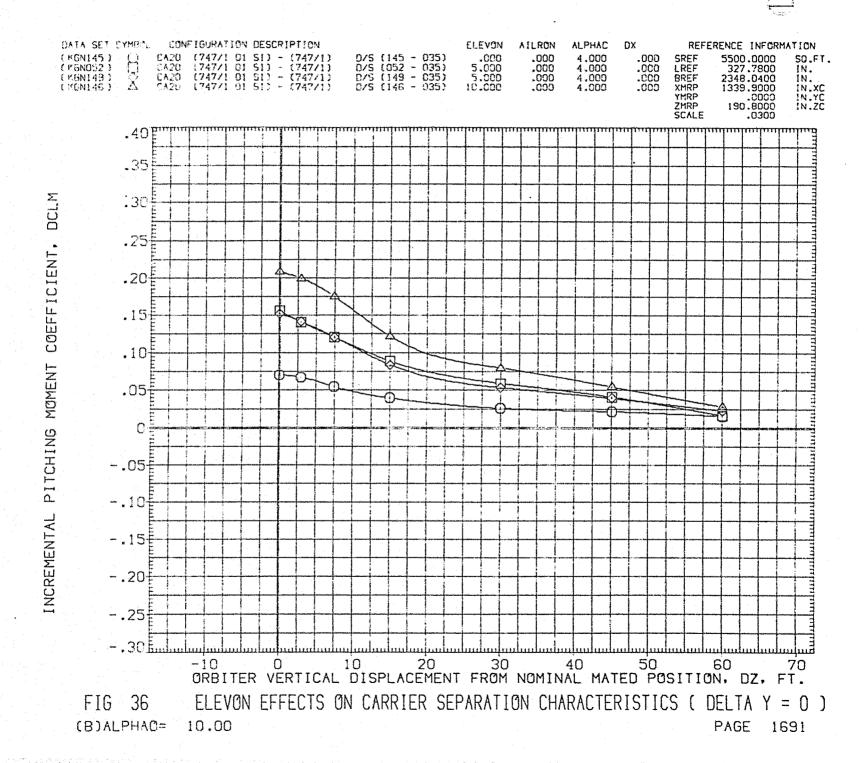


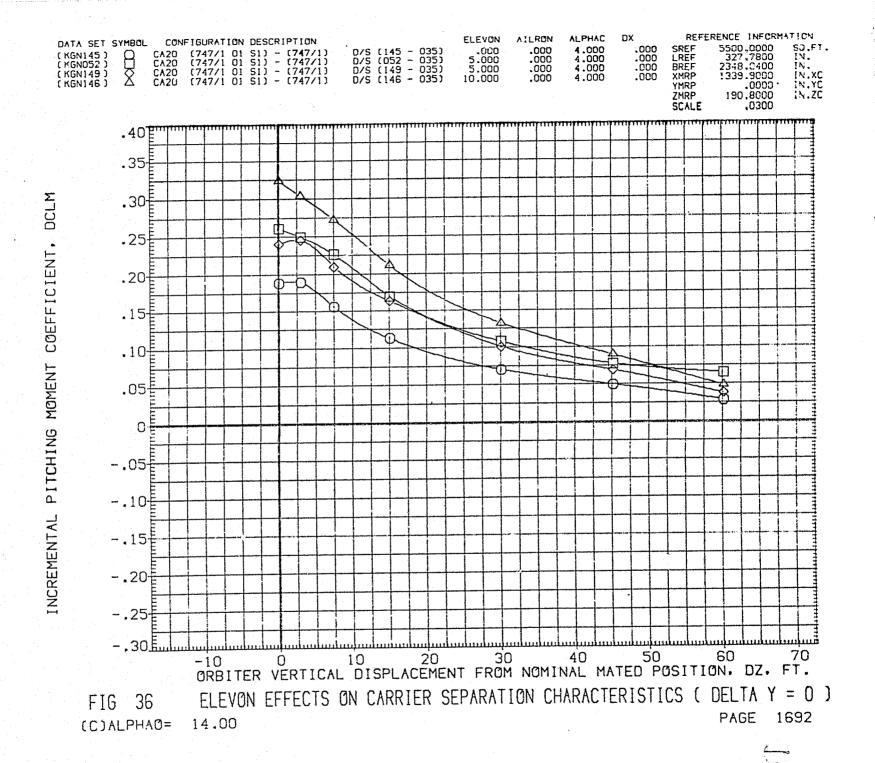
FIG 36 ELEVON EFFECTS ON CARRIER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

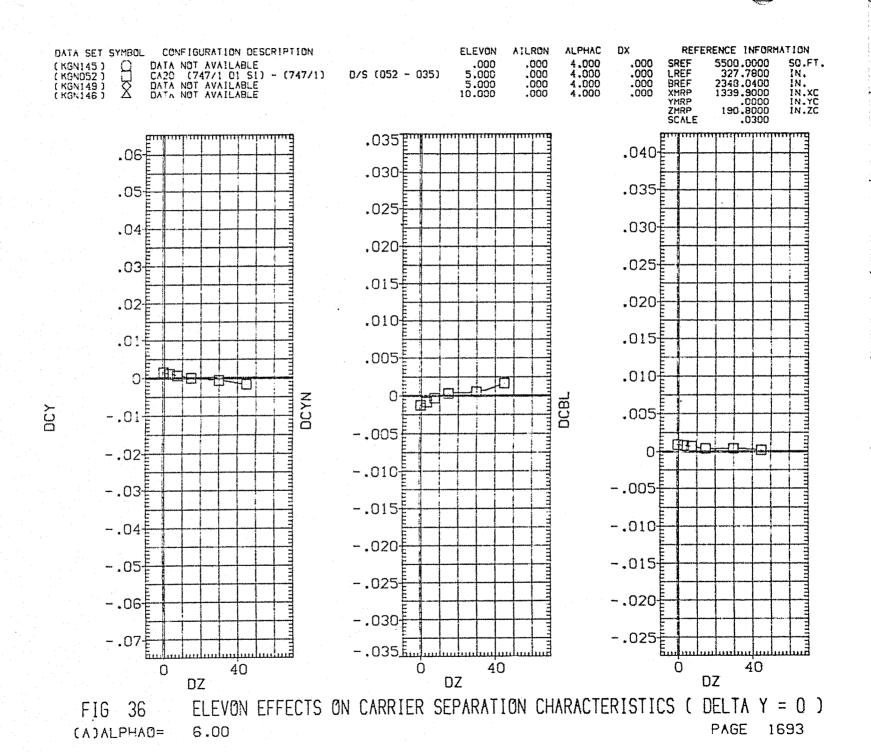
COALPHAG= 14.00 PAGE 1689

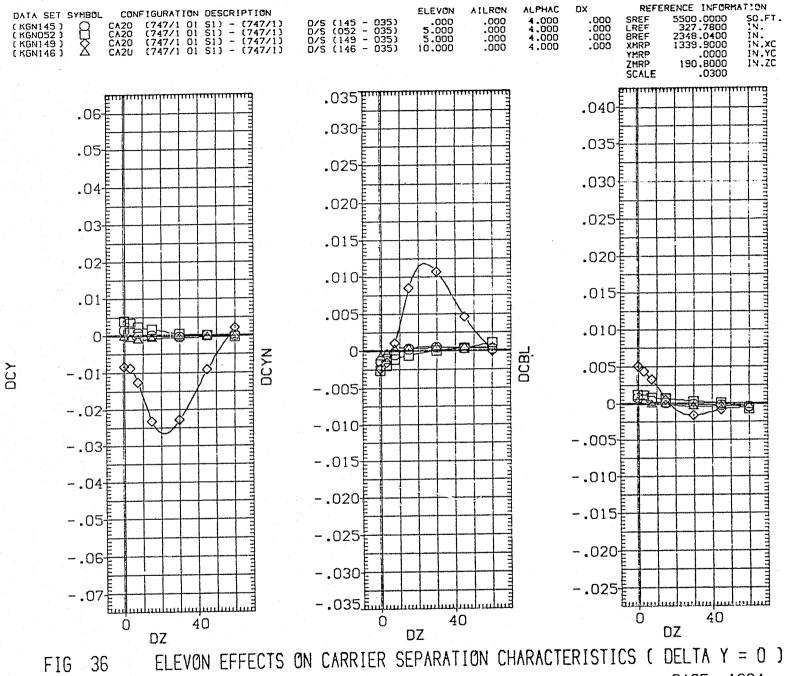




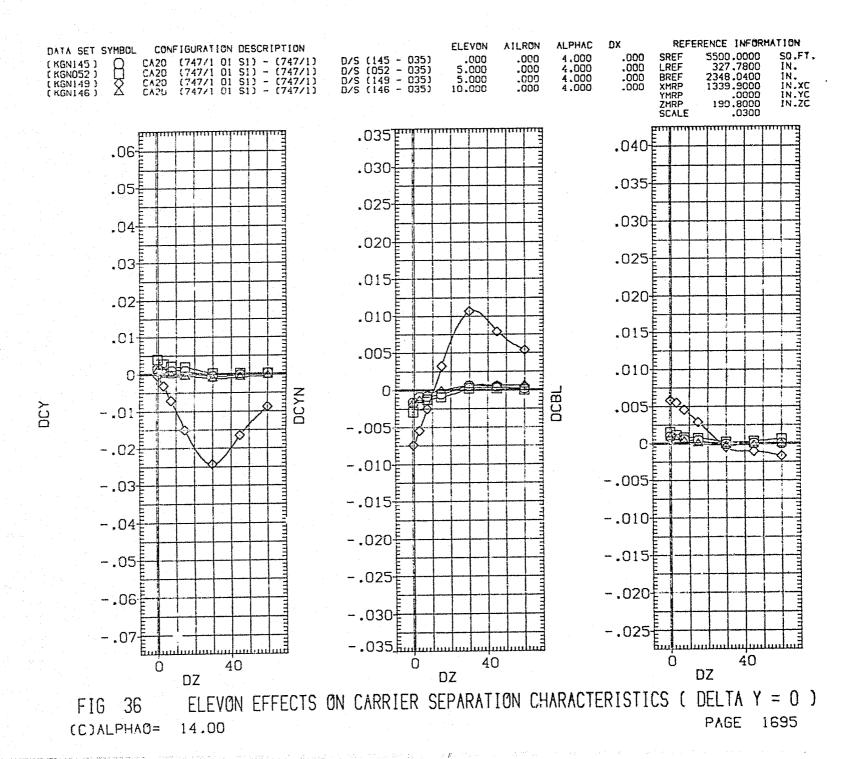
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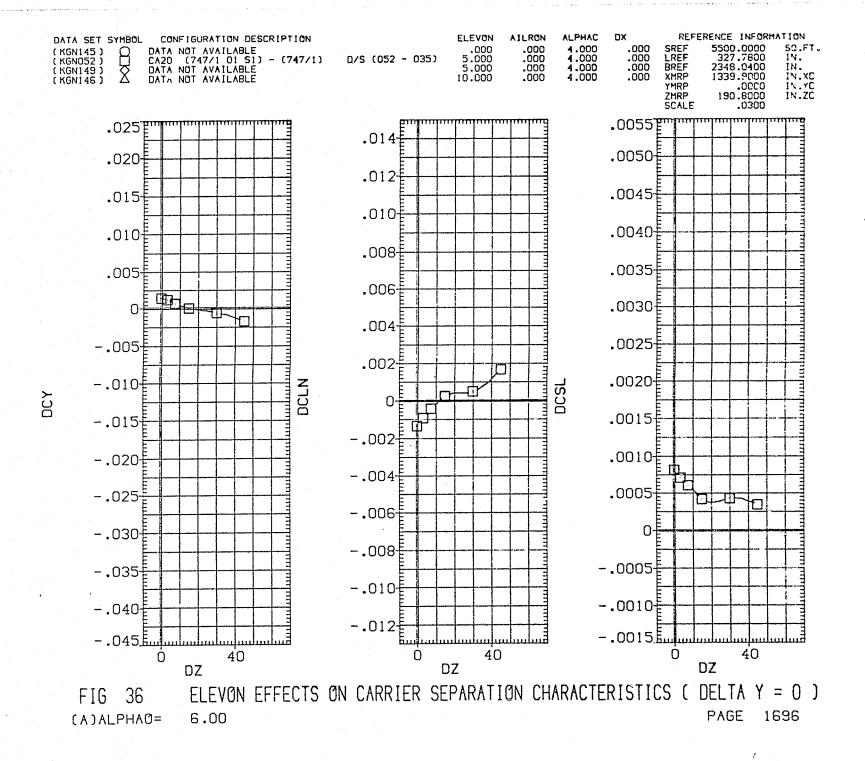


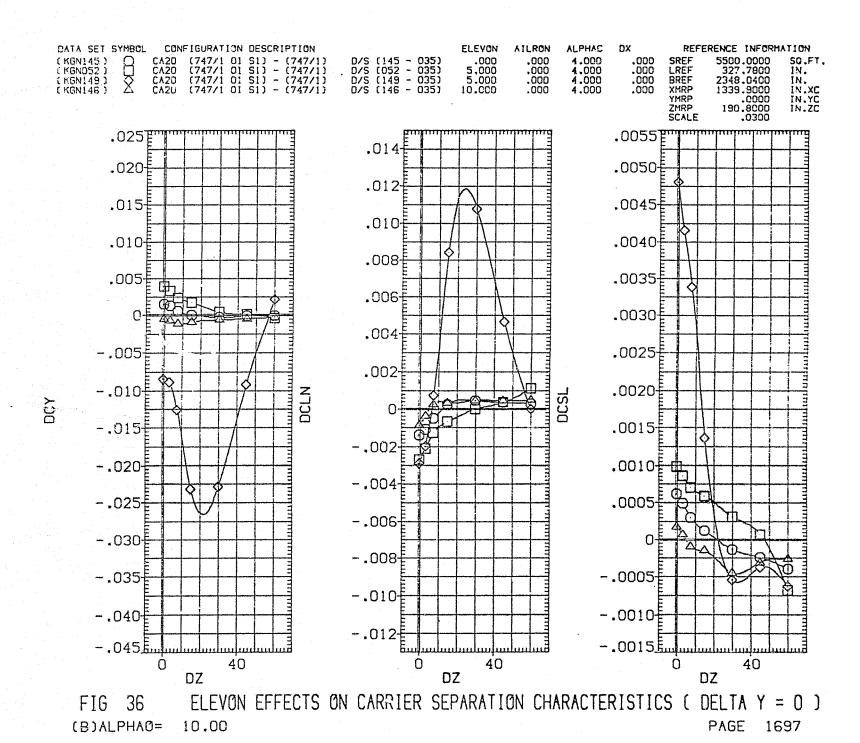


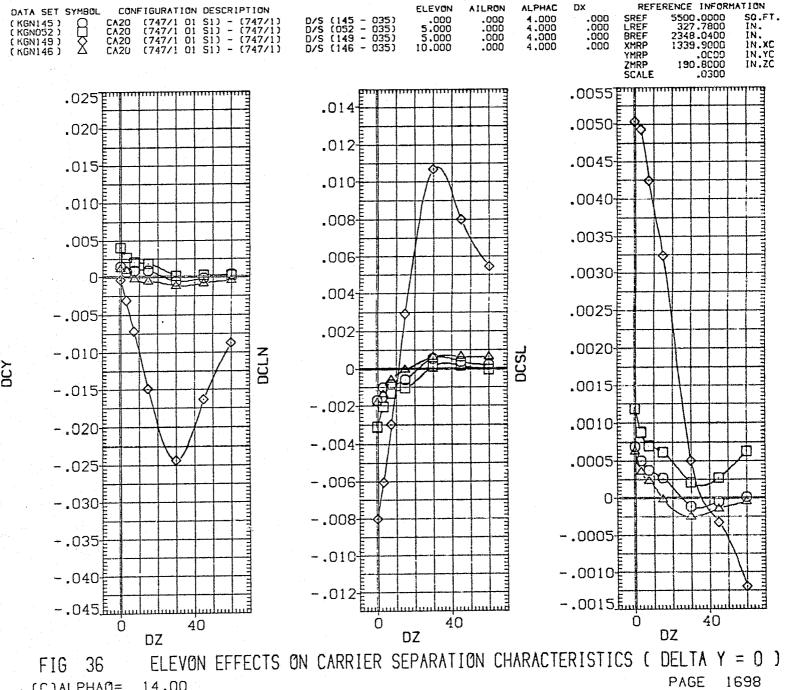


PAGE 1694 (B)ALPHAO= 10.00









- (C)ALPHAO= 14.00

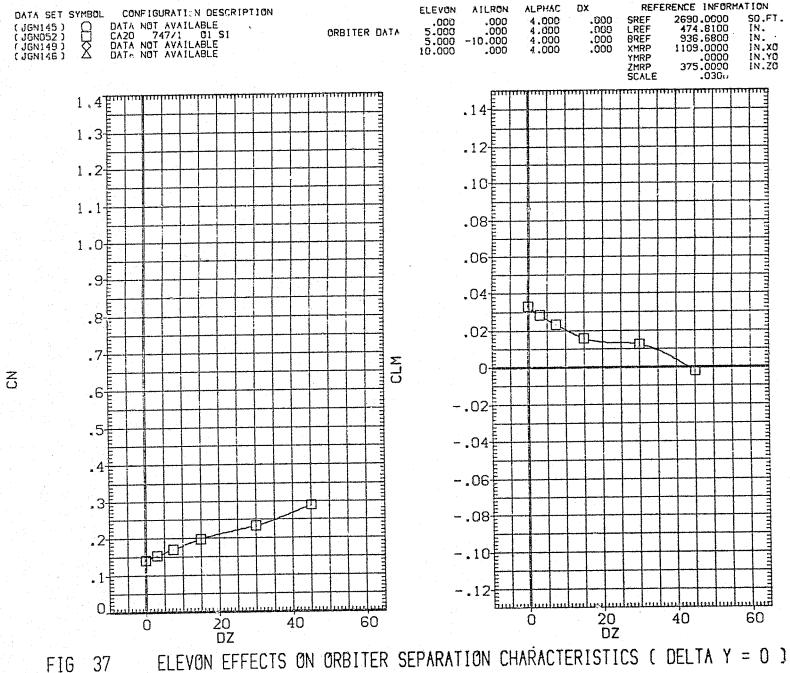


FIG PAGE 1699 (A)ALPHAO= 6.00

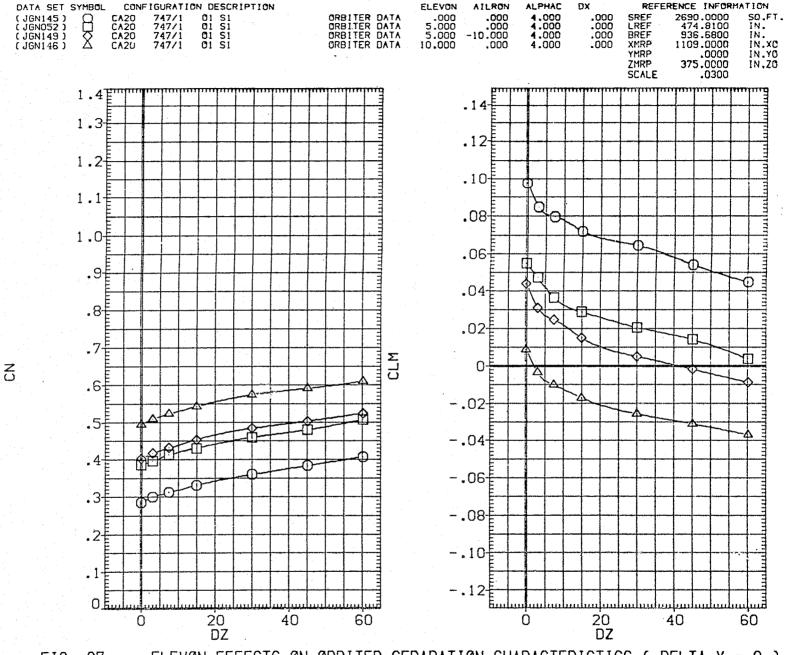


FIG 37 ELEVON EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B)ALPHAO= 10.00 PAGE 1700

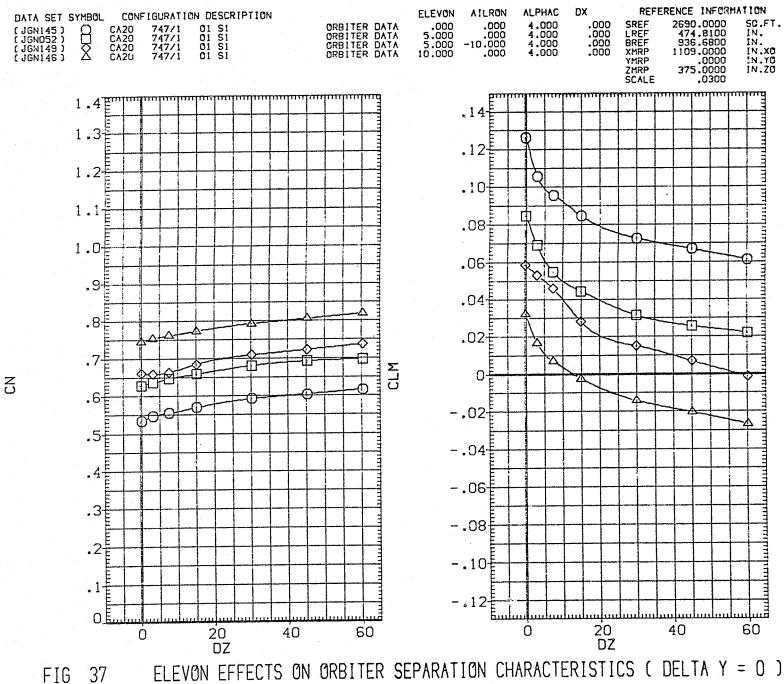
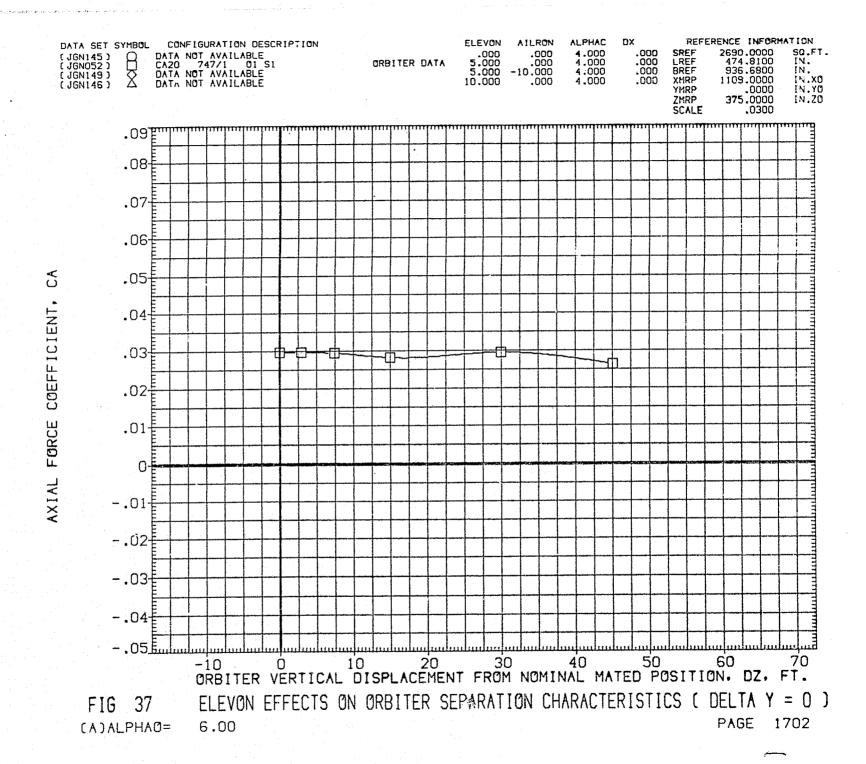
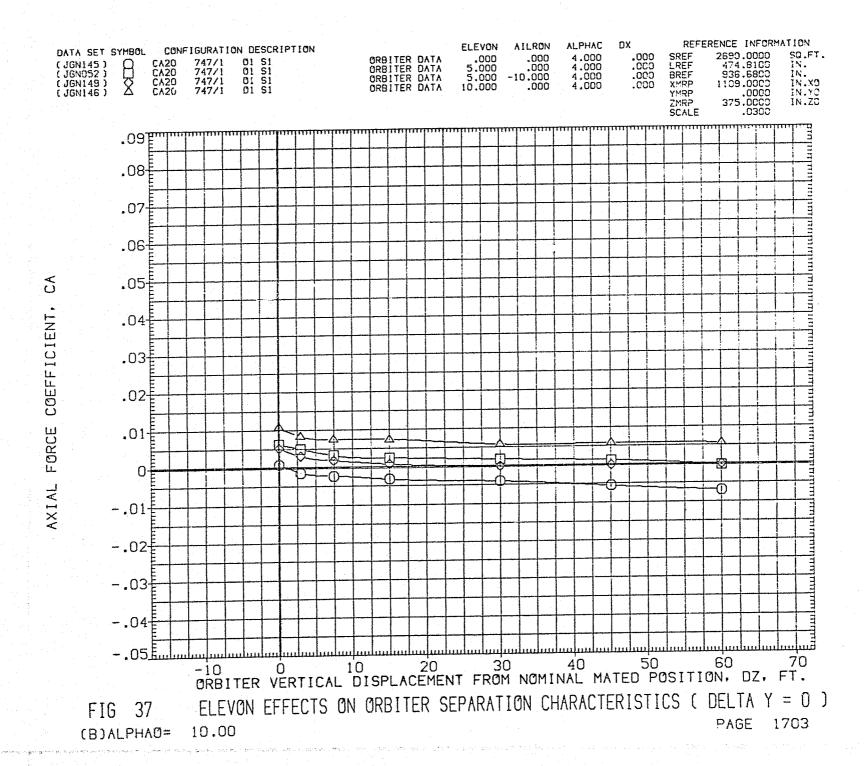
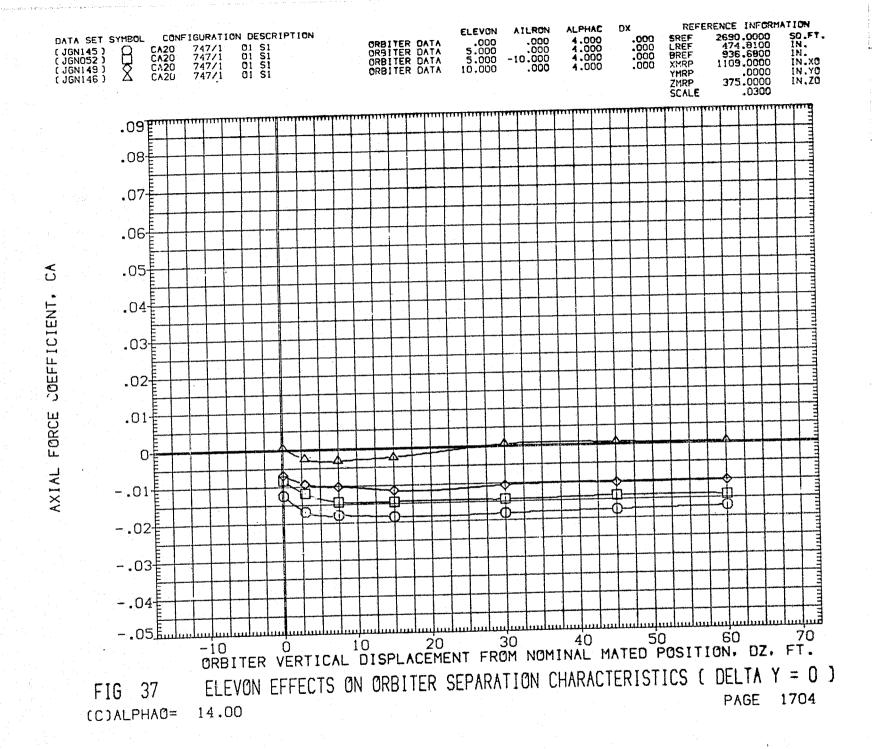
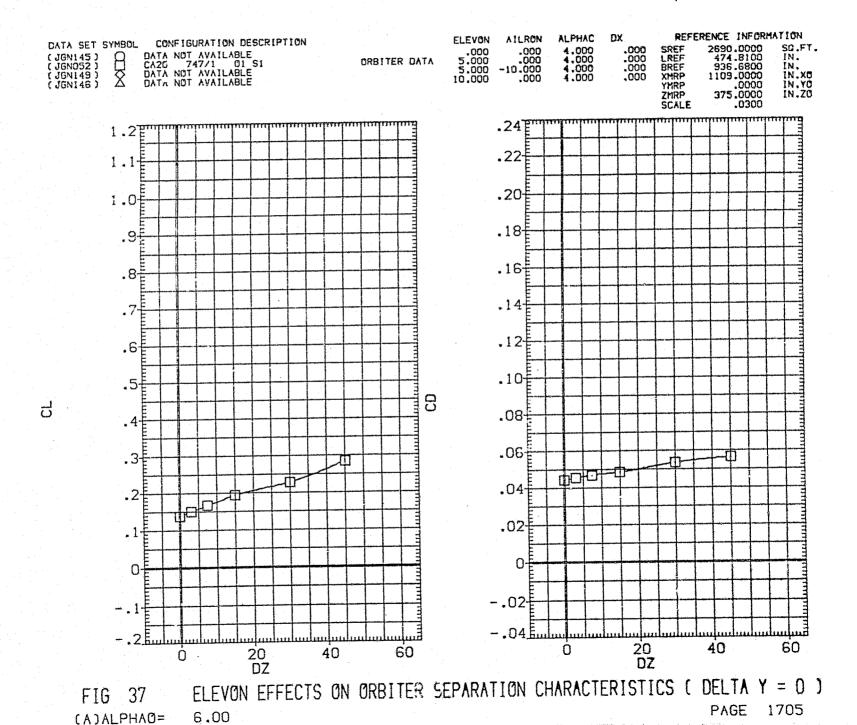


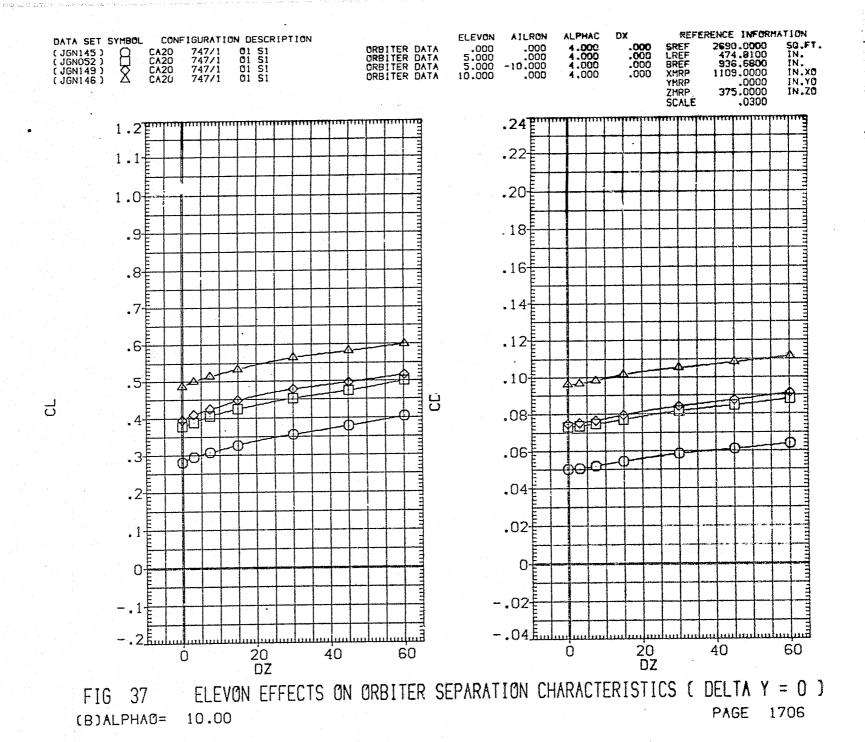
FIG PAGE 1701 (C)ALPHAO= 14.00











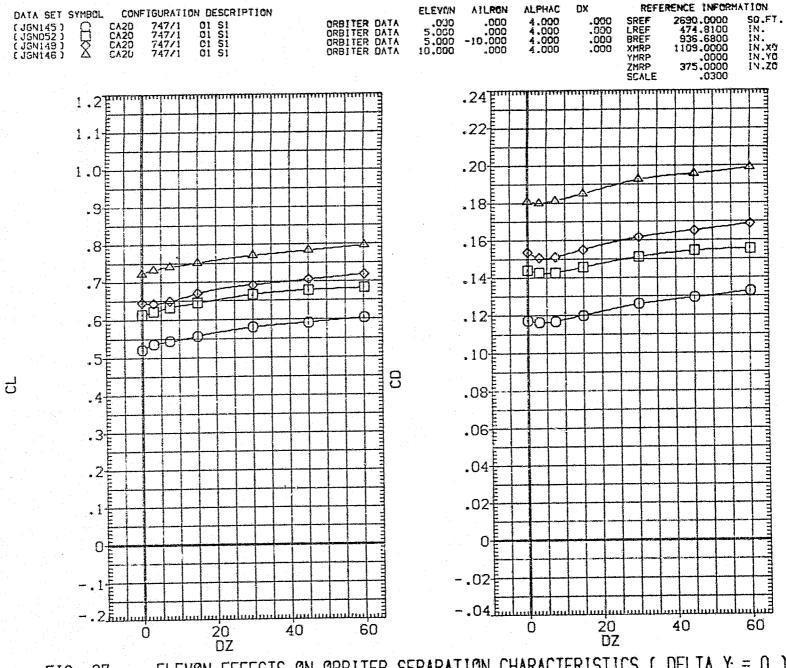


FIG 37 ELEVON EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

COALPHAGE 14.00

PAGE 1707

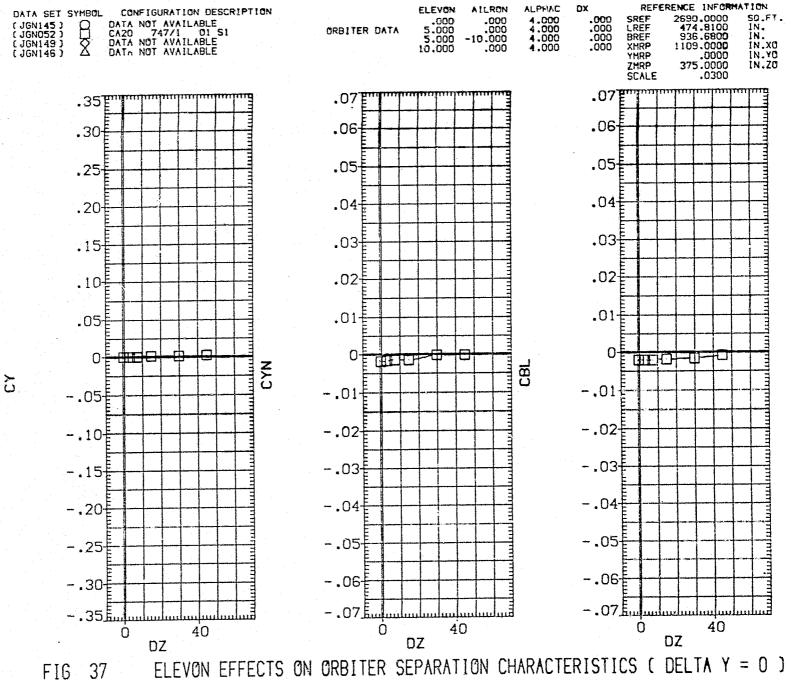


FIG 37 ELEVON EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 PAGE 1708

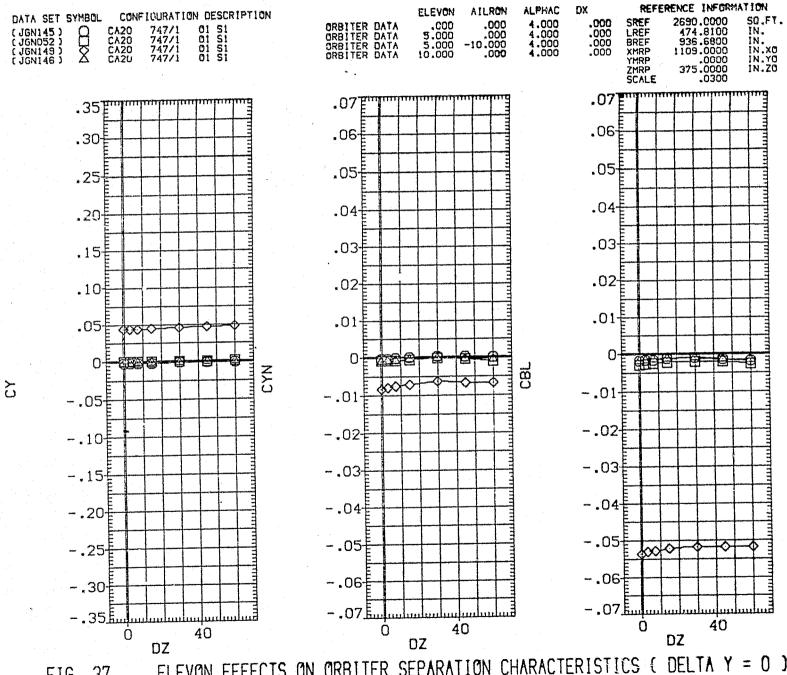
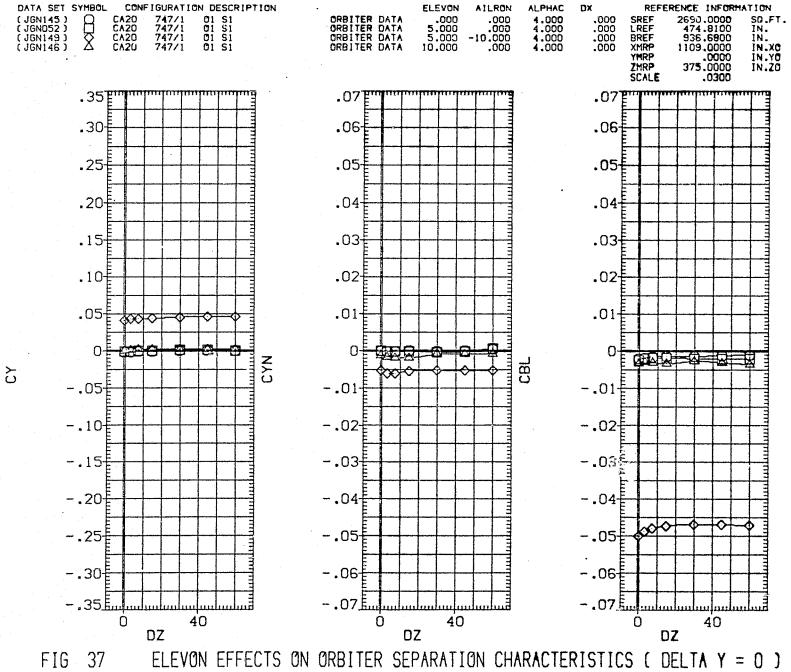


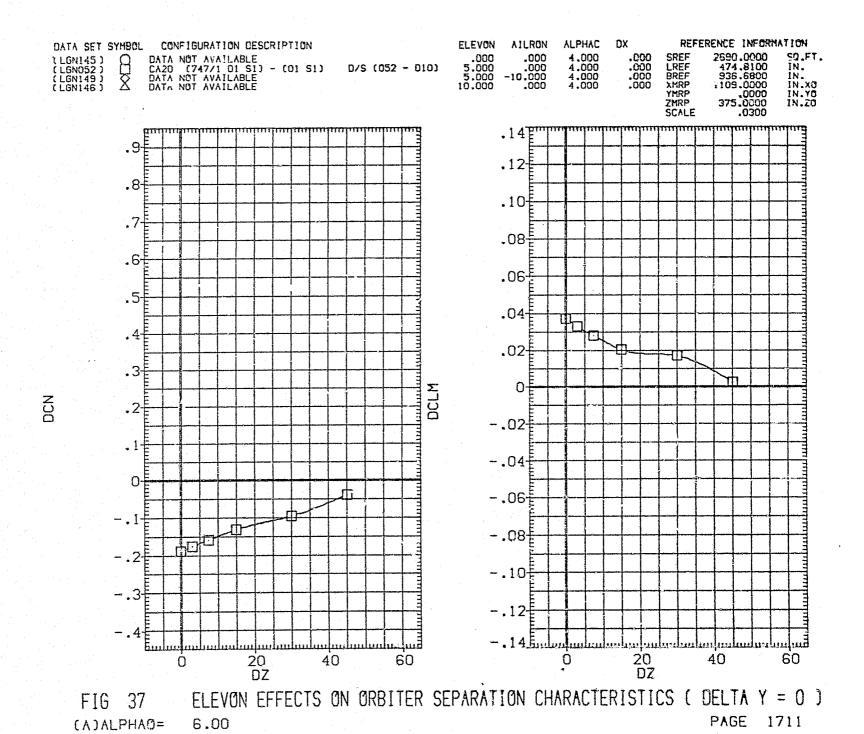
FIG 37 ELEVON EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0

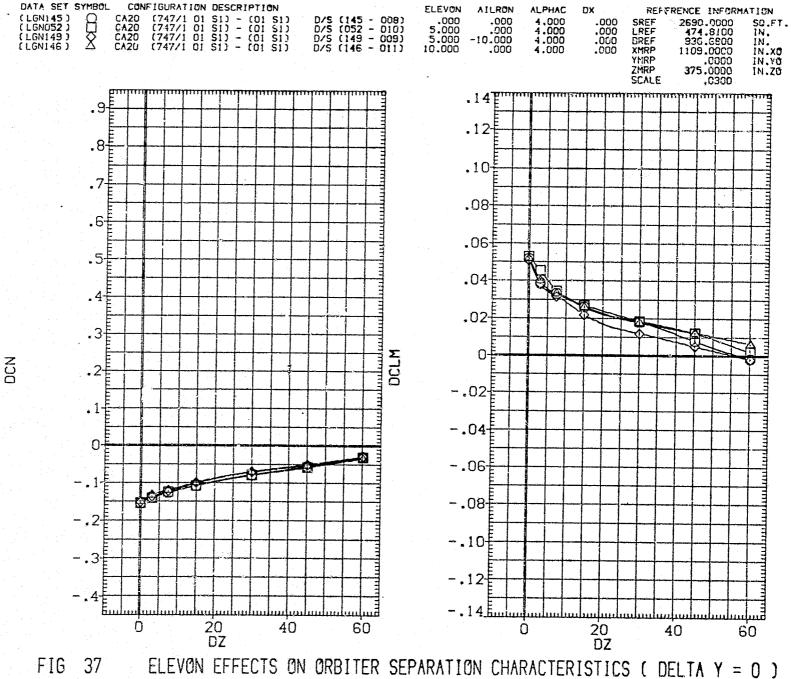
(B) ALPHAO = 10.00

PAGE 1709

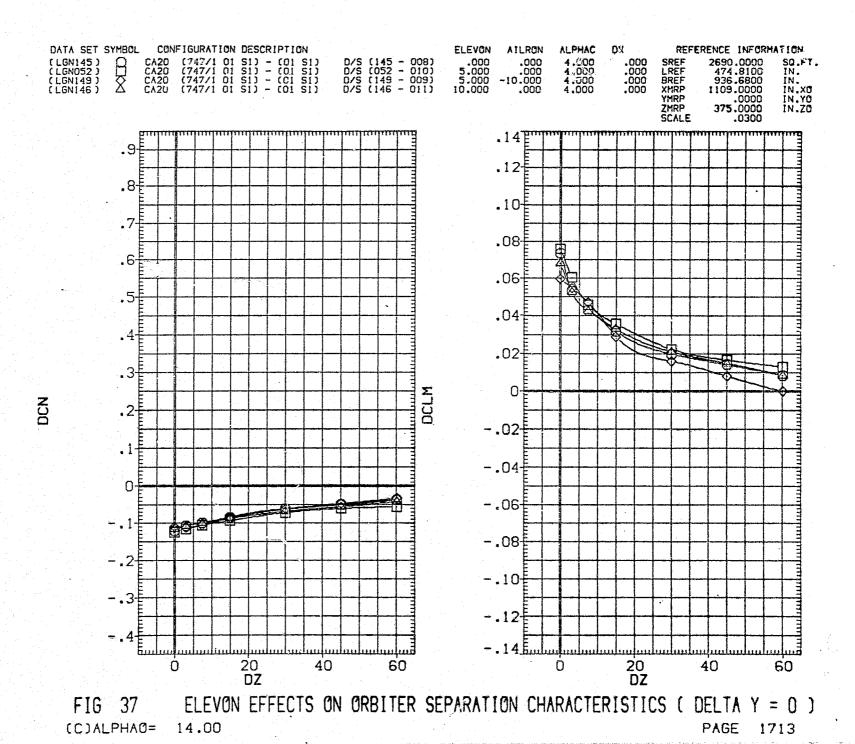


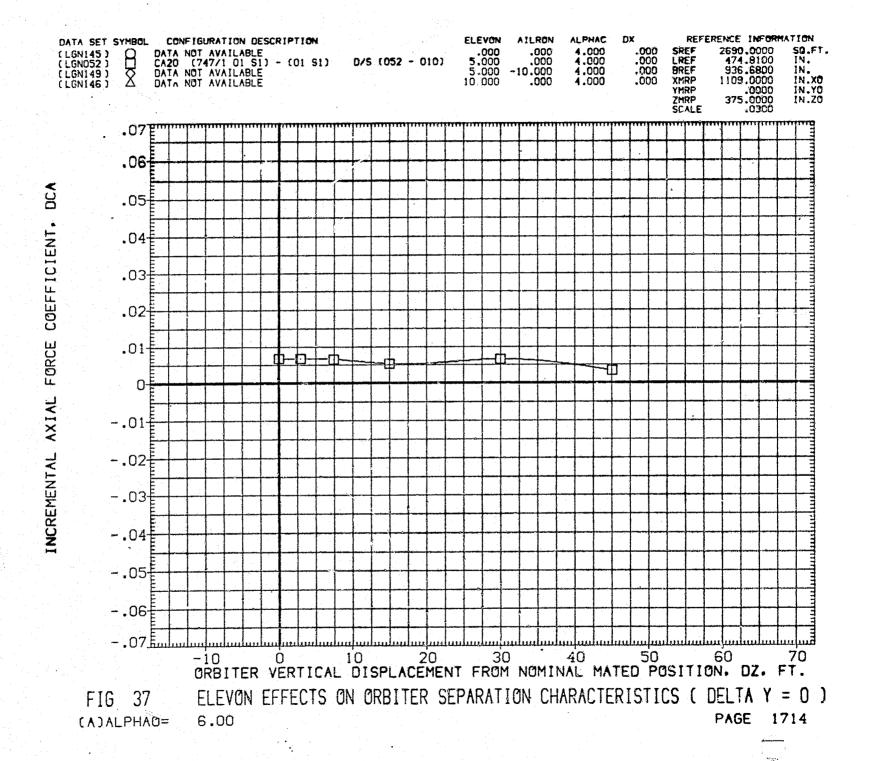
(C)ALPHAO= 14.00 PAGE 1710



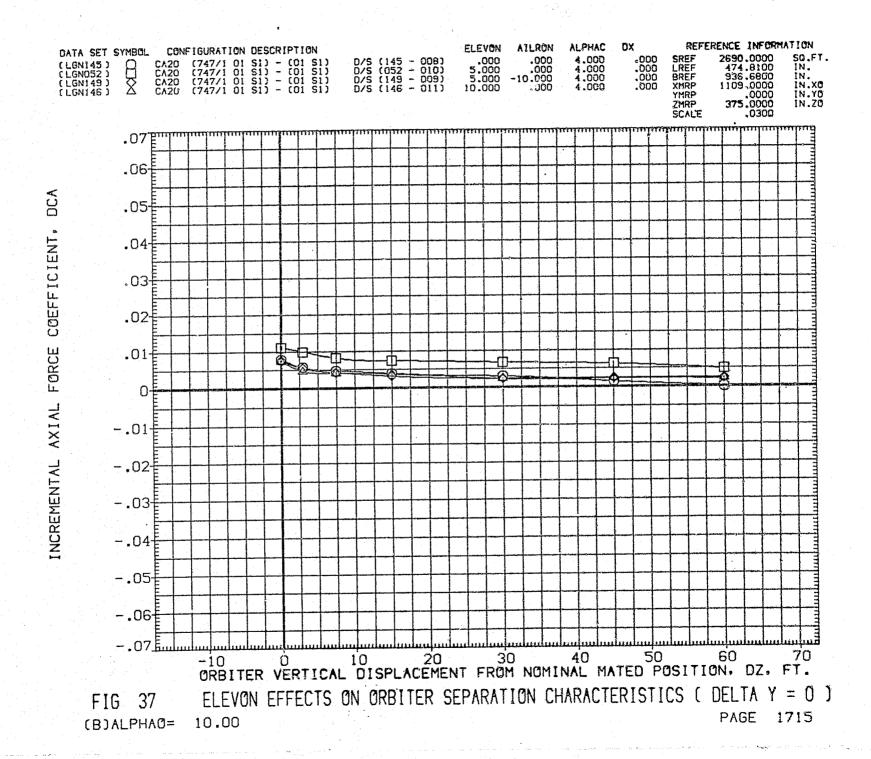


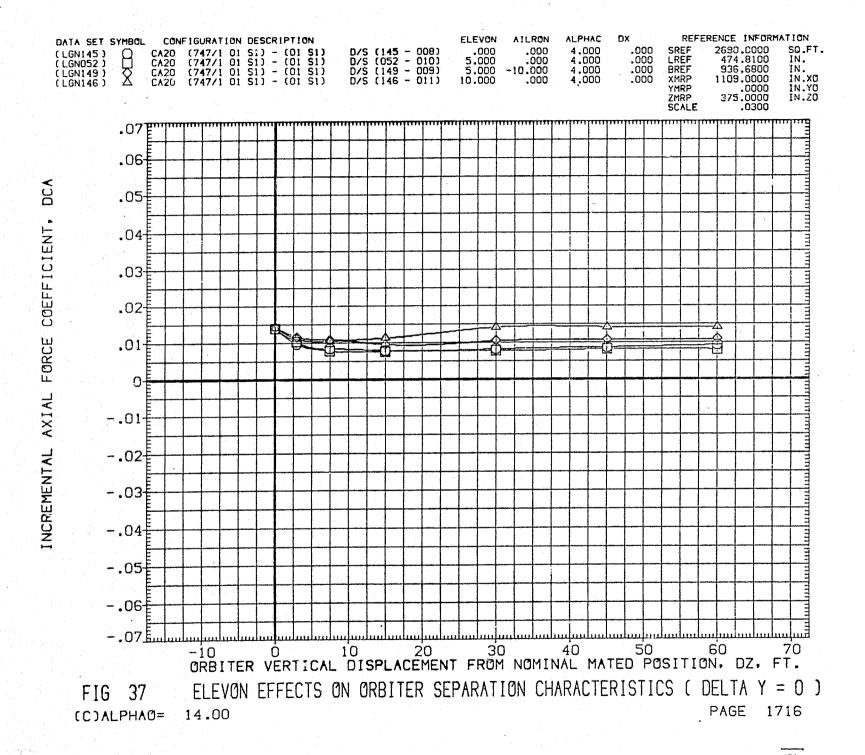
(B)ALPHAO= 10.00 PAGE 1712





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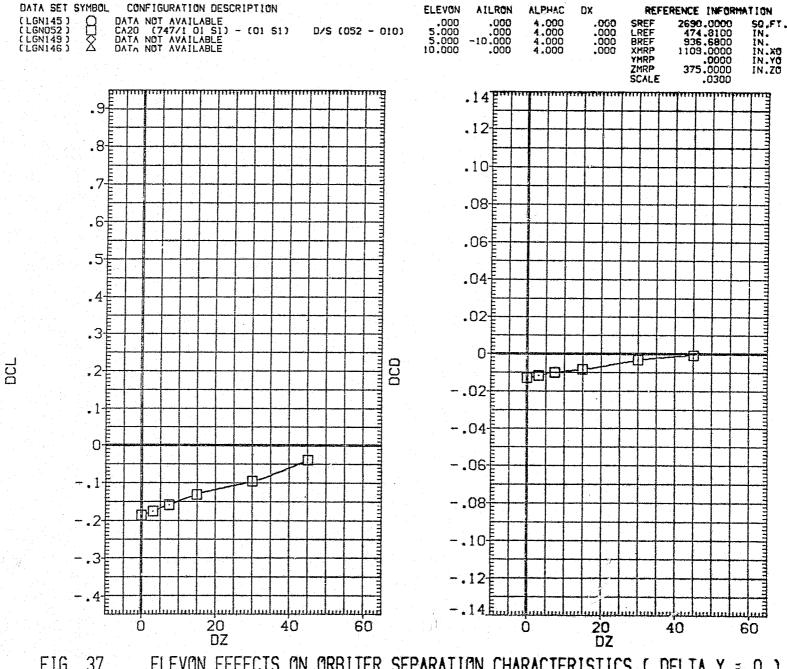


FIG 37 ELEVON EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )
(A)ALPHAO= 6.00

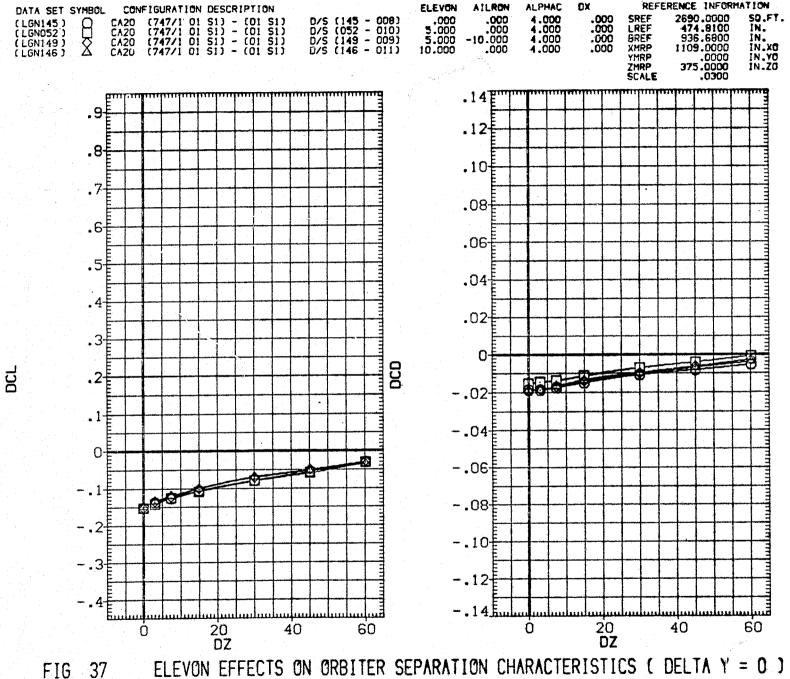
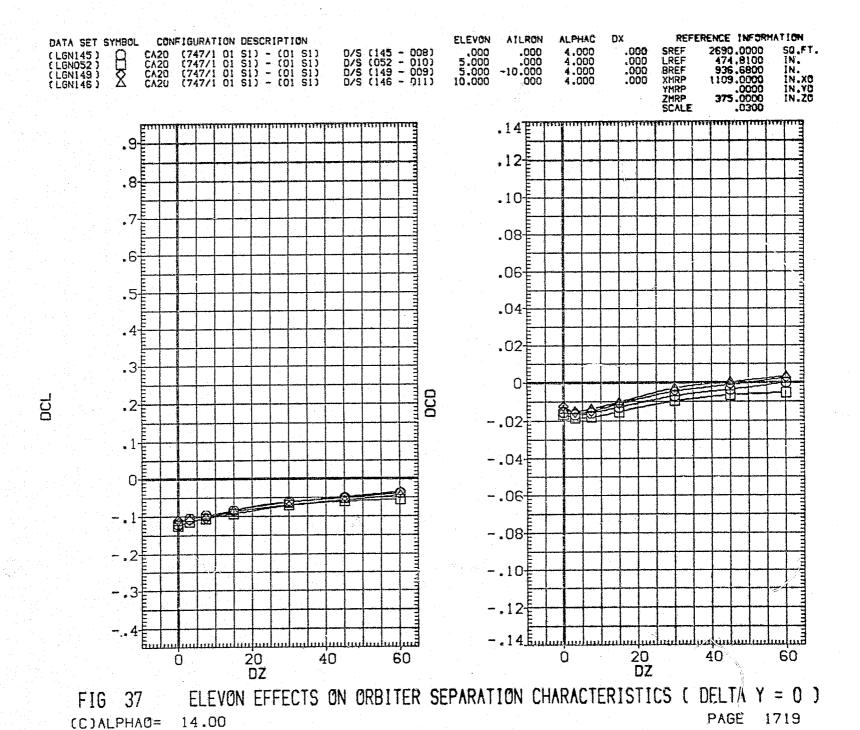


FIG 37 ELEVON EFFECTS ON ORBITER SEPARATION CHARACTERISTICS ( DELTA Y = 0 )

(B)ALPHAO = 10.00 PAGE 1718



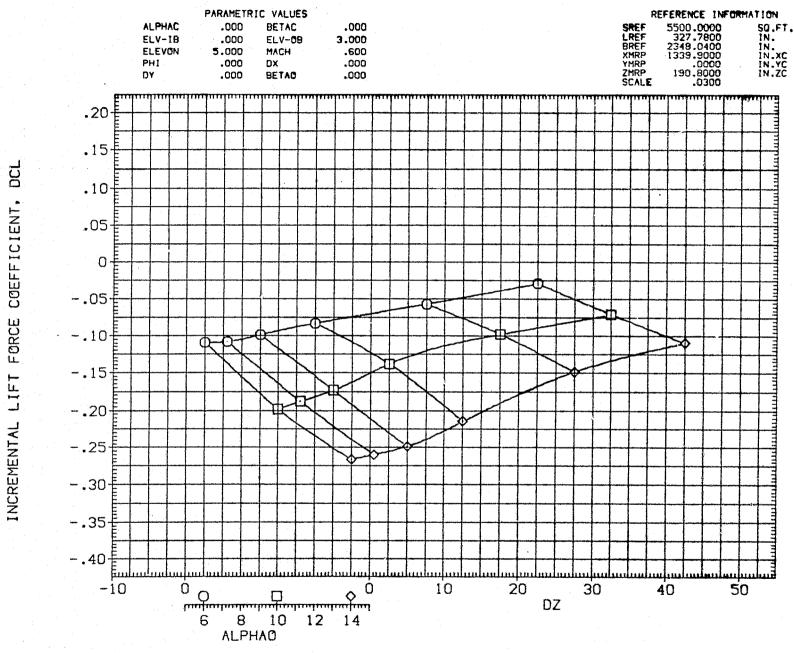
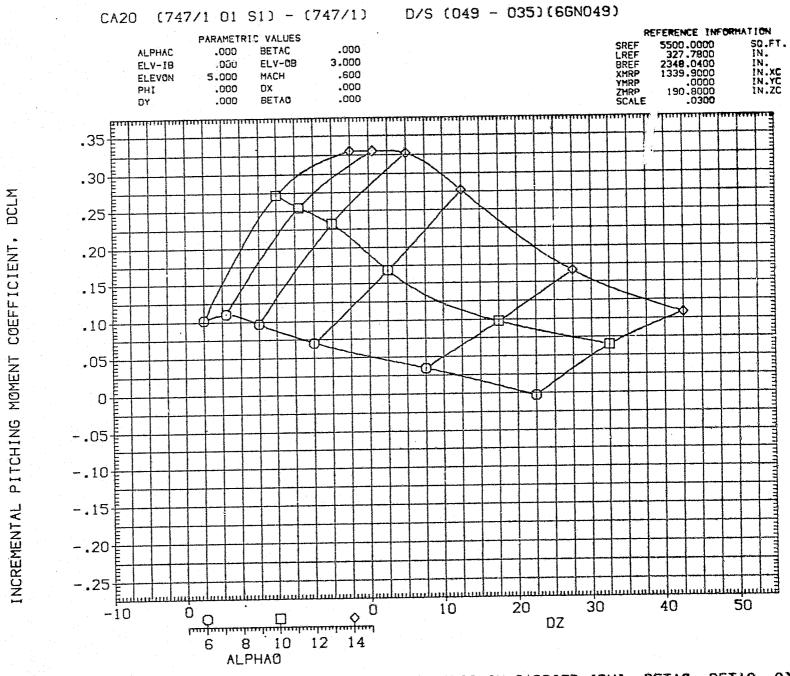


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1720



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FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1721

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

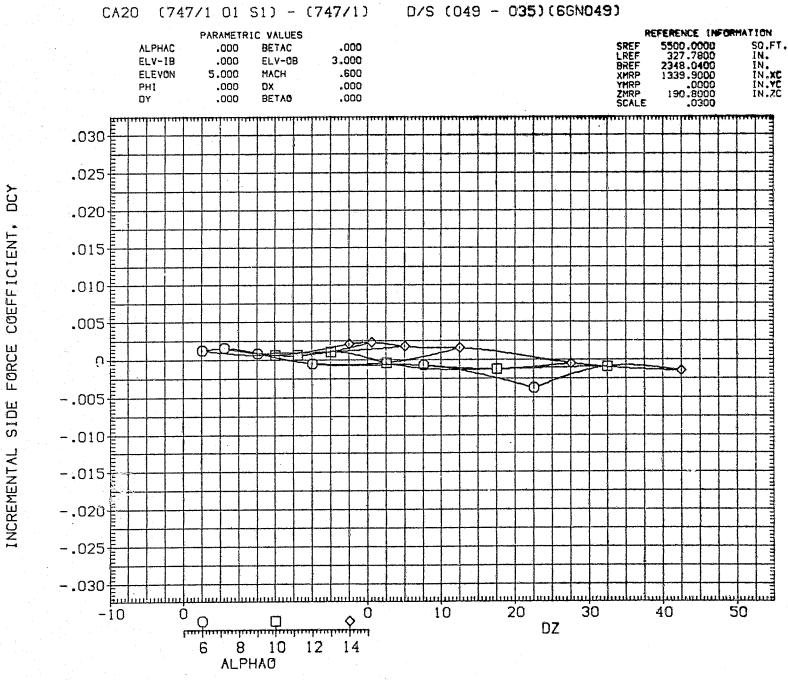


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1723

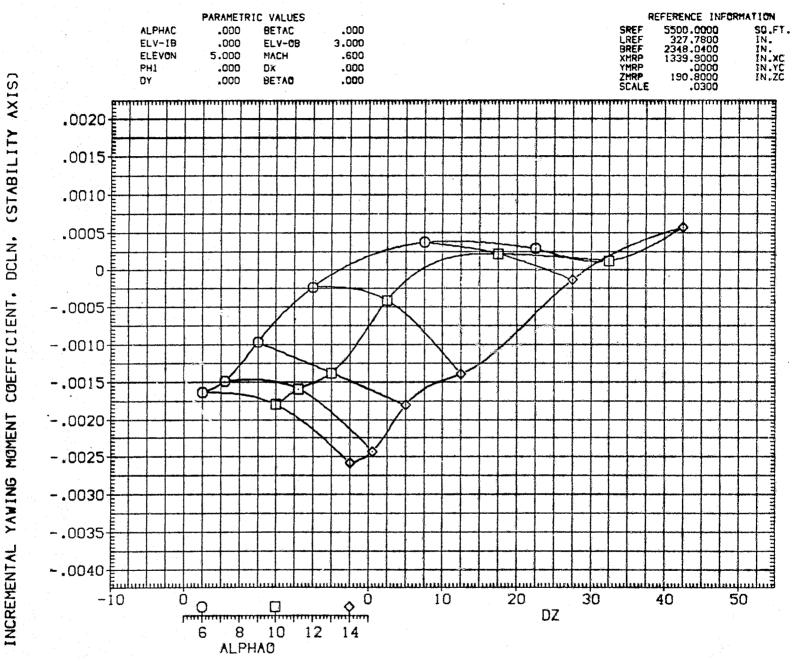


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1724

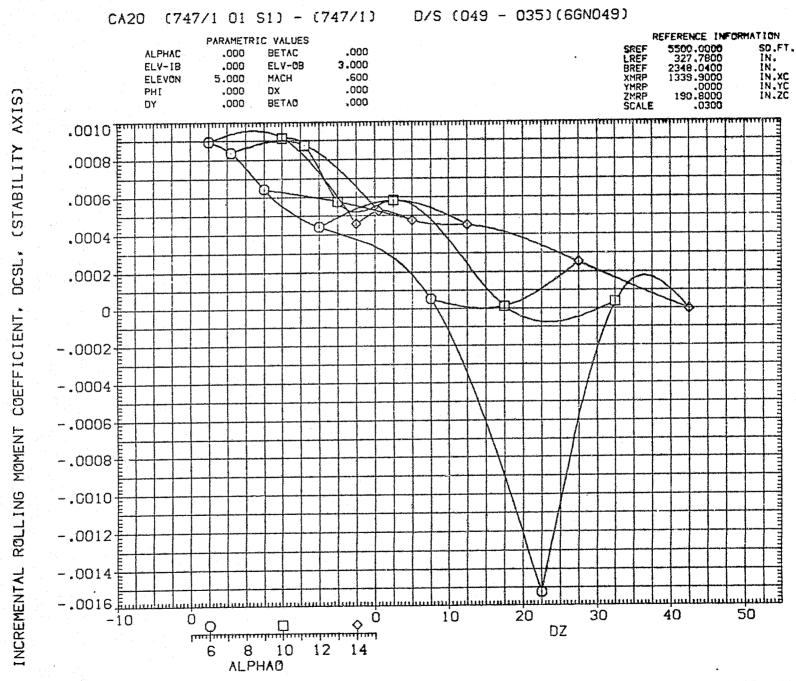


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO, BETAC =0)
PAGE 1725

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO, BETAC =0)
PAGE 1726

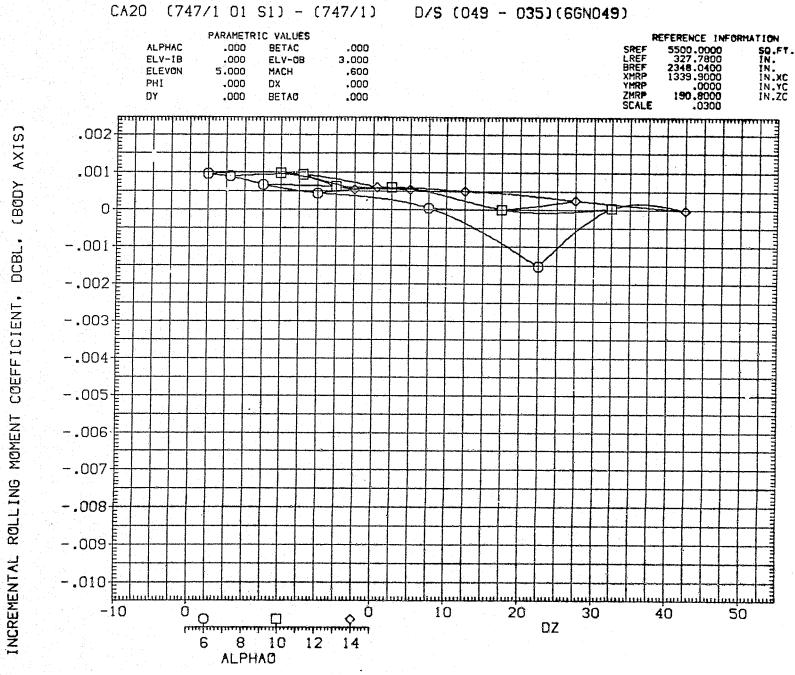


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO, BETAC =0)
PAGE 1727

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO. BETAC =0)
PAGE 1728

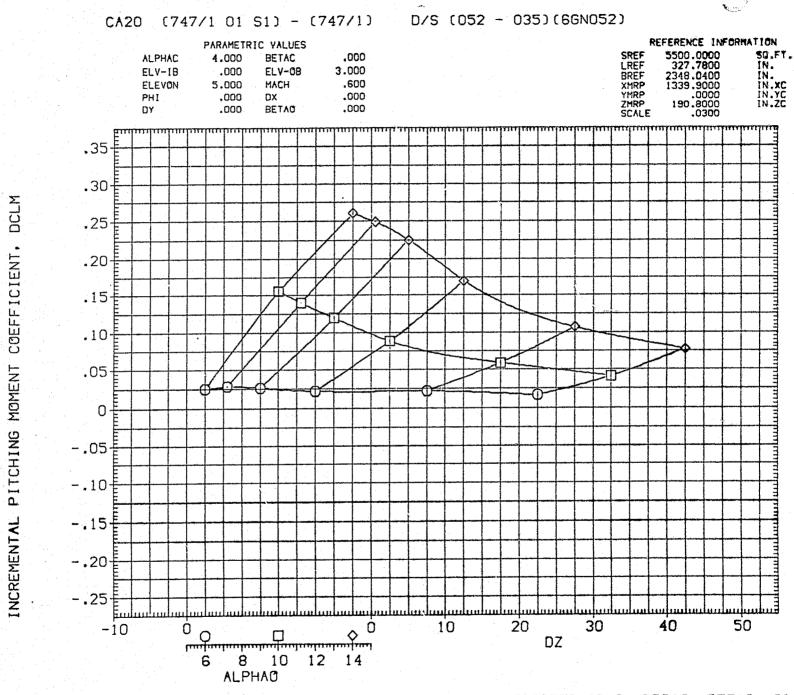


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1729

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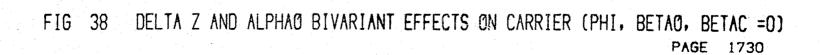
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ALPHAO

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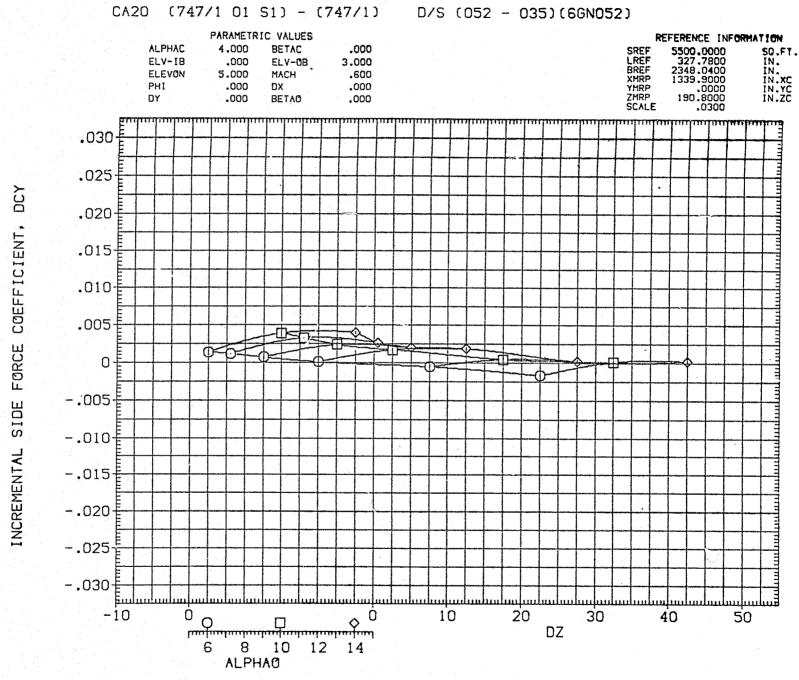
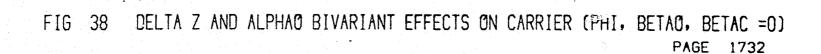


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1731



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ALPHAO

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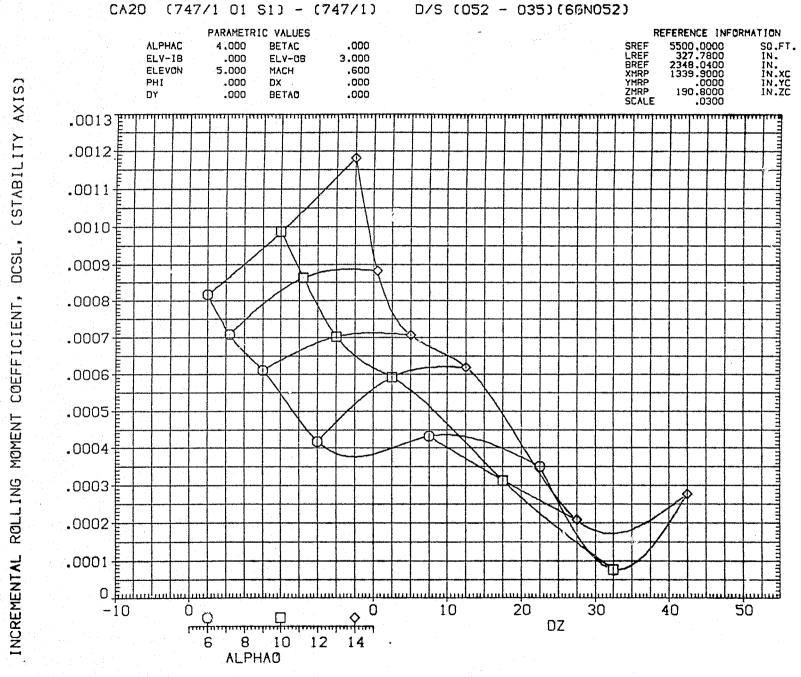


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO, BETAC =0)
PAGE 1733

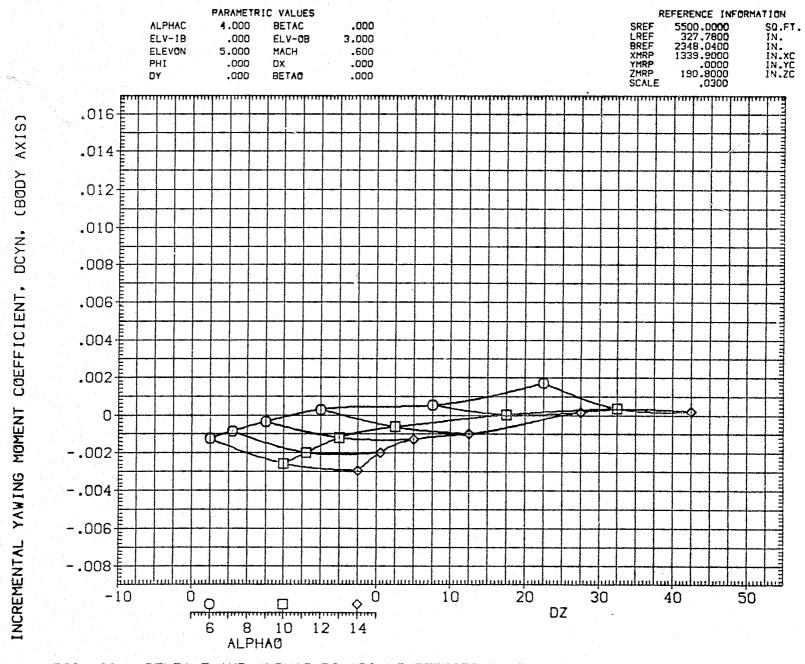


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1734

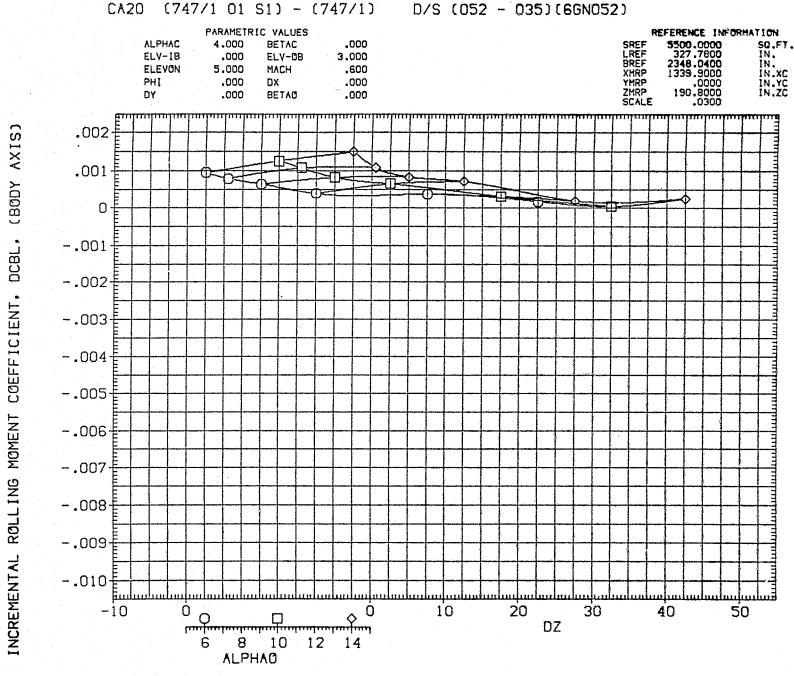


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1735

COEFFICIENT

FORCE

INCREMENTAL

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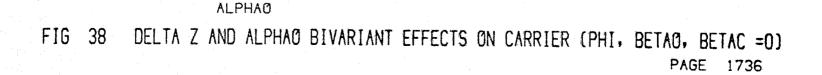
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MOMENT

PITCHING

INCREMENTAL

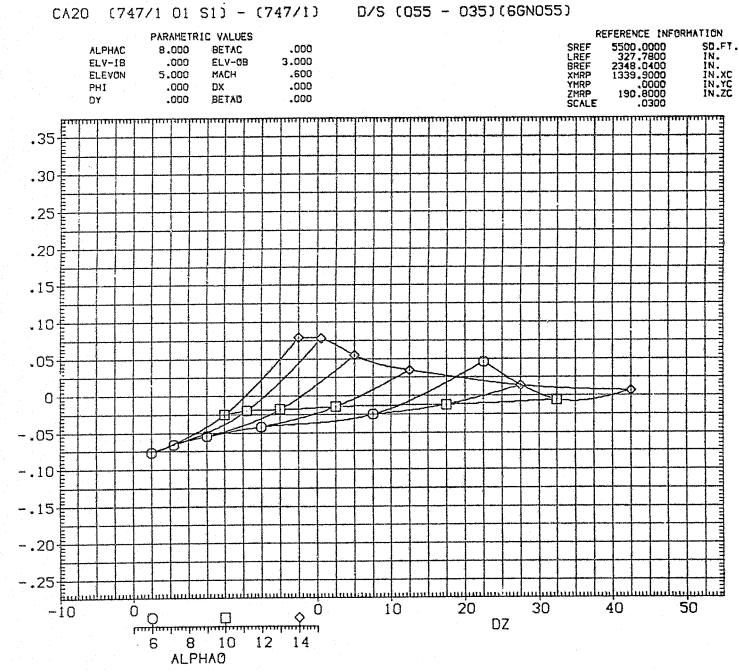


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1737

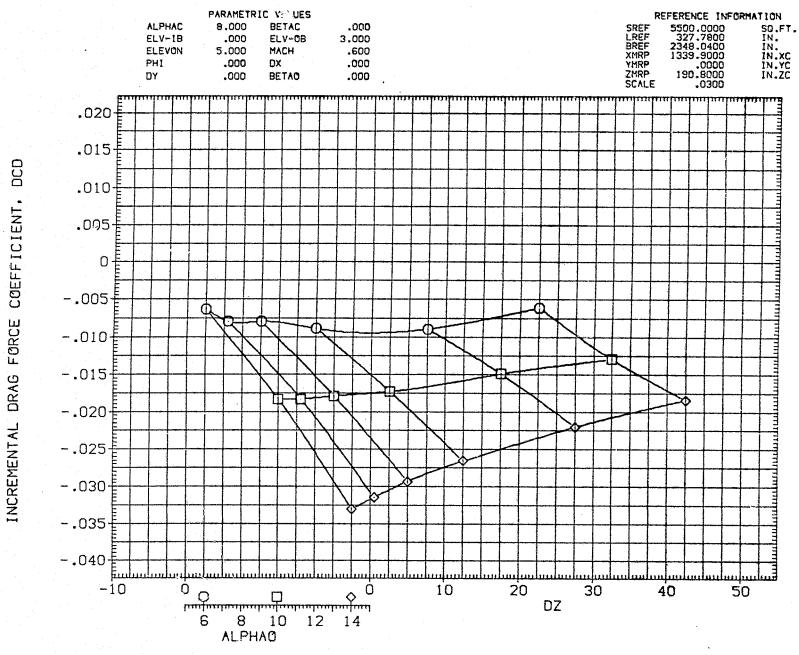


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1738

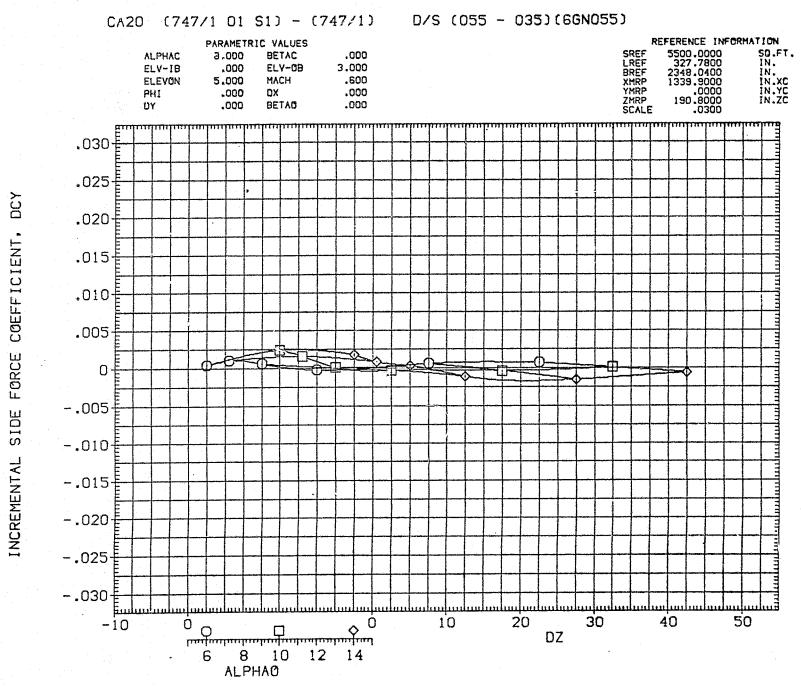


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1739

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1740



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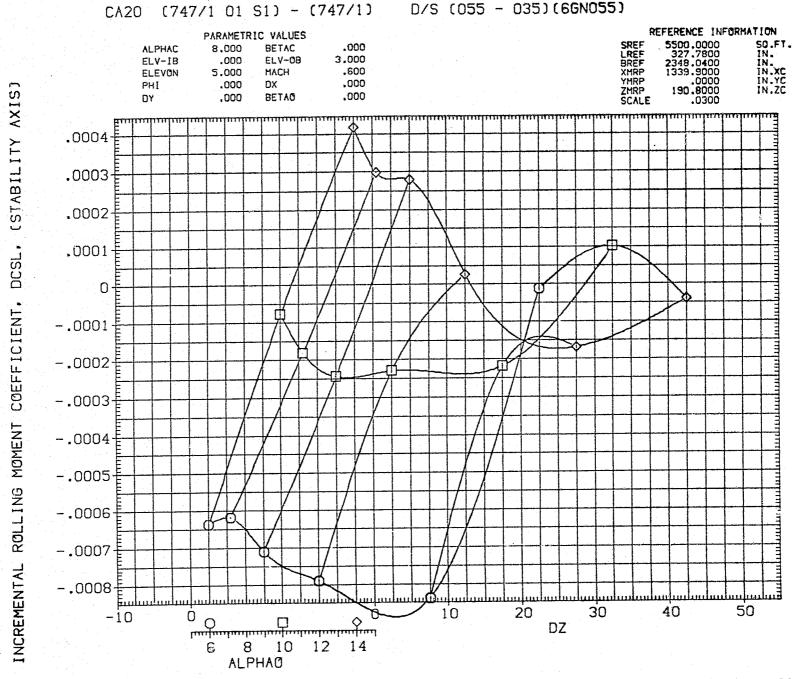


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1741

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0) PAGE 1742

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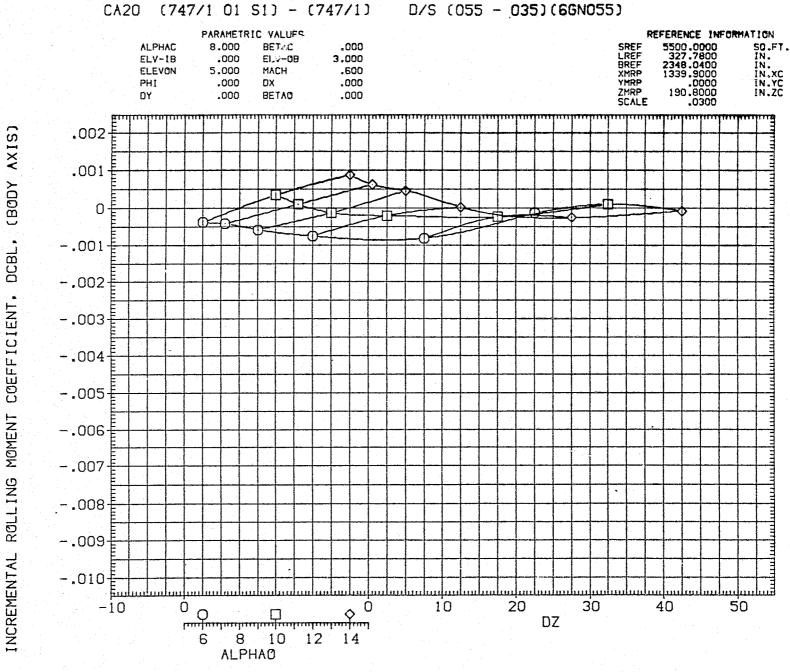


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1743

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
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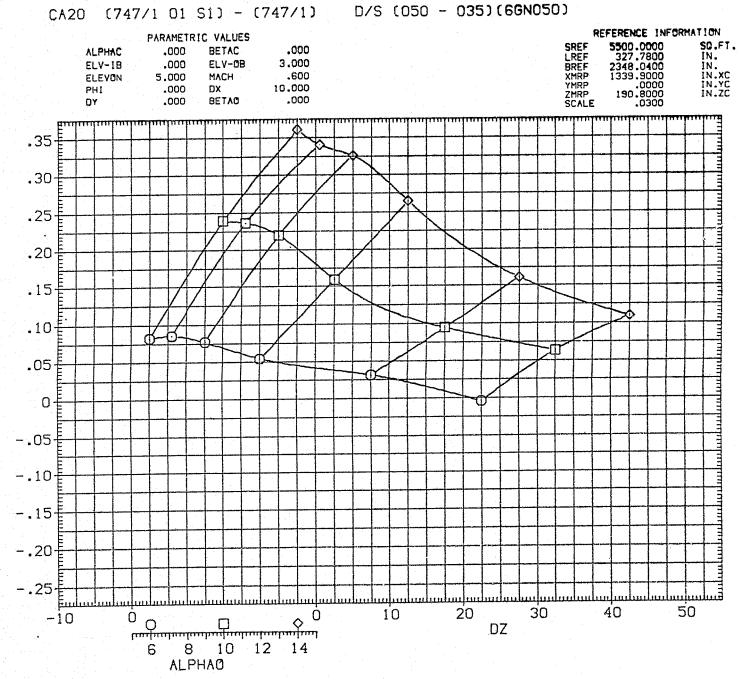


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1745

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1746

ALPHAO



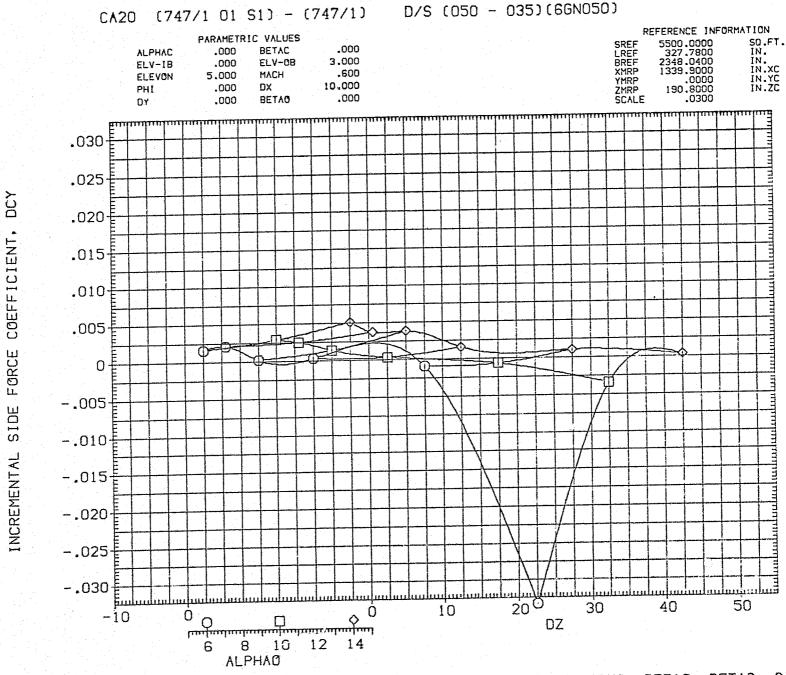


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

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DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0) FIG 1749 PAGE

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1750

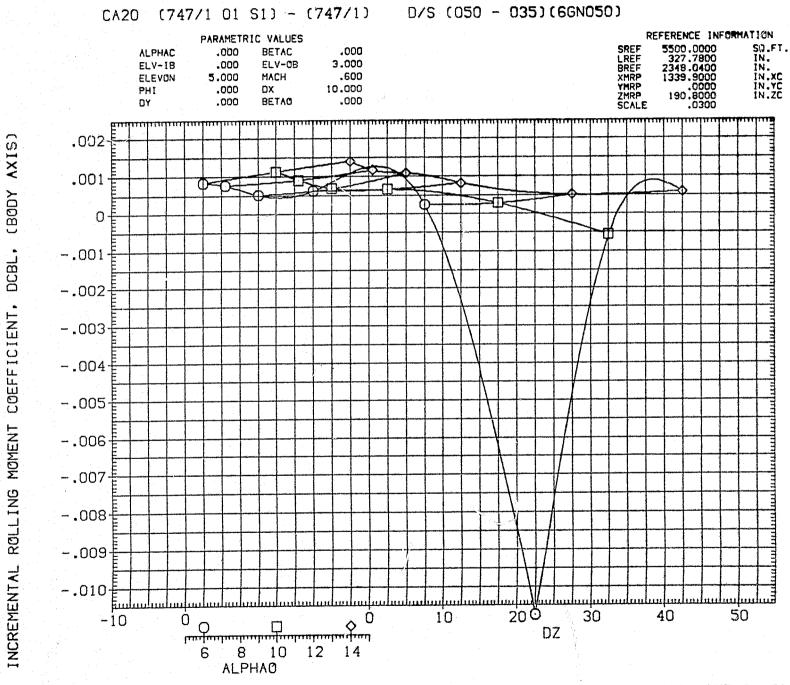


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1752

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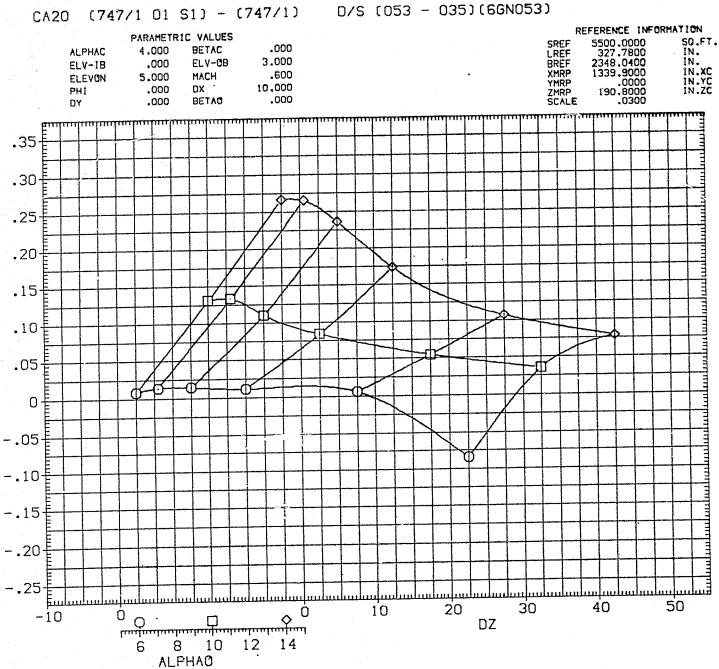


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

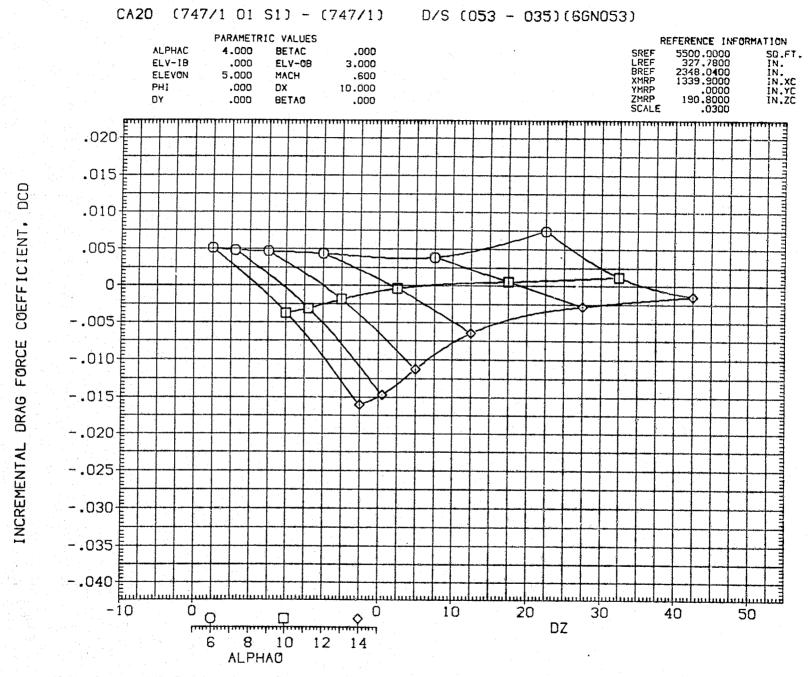


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1754

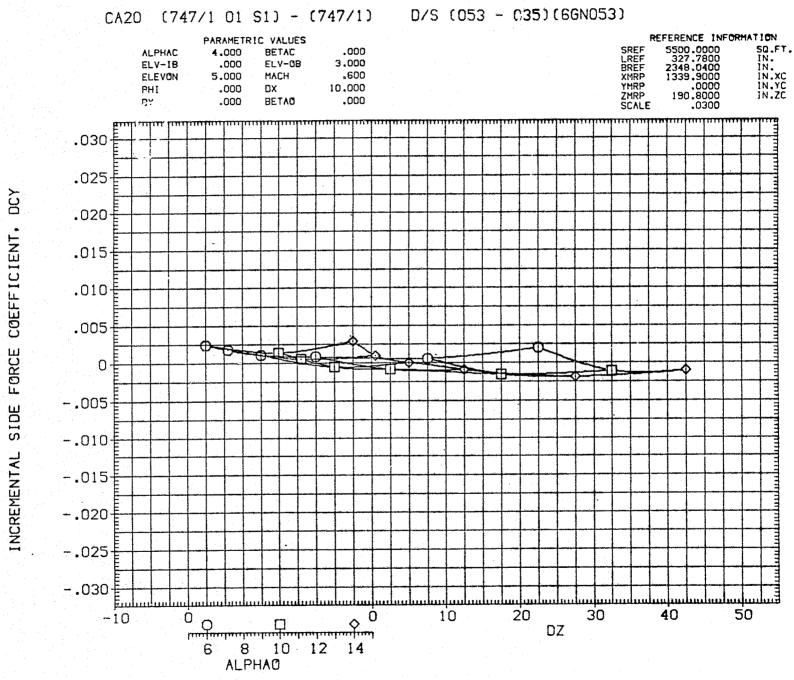


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1755

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
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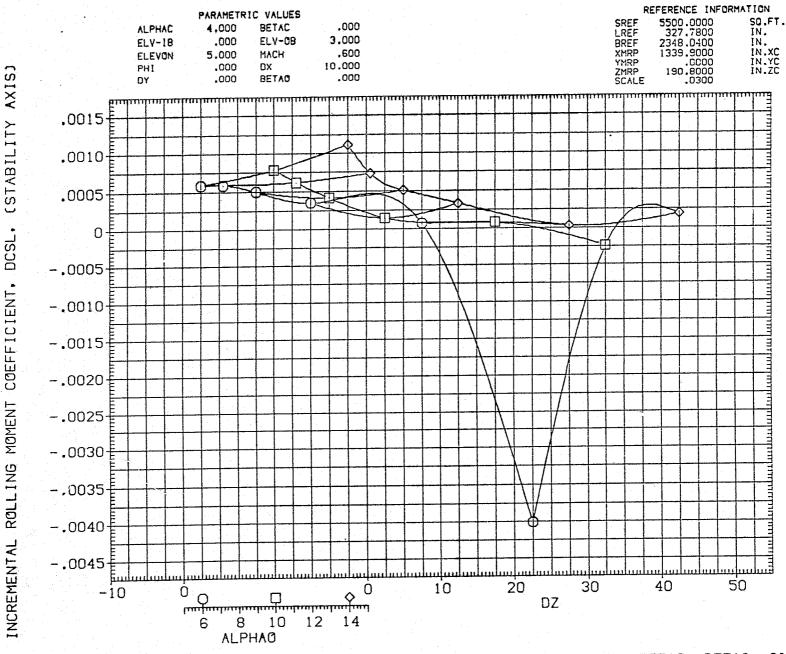


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
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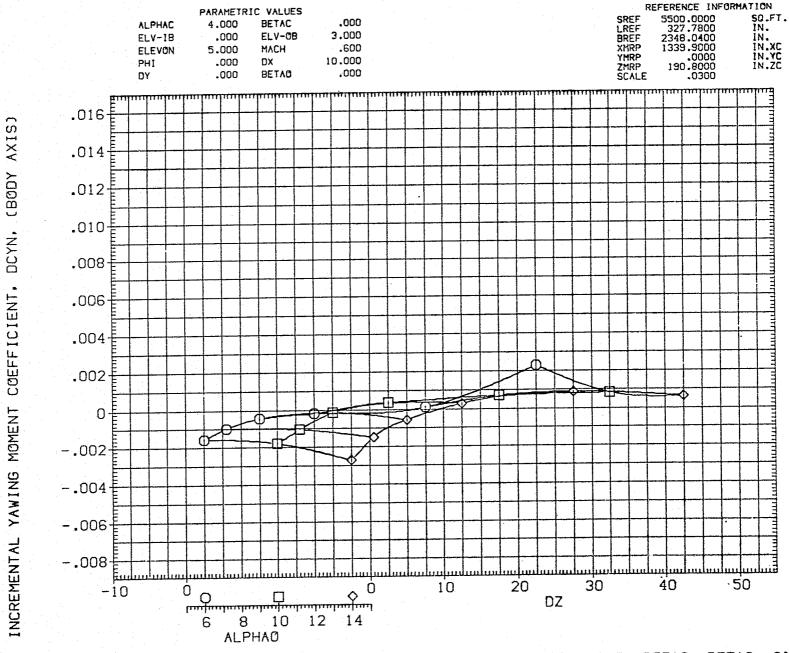


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1758

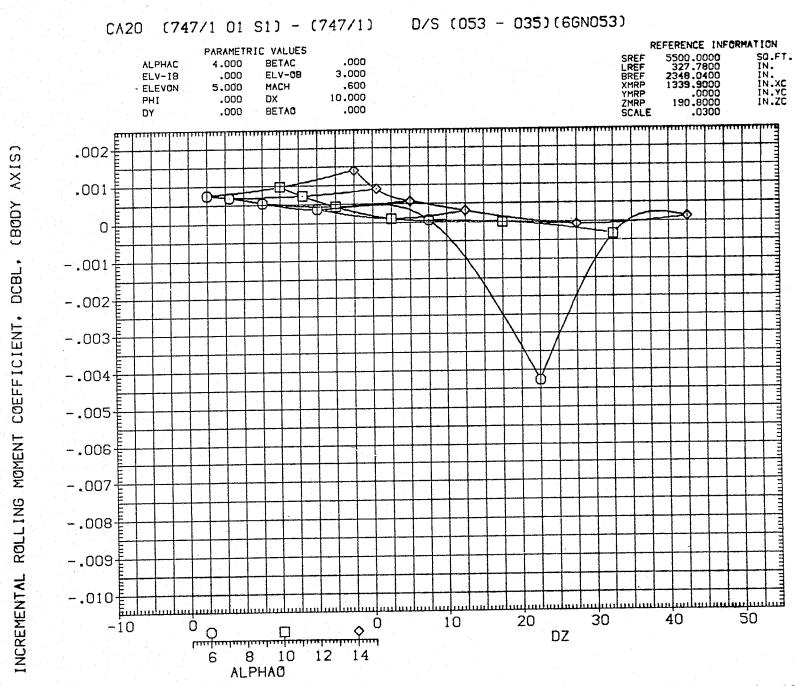
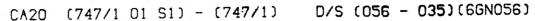


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1759

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1760

**ALPHAO** 



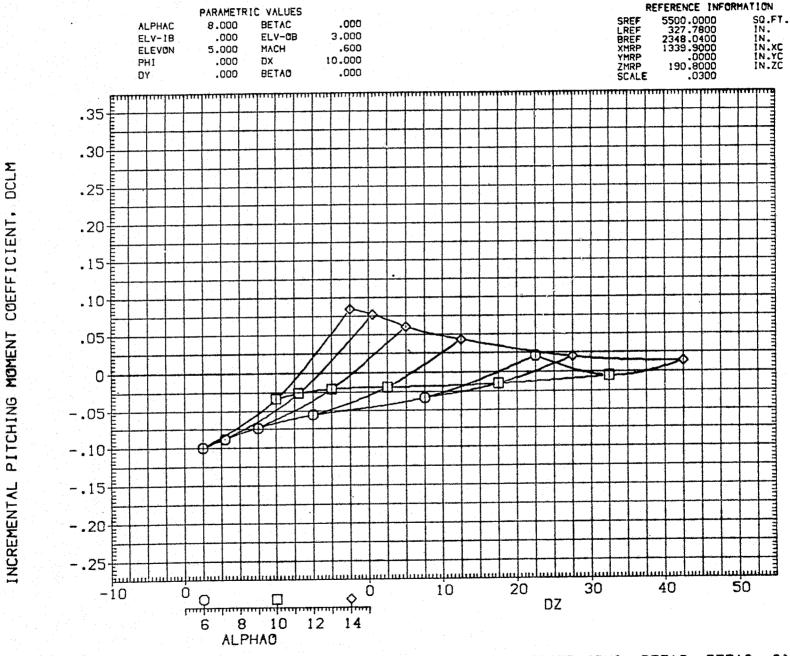


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1761

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
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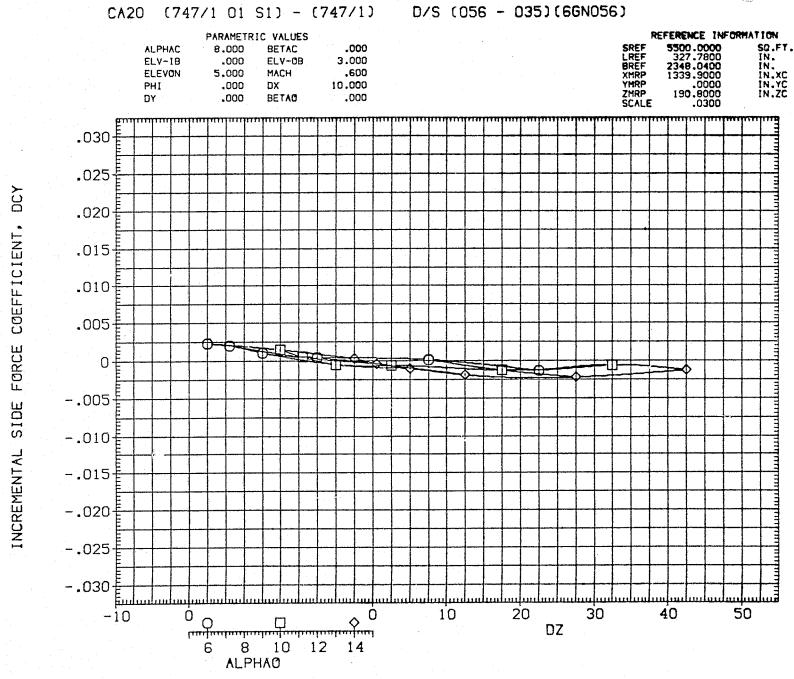


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1763

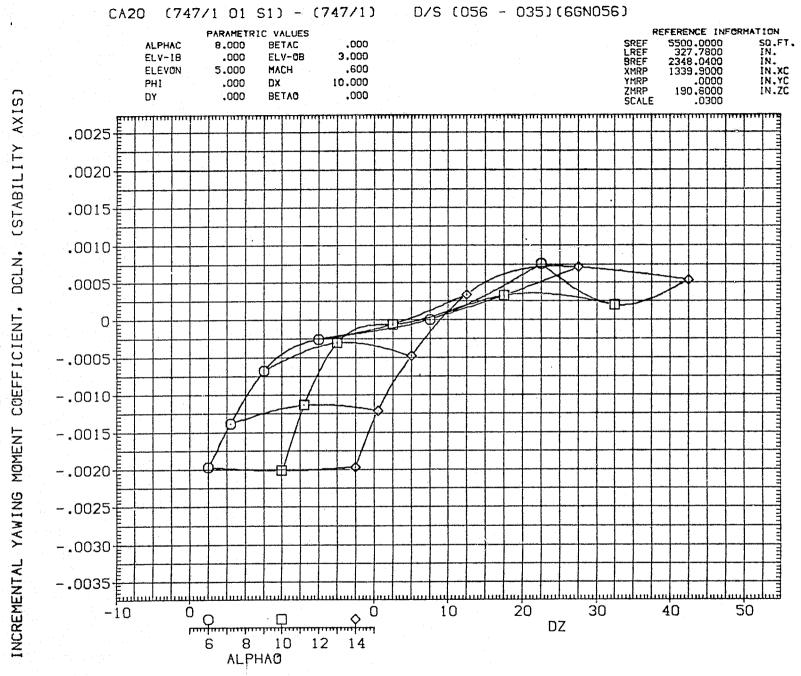


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1764

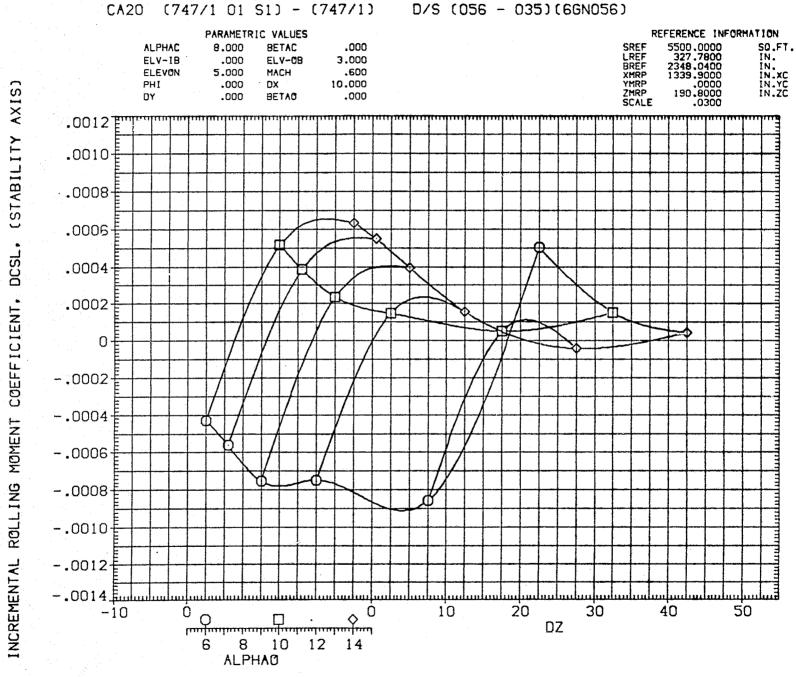


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1765

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO, BETAC =0)

PAGE 36

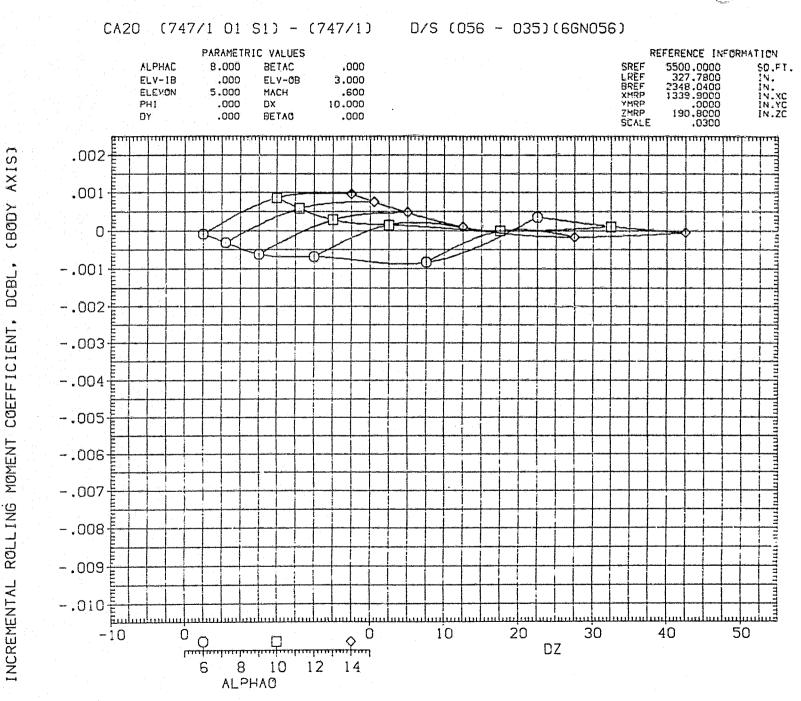


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1767

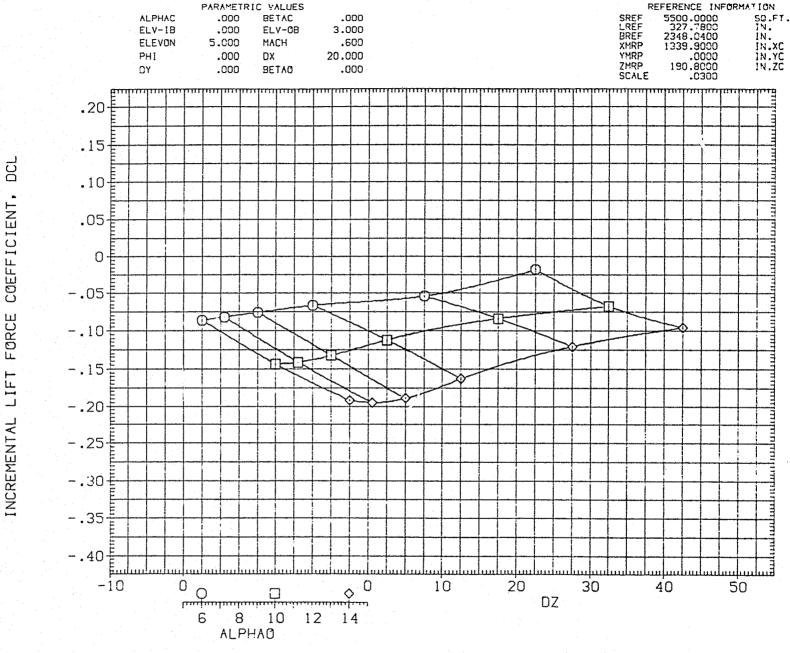


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1768



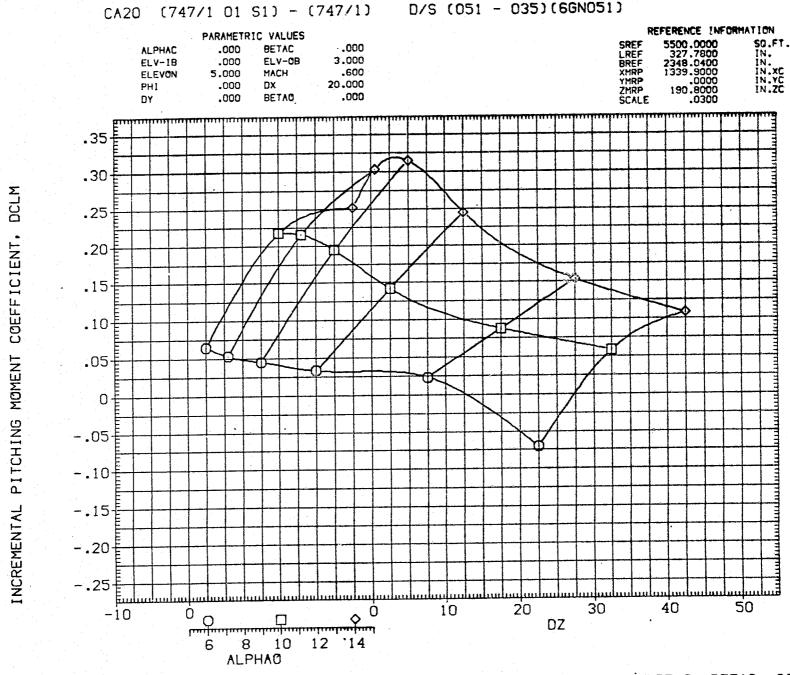


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1769

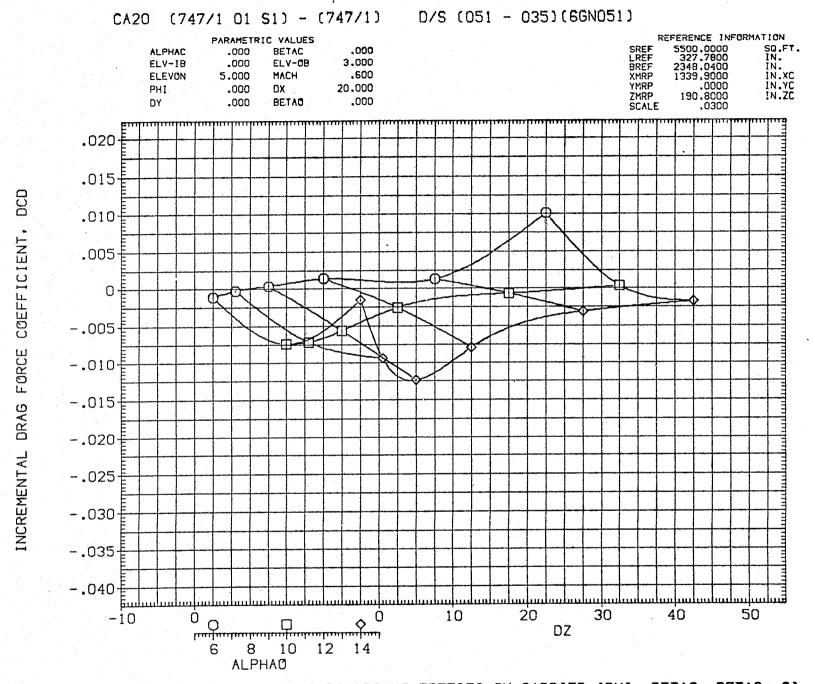


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1770

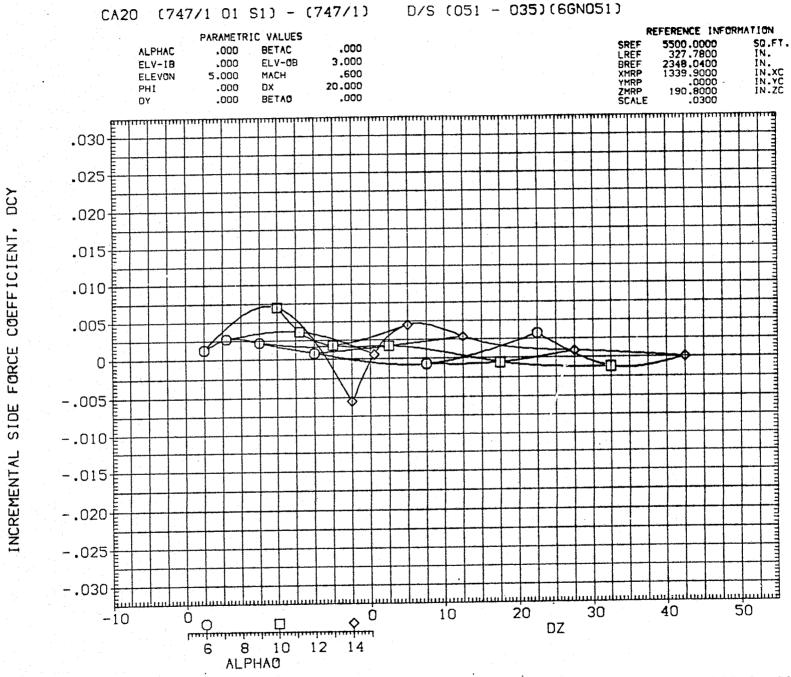


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1772

ALPHAO

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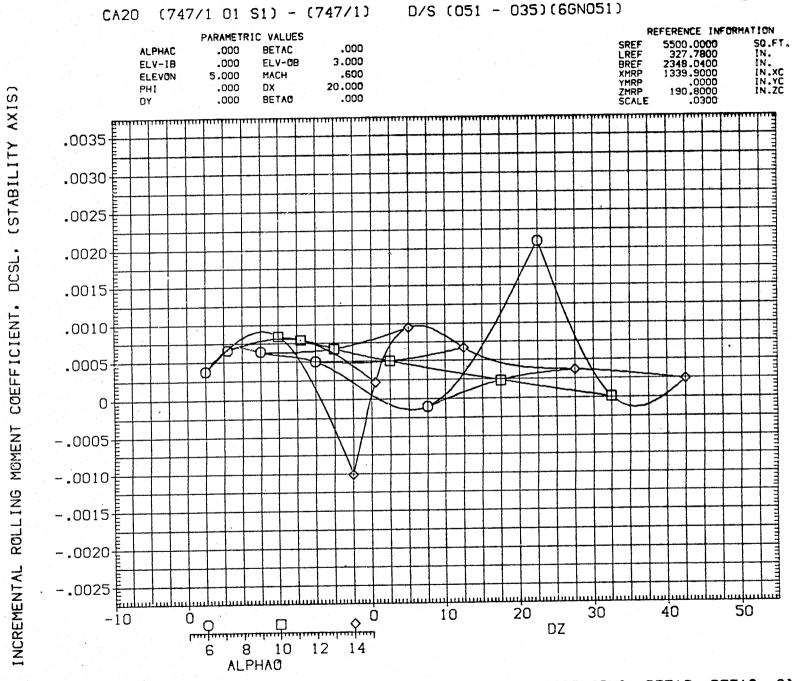


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1773

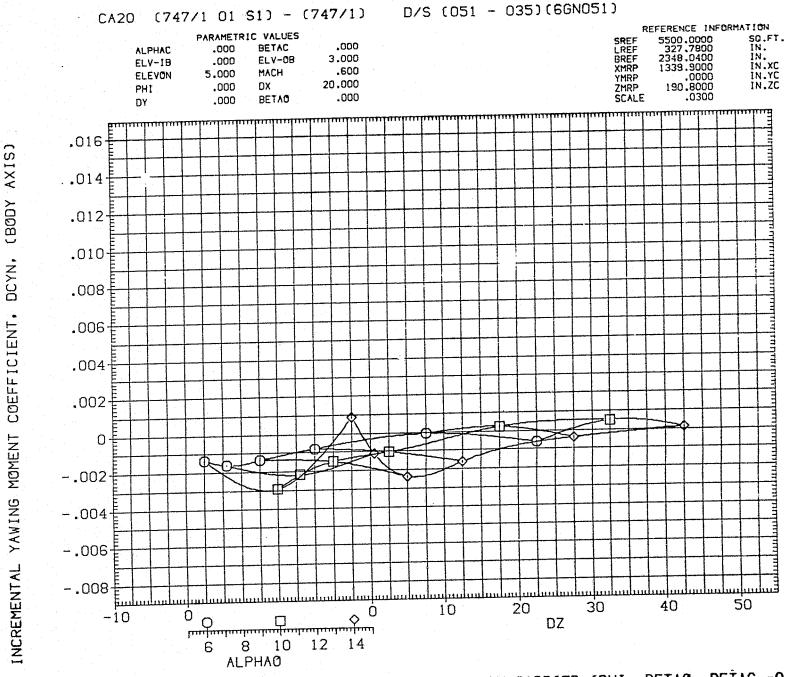


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)



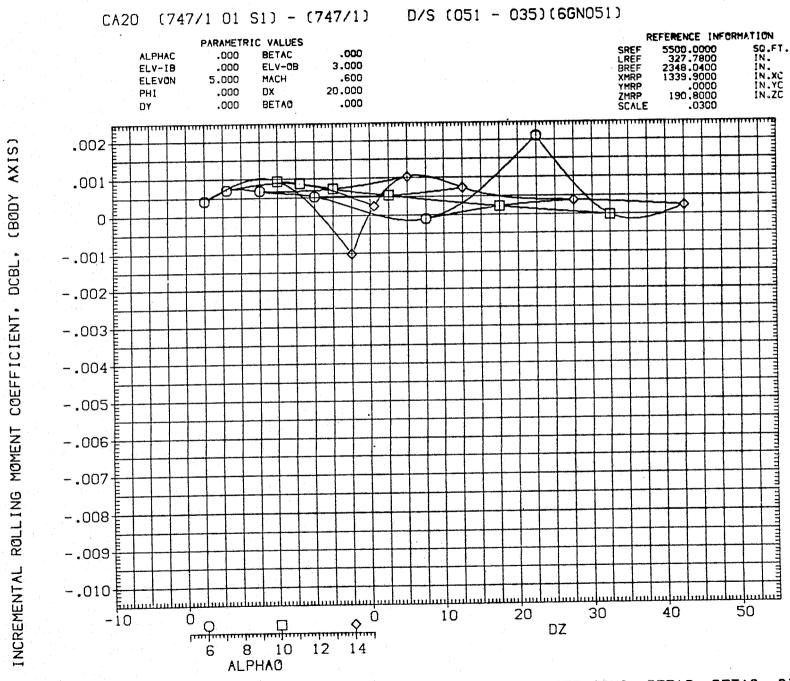


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1775

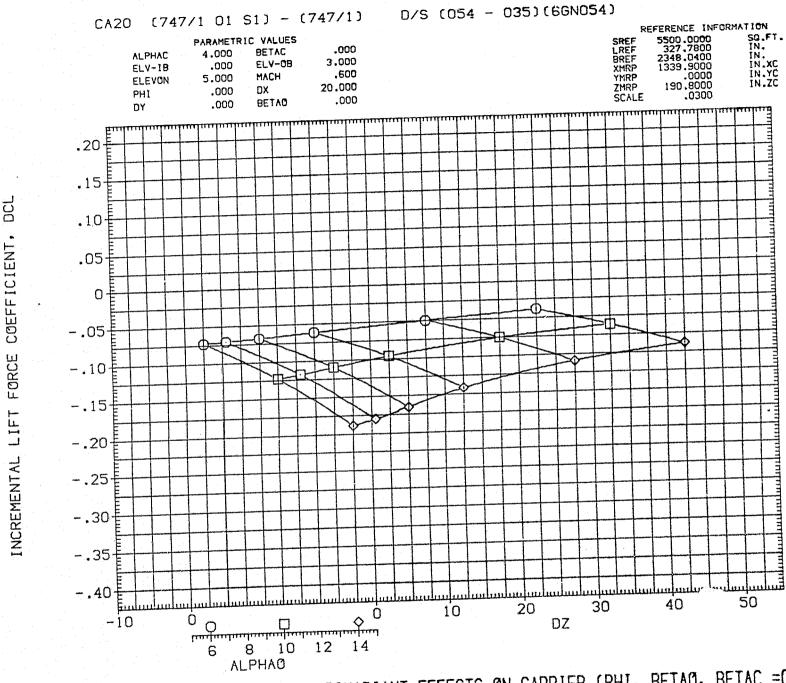


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

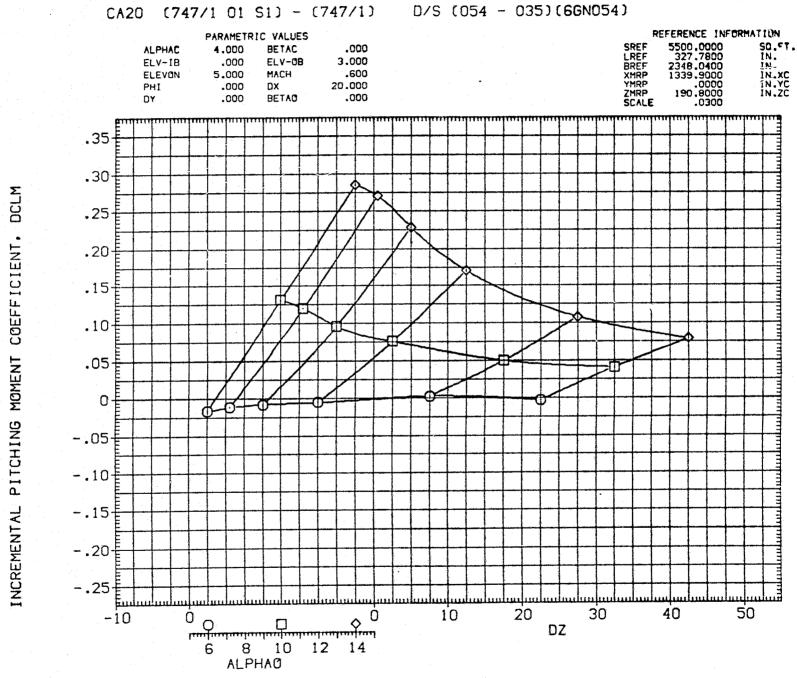
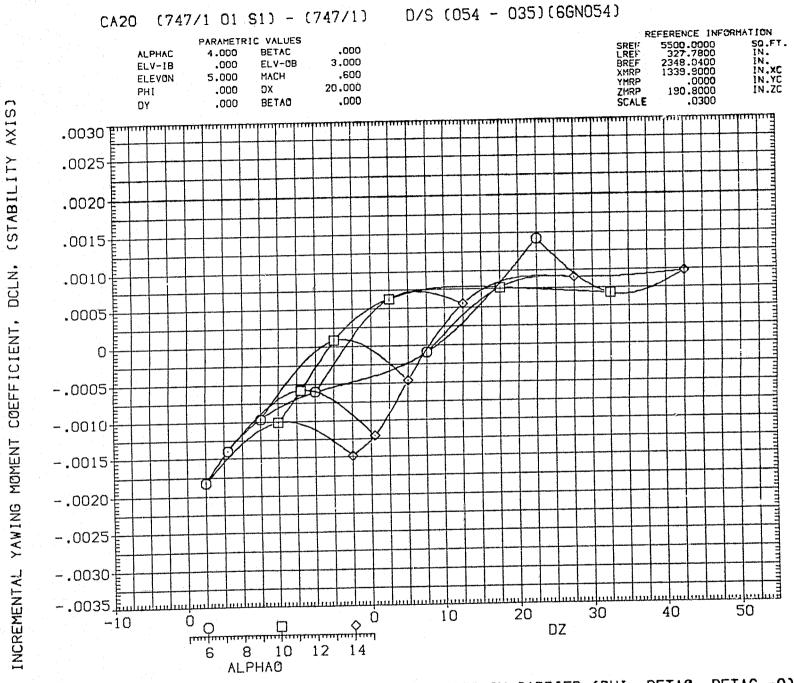


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1777

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1779



DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0) PAGE 1780

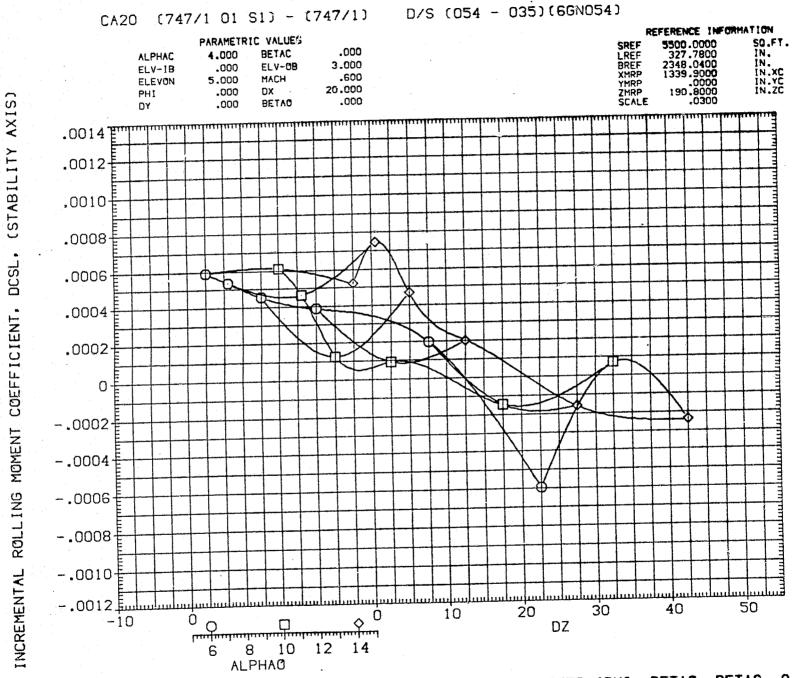


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0) FIG 38 PAGE 1782

ALPHAO

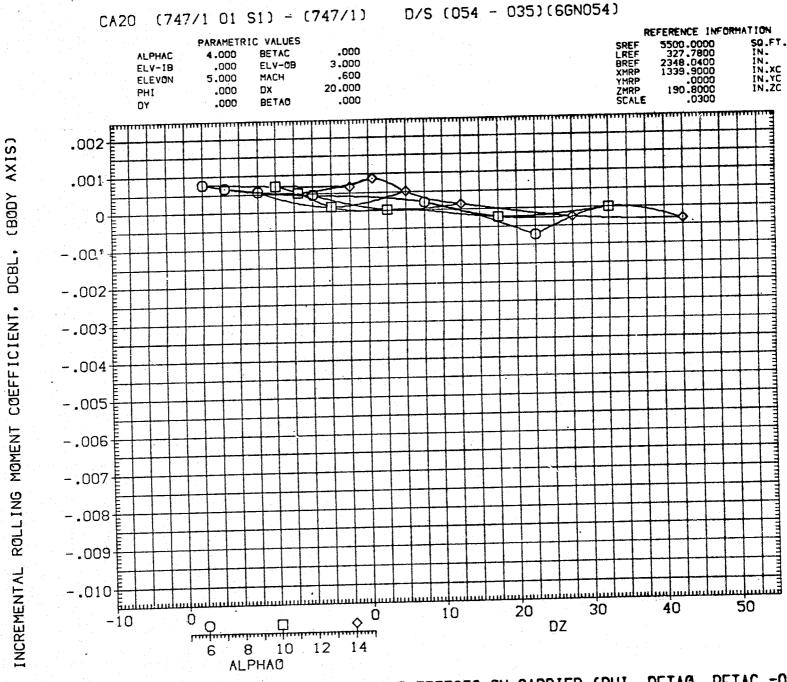


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1783

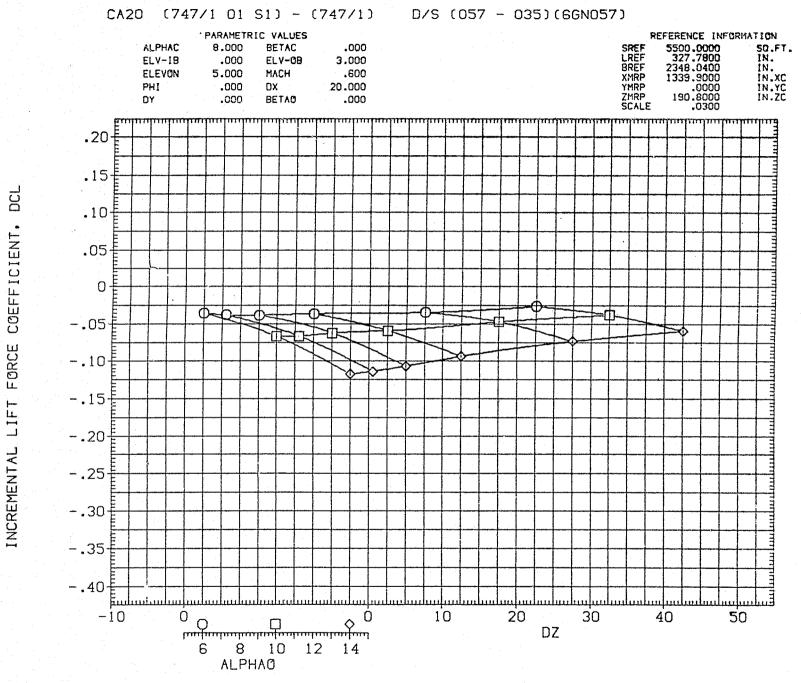


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1784

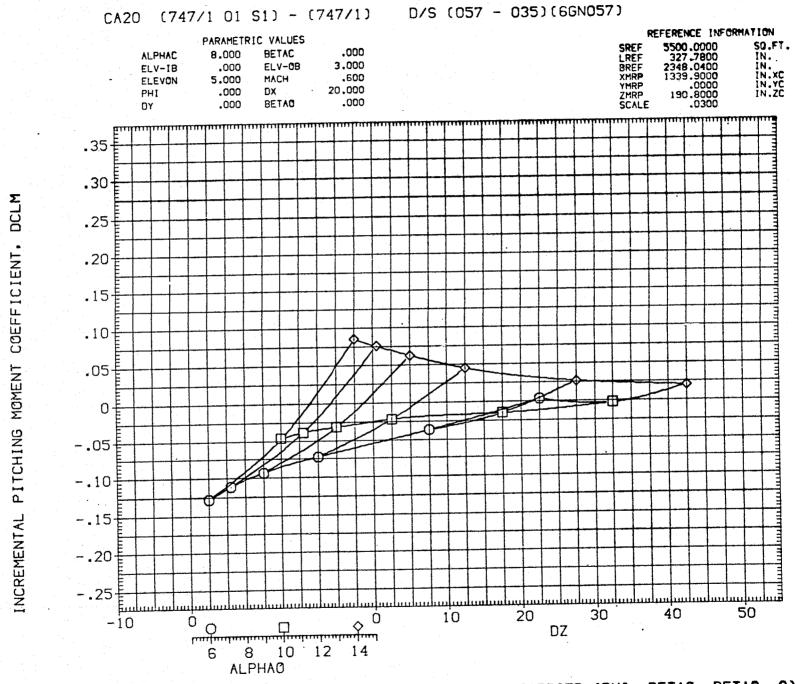


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1785

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI. BETAO, BETAC =0)
PAGE 1786

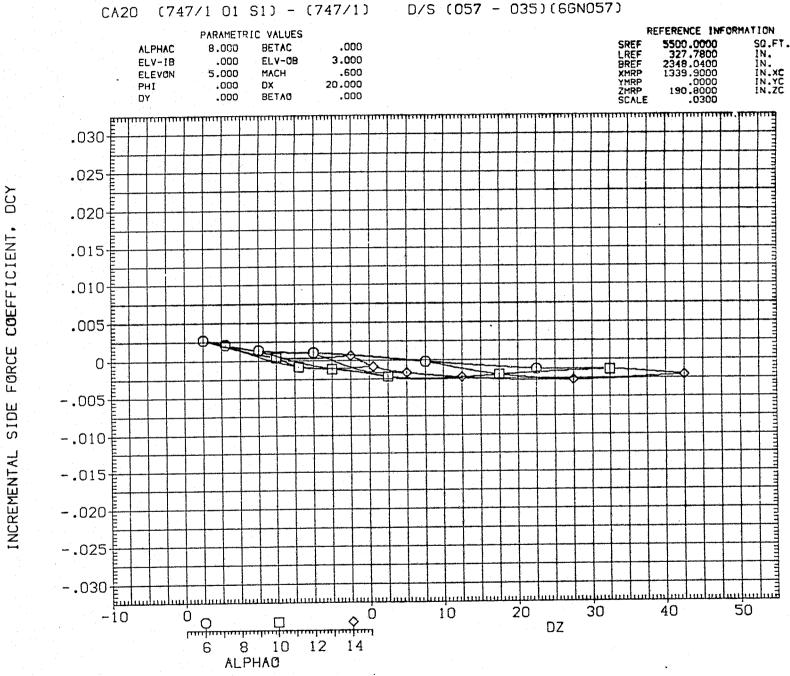


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1787

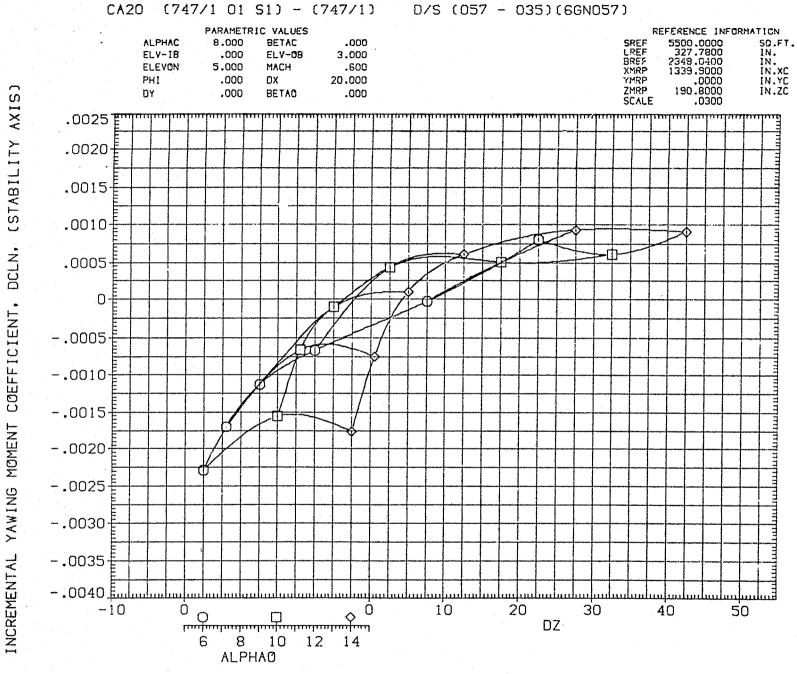


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1788



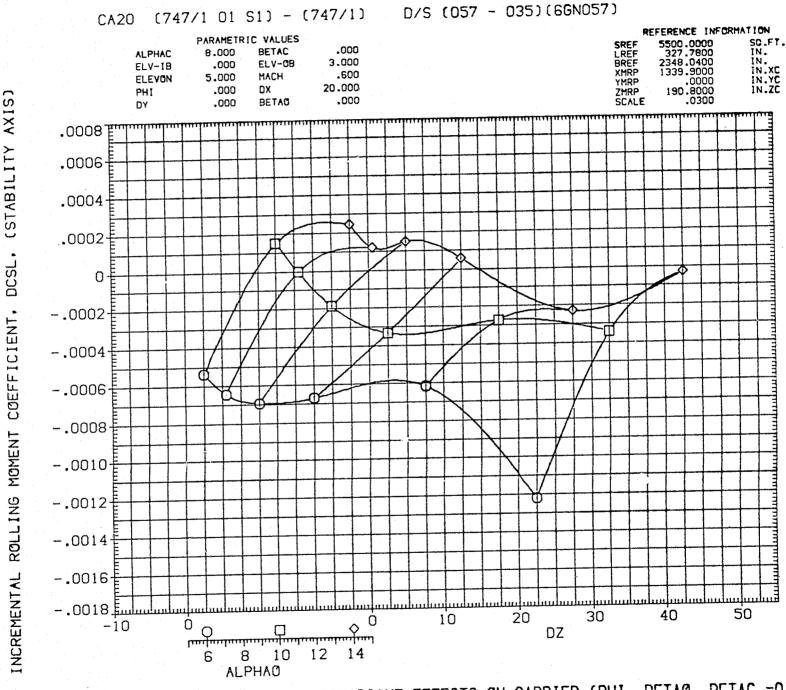


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)
PAGE 1790

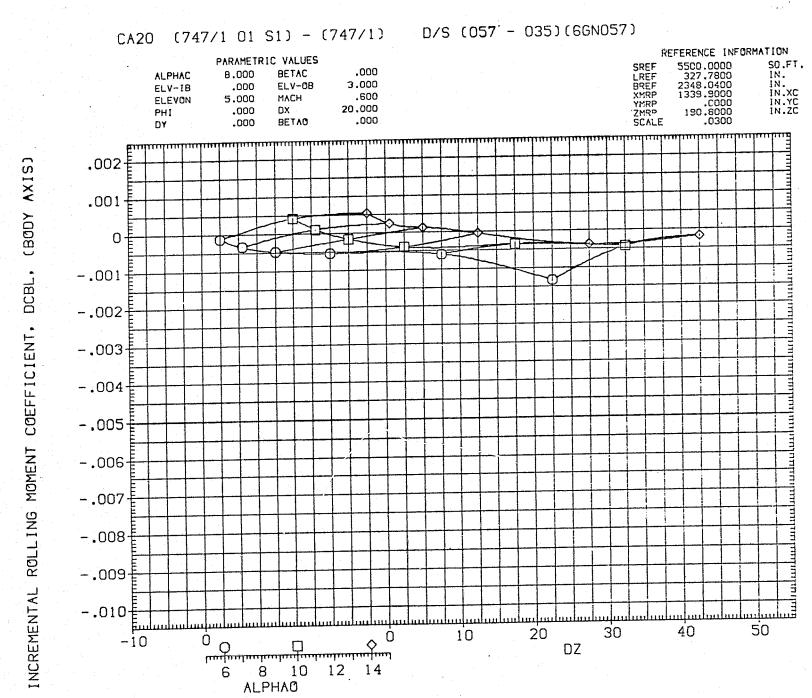


FIG 38 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CARRIER (PHI, BETAO, BETAC =0)

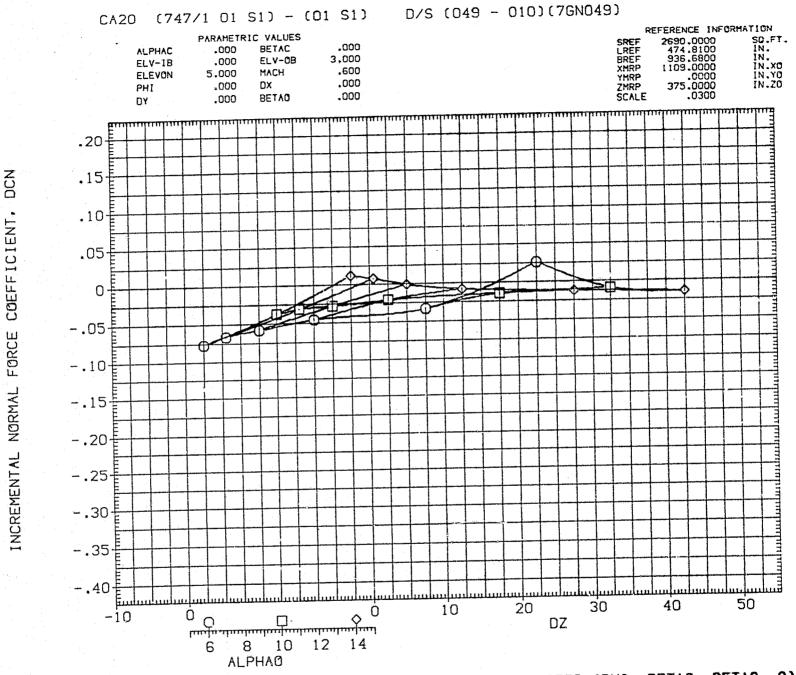


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1792

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1793

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1794

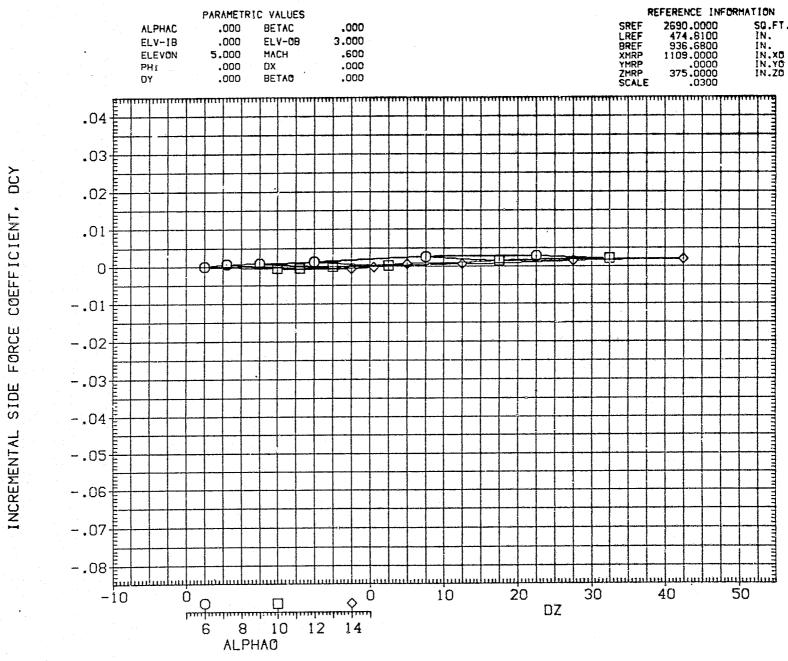


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1795

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

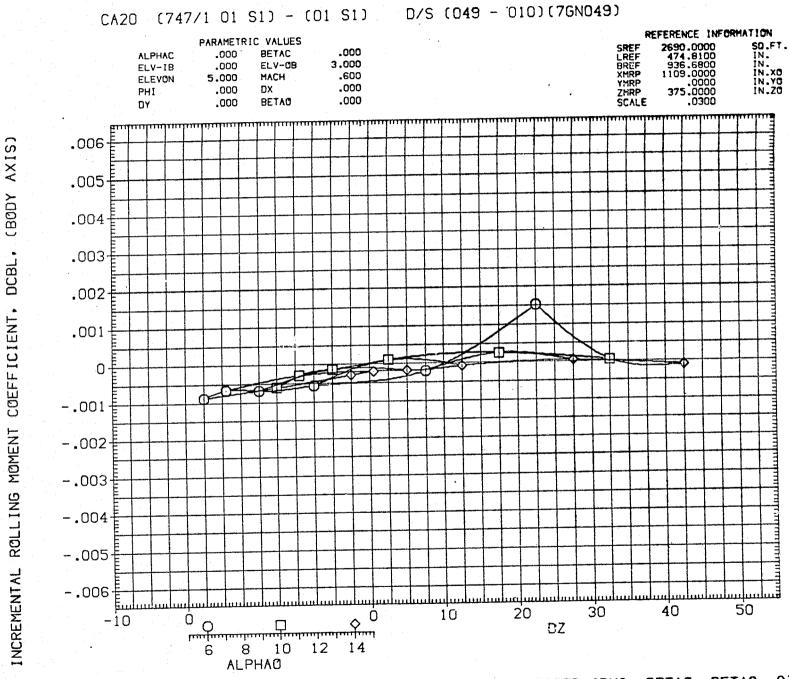


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1798

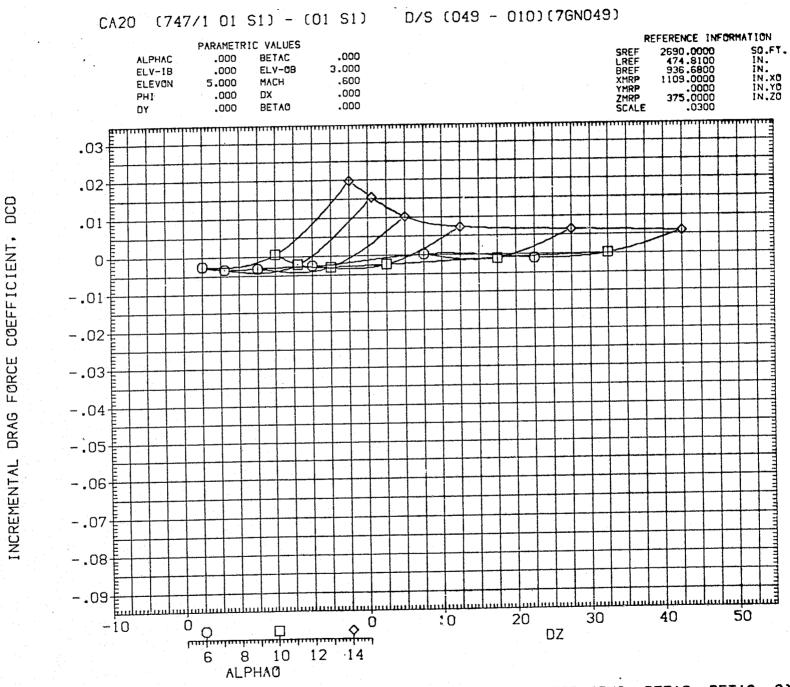


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1799

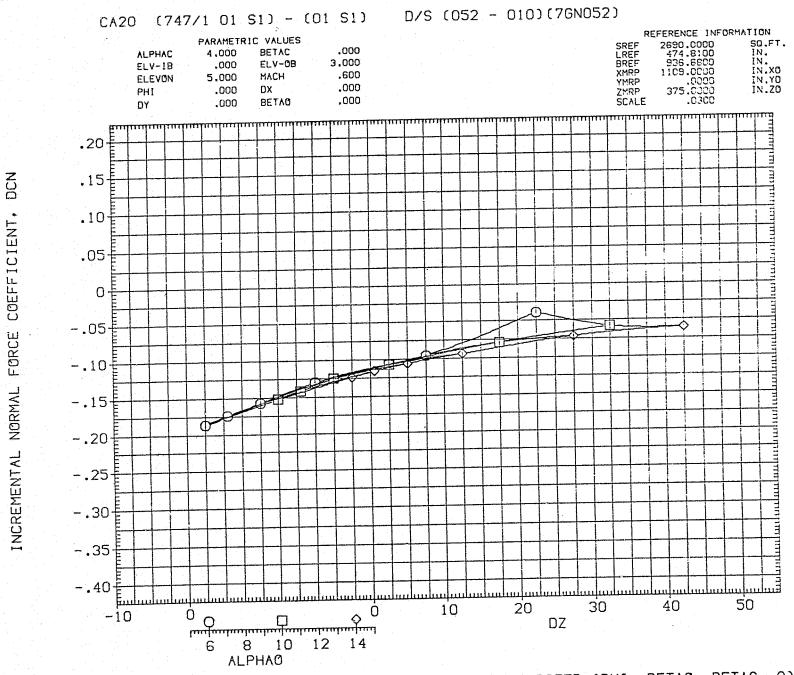


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1800

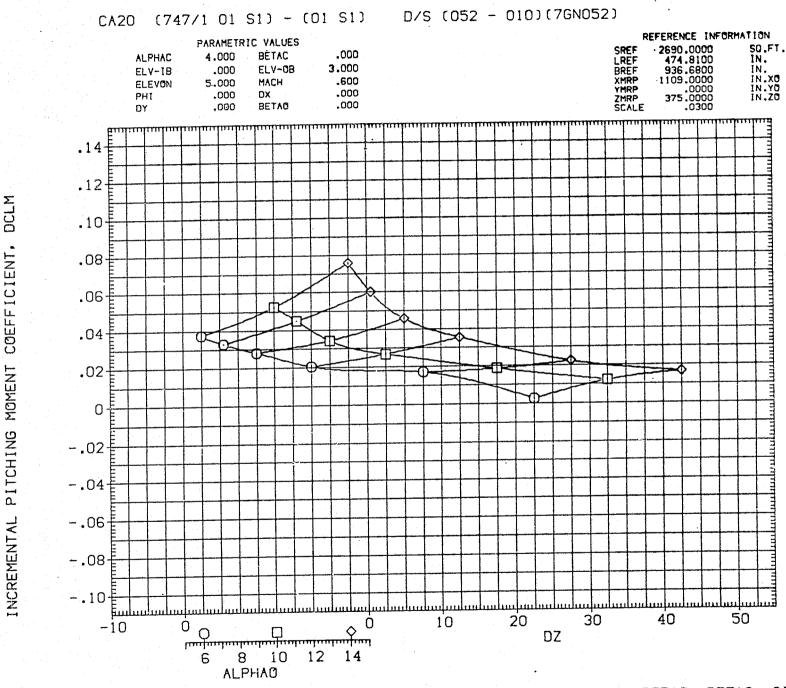


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1801

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1802

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0) FIG. 39 PAGE 1804

ALPHAO

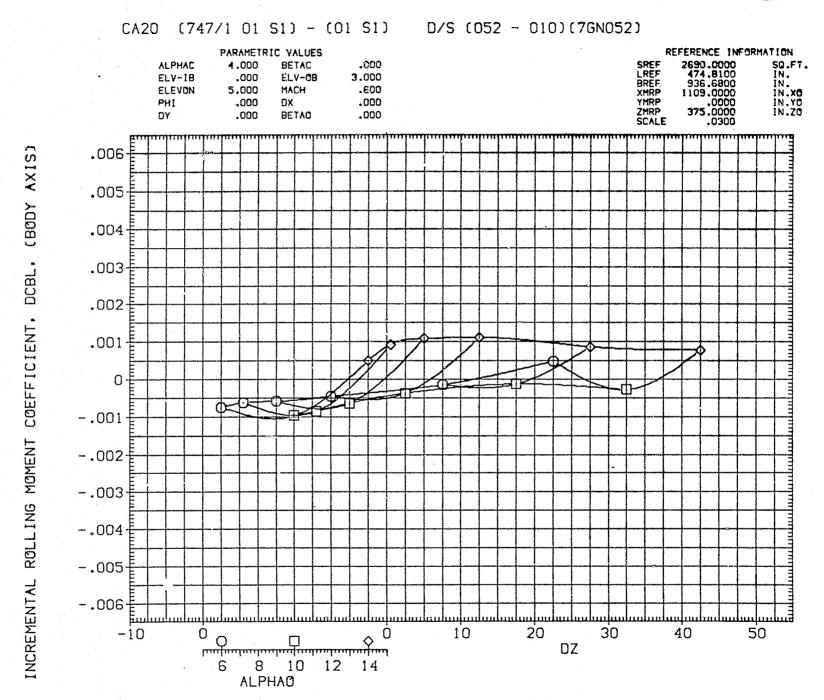


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1805

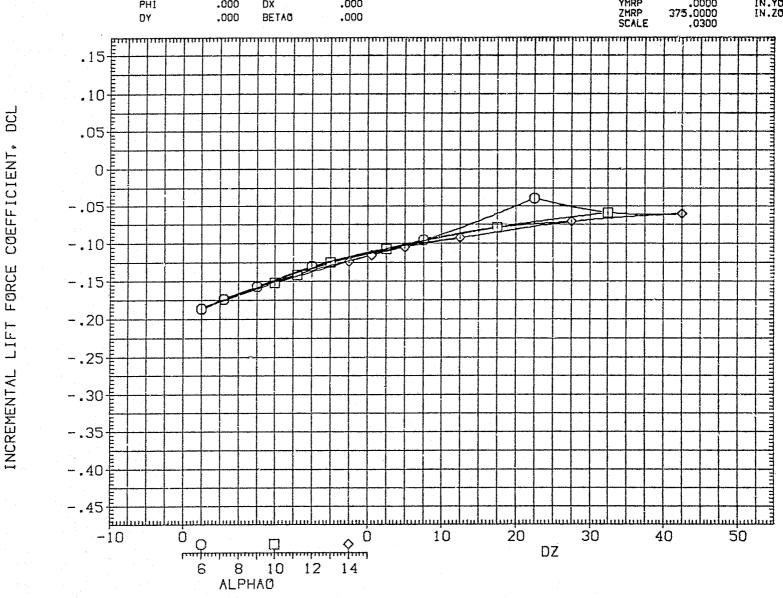


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1806

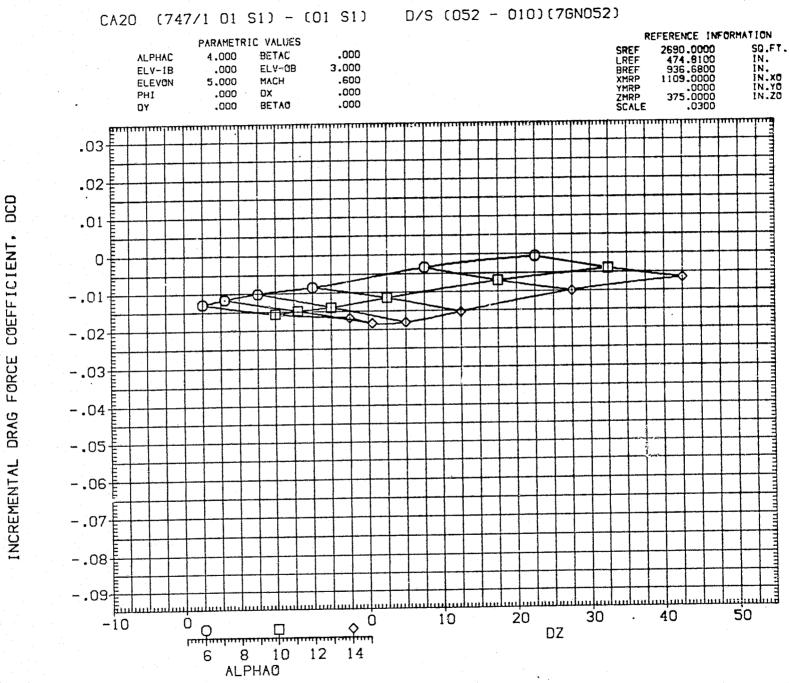


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1807

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
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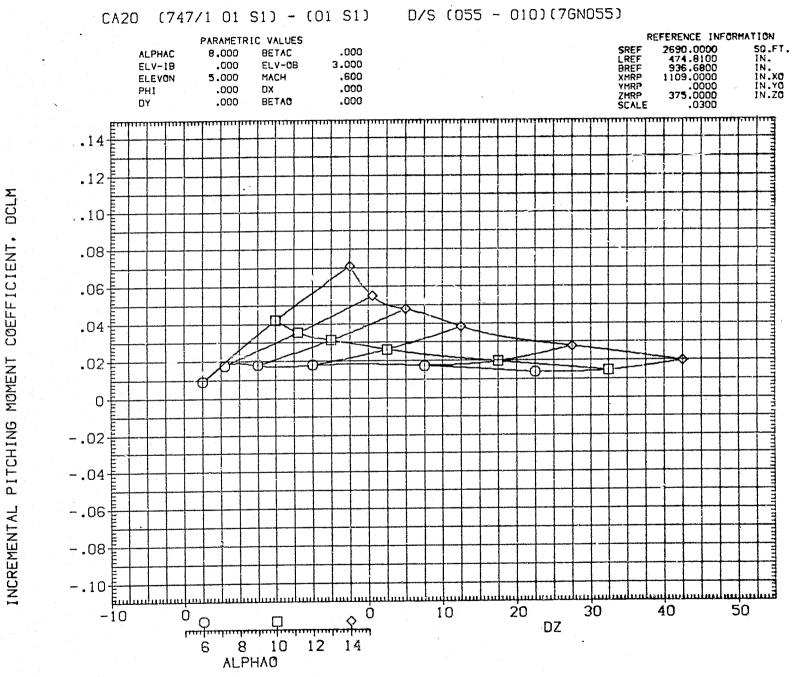
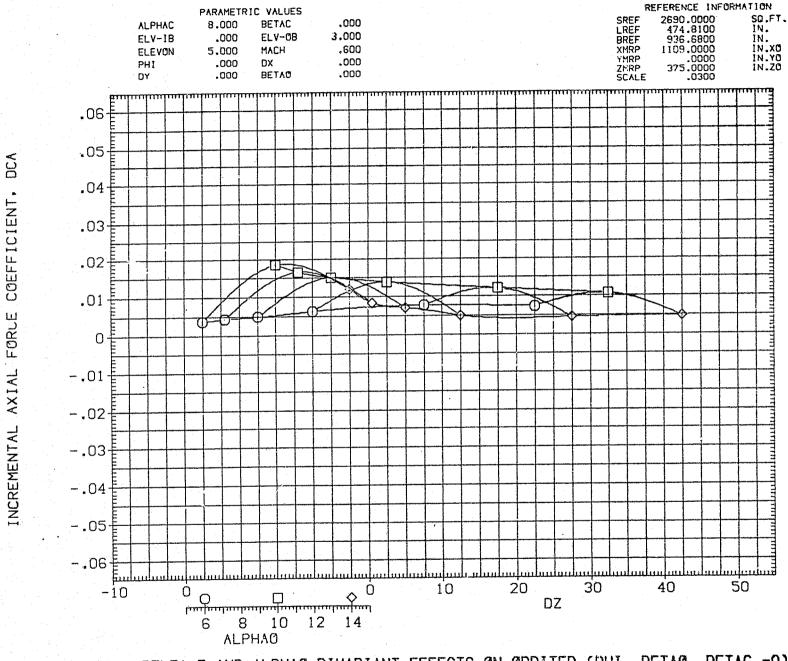


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
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FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
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FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1812

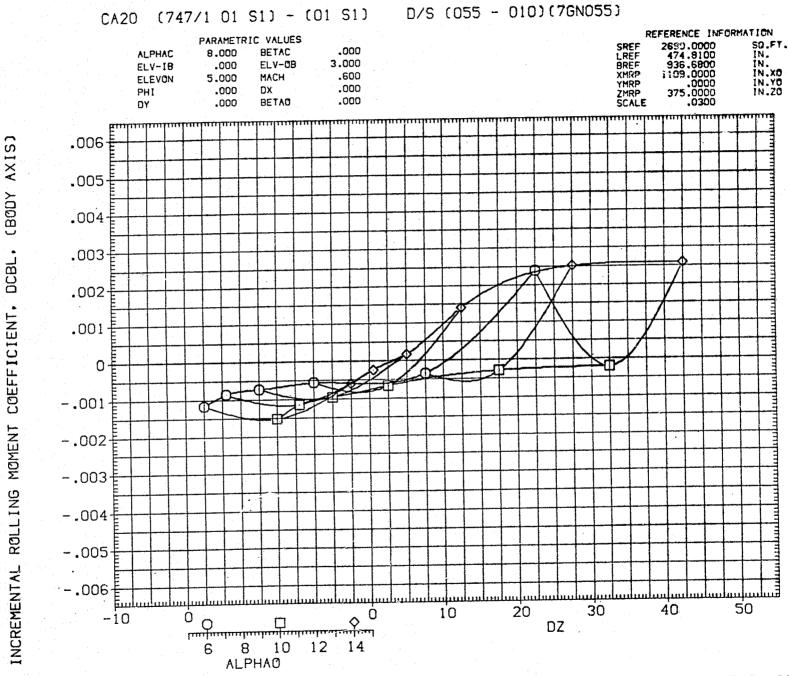


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1813

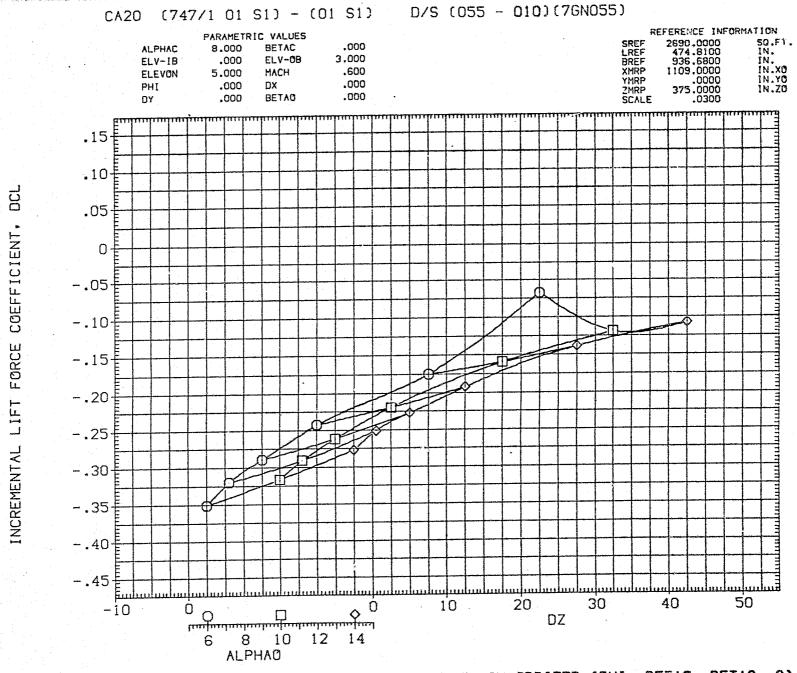


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1814

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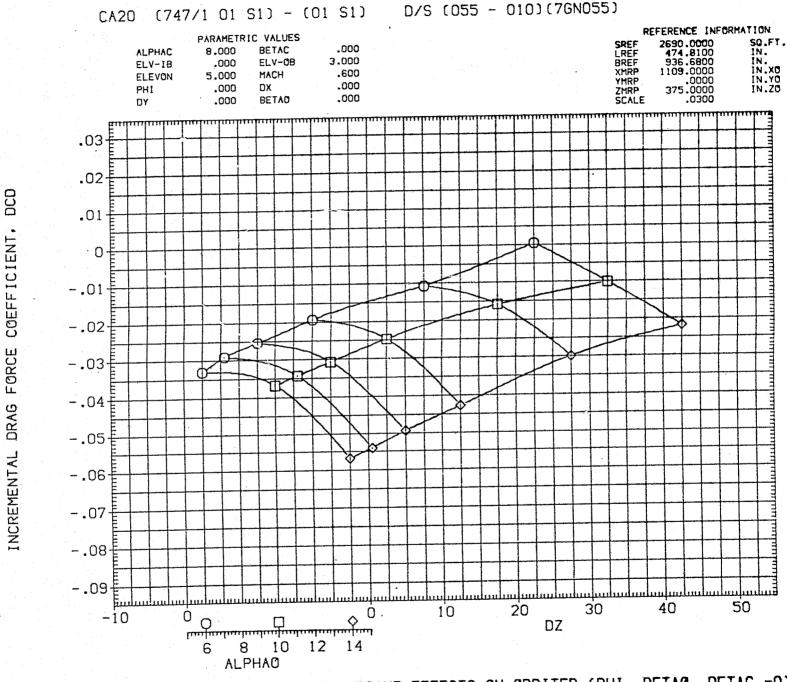
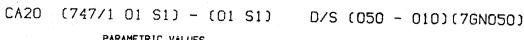


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1815

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)



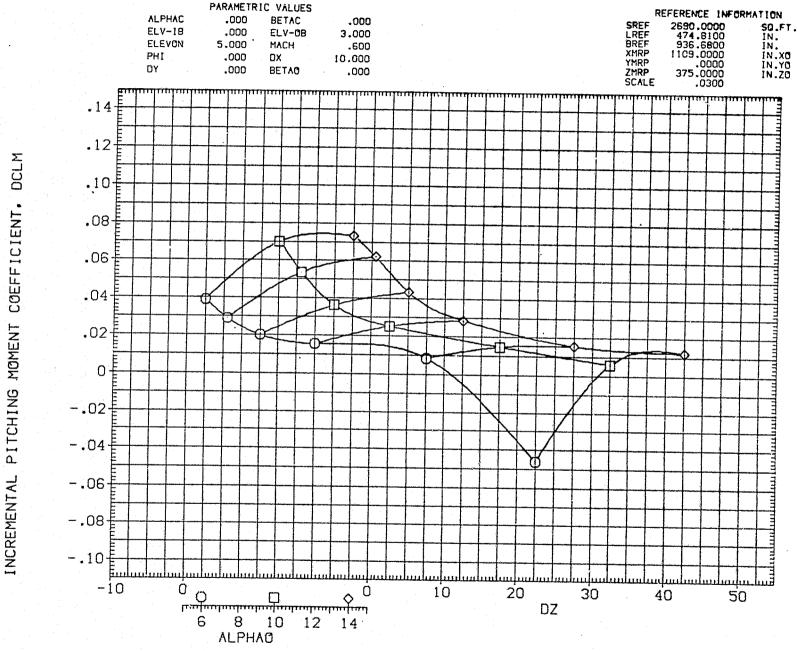


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1817

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
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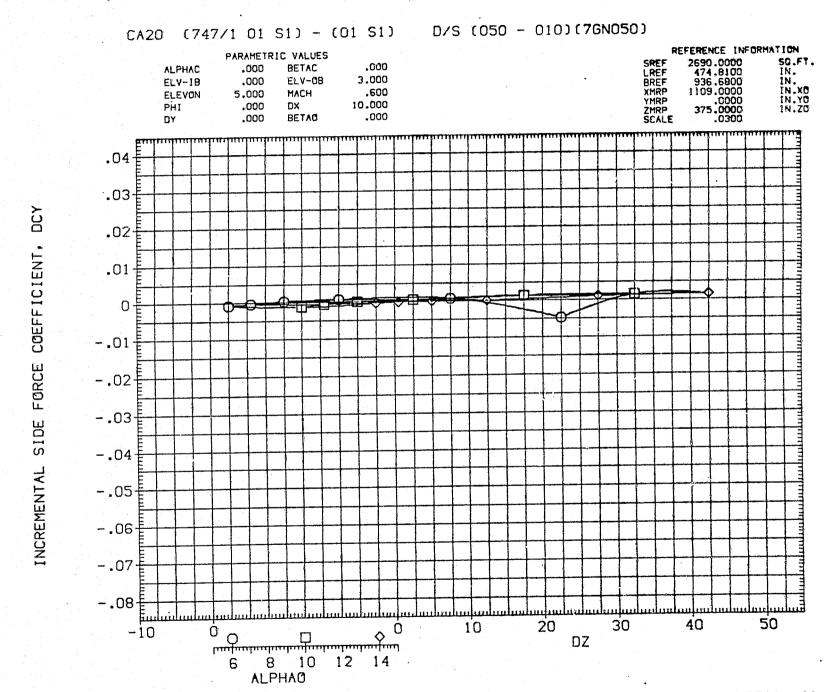


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

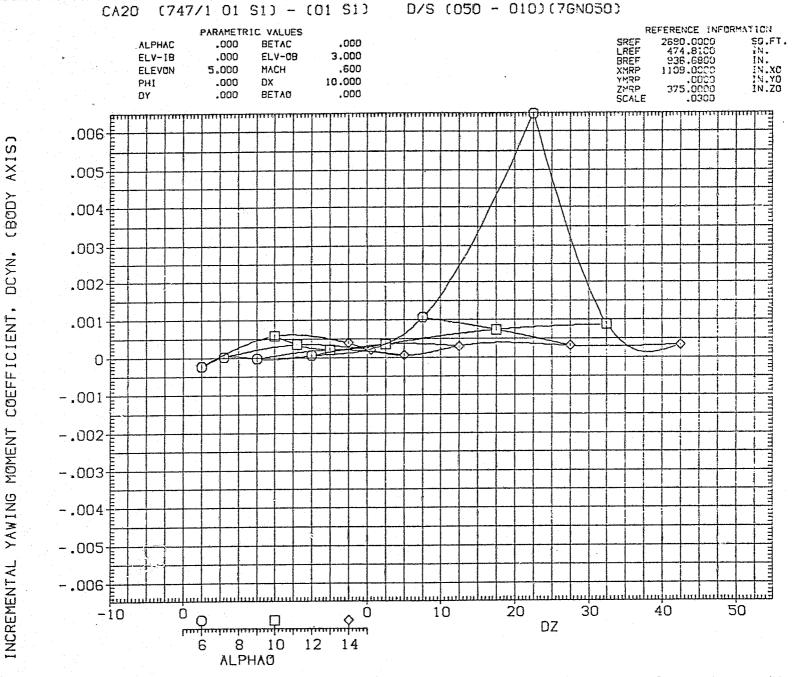


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON CRBITER (PHI. BETAO, BETAC =0)
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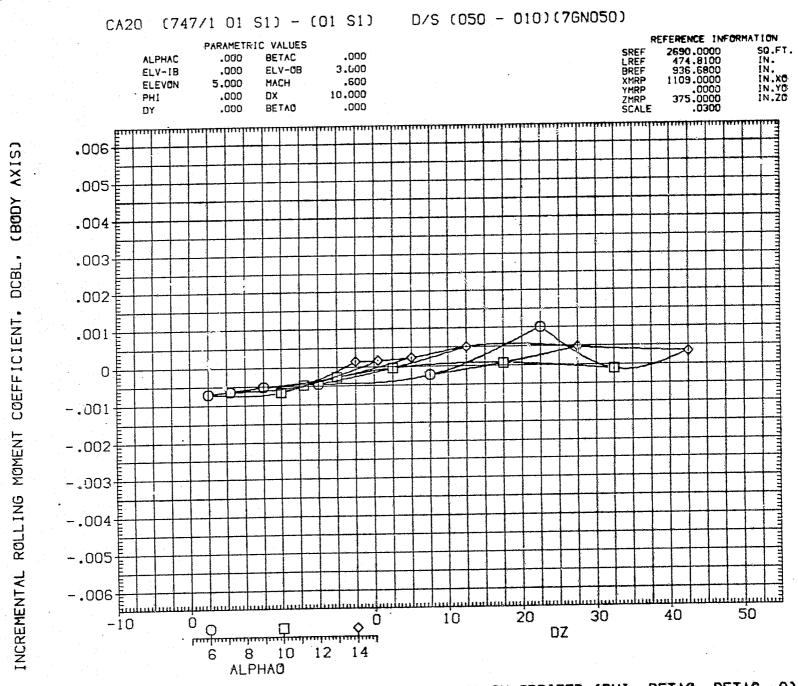


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1821

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)



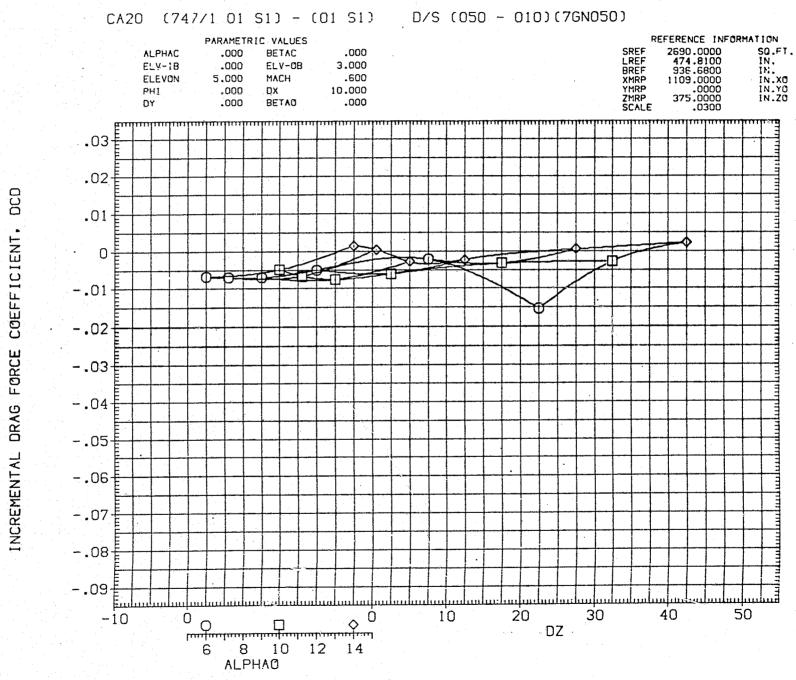


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

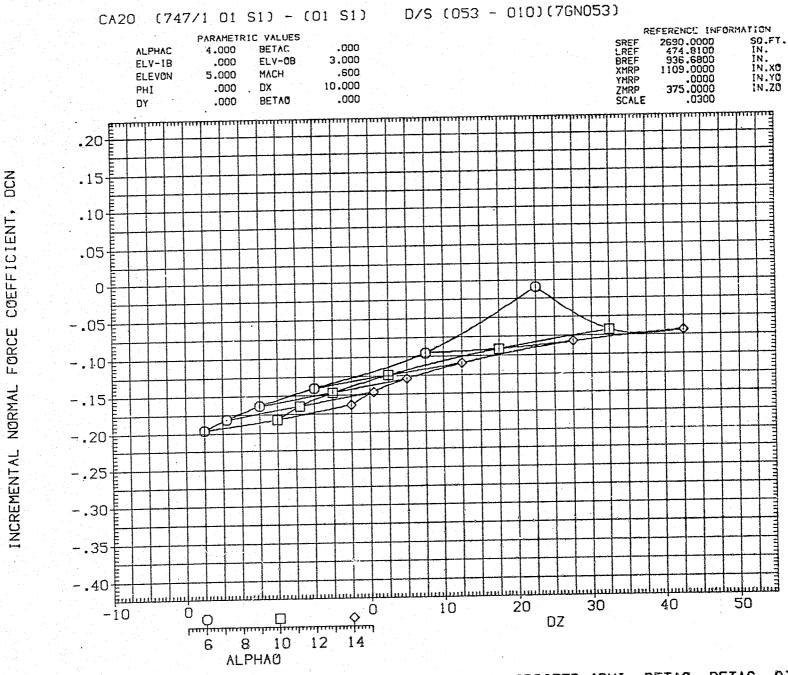


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1825

**ALPHAO** 

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0) PAGE 1826

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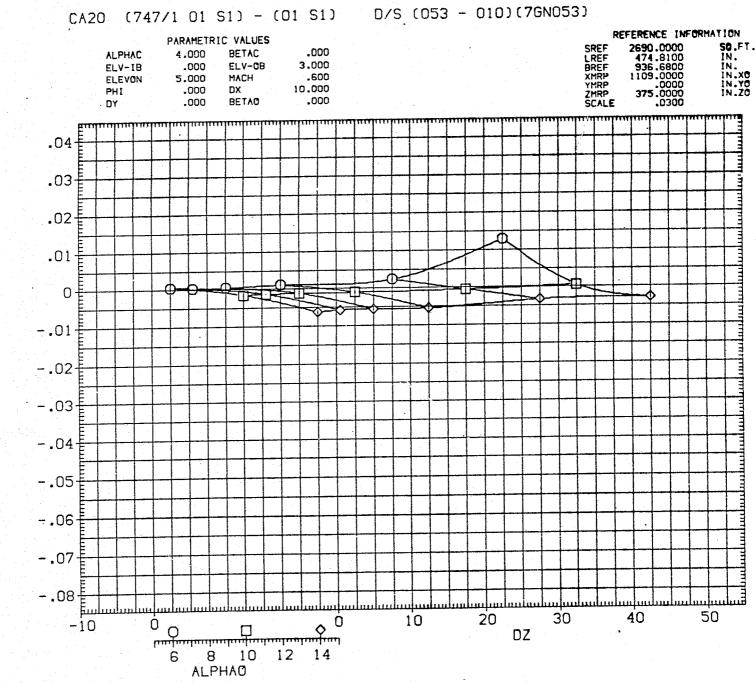


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0) FIG PAGE 1828

**ALPHAO** 

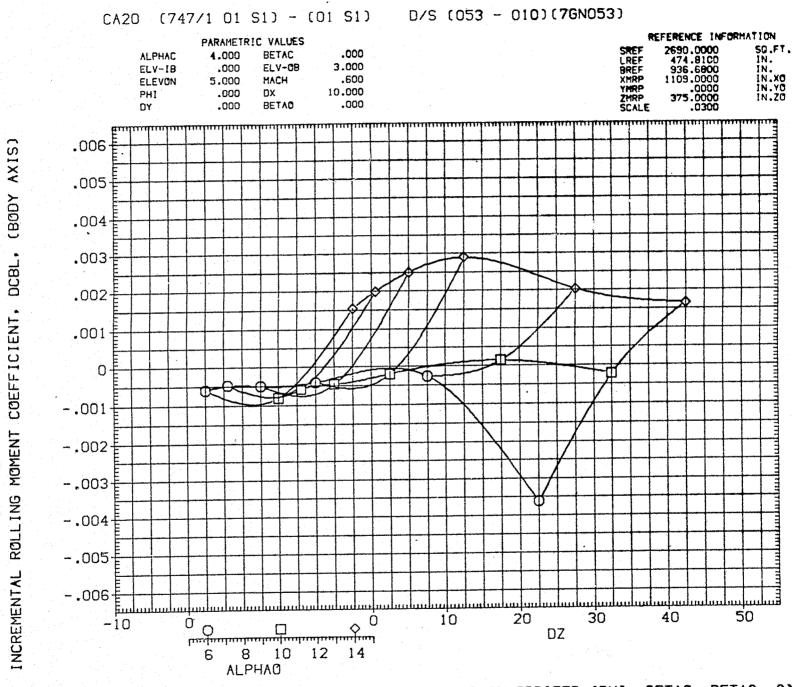


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

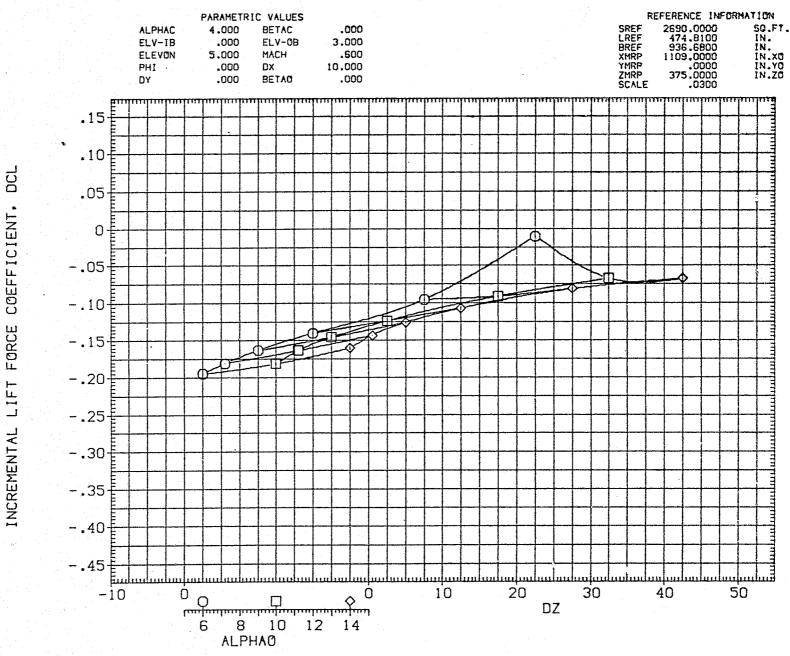


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1830

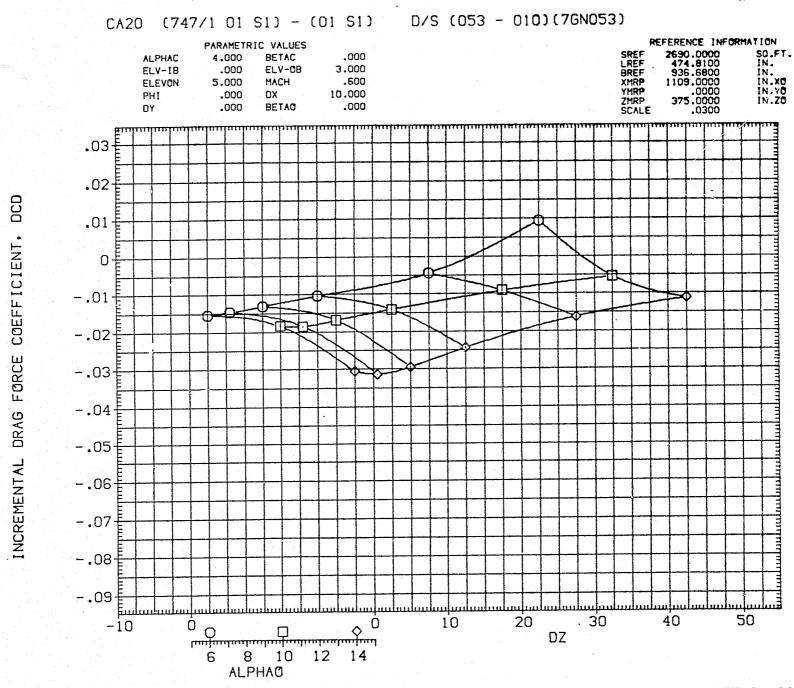


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1831

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1832



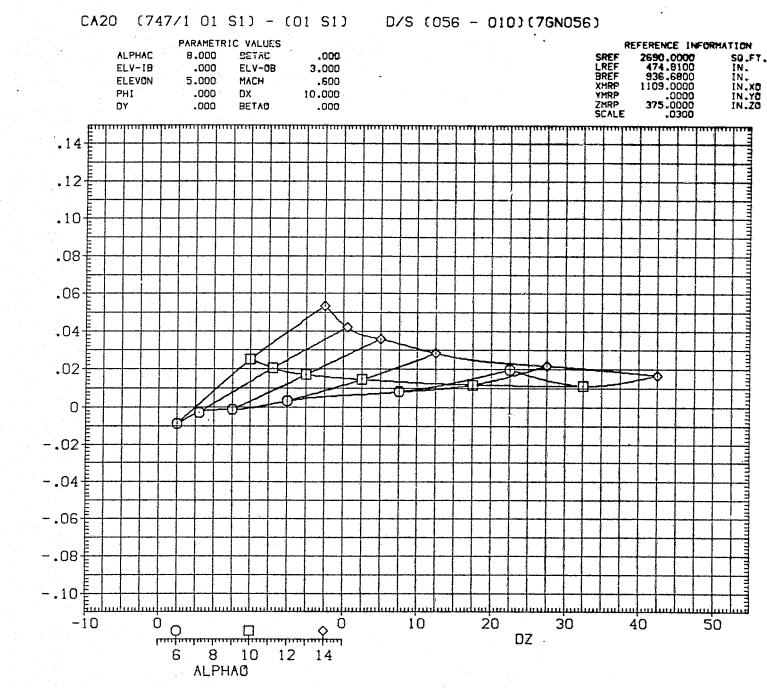


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1833

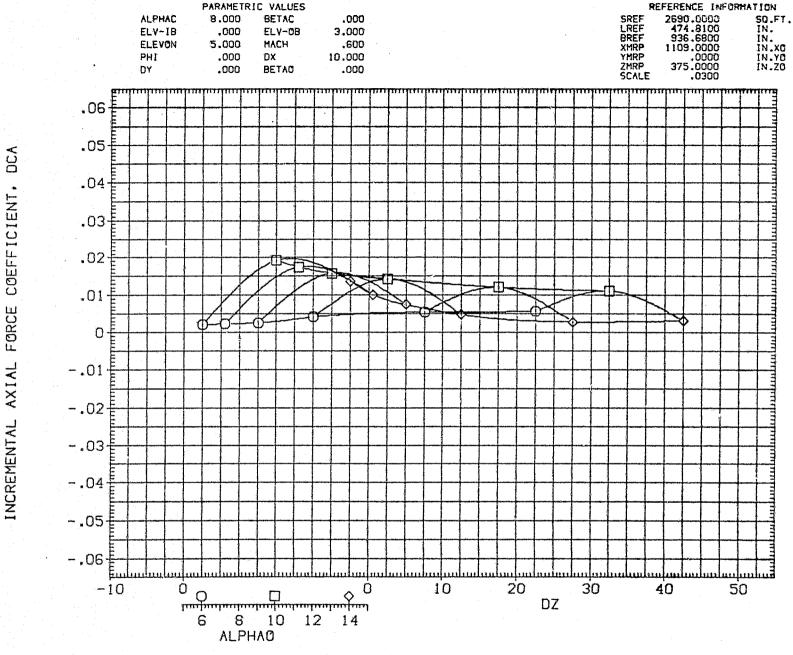


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1834

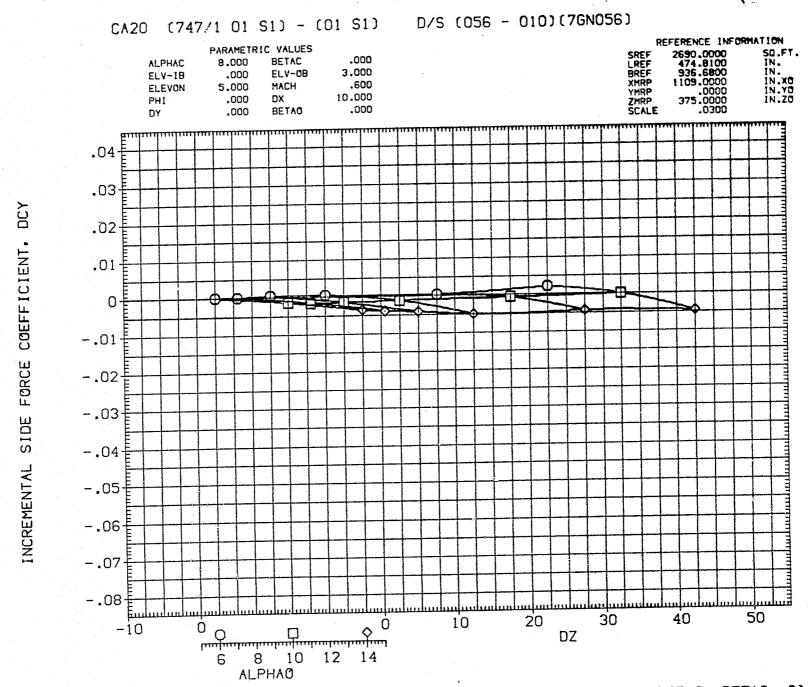


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1835

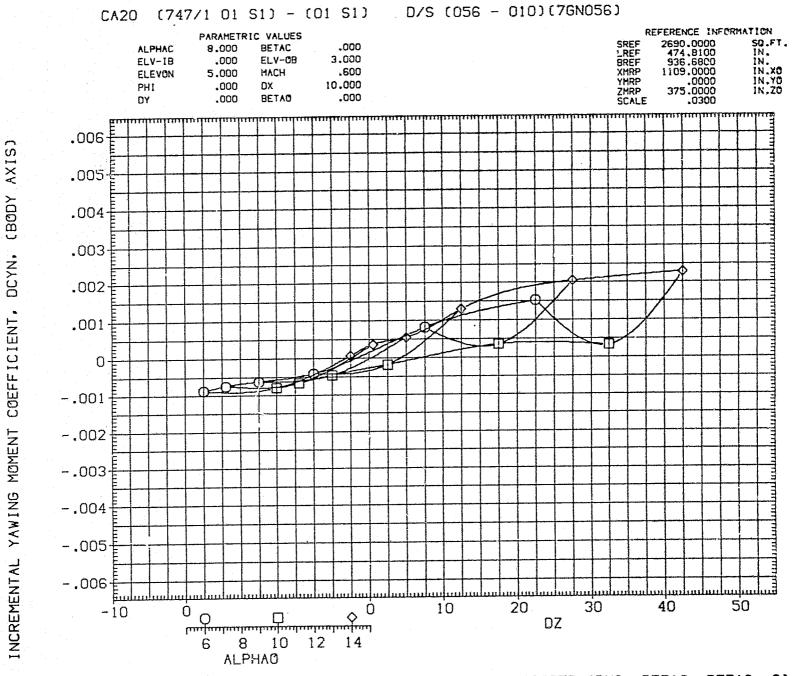


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1836



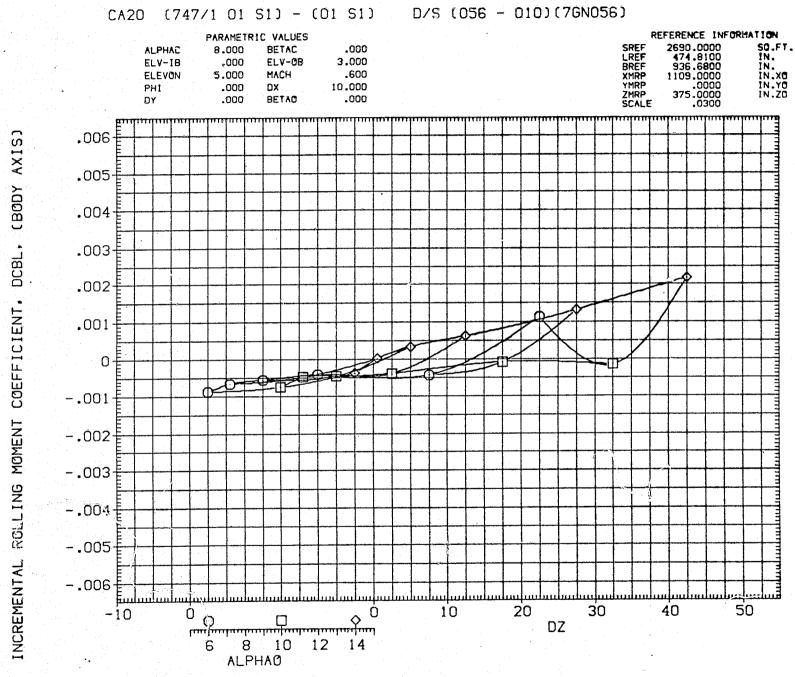


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1837

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)

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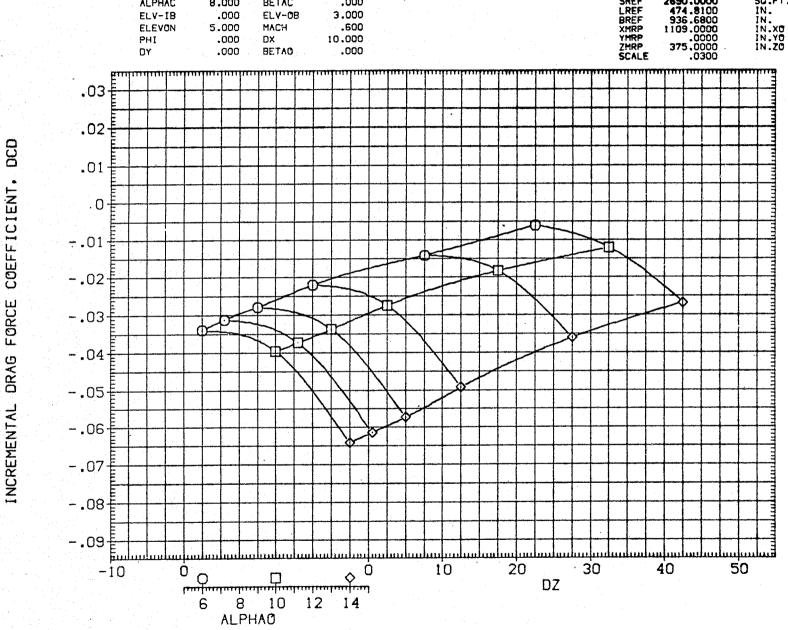


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO. BETAC =0)
PAGE 1839

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1840

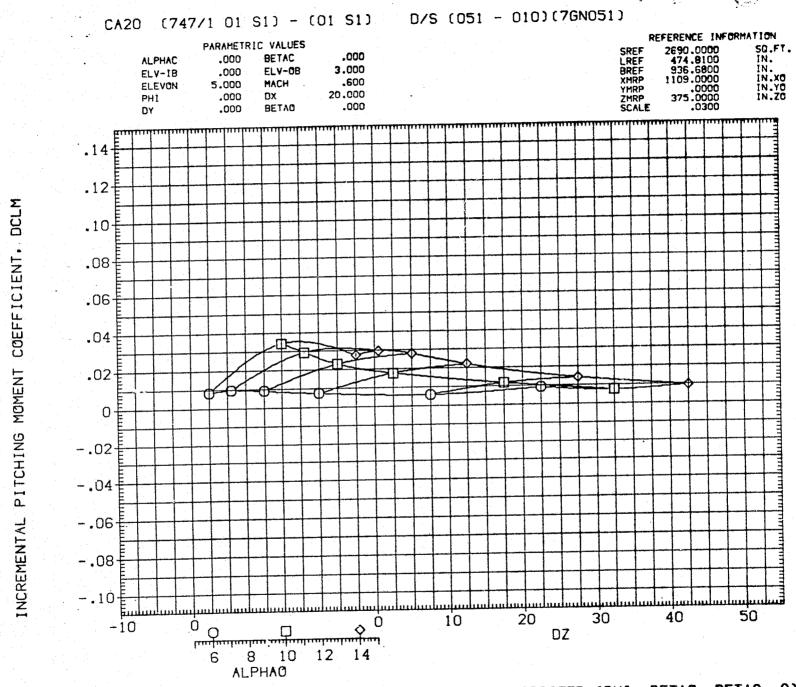


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO. BETAC =0)
PAGE 1841

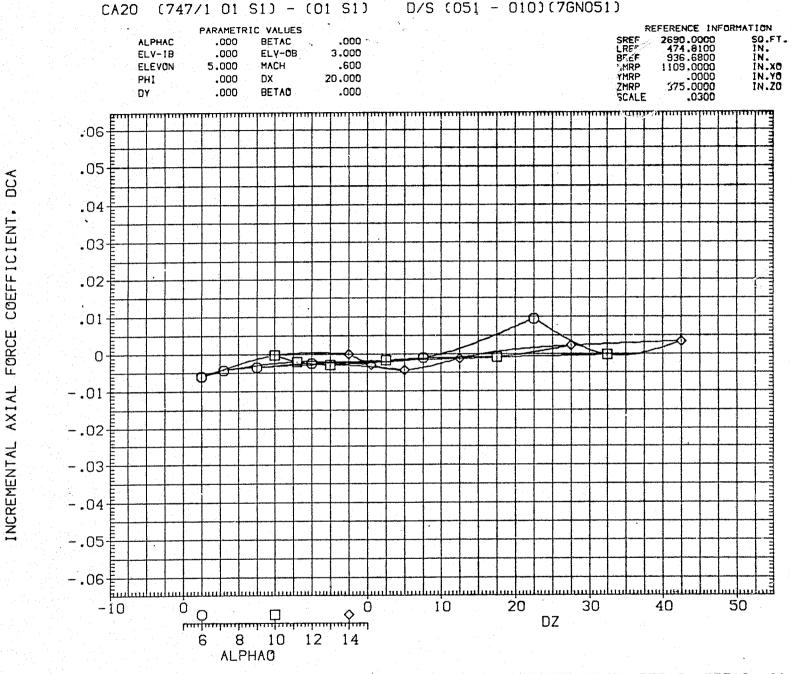


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1842

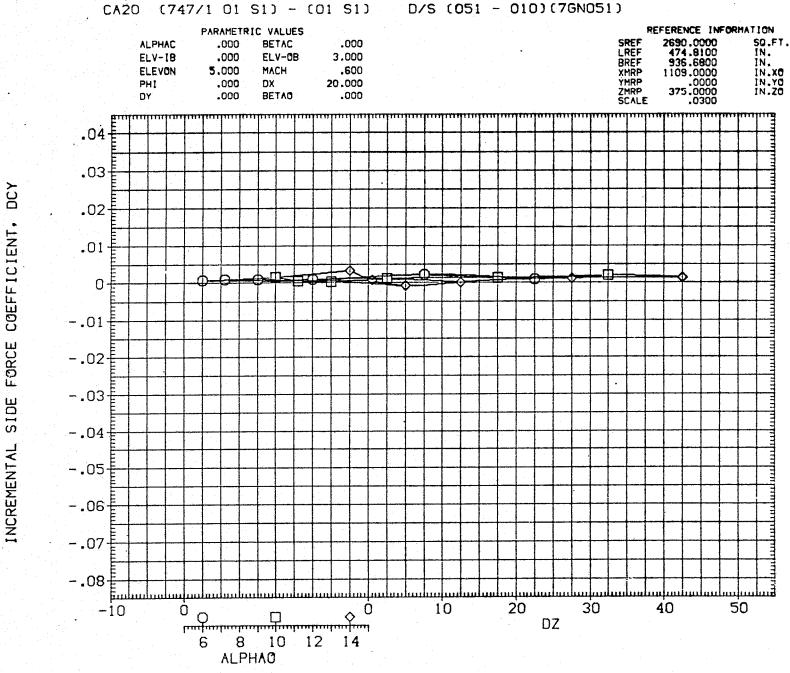


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1843

DELTA Z AND ALPHAU BIVARIANT EFFECTS ON ORBITER (PHI. BETAO. BETAC =0) 39 PAGE 1844

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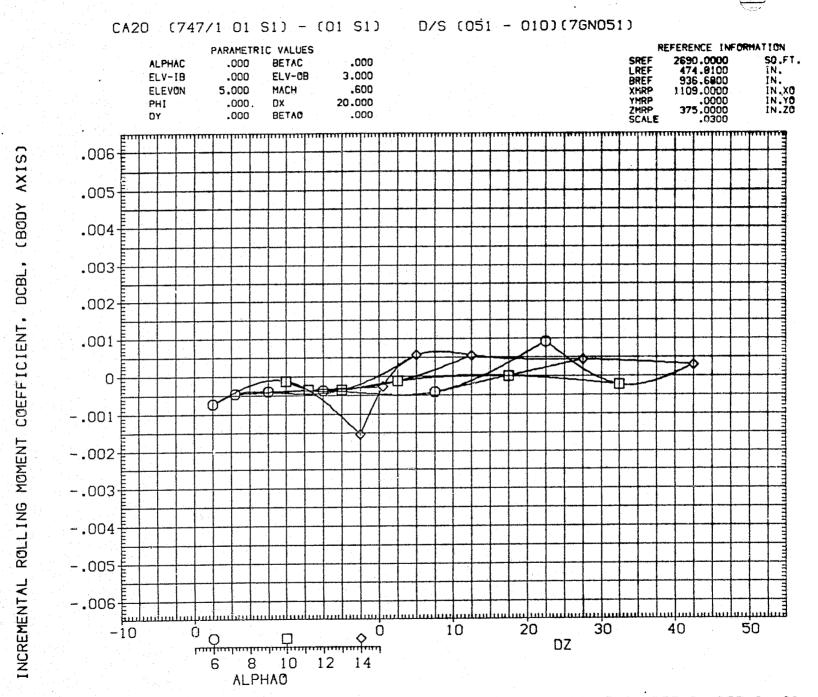


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1845

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0) FIG 39 PAGE 1846

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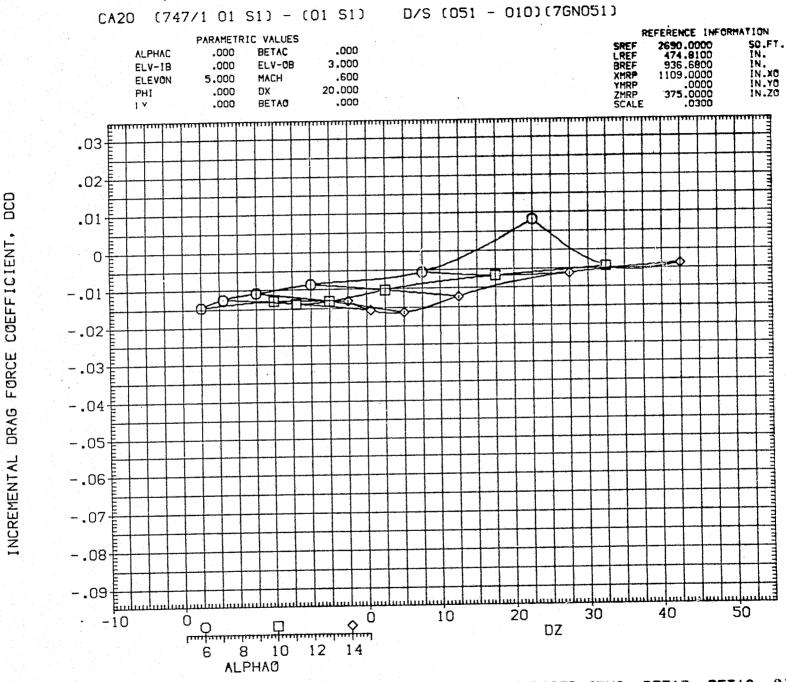


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

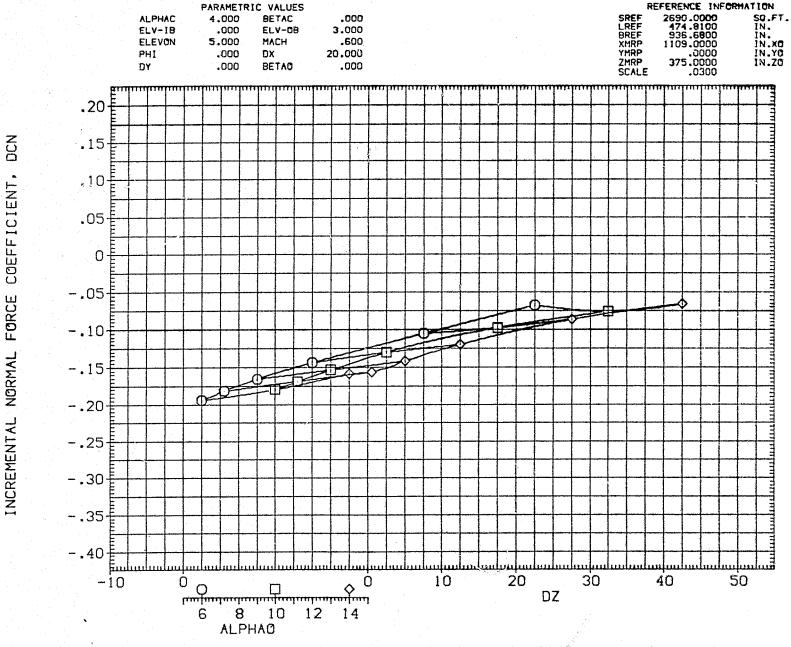


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1848

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

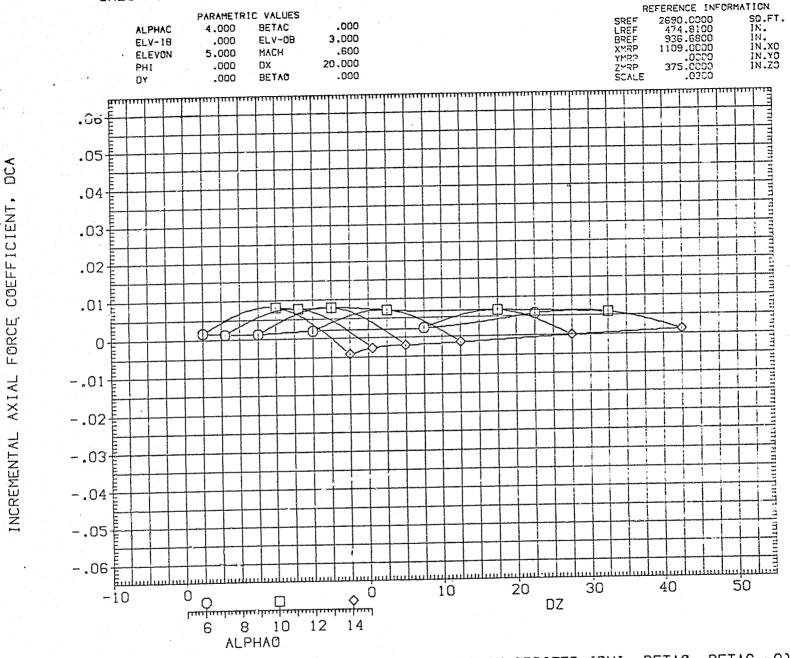


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1850

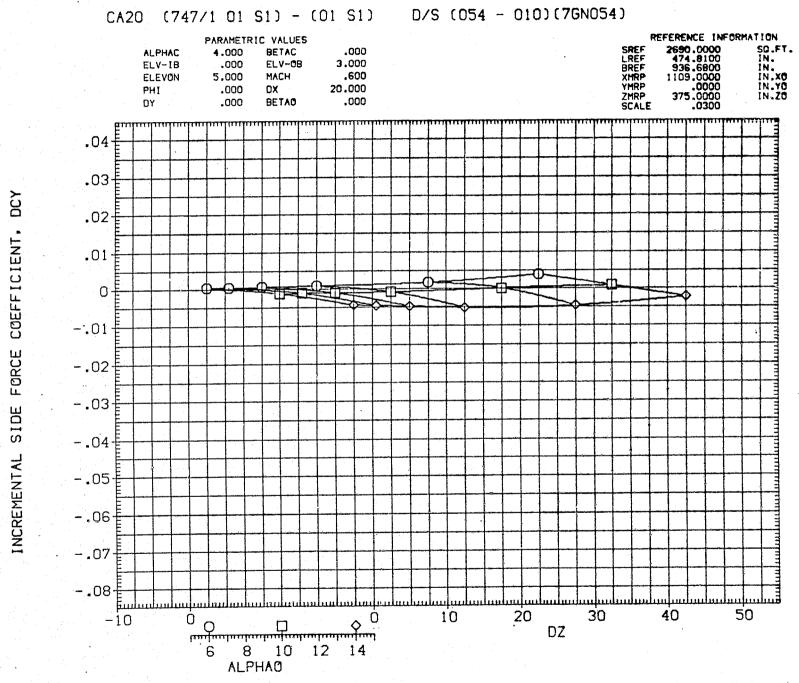


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1851

DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0) 1852 PAGE

**ALPHAO** 

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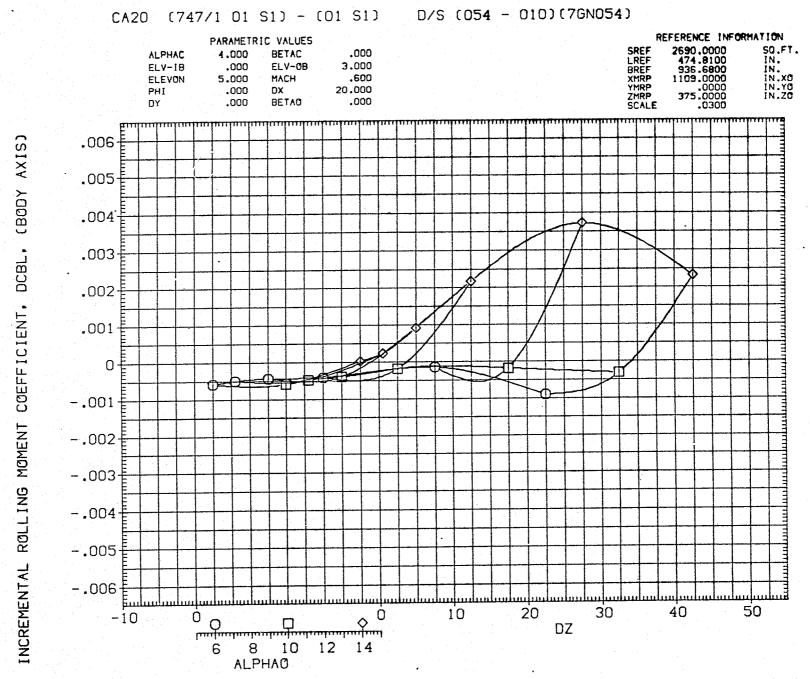


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1853

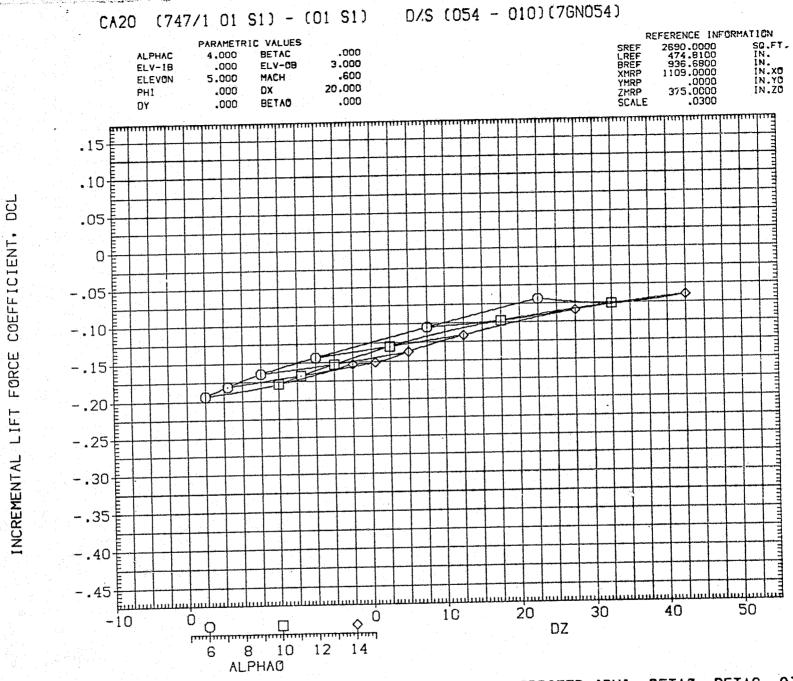


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1854

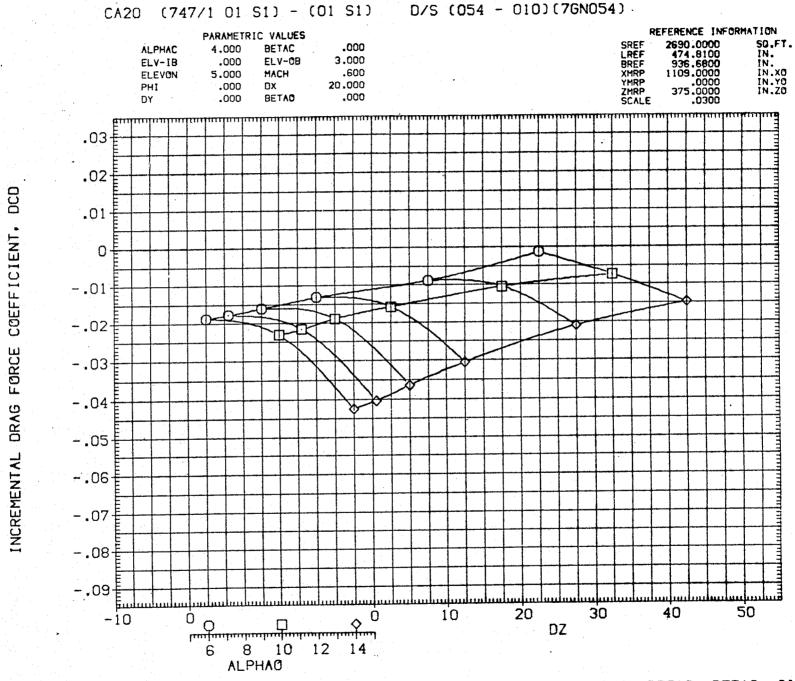


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)

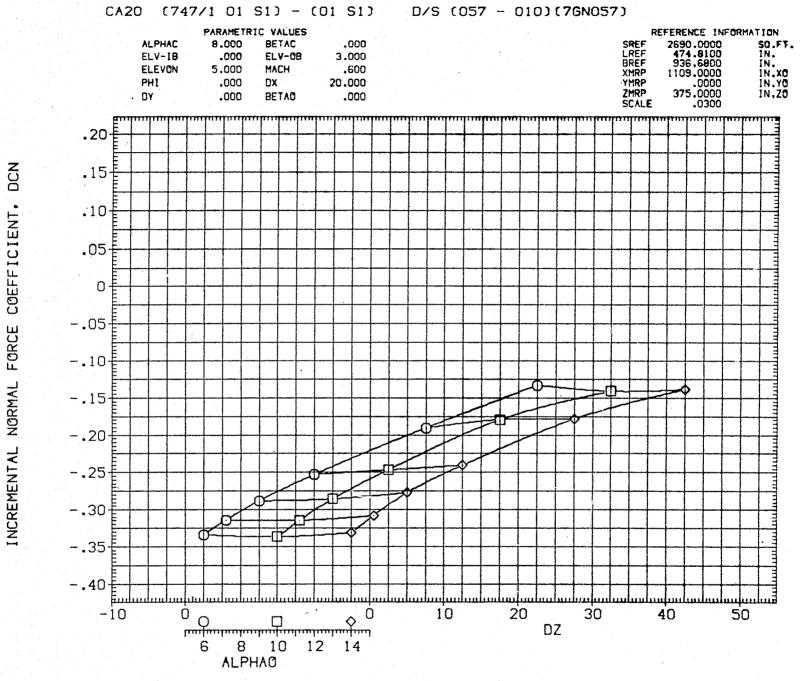


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1856

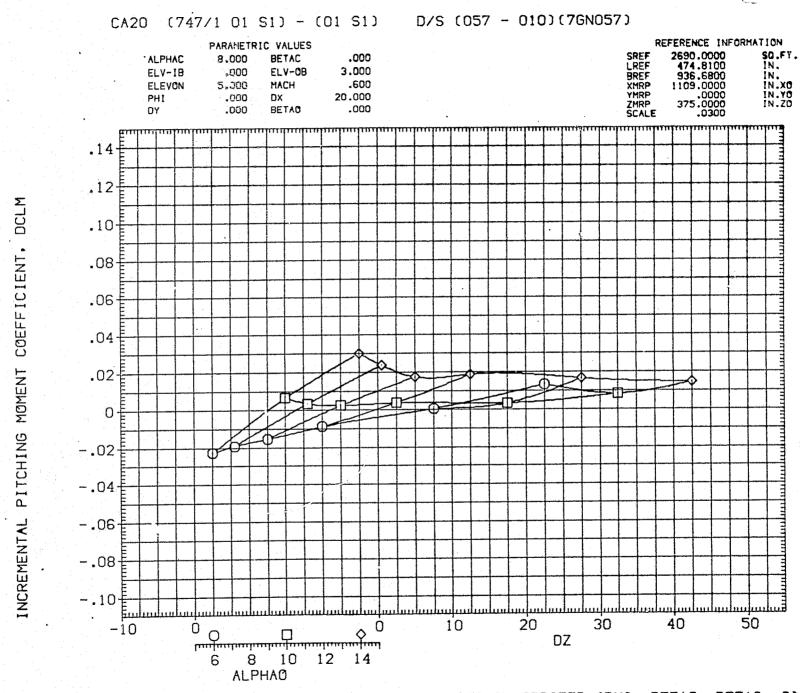


FIG. 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1857

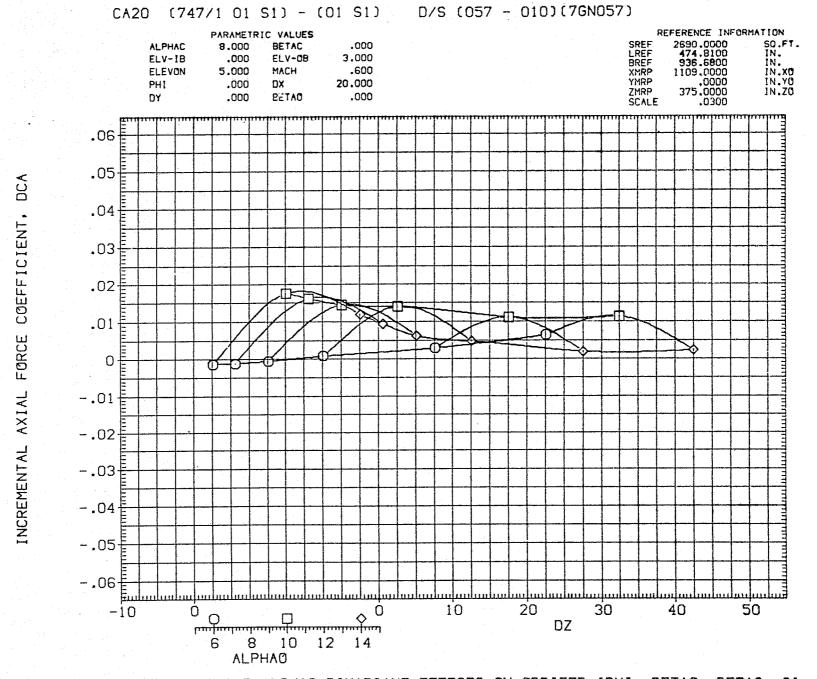


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1858

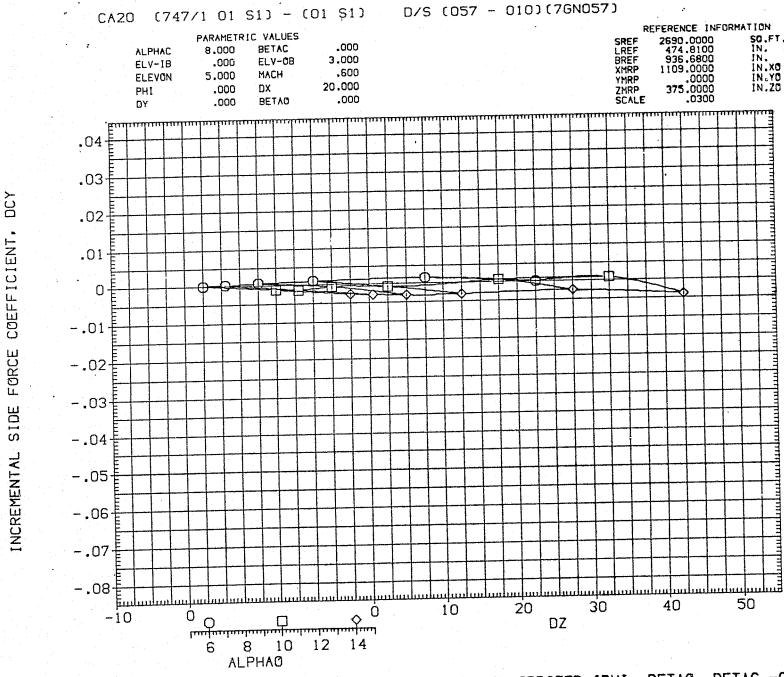
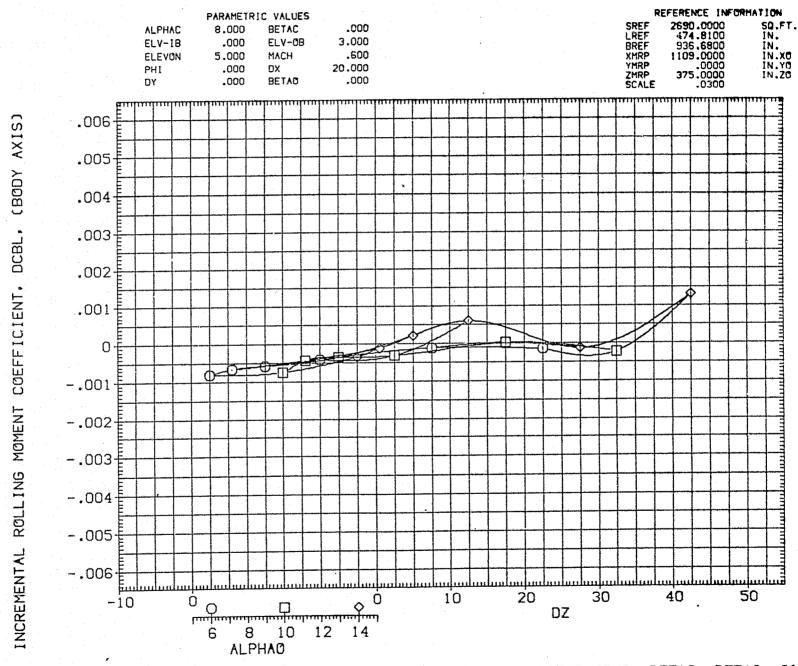


FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1859

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)





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FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI. BETAO, BETAC =0)
PAGE 1861

FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
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FIG 39 DELTA Z AND ALPHAO BIVARIANT EFFECTS ON ORBITER (PHI, BETAO, BETAC =0)
PAGE 1863